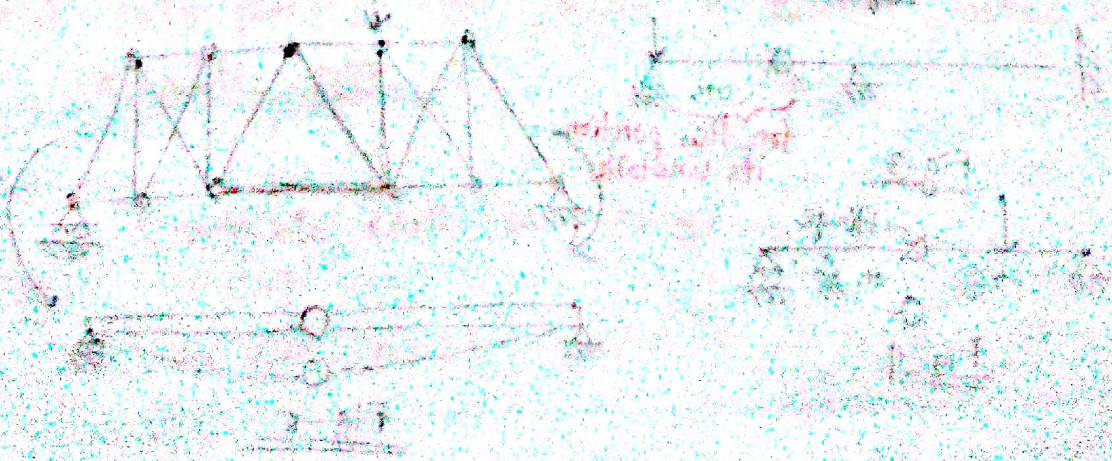


CE-311

Roof Site

# Syllabus

1. Stability
2. Determinacy of structures
3. Analysis of statically determinate trusses
4. " " " " arches
5. influence lines
6. moving loads on beams, frames and trusses
7. analysis of suspension bridge
8. Wind load
9. Earthquake loads
10. Approximate analysis of statically indeterminate structures: braced trusses, portal method, cantilever method and vertical load analysis of multi-storied building frames.
11. Deflection of beams, trusses and frames by virtual work method.



Books :

1) Structural analysis

— by Norris + Willbur + Utku

2) Structural analysis

— By Seld + Vawter

# Stability —

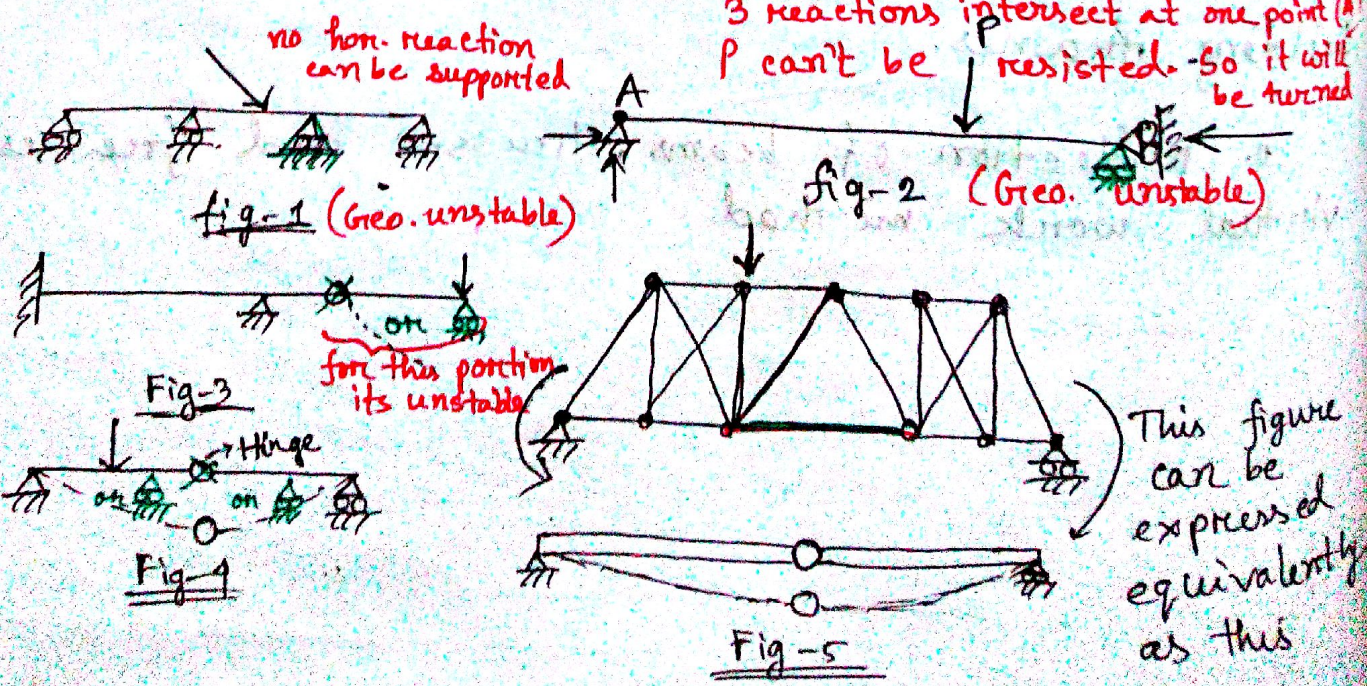
A structure is —

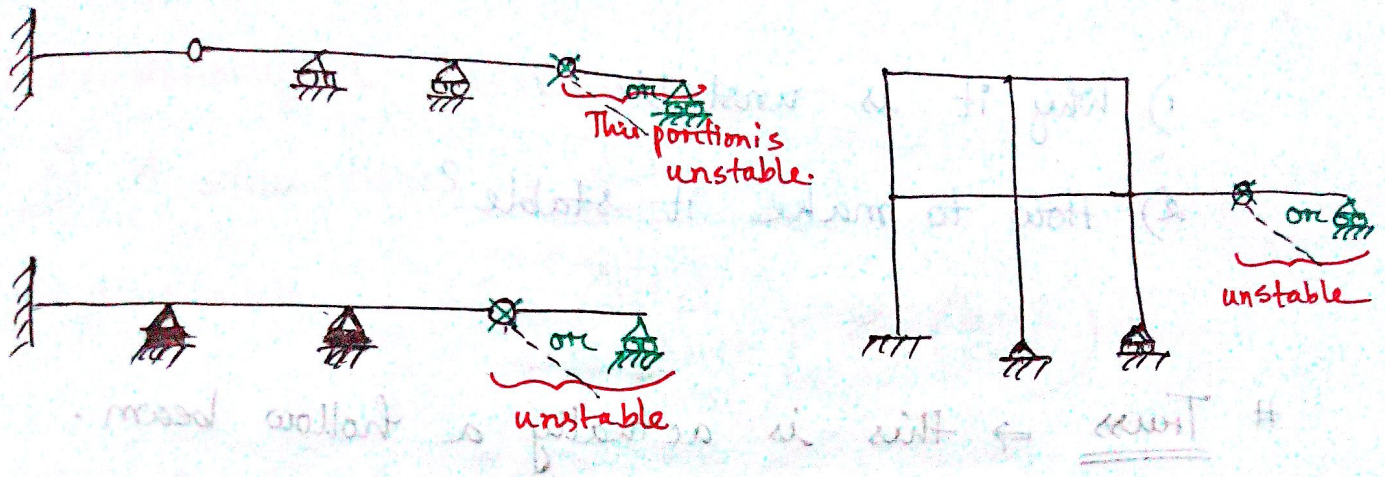
1) Statically unstable

if total eq<sup>n</sup> > total unknown

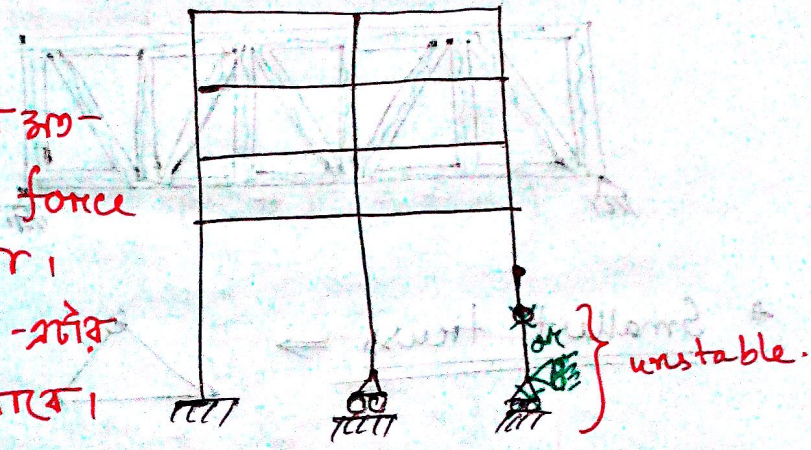
2) Geometrically unstable —

if general arrangement of reactions on members are not properly made.





○ ⇒ Hinge support  
 but দরজার কবজার ঠাট-  
 খোঁটকাতে পারবে but force  
 resist করতে পারবে না।  
 So এর পাবের position-রটার  
 respect -এ দুলবে থাকবে।



All the above structures are geometrically unstable.

\* In fig-1

statically stable →  $cz \text{ eq}^n < \text{unknowns}$   
 but geometrically unstable →  $cz \text{ no such support}$

(\*) সেকোন ছোট portion-ও unstable শরমত total টে  
 unstable হবে।

↓  
 Unstable cant be analysed.

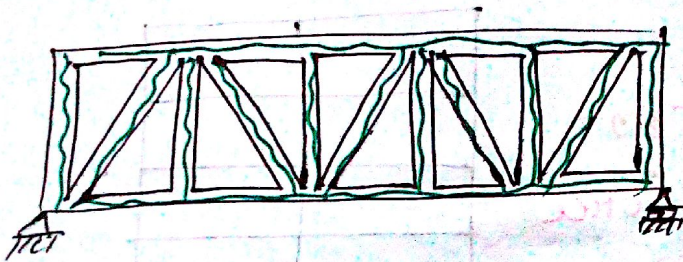
↓  
 so we must make it stable.

⊛ So first to identify —

1) Why it is unstable?

2) How to make it stable?

# Truss ⇒ this is actually a hollow beam.



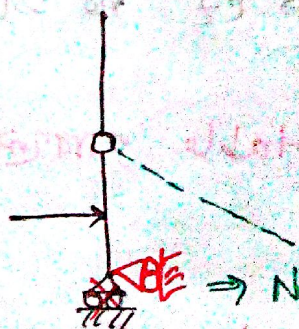
# Smallest truss ⇒



$$b = 2j - 3$$

↓  
minimum no. of bars required for a truss.

# How to understand stable/unstable —



i.e. non-load can not resist

⇒ Now the load can be resisted.

## Structural Stability.

### \* Determinacy & Indeterminacy —

a) A structure is statically :

i. Unstable if  $(\text{total unknown forces}) < (\text{Total Eq}^{\text{ns}})$

ii. Stable and determinate if  $(\text{"}) = (\text{"})$

iii. Stable and indeterminate if  $(\text{"}) > (\text{"})$

AND  $D^{\circ}$  of Indeter-  
-minacy  $= (\text{"}) - (\text{"})$

# # Determinacy and Indeterminacy (Contd.) $\Rightarrow$

b) Total Unknown Forces = Member forces + R's

$$= \underline{1} b + r \quad \text{for a (2D) truss}$$

$$= \underline{3} m + r \quad \text{for a (2D) frame}$$

$\underbrace{\hspace{10em}}_{(AF)}$   
 $\underbrace{\hspace{10em}}_{(AF, SF, BM)}$

$= r$  for a beam/col.

c) Total Eq<sup>n</sup>s = Equilibrium + Eq<sup>n</sup>s of conditions

Eq<sup>n</sup>s for joints

$$= \textcircled{2} j + 0 \quad \text{for a (2D) truss}$$

$(\Sigma F_x, \Sigma F_y)$

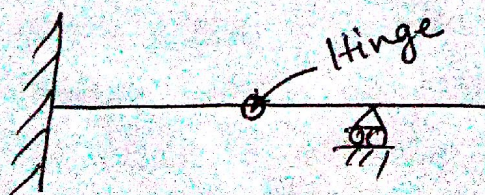
$\Sigma F_x, \Sigma F_y, \Sigma M$

$$= \textcircled{3} j + c \quad \text{for a (2D) frame}$$

$$= 3 + c \quad \text{for a beam/col.}$$

*bee z truss -> special condition 2x 2x c = joint -> hinge.*

- $b$  = No. of bars ;  $r$  = No. of reactions
- $m$  = no. of members ;  $c$  = No. of conditions (e.g. Hinges)
- $j$  = no. of joints



$$Eq^m = 3 + c$$

$$= 3 + 1$$

$$= 4$$

; Here,  $c$  = Condition  $\Rightarrow \Sigma M_{Hinge} = 0$

# # Problems :-

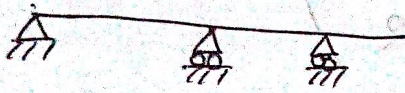
## Beams -

Geo. bility

No. of R's

Indeter-  
minacy

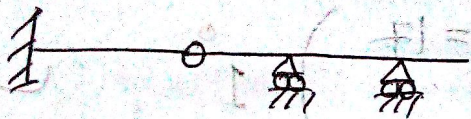
Statically  
Stability ?



$$\gamma = 4$$

$$4 - 3 = 1^{\circ}$$

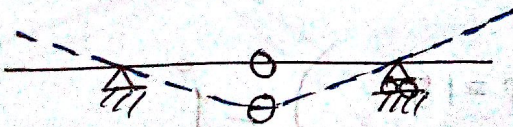
✓



$$\gamma = 5$$

$$5 - (3 + 1) = 1^{\circ}$$

✓

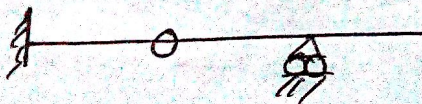


$$\gamma = 3$$

$$3 - (3 + 1) = -1^{\circ}$$

X

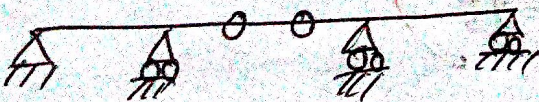
-ve means this is statically unstable.



$$\gamma = 4$$

$$4 - (3 + 1) = 0^{\circ}$$

✓



$$\gamma = 5$$

$$5 - (3 + 2) = 0^{\circ}$$

✓

Unknown

Eq<sup>n</sup>s  
(3 + e)

D<sup>o</sup>

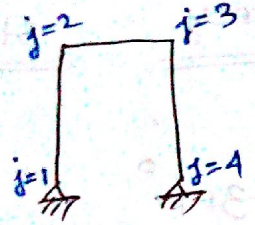
# Plane (2D) Frames :

Geo. Stability

Indeter. minancy

Stat. Stability

✓



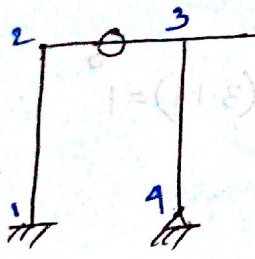
$m=3$   
 $r=4$   
 $j=4$   
 $c=0$

Total unk. =  $3m+r=13$   
 Total eq's.  $3j+c=12$

1°

✓

✓



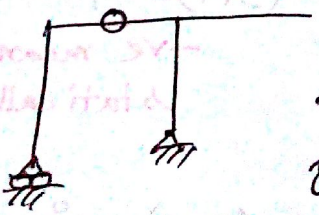
$m=4$   
 $j=5$   
 $r=5$   
 $c=1$

$3m+r=17$   
 $3j+c=16$

1°

✓

X



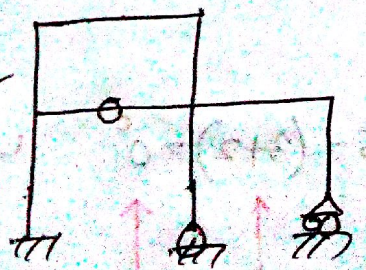
$m=4$   
 $r=3$   
 $j=5$   
 $c=1$

$3m+r=15$   
 $3j+c=16$

-1°

X

✓



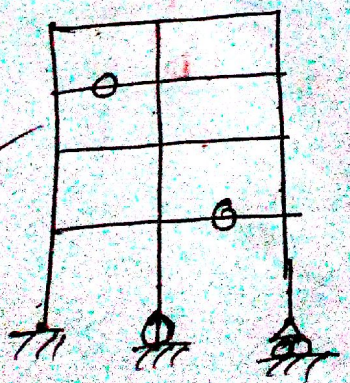
$m=8$   
 $r=6$   
 $j=6$   
 $c=1$

$3m+r=30$   
 $3j+c=25$

5°

✓

✓



$m=20$   
 $r=6$   
 $j=15$   
 $c=2$

$3m+r=66$   
 $3j+c=47$

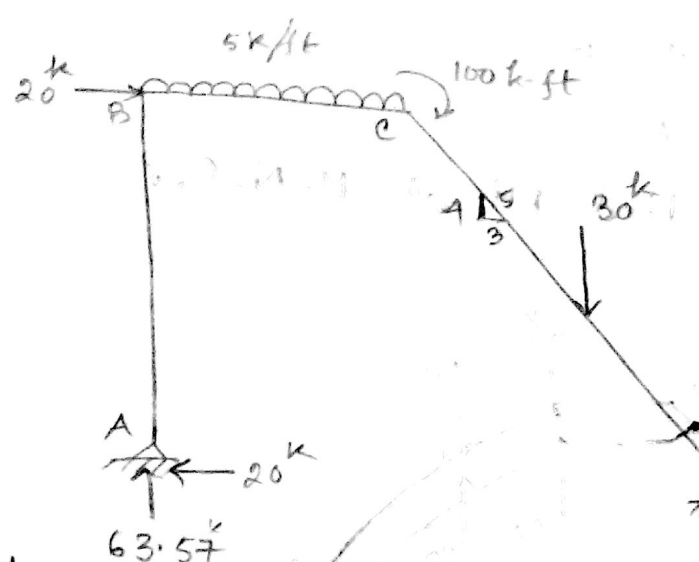
19°

✓





H.W



$$\sum M_A = 0 \quad \uparrow +ve$$

$$\Rightarrow 20 \times 20 + 100 \times 10 + 30 \times 27.5 + 100 - R_D \times 35 = 0$$

$$\Rightarrow R_D = 66.4 \text{ k} (\uparrow)$$

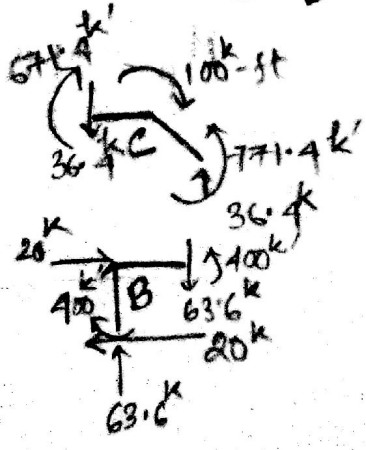
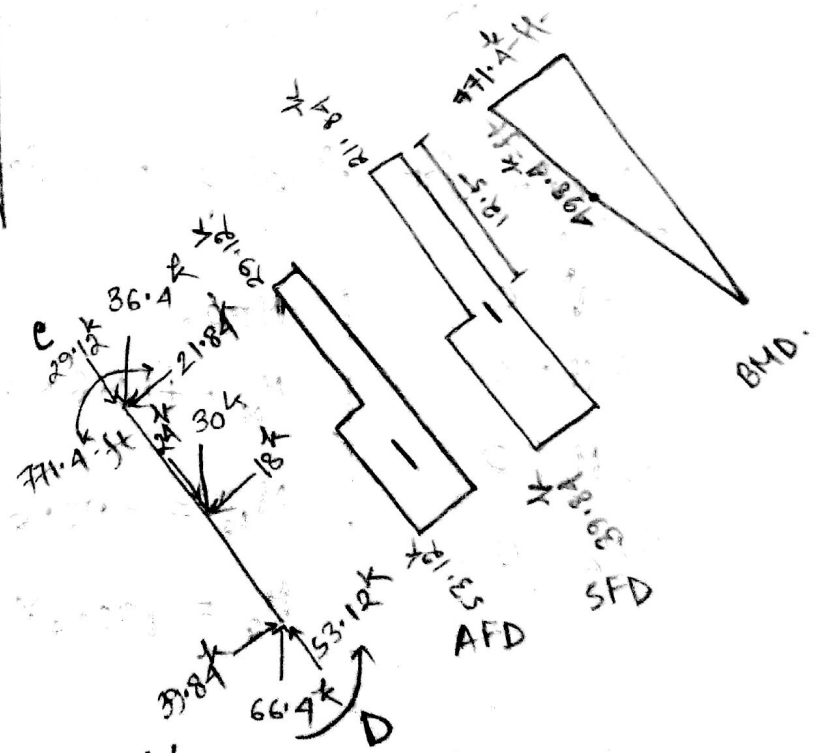
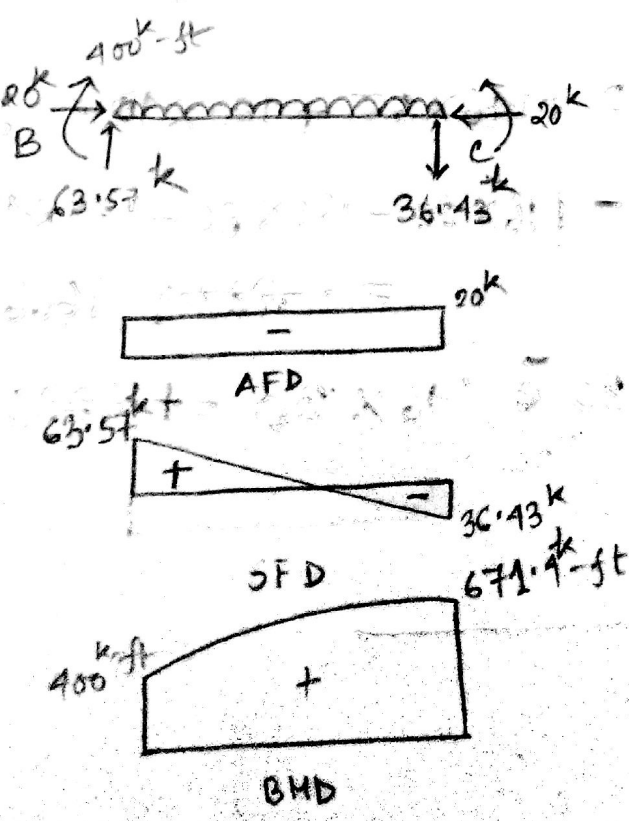
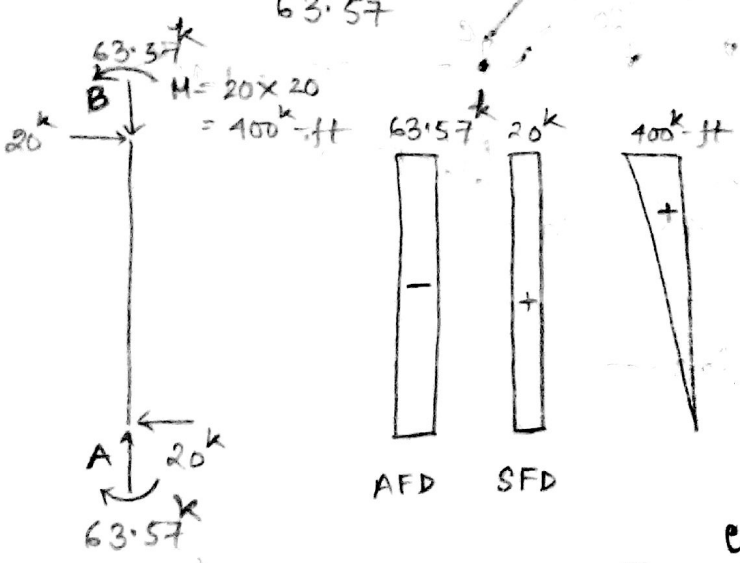
$$\sum F_y = 0 \quad \uparrow$$

$$\Rightarrow R_A + 66.4 - 100 - 30 = 0$$

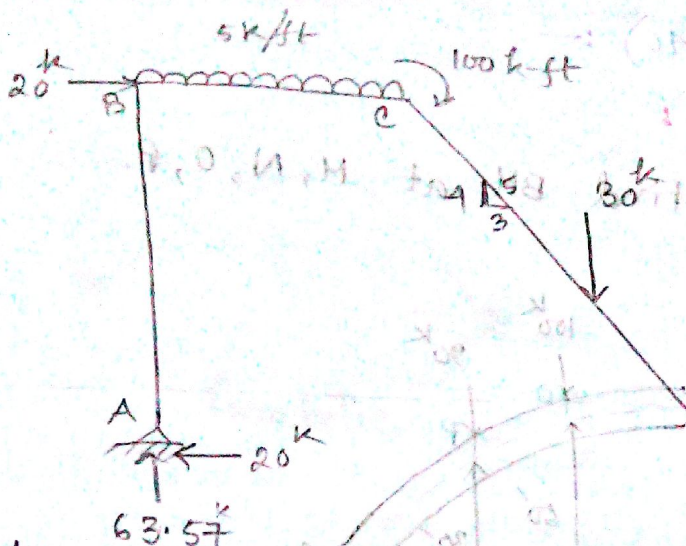
$$\Rightarrow R_{Ay} = 63.57 \text{ k} (\uparrow)$$

$$\sum F_x = 0$$

$$\Rightarrow -R_{Ax} + 20 = 0 \rightarrow R_{Ax} = 20 \text{ k}$$



H.W



$\Sigma M_A = 0 \quad \uparrow +ve$

$$\Rightarrow 20 \times 20 + 100 \times 10 + 30 \times 27.5 + 100 - R_D \times 35 = 0$$

$$\Rightarrow R_D = 66.4 \text{ k} \quad (\uparrow)$$

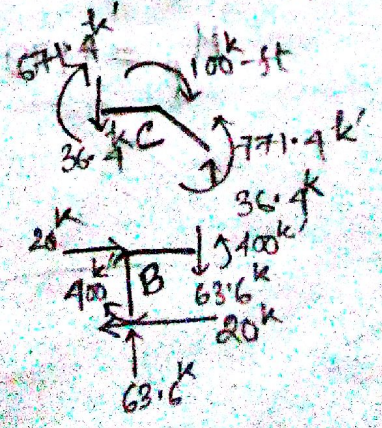
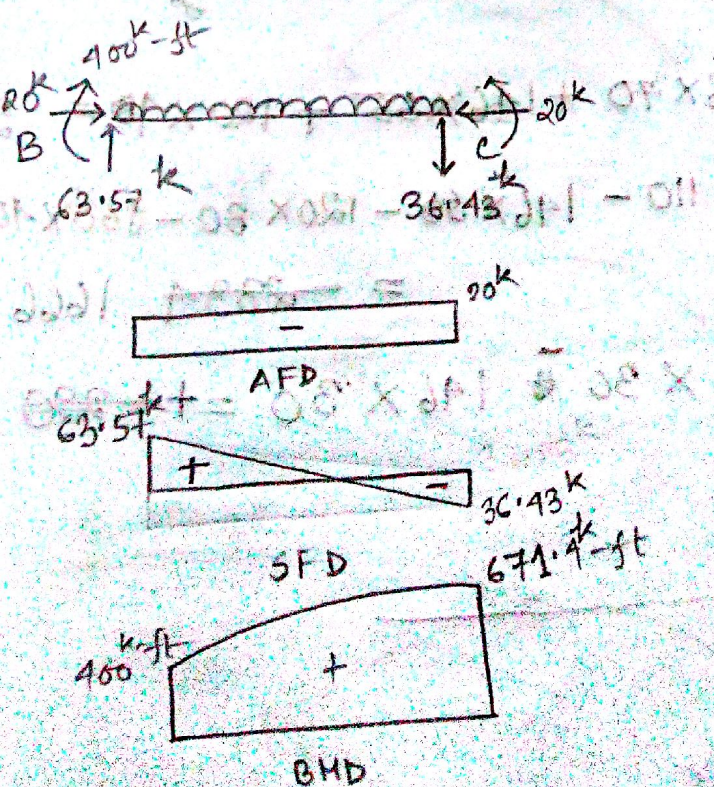
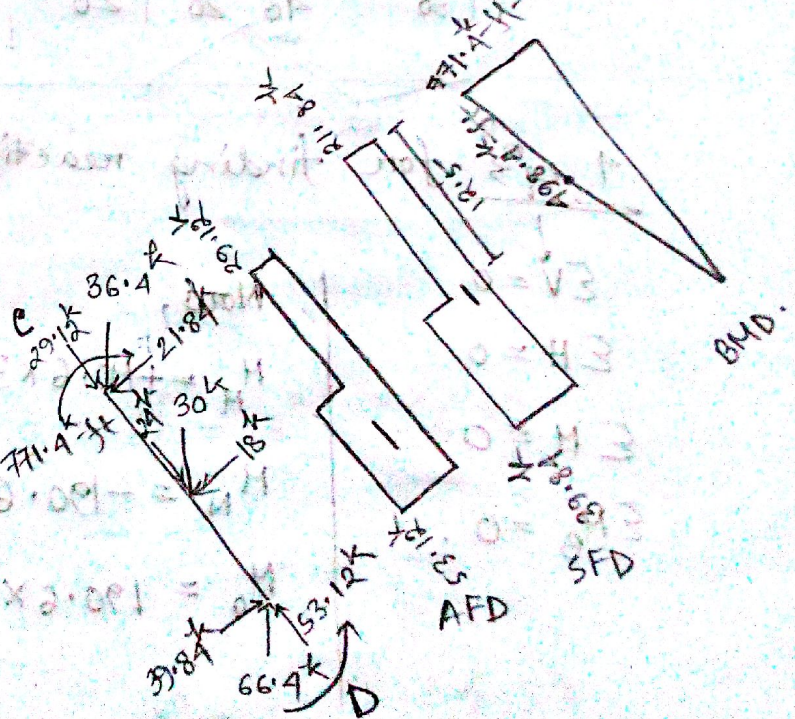
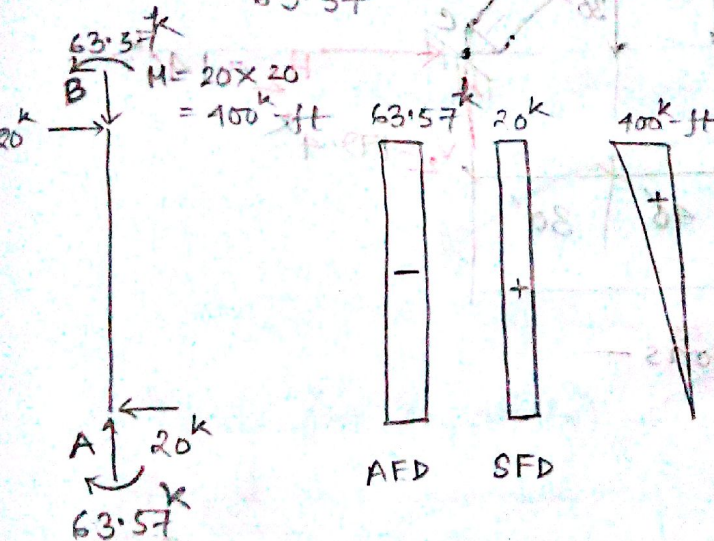
$\Sigma F_y = 0 \quad \uparrow$

$$\Rightarrow R_A + 66.4 - 100 - 30 = 0$$

$$\Rightarrow R_A = 63.57 \text{ k} \quad (\uparrow)$$

$\Sigma F_x = 0$

$$\Rightarrow -R_{Ax} + 20 = 0 \Rightarrow R_{Ax} = 20 \text{ k}$$

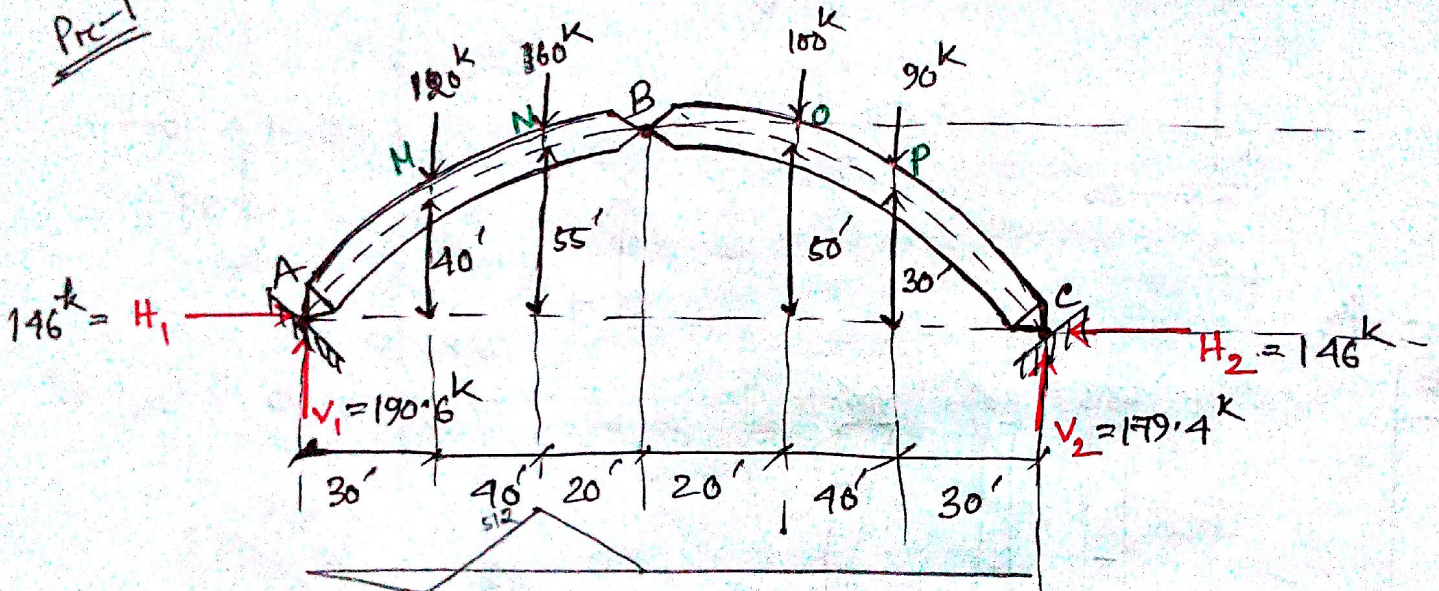


# Shed Vawter (Pg-46) :-

• Analysis of Arch :

Find BM at M, N, O, P.

Pr-1



4 eq<sup>s</sup> for finding reactions -

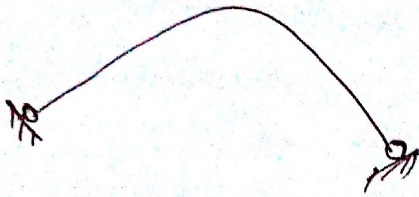
$\Sigma V = 0$   
 $\Sigma H = 0$   
 $\Sigma M = 0$   
 $\Sigma M_B = 0$

Now,

$M_M = +190.6 \times 30 - 146 \times 40 = -122$   
 $M_N = -190.6 \times 70 + 146 \times 55 + 120 \times 40 = 512$   
 $M_O = 190.6 \times 110 - 146 \times 50 - 120 \times 80 - 160 \times 40 = -2334$   
 $M_P = +179.4 \times 30 - 146 \times 30 = +100$



• Types of arches — (on the shape of axis)



Parabolic



Elliptical



Circular

• Types (on supports)

Fixed



$$D^{\circ} = 3^{\circ} \quad (6-3)$$

2-Hinge



$$D^{\circ} = 1^{\circ} \quad (4-3)$$

3-Hinge

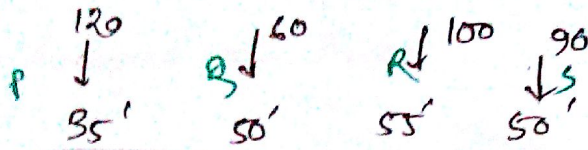


$$D^{\circ} = 0^{\circ} \quad (3-3)$$

Shedd Vawter - Pg-47.

Q. Find the BM at the loads

Pr-2



$$V_1 = 210^k, V_2 = 160^k$$

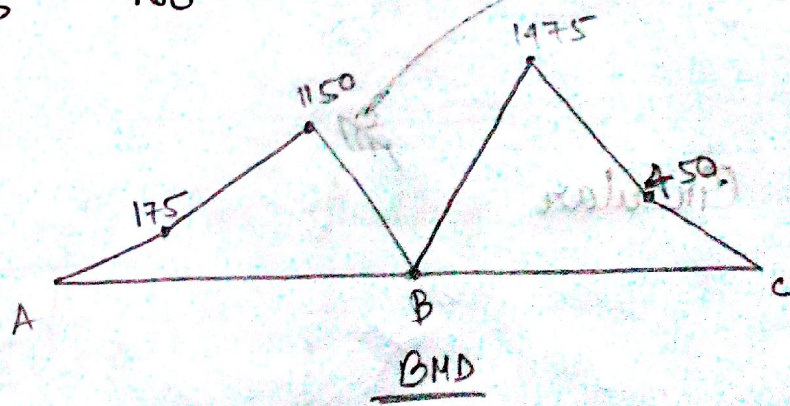
$$H_1 = 175^k, H_2 = 175^k$$

$$M_P = 210 \times 30 - 175 \times 35 = +175$$

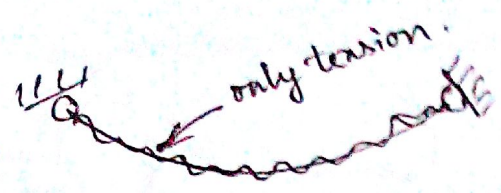
$$M_Q = 210 \times 70 - 175 \times 50 - 120 \times 40 = +1150$$

$$M_R = 210 \times 110 - 175 \times 55 - 120 \times 80 - 60 \times 40 = +1475$$

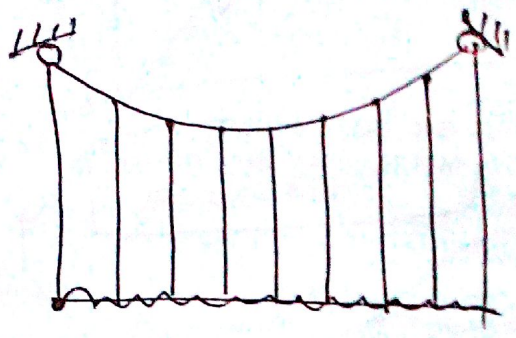
$$M_S = -160 \times 30 + 175 \times 30 = 450$$



Cable



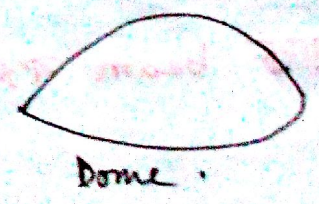
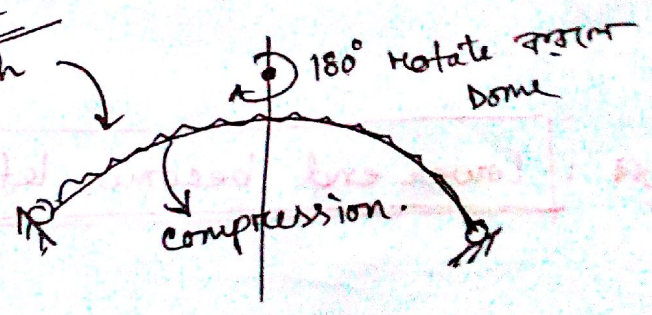
Cable shape  $\Rightarrow$  Catenary  
Fig-1



Cable shape  $\Rightarrow$  Parabola.  
Fig-2

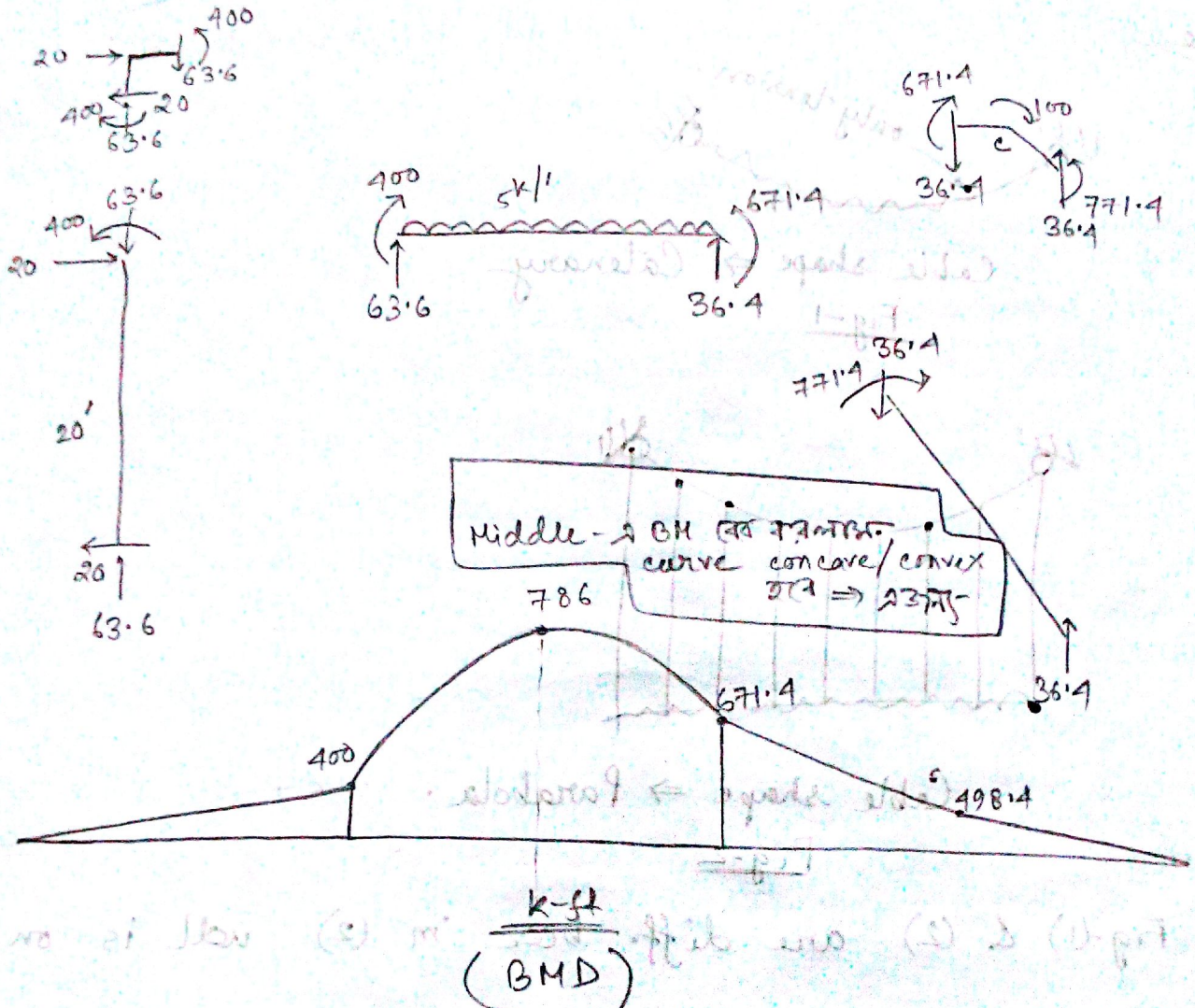
Fig-1) & (2) are diff. beca in (2) udl is on horizon

Cable = तार  
अभिनत  
Arch



⊛ Arch/Dome तार durable cz here no steel.  
 $\Downarrow$   
 But in Rec. steel will corrode after long time.

In all parts  $\sum F_x, \sum F_y, \sum M = 0$  must satisfy.



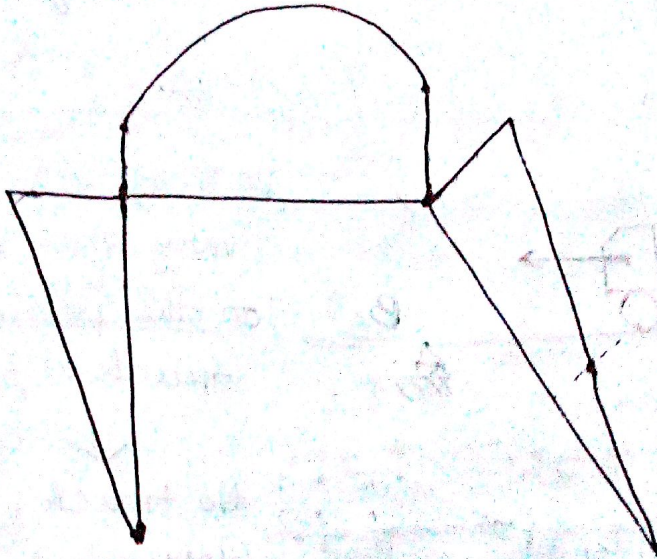
Column-के द्वारा beam करार | Lower end becomes left end.

BM (sign convention)

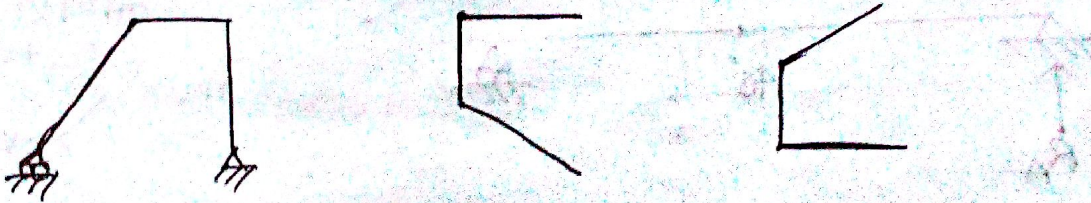
BM due to upward load is +ve  $\Rightarrow$  (in Beam)

9.9.15

Combined BMD



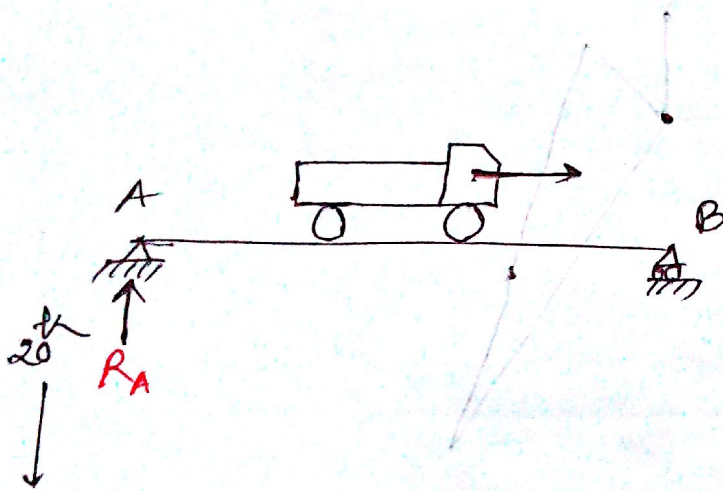
Problems



(BMD) for a beam

(Shedd Vauter)  $\rightarrow$  Pg (100)

Influence Lines: (for internal forces under moving loads)



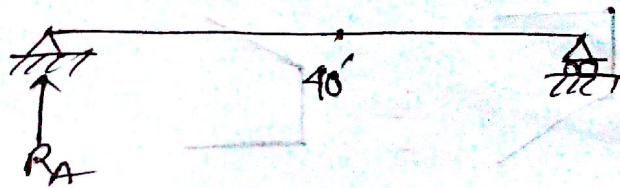
Let  $R_A$  value vary depending on the position of the truck.

No truck,  $R_A = 0$   
closest to A,  $R_A = \max^m$

So design

max<sup>m</sup> & min<sup>m</sup>

max<sup>m</sup>  $\rightarrow$  +ve (upward)  
min<sup>m</sup>  $\rightarrow$  -ve (downward)

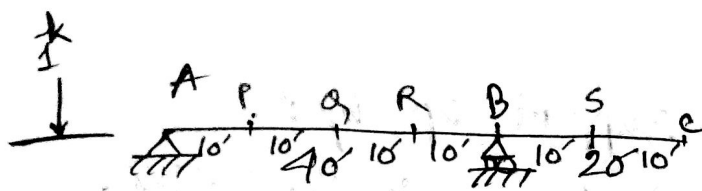


IL is drawn for a unit load (1 lb / 1 k / 1 kg)

⊛ for variation of loads, IL is drawn.

⊛ Our Aim  $\Rightarrow$  how to draw Influence Lines and how to work with IL.

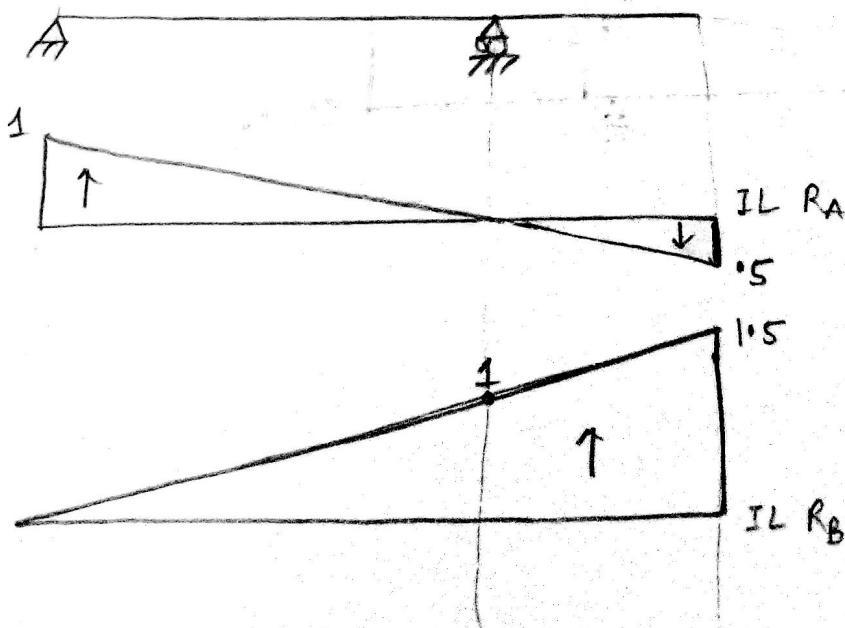
I-Line (This is crude method)



unit (k/k or #/# or kg/kg)  
 Draw I-Line for  $R_A$  &  $R_B$ .

I load at

	<u><math>R_A</math></u>	<u><math>R_B</math></u>
A	1 ↑	0
P	.75 ↑	.25 ↑
Q	.5 ↑	.5 ↑
R	.25 ↑	.75 ↑
B	0	1 ↑
S	.25 ↓	1.25 ↑
e	.5 ↓	1.5 ↑

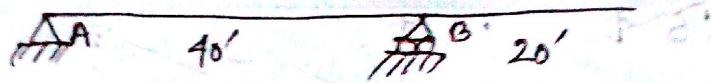


# # Influence line for beams - (to draw)

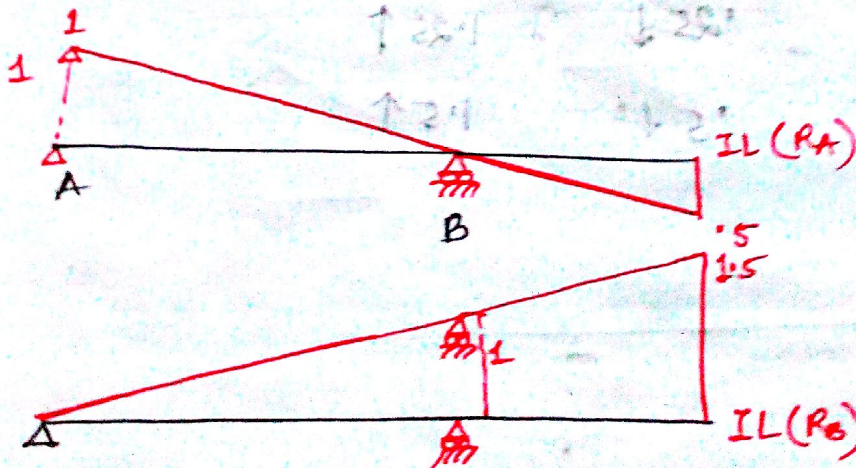
## 1) Reactions -

- i. Push the support up by 1.
- ii. Draw the S-shape - is the I-Line.

Beam-1

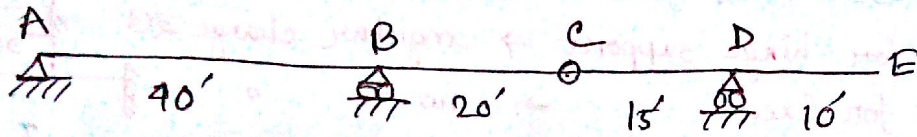


for drawing  $R_A$ , we will push A support.



Beam-2

Here 4 reactions, but for IL, load is only one i.e. (1k)  
 (normally this unit load is vertical. But sometimes it can be hori. or inclined)

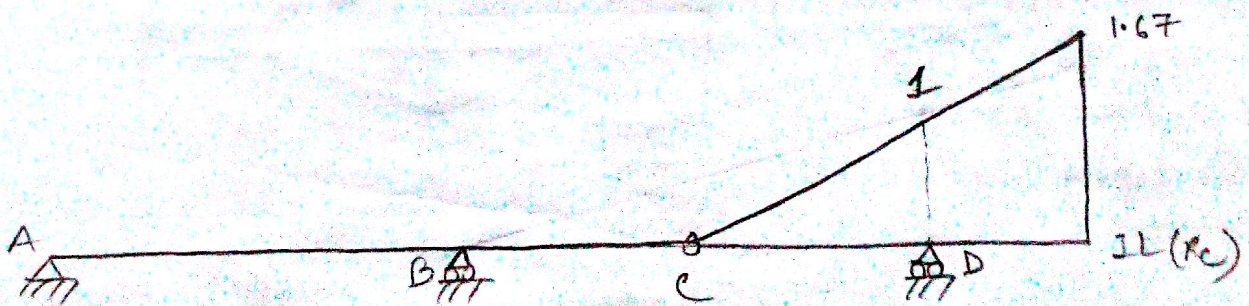
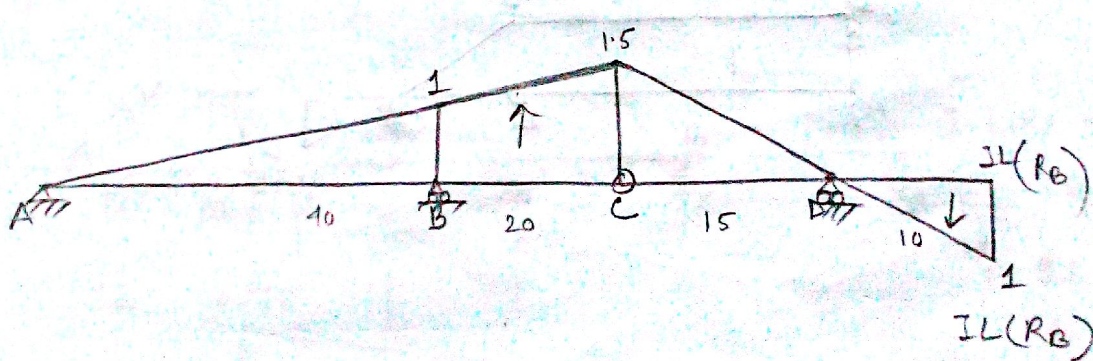
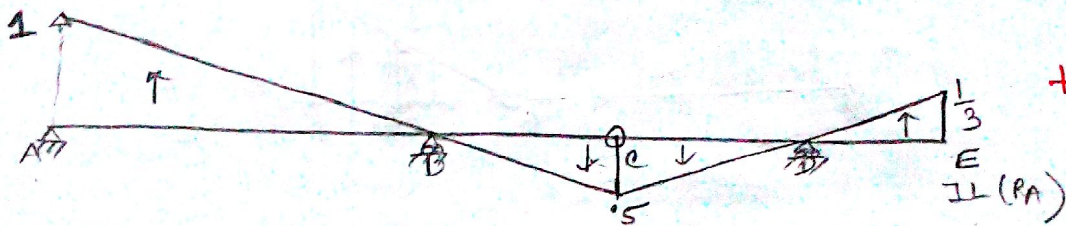


But for only ver. load no  $R_{Ax}$   
 So the reactions are  $R_A, R_B, R_D$

original beam - 1  
 दोस्रोत straight  
 line दोस्रोत IL  
 राखि

for this beam  
 two straight  
 lines.

two sides of  
 the hinges.

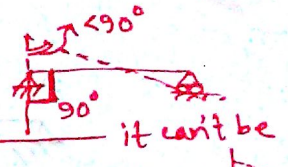


Chp-1 (from book for practice)

# Influence Line for reactions in Beams -

for hinge support  $\Rightarrow$  angular change  $\Rightarrow$   $90^\circ$

but for fixed  $\Rightarrow$  no " "  $\Rightarrow$   $90^\circ$

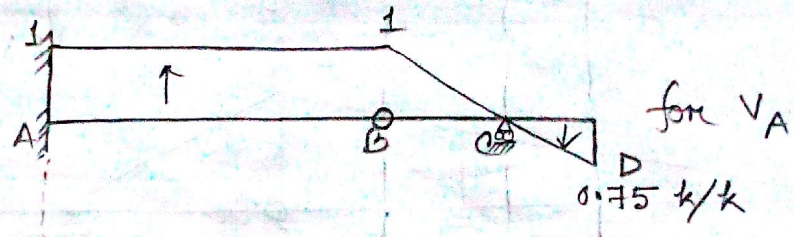
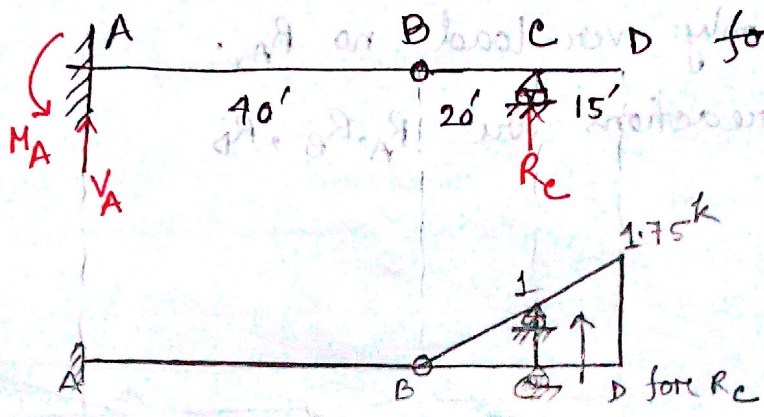


Draw I. Lines

for  $V_A, R_C, (M_A \text{ later})$

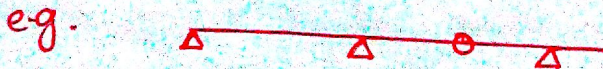
Beam-3

i.e.  $90^\circ$  &  $90^\circ$  should be maintained.



Step

Alltime first - L stable part draw করতে হবে।



এই left side stable so left first draw



" right " " so right draw

# \* I-Lines for shear force in beams -

Steps -

1) Cut at the section X

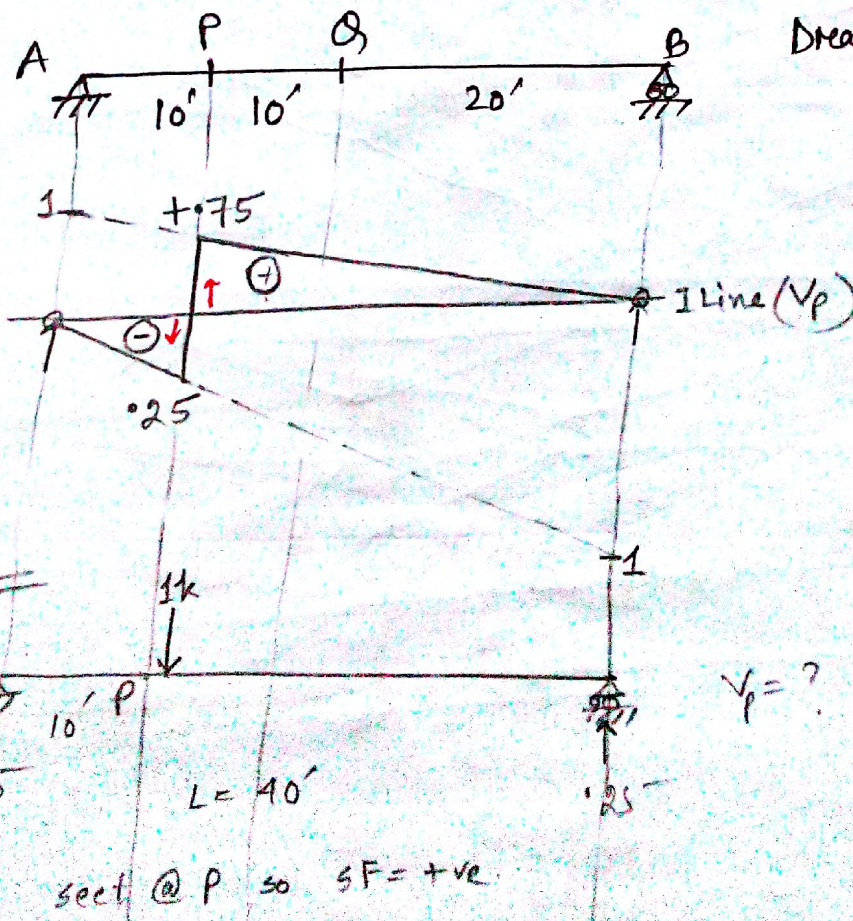
2) Push the left part ~~up~~ <sup>down</sup> and right part ~~down~~ <sup>up</sup>.

(by two sides of X)

3) Draw  $\delta$ -shape  $\rightarrow$  is the I-Line

4) Put the ordinates and sign.

Beam-1



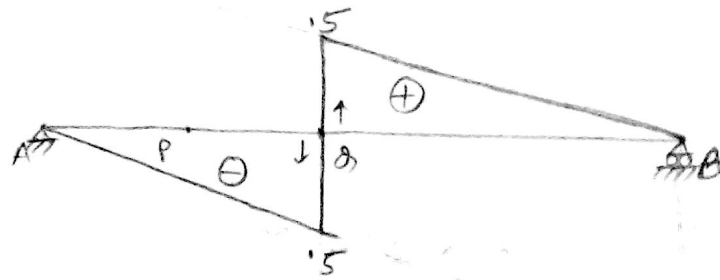
Draw I-Line for  $V_{A-Rt}$ ,  $V_p$ ,  $V_q$  &  $V_{B-Rt}$ .

$\downarrow$   
 exactly support  $\rightarrow$  1  
 SF  $\rightarrow$  2.25 at B  
 so either left/right draw  
 बराबर रख।

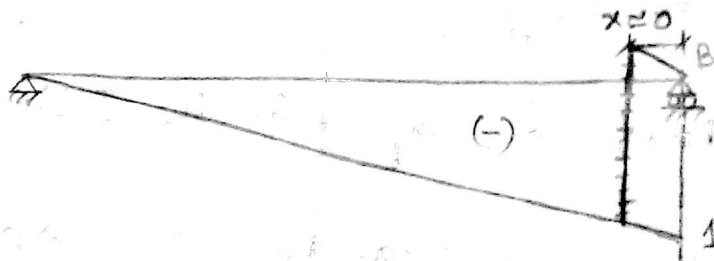
for sign

$V_p = ?$

sect @ P so SF = +ve

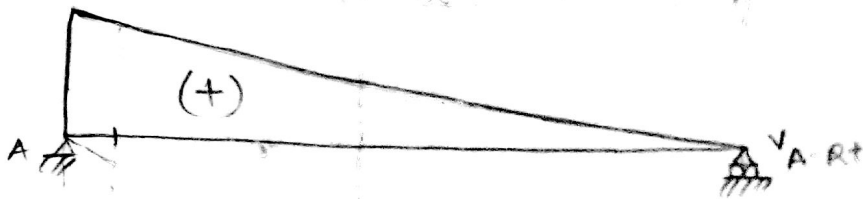


I Line (VQ)



I Line  $V_e - I_f$

अध्यात (सकार) अवधि  
 को  $x$  एक दूर  
 कम मात्र but it  
 tends to 0

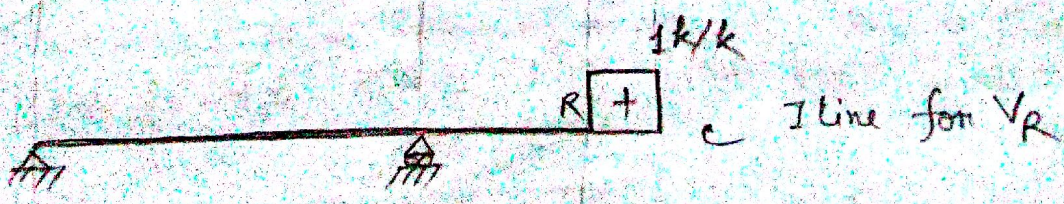
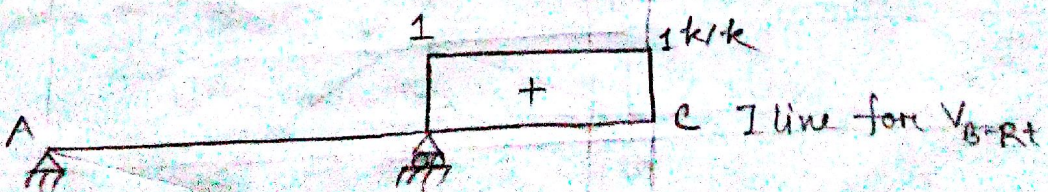
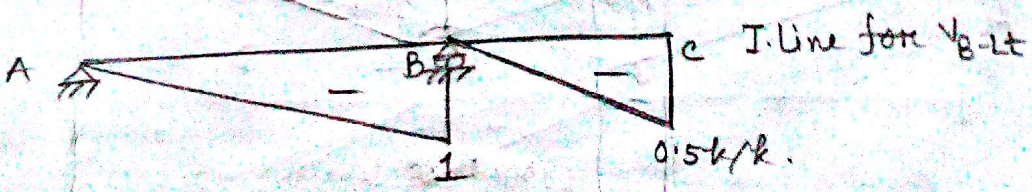
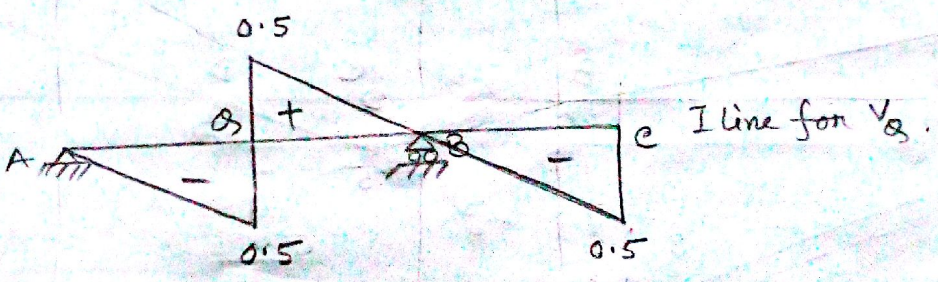
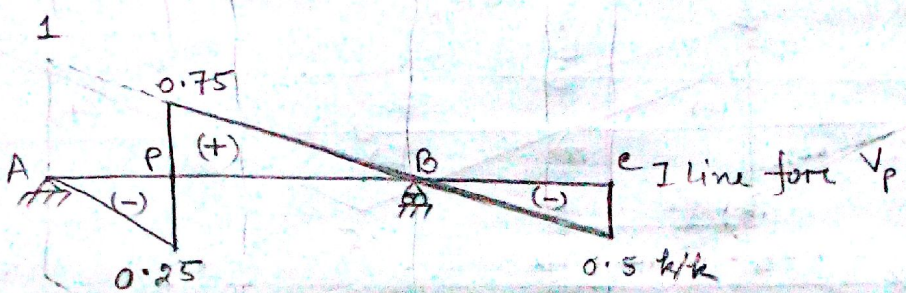
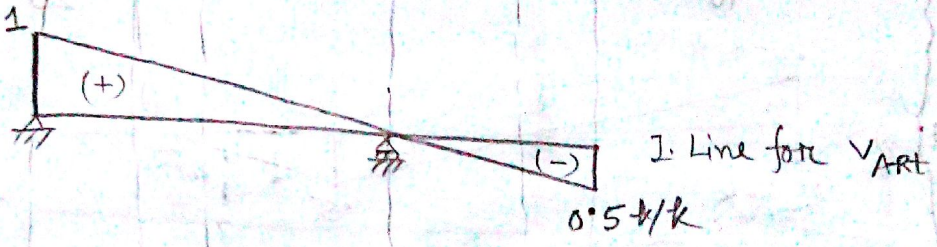
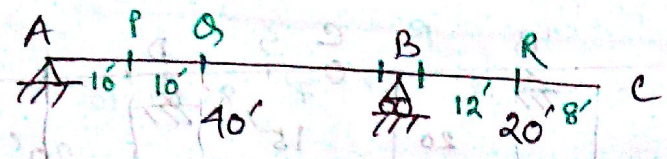


↓  
 So only किस  
 line के कारण

Beam-2

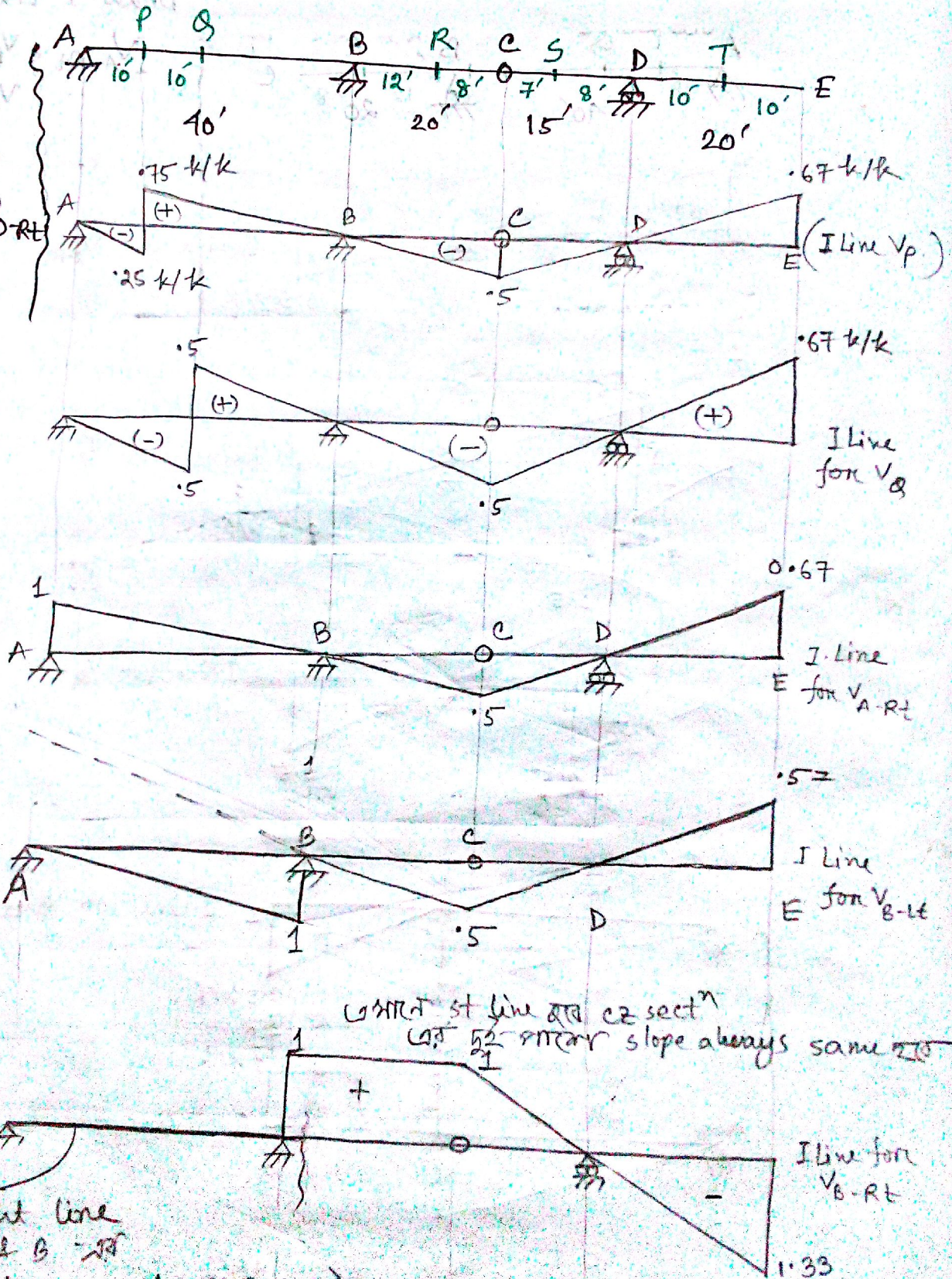
Draw I-Line for

$V_{A-RT}$ ,  $V_P$ ,  $V_Q$ ,  $V_{B-LT}$ ,  
 $V_{B-RT}$ ,  $V_R$



Beam-3

$V_{A-RT}, V_P, V_Q,$   
 $V_{B-Lt}, V_{B-RT},$   
 $V_R, V_S, V_{D-Lt}, V_{D-RT}$   
 $V_T$



actually dotted line is portion

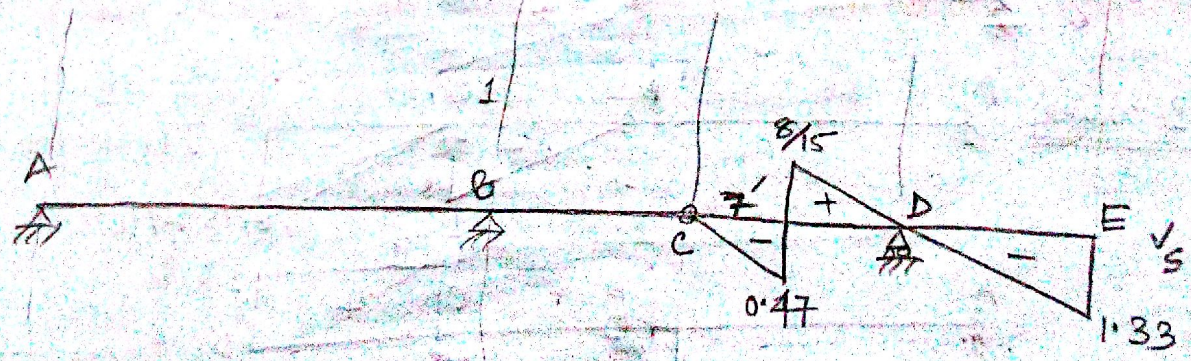
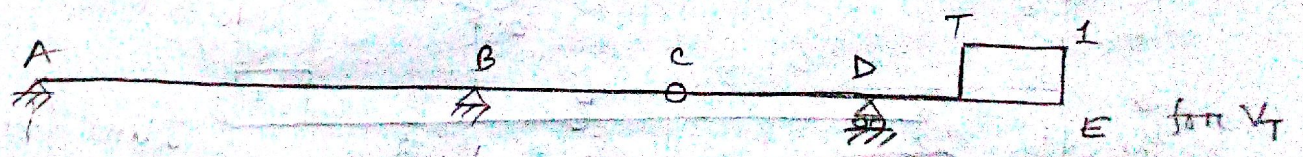
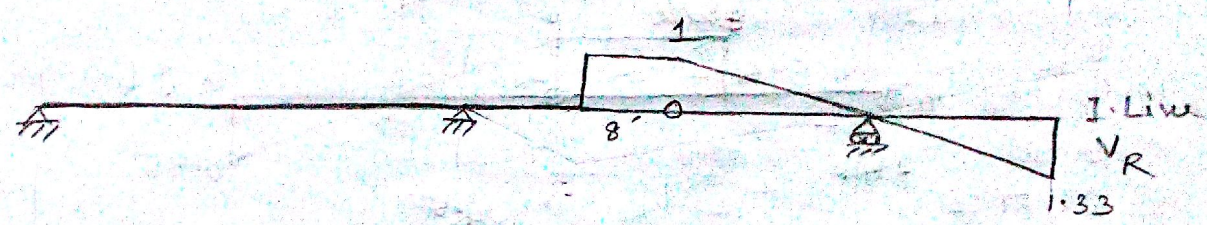
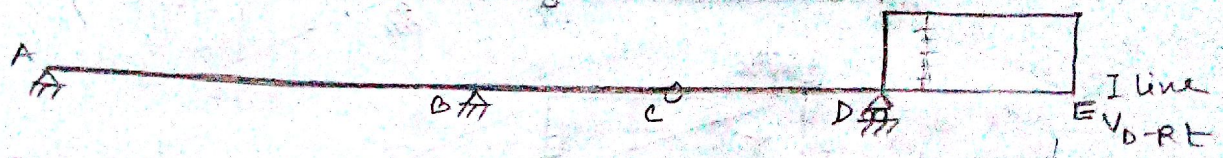
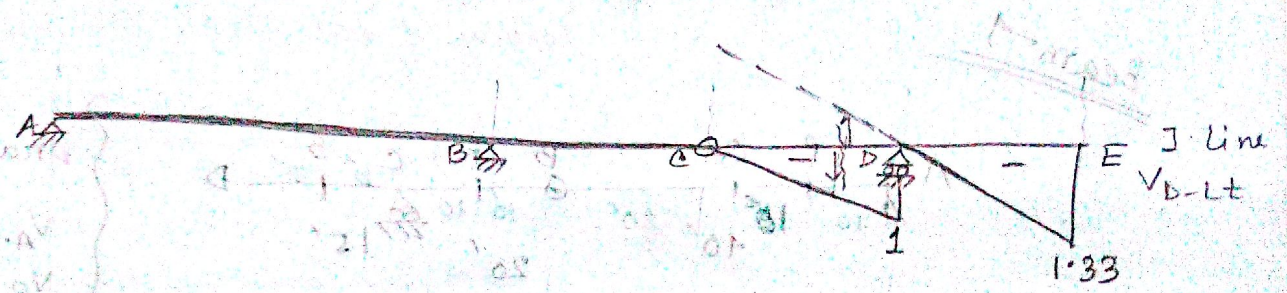
constant slope always same

straight line at A & B  
 support change

and constant slope always same EF of left & Right diff is 1

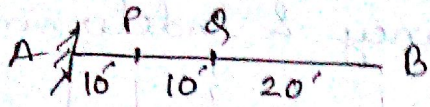
$$\sum F = (F_{right} + F_{left})_{section} = 1$$

right line  
 total  
 joint  
 high  
 low  
 high  
 low





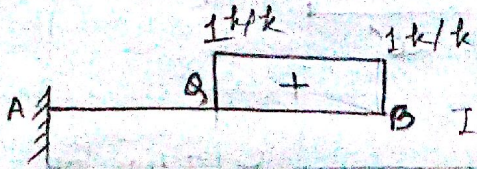
Beam 5



I Line for  $V_{AP}$



I Line for  $V_P$



I Line for  $V_B$

\* To draw I line for shear force boundary conditions —

1. Slope same ~~एक~~ section - ~~एक~~ ~~दो~~ side -  $\hookrightarrow$
2. ~~दो~~ side -  $\hookrightarrow$   $sum = 1$  ~~एक~~

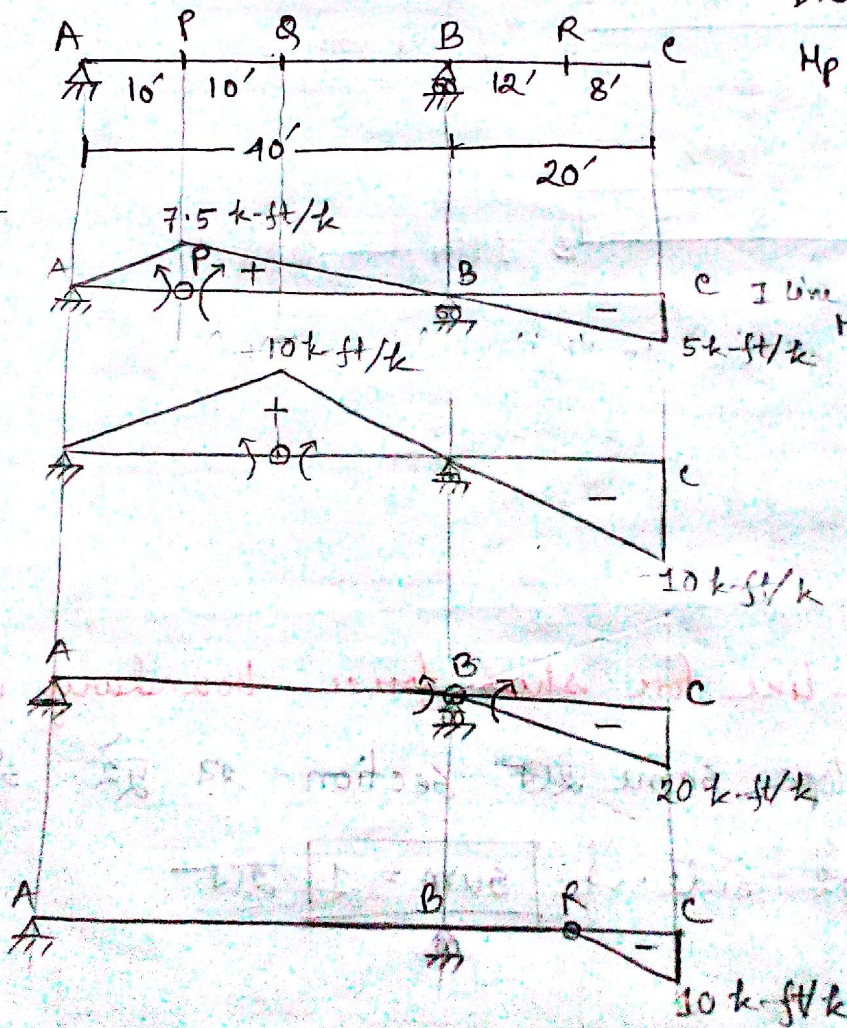


Class Test → Determinancy & Indeterminancy, Stability  
 Wednesday

# I Lines for Bending Moments in Beams :-

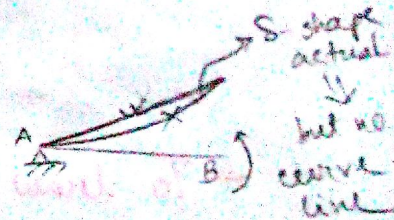
Beam 2

[ I line for BM - प्रकृत के point - I line दबकार अमान HINGE [0] आवा ]



Draw I lines for  $M_p, M_q, M_B, M_R$ .

\* In original beam  
 ↓  
 I straight line  
 ↓  
 but hinge अमान 2 parts. So I line अमान 2 lines.

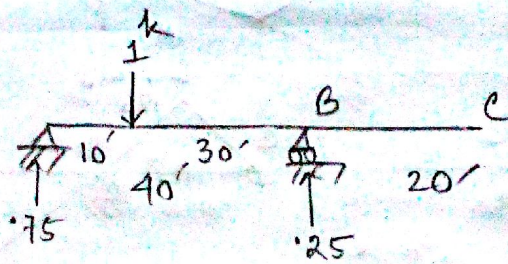


Steps — I line for BM at X →

1. Put a hinge at X
2. Apply B. Moments ↷ ↶
3. Draw S-shape
4. Find the ordinates

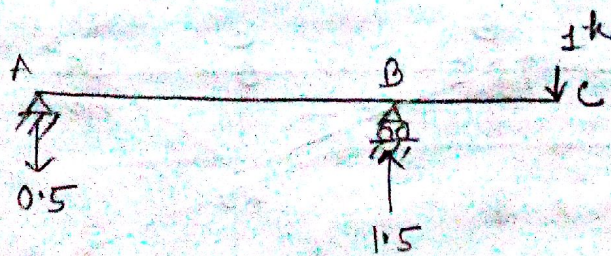
• Calculation of ordinates at P for  $M_p$  —

Place unit load at P and calculate  $M_p$ .



$$M_p = \underline{\underline{7.5 \text{ k}'}}$$

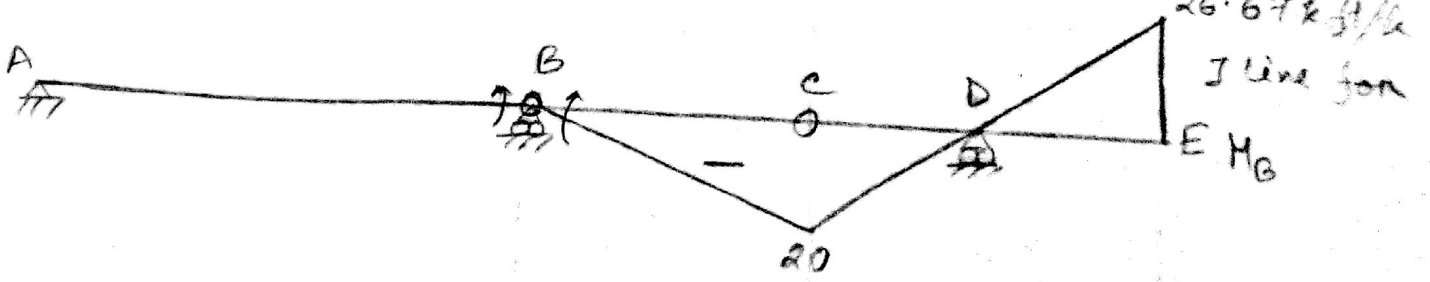
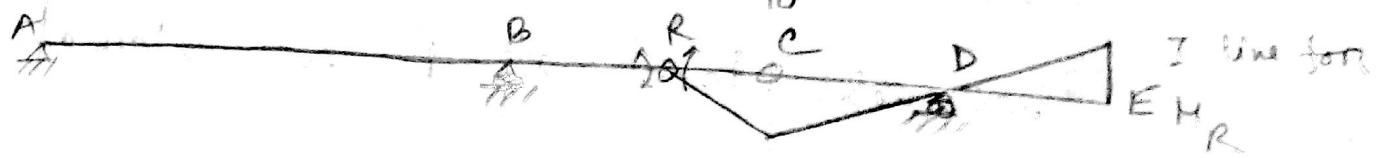
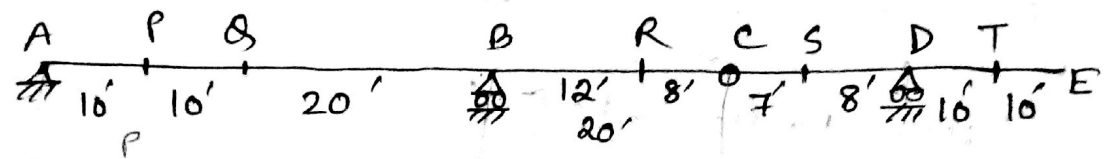
for  $M_A$  &  $M_B$



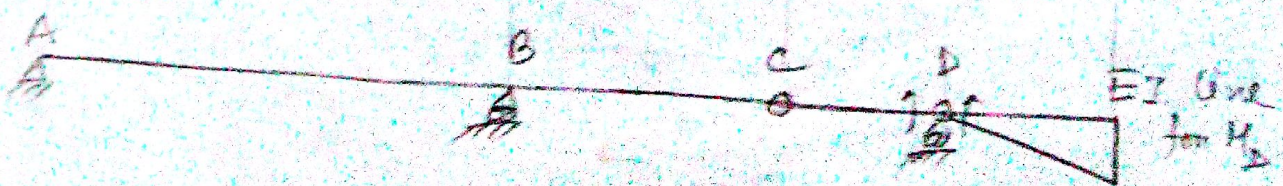
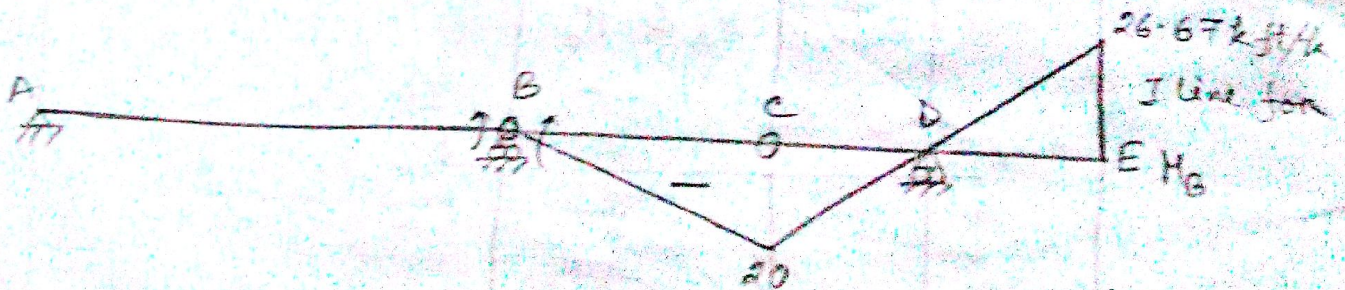
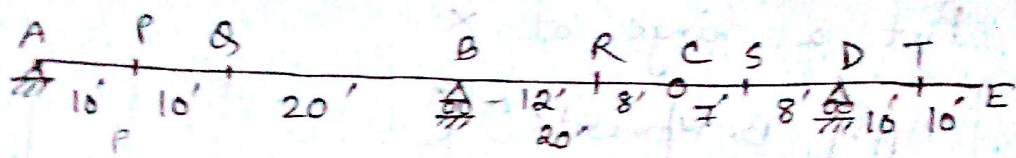
$$M_A = -10$$

$$M_B = -20$$

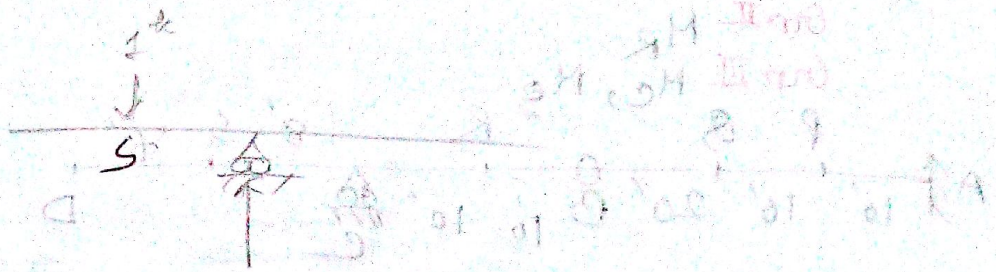
Beam-3 Draw I Lines for BM at  $M_P, M_Q, R, S, T, B, M_D$ ,



Beam-3 Draw I lines for BH at  $H_P, H_B, H_R, H_S, H_T, H_B, H_D,$



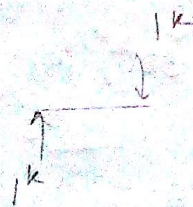
13/10/22



$$1 \times 7 - R \times 15 = 0$$

$$\Rightarrow R = \frac{7}{15}$$

$$\therefore M = \frac{7}{15} \times 8.$$



12.10.15

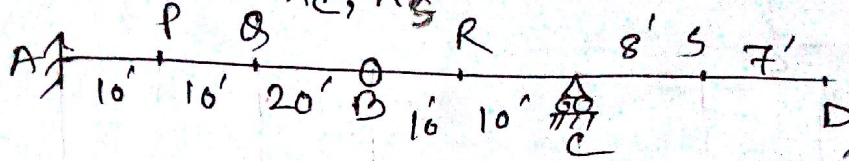
Draw I. line for

Grp I  $M_A - M_p, M_B$

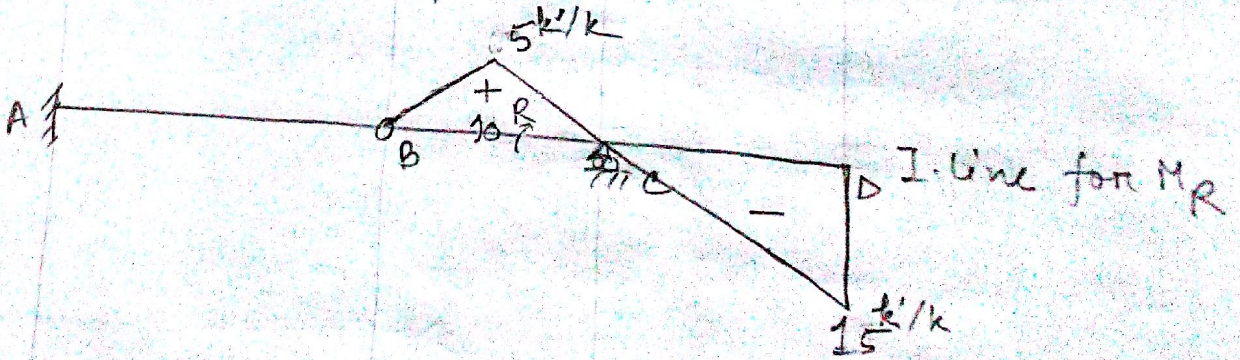
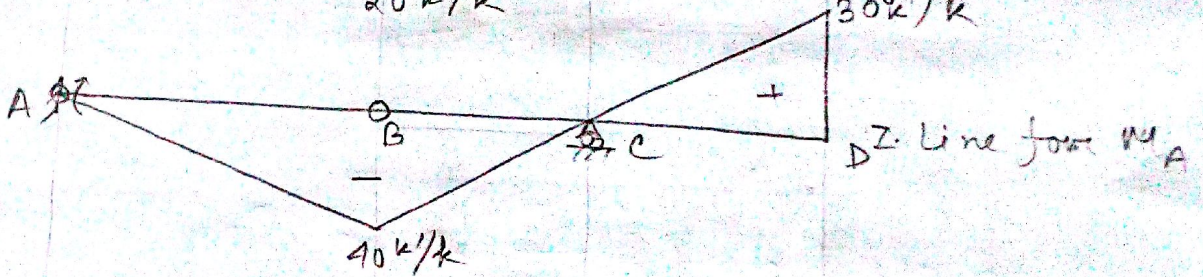
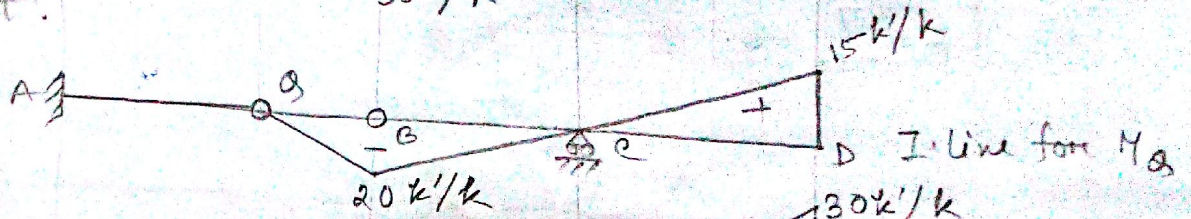
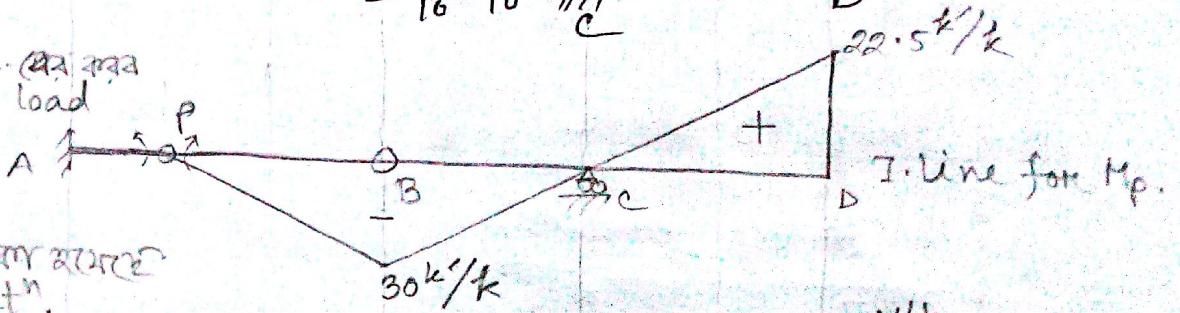
Grp II  $M_R$

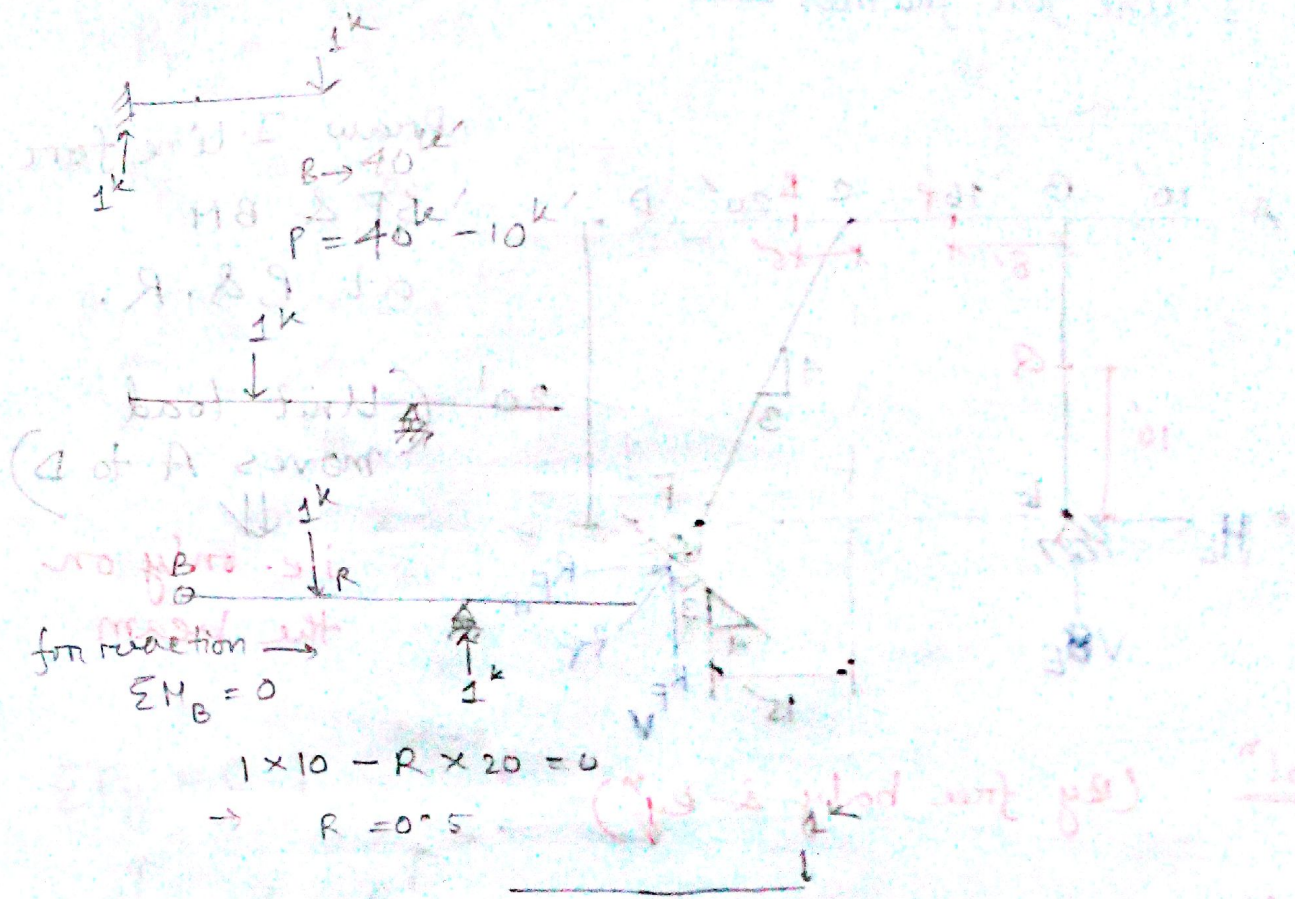
Grp III  $M_c, M_D$

Beam - 1



or ordinate  $\frac{P}{2}$  unit load  
 फिर I  
 ↓  
 अर्थात् I. line काटने वाले  
 समतल सेक्शन.



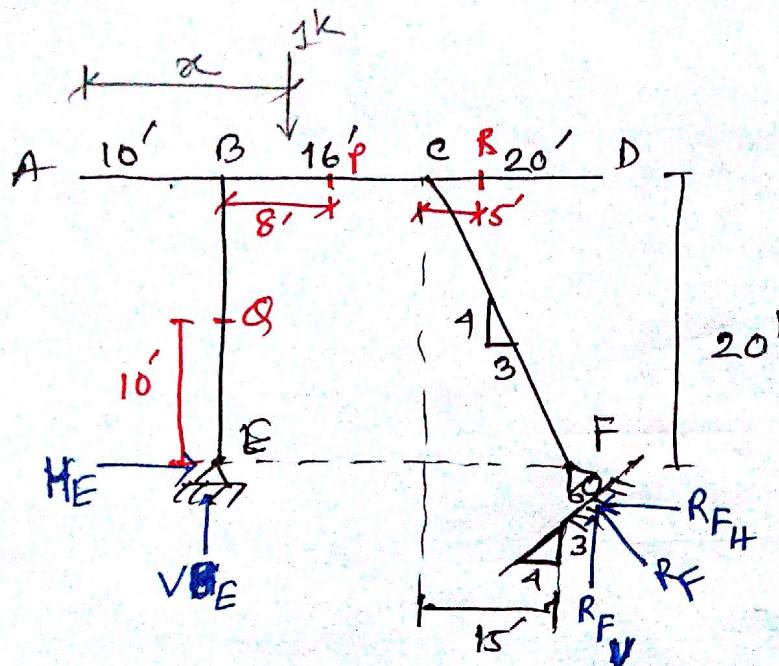


$\Sigma M_B = 1 \times 35 - R \times 20 = 0$

Find eqs for  $R_2$  — a beam like for  $R_2$ .  
 Find eqs for unknown functions.  
 (i.e.  $\Sigma F_{EM}$ )

$\Sigma M_B = 0 = 1 \times 35 - R \times 20 = 0$   
 $\frac{1 \times 35}{20} = R$   
 $R = 1.75$

## # I-line for frames —



Draw I-line for SF & BM at P, Q, R.

(Unit load moves A to D)  
 ↓  
 i.e. only on the beam

Sol<sup>n</sup> (By free body & eq<sup>n</sup>)

### # Steps

1. Find eq<sup>n</sup>s for R's — & draw I-line for R's.
2. Find eq<sup>n</sup>s for required functions.  
 (i.e. SF & BM)

\* For eq<sup>n</sup> ⇒ assume unit load is somewhere at a distance x.

$$\begin{aligned} \sum M_E = 0 \quad \downarrow + \quad & \Rightarrow 1 \times (x-10) - R_{FV} \times 31 = 0 \quad \Rightarrow \text{eq}^n \rightarrow \text{direct}^n \\ & \Rightarrow R_{FV} = \frac{x-10}{31} \quad \left. \begin{array}{l} x = 46 \text{ at D} \\ x = 0 \text{ at A} \end{array} \right\} \end{aligned}$$

$$\Rightarrow \frac{R_{FV}}{R_{FH}} = \frac{1}{3}$$

$$\therefore R_{FH} = \frac{R_{FV}}{4} \times 3$$

$$\Rightarrow R_{FH}^{\leftarrow} = \frac{x-10}{31} \times 0.75 \Bigg)^{46}_{x=0}$$

$$\text{By } \Sigma F_x = 0 \rightarrow$$

$$H_E^{\leftarrow} = -R_{FH}^{\leftarrow}$$

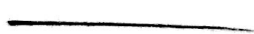
$$\Rightarrow H_E^{\leftarrow} = \ominus \frac{x-10}{31} \times 0.75 \Bigg)^{46}_{x=0}$$

$$\Sigma F_y = 0 \uparrow$$

$$\Rightarrow V_E^{\uparrow} = 1 - R_{FV}^{\uparrow}$$

$$\Rightarrow V_E^{\uparrow} = 1 - \frac{x-10}{31} \Bigg)^{46}_{x=0}$$

Then Draw the I-Line for AD  $\Rightarrow$  (x-AB value 33.75  
ordinate 0.75)

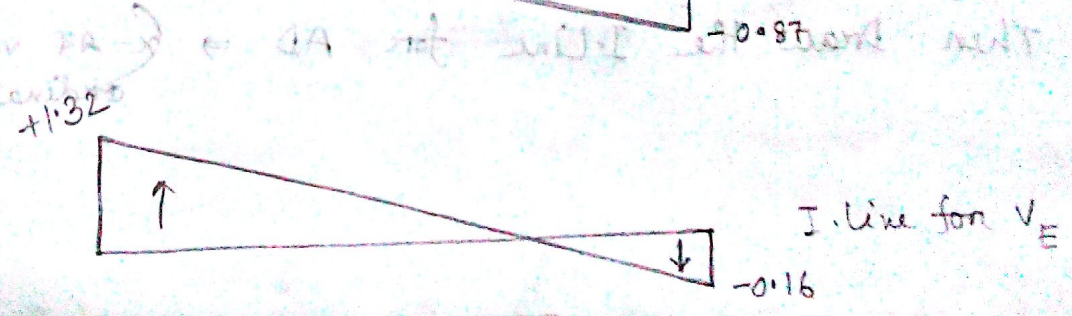
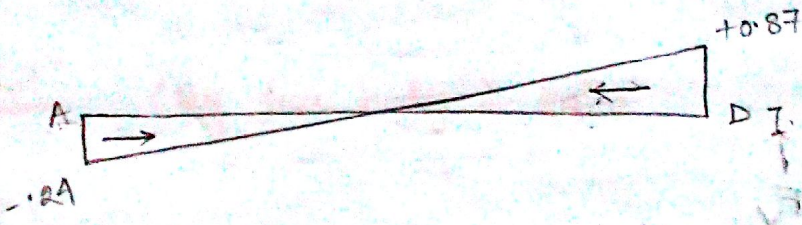


14.10.13

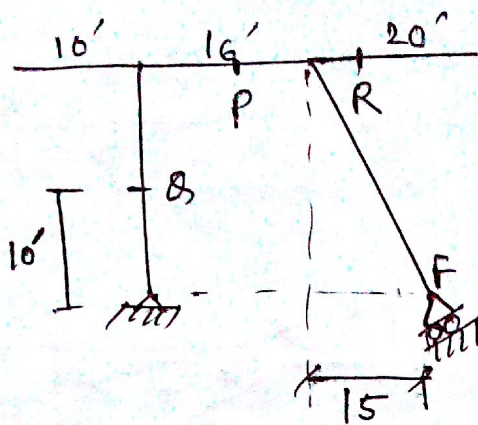
Eqs of reactions

$$R_{FV}^{\uparrow} = \frac{x-10}{31} \Big|_{x=0}^{46} ; R_{FH}^{\leftarrow} = 0.75 \frac{x-10}{31} \Big|_{x=0}^{46}$$

$$H_E^{\leftarrow} = -0.75 \frac{x-10}{31} \Big|_{x=0}^{46} ; V_E^{\uparrow} = 1 - \frac{x-10}{31} \Big|_{x=0}^{46}$$



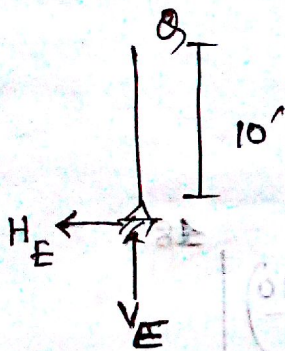
total area for  $R_{FV}$  = 0.4  
 (area shaded)



I-Line for shear force & BM at Q  $\Rightarrow$

1. First section at Q

(unit load - 23 path karo aur  
 aur na so karo  
 = eqn 23) = V



$$\therefore V_Q^+ = H_E = -0.75 \frac{x-10}{31} \Big|_{x=0}^{46}$$

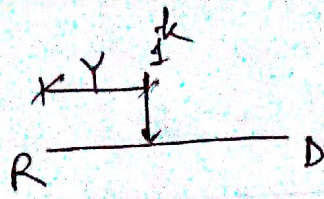
$$\& M_Q^+ = H_E \times 10 = -10 \times 0.75 \frac{x-10}{31} \Big|_{x=0}^{46}$$

I-Line for SF, BM at R  $\Rightarrow$

• unit load - 23 path karo aur karo. So. 23 ko karo karo  
 23  $\Rightarrow$

1. Unit load A-R  $\Rightarrow$  Consider Rt part R-D  $\Rightarrow V_R=0, H_R=0$
2. " " R-D

Consider also Rt part ez only 1 unit load is there.



$$\therefore V_R^+ = 1$$

$$M_R^- = 1 \cdot Y \Big|_{Y=0}^{15}$$

I. line for SF, BM at P  $\Rightarrow$

1. Unit load A-P

(Consider Rt part)

2. Unit load P-D

(Consider left part)

$$V_P^- = R_{FV}^{\uparrow} = \frac{x-10}{31} \Big|_{x=0}^{18}$$

$$M_P = 23 R_{FV}^{\uparrow} - R_{FH}^{\leftarrow} \times 20$$

$$= \left( 23 \frac{x-10}{31} - 20 \times 0.75 \frac{x-10}{31} \right) \Big|_{x=0}^{18}$$

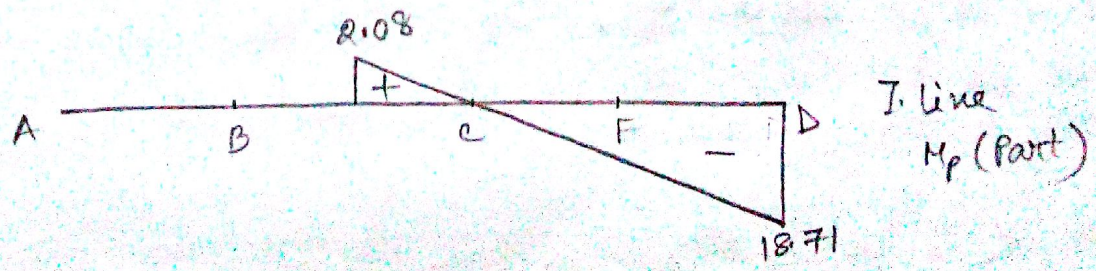
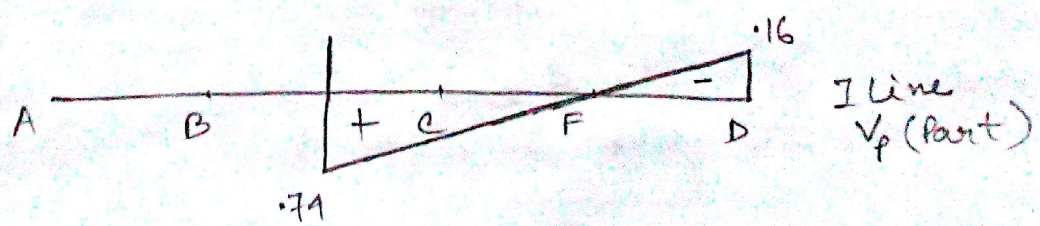
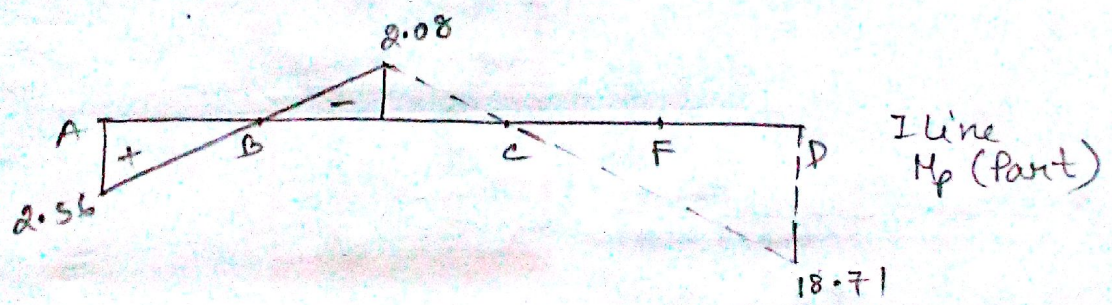
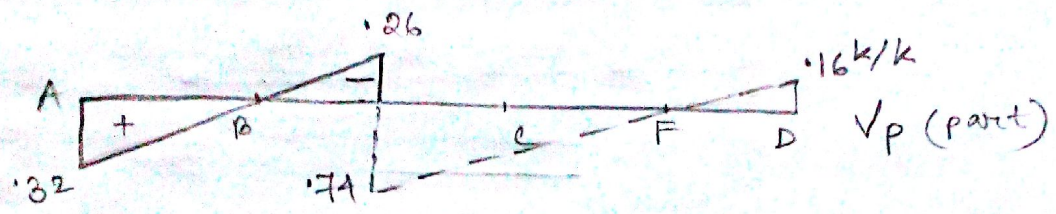
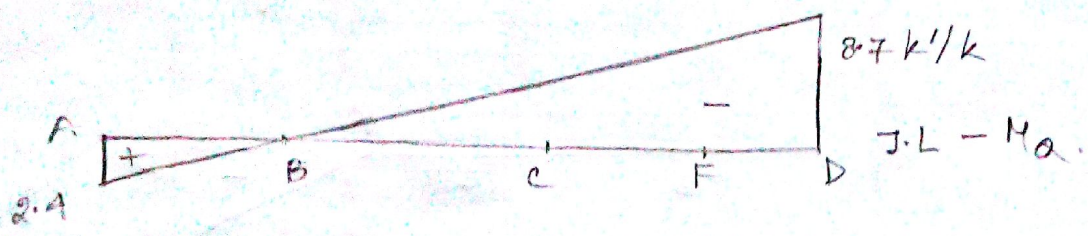
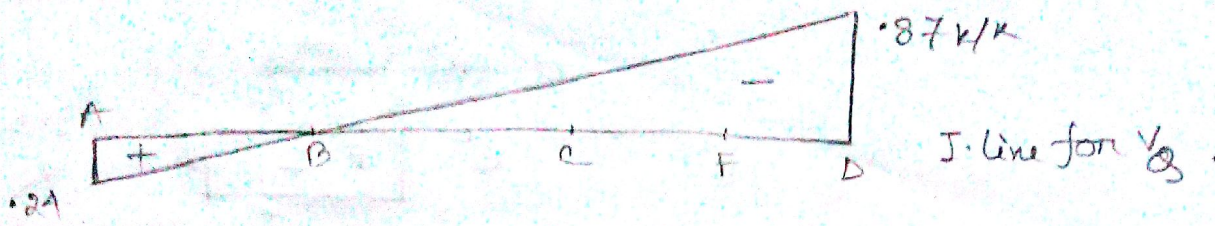
$$= 8 \frac{x-10}{31} \Big|_{x=0}^{18}$$

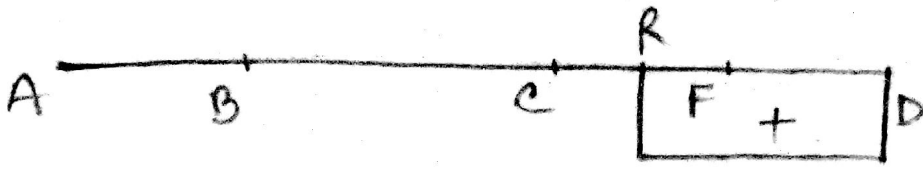
2. Unit load P-D

$$V_P^+ = V_E^{\uparrow} = 1 - \frac{x-10}{31} \Big|_{x=0}^{28}$$

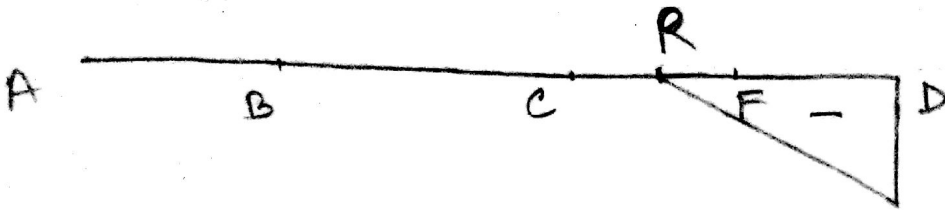
$$M_P = 8 V_E^{\uparrow} - 20 H_E^{\leftarrow} = 8 \left( 1 - \frac{x-10}{31} \right) - 20 (-0.75) \frac{x-10}{31} \Big|_{x=0}^{28}$$

$$\therefore M_P = 8 + 7 \frac{x-10}{31} \Big|_{x=0}^{28}$$





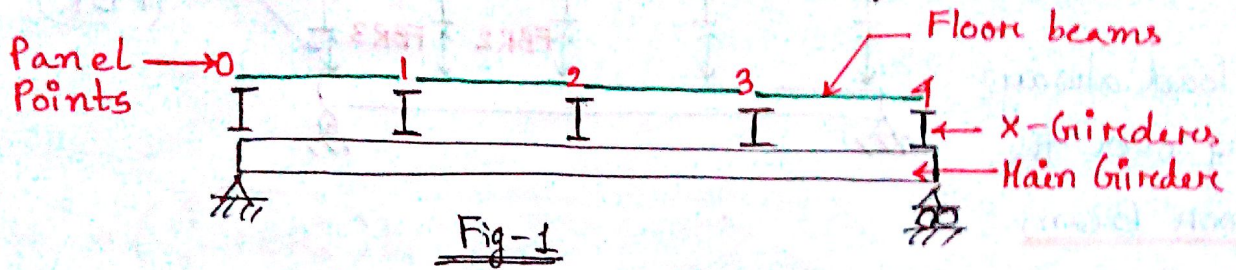
J. line  
 $V_R$



J. line  
 $M_R$

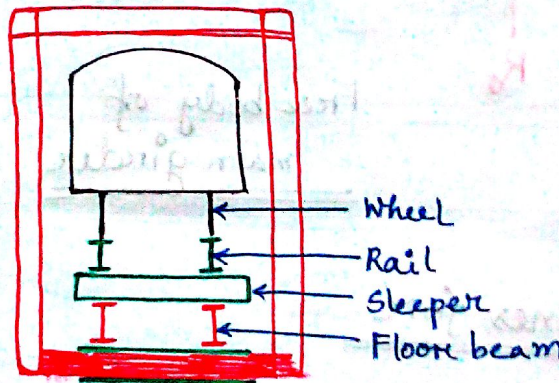


I-line for  
Girders with floor beams



Notes

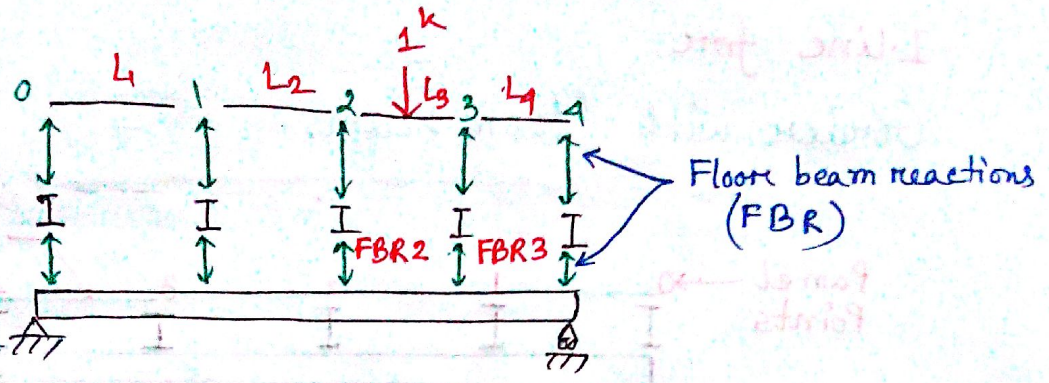
1.6.4 Bracing Systems.



\* truss only joint - 4 force or बल का 4 member - 3  
middle - 1 load पड़ने bending है ।

↓  
To avoid this we need floor system. to take the load only to the joints of the bottom chord

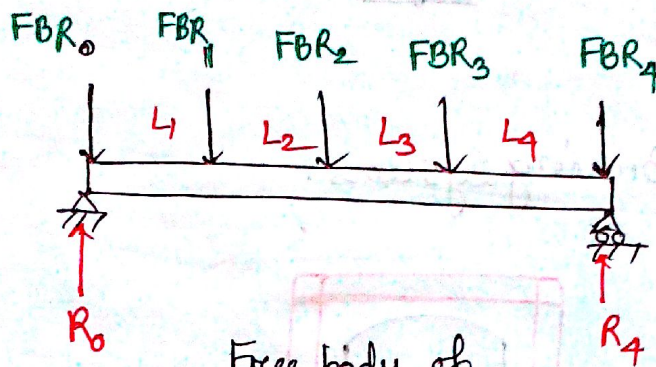
Free body of fig-1



Unit load always moving over the floor beam

↓

It will never directly come to the girder/main girder.



Free body of main girder

0 & 4 Panel Points at girder (unit load always constant)  
 ↓  
 So SF at every point is always constant

• Draw I-lines for

1. FBR's
  2.  $R_0$  &  $R_4$
- > as for R's in beams.

3. SF within Panel Points ( $V_{0-1}, V_{1-2}, V_{2-3}, V_{3-4}$ )

1. BM at PP's ( $M_1, M_2, M_3$ ) → BM always change between two successive panel points.  
 from freebody:

$M_0$  &  $M_4$  ⇒ Iline is zero line cz at 0 & 4,  $M=0$ .

SF eq<sup>n</sup>s :  $V_{0-1}^+ = R_0^{\uparrow} - FBR_0^{\downarrow}$

$V_{1-2}^+ = R_0^{\uparrow} - FBR_0^{\downarrow} - FBR_1^{\downarrow} \Rightarrow$  sect<sup>n</sup> काटेर bet<sup>n</sup> ① & ②

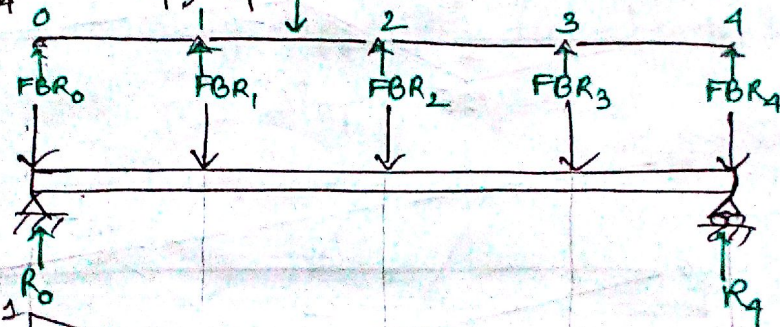
$V_{3-4}^- = R_4^{\uparrow} - FBR_4^{\downarrow} \Rightarrow$  sect<sup>n</sup> काटेर bet<sup>n</sup> ③ & ④

BH eq<sup>n</sup>s :

$M_1^+ = (R_0^{\uparrow} - FBR_0^{\downarrow}) L_1$

$M_2^+ = (R_0^{\uparrow} - FBR_0^{\downarrow})(L_1 + L_2) - FBR_1^{\downarrow}(L_2)$

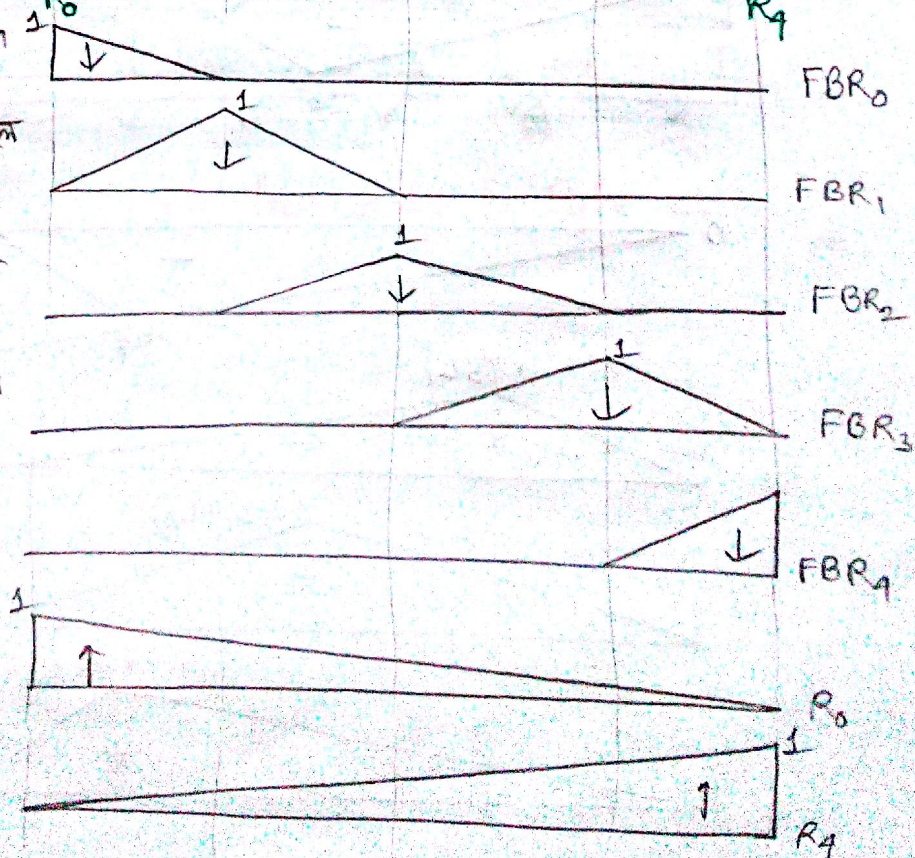
$M_3^+ = (R_4^{\uparrow} - FBR_4^{\downarrow}) L_4$



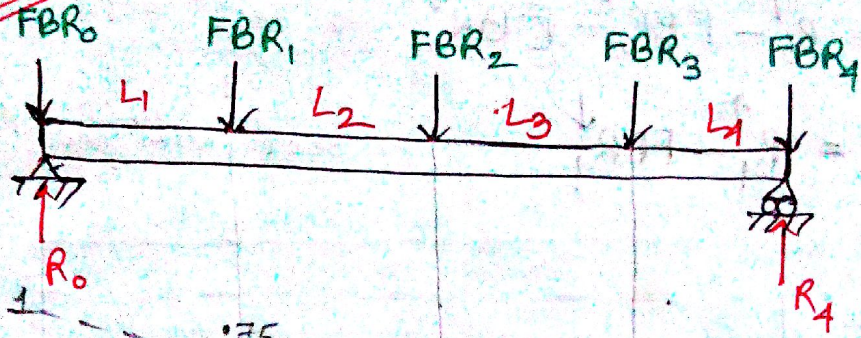
Let's assume floor beam as panel point - support or cross girder etc.

direct<sup>n</sup> floor beam छिन्ना कराल (↑) कर but main girder छिन्ना कराल (↓) कर

↓  
mainly main girder-कें लक्षणकर जोडाए सो, अतिर main girder-कें direct<sup>n</sup> use करवा



असम. भारों का  
Pattern

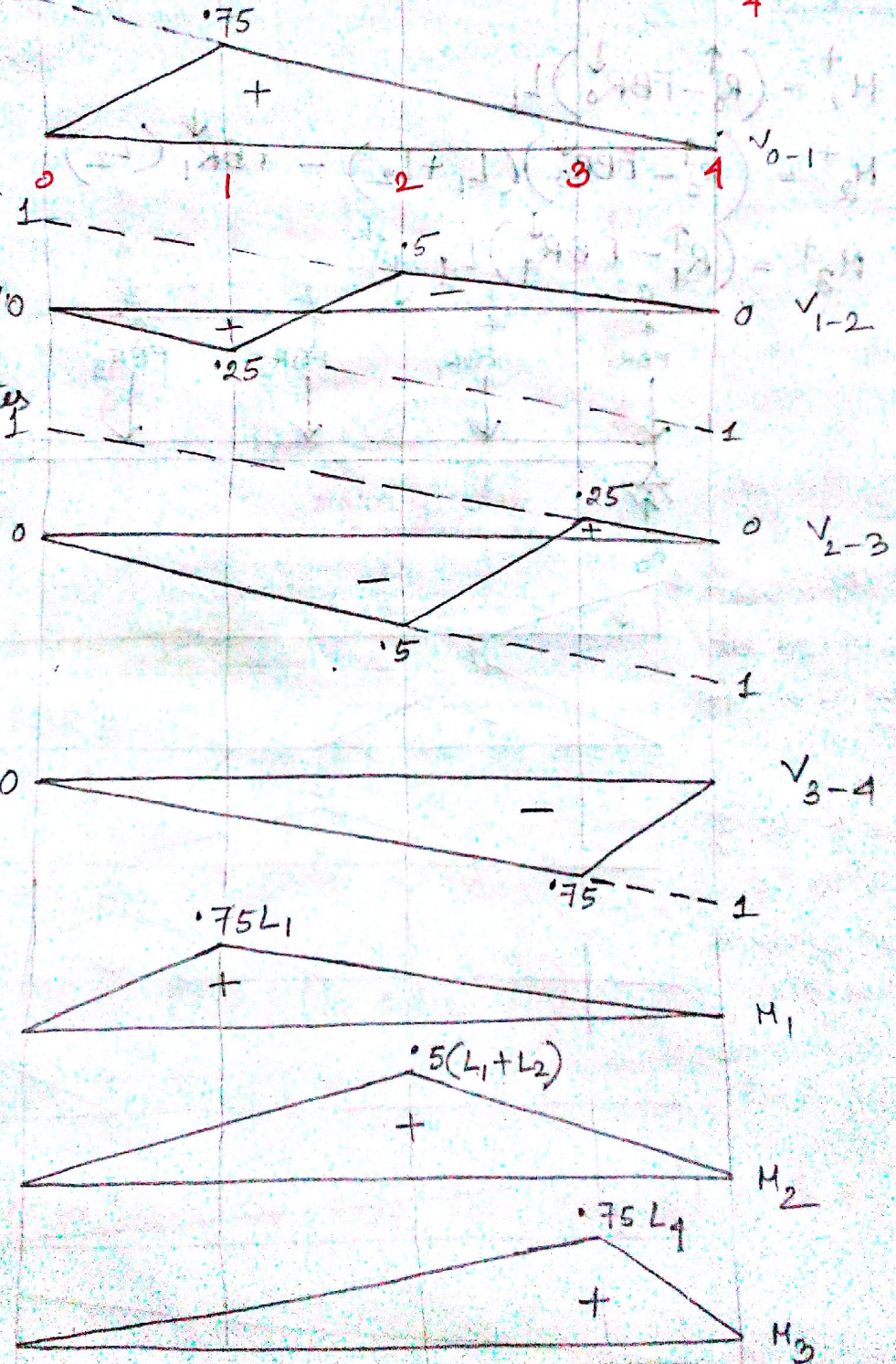


⊛ first parallel line दूरी लेना।

Then in दूरी point  $L_1$  SF काकरा करे point दूरी add करे।

↓

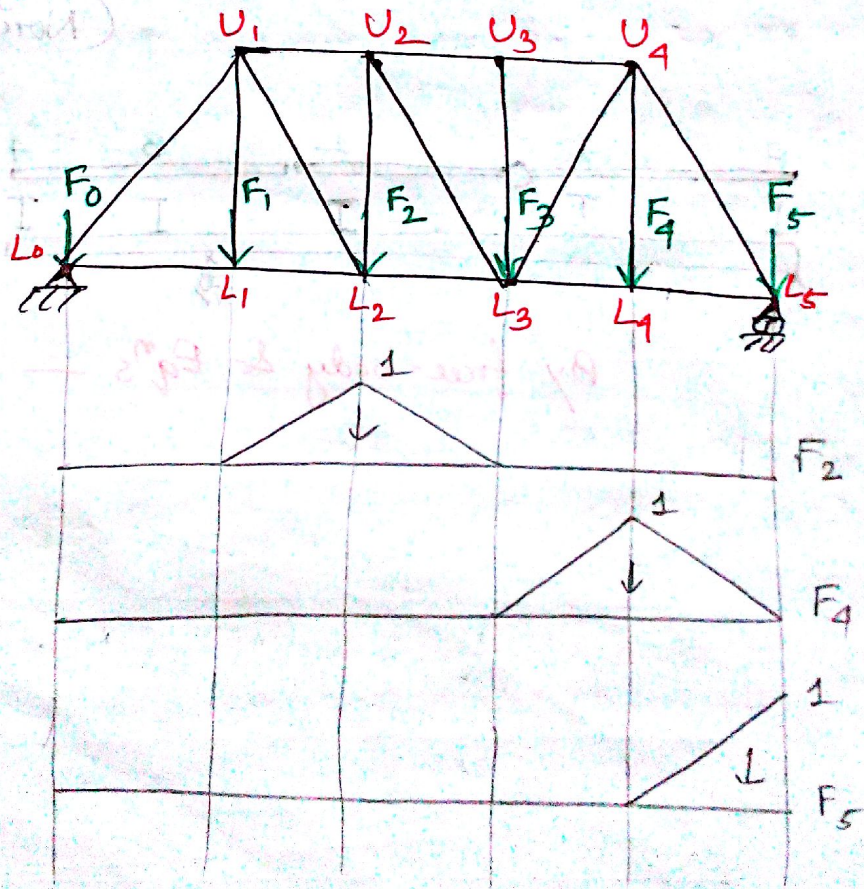
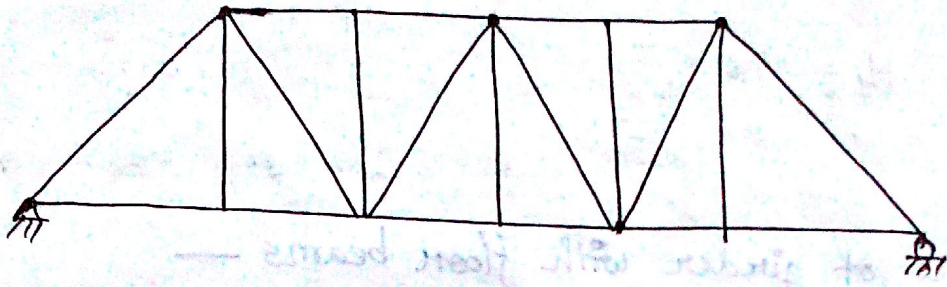
Then find ordinates by similar  $\Delta$ s.





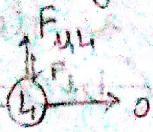
# I-Lines for a truss —

(\*) প্রতিটি truss - ১২ সারিতে - type-A floor beams থাকবে।



Find the I-line for  $U_1 L_1$

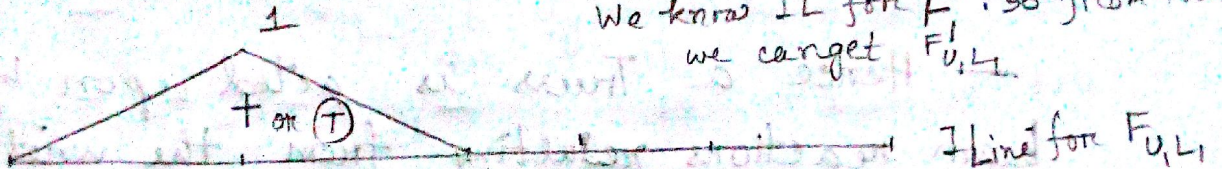
⊙ for finding  $U_1 L_1$  force,  $L_1$  joint analysis



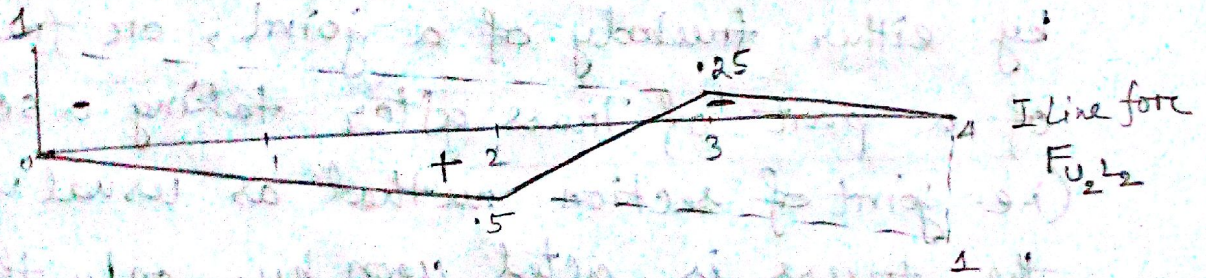
$$\sum V = 0$$

$$\Rightarrow F_{U_1 L_1}^{\uparrow} = F_{U_1 L_1}^{\downarrow}$$

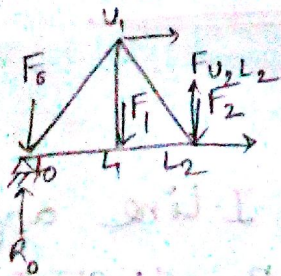
↓ We know IL for  $F$ . So from this we can get  $F_{U_1 L_1}^{\uparrow}$



Find the Iline for  $F_{U_2 L_2}$



for  $L_2 U_2$



$$\sum V = 0 \Rightarrow F_{U_2 L_2} + R_0 - F_0 - F_1 - F_2$$

$$\Rightarrow F_{U_2 L_2} = F_0 + F_1 + F_2 - R_0 \quad \text{--- (2)}$$

This eq<sup>n</sup> (2) is similar to  $V_{2-3}$

So the Iline will be equal to  $V_{2-3}$

- Influence lines for truss members —
  - A bridge is fitted with a floor systems either at the bottom chord level (~~Deck Type~~) (Through type) or at the Top chord level (Deck Type).
  - Hence a Truss is acted upon by floor beam reactions resulting from the unit load <sup>Moving</sup> live load.
  - Force in a member of Truss is calculated by either freebody of a joint, or freebody of a part of Truss after taking a section (i.e. joint of section method as usual while the truss is acted upon by only the

### FLOOR BEAM REACTIONS

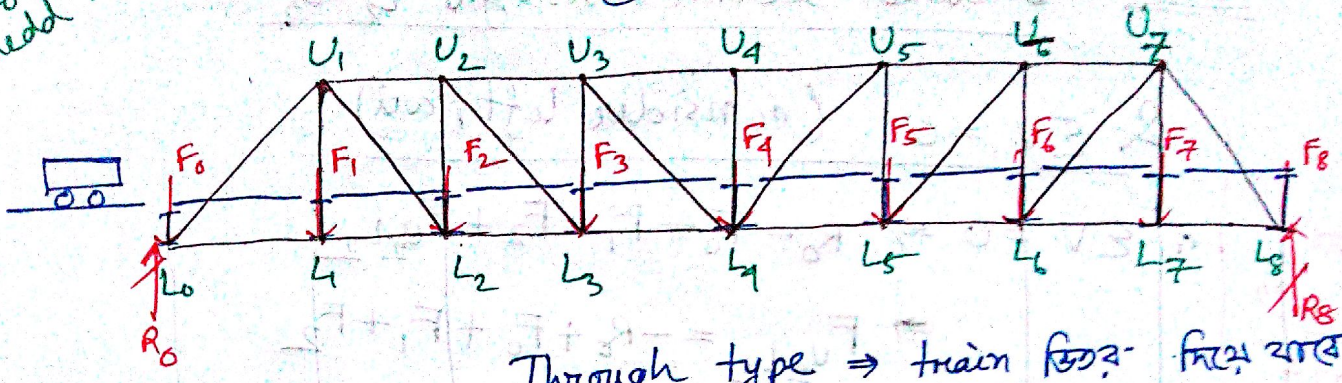
#### • Steps —

1. Find the shape of I. line applying TRUSS ANALYSIS BY JOINT OR SECTION METHOD.
2. Then find the controlling ordinates by truss analysis for specific position of UNIT LOAD.

Fig-15  
Pg. 264  
Shedd Vault

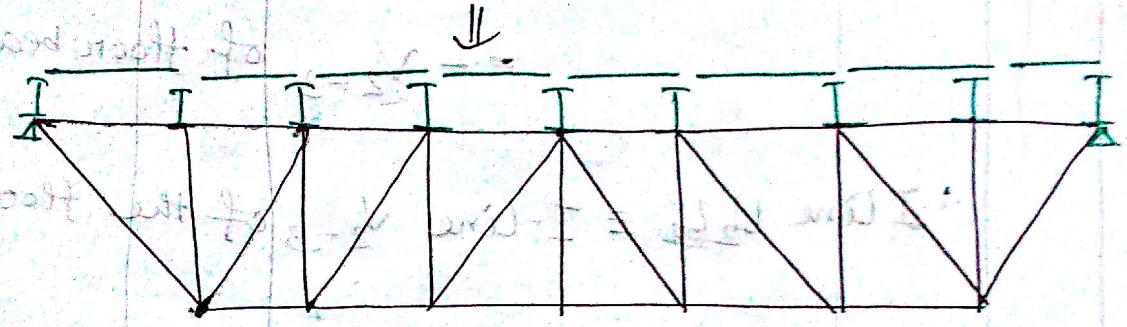
Truss - 1  $\Rightarrow$  Parallel Chord Truss

8' @ 25' = 200'



Through type  $\Rightarrow$  train kee? - free raab

Deck "  $\Rightarrow$  " truss kee? keeb " "

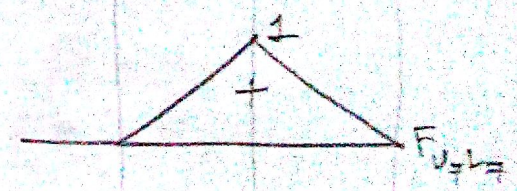
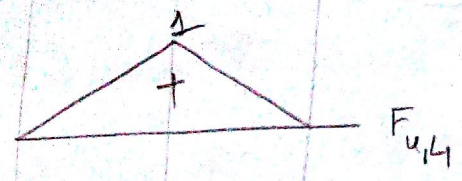


I line for  $U_1 L_1 \Rightarrow \sum J + L_1 \& \sum V = 0$

$$F_{U_1 L_1}^{\oplus} = F_1^{\downarrow}$$

Similarly,  $U_7 L_7$  &  $U_4 L_4$

$$F_{U_7 L_7}^{\oplus} = F_7^{\downarrow}, \quad F_{U_4 L_4} = 0$$



$U_2L_2$   $\rightarrow$  Inclined section through  $L_2-L_3$

$\sum V_i = 0$  (consider left part)

$\sum V = 0 \Rightarrow R_0 - F_0 - F_1 - F_2 + F_{U_2L_2}$

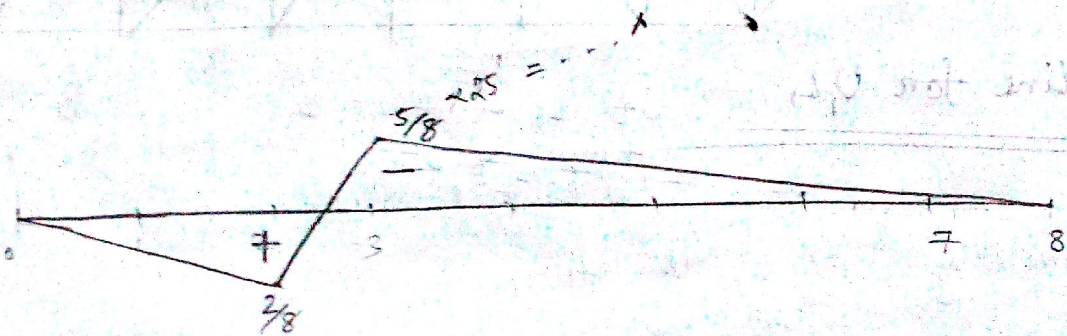
$\Rightarrow F_{U_2L_2} = -R_0 + F_0 + F_1 + F_2$

$= - (R_0 - F_0 - F_1 - F_2)$

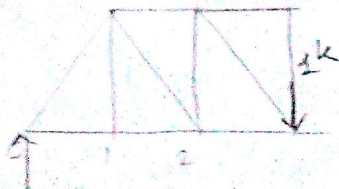
$= -V_{2-3}$  of floor beam system

$\therefore$  I line  $U_2L_2 \equiv$  I-line  $V_{2-3}$  of the floor system

eqn for  
for  $V_{2-3}$   
force  
line at  
 $V_{2-3}$



Load (3)  $\Rightarrow$  3 k



$\frac{5}{8} \Rightarrow$  on right side 5 in part  $\uparrow$  reaction  
& left " 3 in " " "

$\uparrow = \frac{5}{8}$  & in the member this is  $\Rightarrow \sum V = 0 \Rightarrow$  reaction  $\uparrow$   
compression — so -ve so force at  $U_2L_2$  is  $\downarrow$  i.e. comp -ve

Iline for  $U_3 - U_4 \Rightarrow$  Ver Sect<sup>n</sup> Truss  $L_3 - L_4$   
 &  $\Sigma M_{L_4} = 0$ .

$$\Sigma M_{L_4} = 0 = \text{---}$$

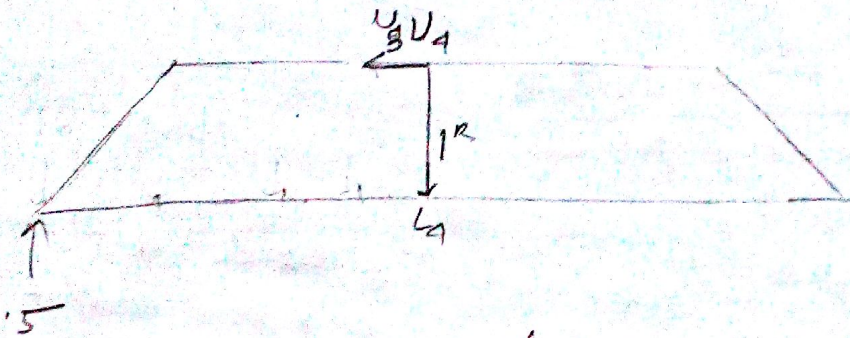
$$= \text{---}$$

area under the parabola

$$\therefore U_3 U_4 \equiv M_{L_4}$$

for ordinate  $\Rightarrow$

unit load at  $L_4$  & taking left —



$$\Sigma M_{at L_4} = '5 \times 4 \times 25' - U_3 U_4 \times 35$$

$$\Rightarrow M_{U_3 U_4} = \frac{5 \times 4 \times 25}{35}$$

Similarly Top Chord Members

BC Members

at point  $\rightarrow$   
 moment  $\rightarrow$   
 bar force  
 (BA)  $\rightarrow$  25  
 (AB)  $\rightarrow$  M dia  
 (BA)  $\rightarrow$  25

$U_1 U_2 = M_2$   
 $U_2 U_3 = M_3$   
 $U_3 U_4 = M_4$   
 $U_4 U_5 = M_4$   
 $U_5 U_6 = M_5$   
 $U_6 U_7 = M_6$

$L_0 L_1 = M_1$   
 $L_1 L_2 = M_1$   
 $L_2 L_3 = M_2$   
 $L_3 L_4 = M_3$   
 $L_4 L_5 = M_4$   
 $L_5 L_6 = M_5$   
 $L_6 L_7 = M_6$   
 $L_7 L_8 = M_7$

Inclined member  $\Rightarrow (U_3 L_4)$  Vertical sect<sup>n</sup> through  $L_3-L_4$  &  $E V = 0$

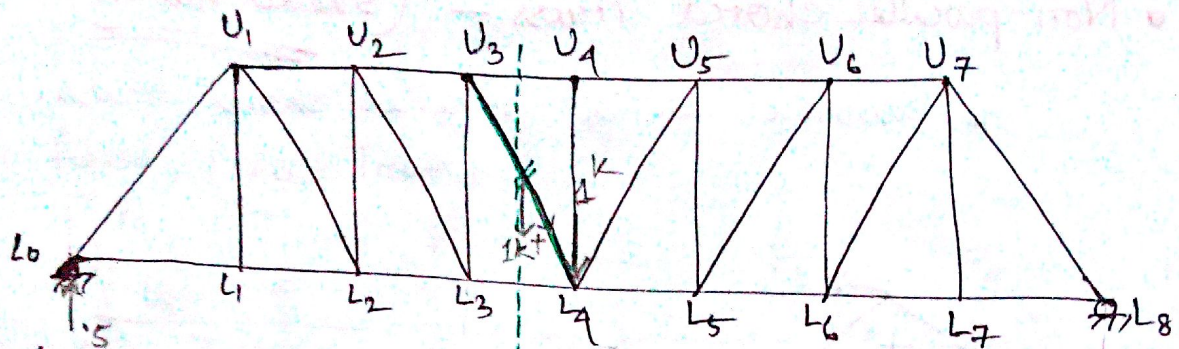
$\therefore U_3 L_4 = V_{3-4}$

ordinate  $\Rightarrow$  next class

$M = P U V$   
 for ordinate

Gradient of vertical height of chord member

$M = P U V$ $u = \frac{M}{P U}$ $\frac{dM}{dP} = \frac{M}{P}$ $\frac{dM}{dU} = \frac{M}{U}$ $\frac{dM}{dV} = \frac{M}{V}$ $\frac{dM}{dW} = \frac{M}{W}$ $\frac{dM}{dX} = \frac{M}{X}$ $\frac{dM}{dY} = \frac{M}{Y}$ $\frac{dM}{dZ} = \frac{M}{Z}$	$M = P U V$ $H = P U V$ $P U = \frac{H}{V}$ $u = \frac{H}{V U}$ $\frac{dH}{dP} = \frac{H}{P}$ $\frac{dH}{dU} = \frac{H}{U}$ $\frac{dH}{dV} = \frac{H}{V}$ $\frac{dH}{dW} = \frac{H}{W}$ $\frac{dH}{dX} = \frac{H}{X}$ $\frac{dH}{dY} = \frac{H}{Y}$ $\frac{dH}{dZ} = \frac{H}{Z}$
---	---



for  $U_3L_4 \Rightarrow$

$$\therefore F_{U_3L_4} = .5 \times \frac{\sqrt{35^2 + 25^2}}{35} = .614$$



for ordinate put unit load  $U_3L_4$  at pt 4

Similarly,

$$U_1L_0 \equiv V_{0-1}$$

$$U_1L_2 \equiv V_{1-2}$$

$$U_2L_3 \equiv V_{2-3}$$

$$\underline{U_3L_4 \equiv V_{3-4}}$$

$$U_5L_4$$

$$U_6L_5$$

$$U_7L_7$$

$$U_7L_8$$

Ordinate for T.C. Members

for  $U_3U_4 \Rightarrow$  Ver. sect<sup>n</sup> through  $L_3L_4$  & Left part &  $\Sigma M_{L_4} = 0$

$$\therefore U_3U_4 \equiv M_4$$

• point 4 - 1 unit load

$$\begin{aligned} \Sigma M_{L_4} = 0 &\Rightarrow .5 \times 100 + F_{U_3U_4} \times 35 \\ &\Rightarrow F_{U_3U_4} = \underline{\underline{1.428}} \end{aligned}$$

Ordinate for B.C. Members

for  $L_3L_4 \Rightarrow$  sect<sup>n</sup> (vert.) &  $\Sigma M_{U_3}$

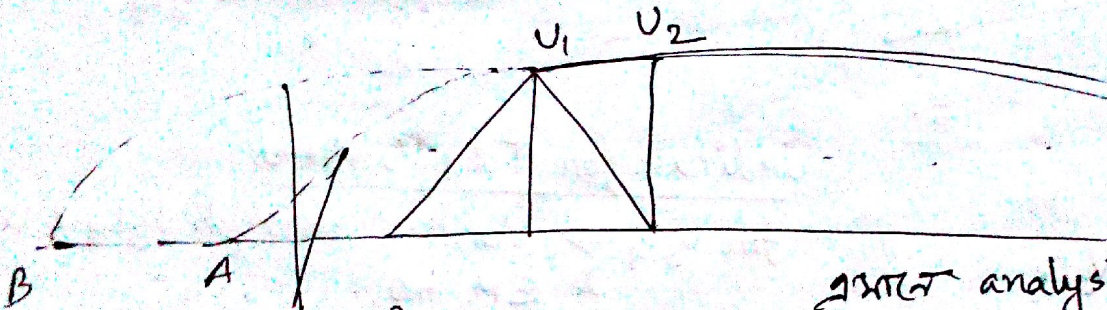
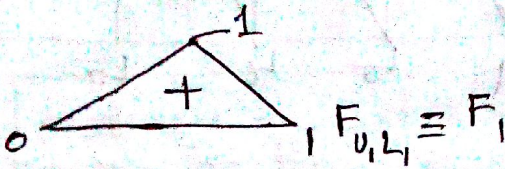
$$\therefore L_3L_4 \equiv M_3$$

• point 3 @ unit load -

$$\begin{aligned} \Sigma M_{U_3} = 0 &\Rightarrow \frac{5}{8} \times 75 = F_{L_2L_3} \times 35 \\ &\Rightarrow F_{L_2L_3} = \underline{\underline{1.339}} \end{aligned}$$

• Non-parallel chorded Truss - (Shedd Vault)

↪ assume as a parallel chord truss.



$U_2 U_3$  &  $U_1 U_2$  - projected length

( $A = U_1 U_2$  - project  $U_1 U_2$  meet  $U_1 U_2$  and  $U_1 U_2$  मिल force  $U_1 U_2$ )

Similarly  $B =$  for  $U_2 U_3$

असमत analysis -

Top chord -

ver. & hor. member

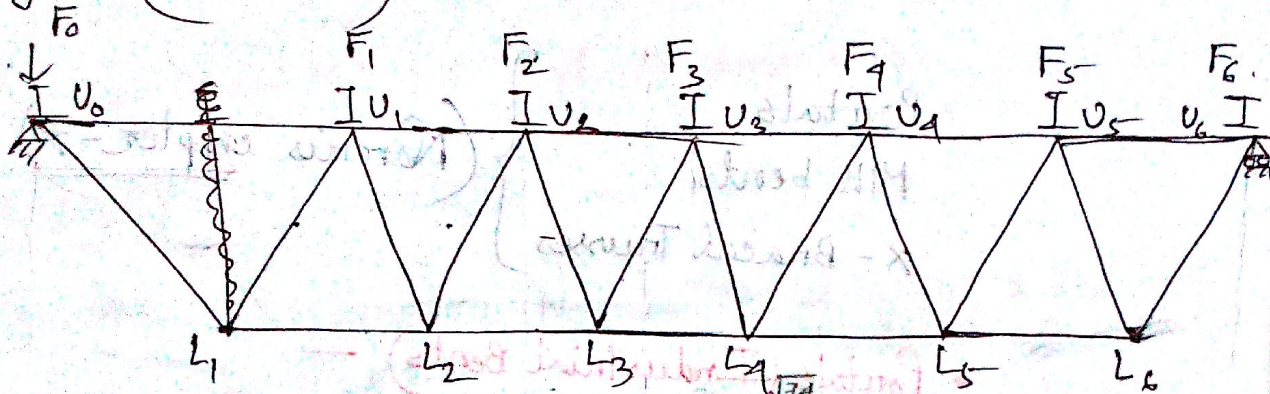
असमत & so analysis  $U_1 U_2$  tough

So projection is used.

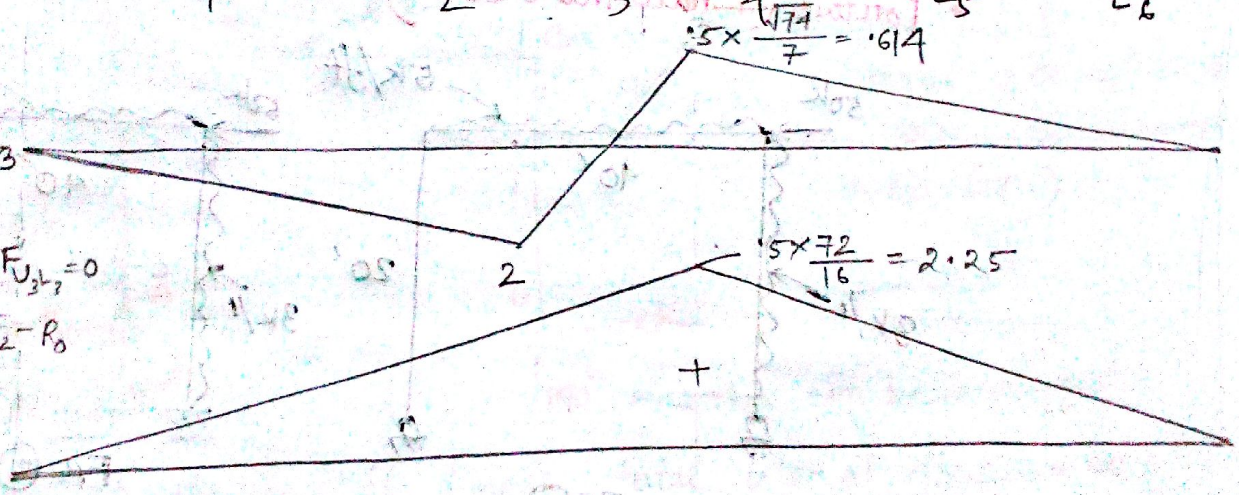
• Truss without verticals —

Fig. 151 (shedded Vauster)

1st figure, (Deck Truss)



$\sum V = 0$   
 $\Rightarrow R_0 - F_0 - F_1 - F_2 + F_{U_3L_3} = 0$   
 $\Rightarrow F_{U_3L_3} = F_0 + F_1 + F_2 - R_0$   
 $= 2.25$



③ for  $U_2$  and  $U_2 U_3 \Rightarrow EM_{L_2}$  but  $L_2$  शरत  $U_1 U_2$  - रा शरत, not ver. So actually  $H_{L_2}$  रा शरत  $U_2$  शरत शरत but  $L_2$  is not a panel point with girder.



(So शरत peak  
 ये शरत शरत  
 point शरत add  
 शरत शरत.)  
 Ordinate same  
 way शरत शरत

(\*) class test  $\Rightarrow$  I lines for beams

(23.11.2015)

11.11.15

# # Approximate Analysis of Indeterminate Structures

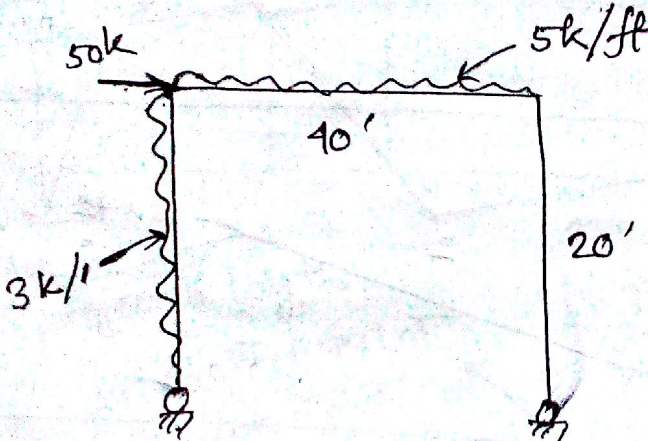
Portals

Mill bents

X-Braced Trusses

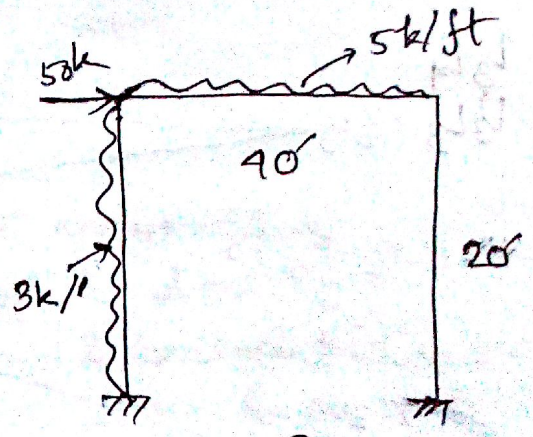
(Norris chapter-7)

## • Portals (Industrial Bents) —



FR ①

$$D^{\circ} \text{ Ind} = 1^{\circ}$$



FR ②

$$D^{\circ} \text{ Ind} = 3^{\circ}$$

\* Assumptions of identifying extra eq's for Portals :

1. There is a hinge at the mid-height of each column.

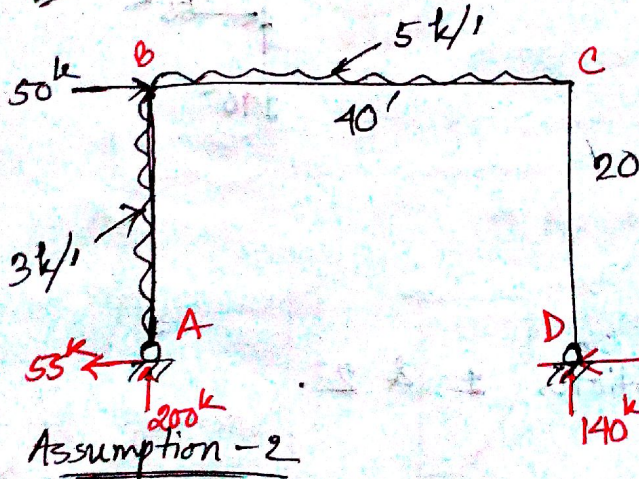
2. Total lateral force at the level of hinges are equally divided among two columns.

\* Steps of Analysis —

1. Make necessary assumption(s)
2. Find forces from freebody
3. Draw SF, BM and AF diagrams.

FRAME ①

1. Make assumption No-2  $\Rightarrow$



(Cz 2nd part 1<sup>o</sup> Ind.  $\Rightarrow$  SO only 1 eq<sup>n</sup> is needed. But Ass. ① यदि तबे लक्षण 2 के eq<sup>n</sup> चलें पाएँ। So assumption ② वि।)

③ SMART

Now this can be analyzed.

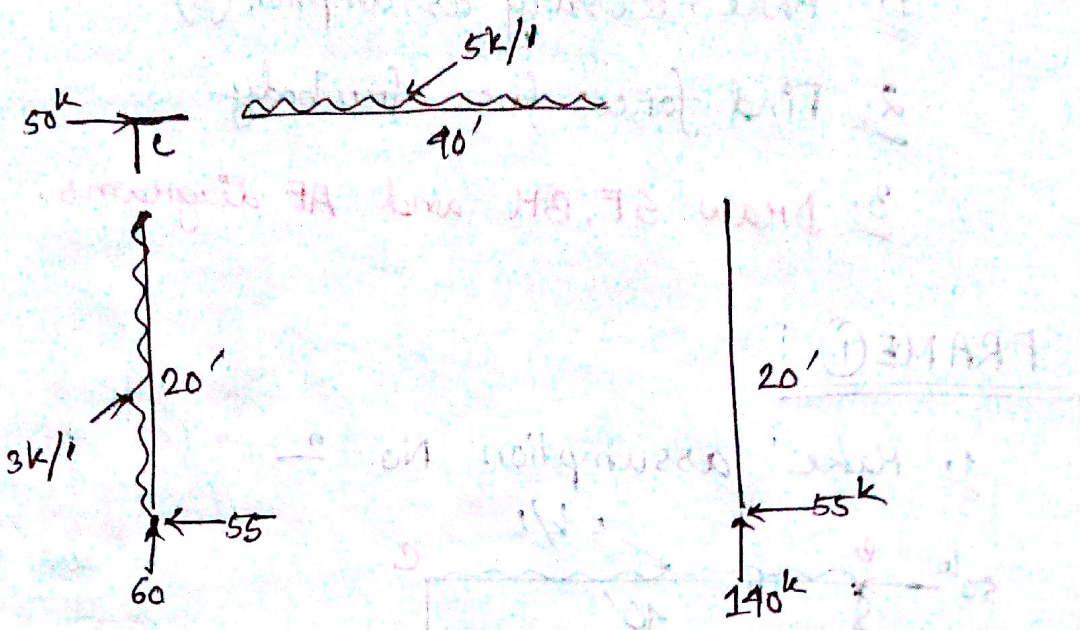
at the level of hinges, total lateral forces =  $50 + 60 = 110k$

It will be equally distributed to the hinges in the opposite direction.

$$EM_A = 0 \Rightarrow 50 \times 20 + 60 \times 10 + 200 \times 20 = R_D \times 40$$

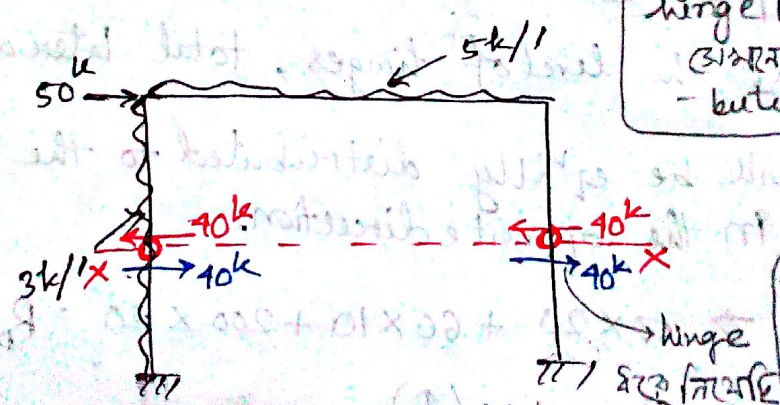
$$\Rightarrow R_D = 140 (\uparrow)$$

Now free-body -



FRAME (2)

1: Make assumptions 1 & 2.



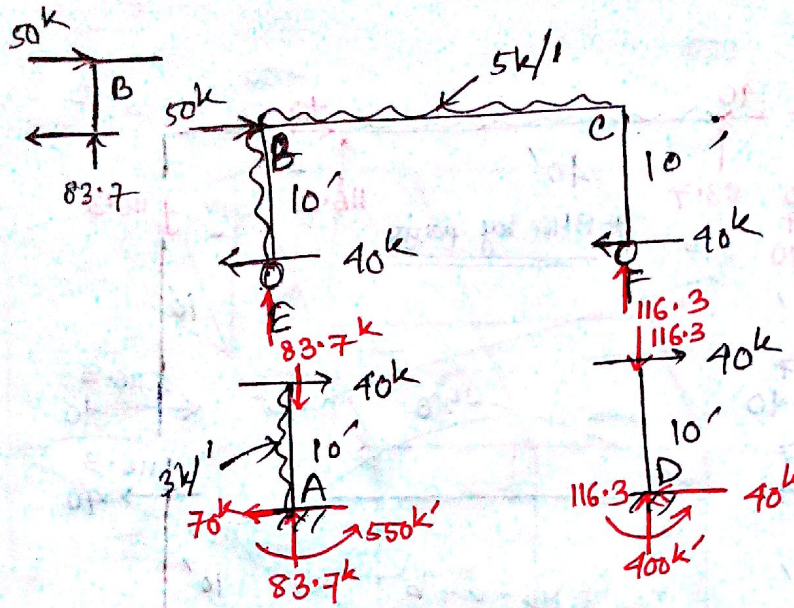
बातचीत करी support hinge कि 50' त बायत equally distributed राखे।

But अशात hinge is inserted So अहे level lateral load divide राव।

total lateral load at XX  $\Rightarrow$   
 $50 + 3 \times 10 = 80k$

hinge अहे कालात बायत कालात अहे कालात

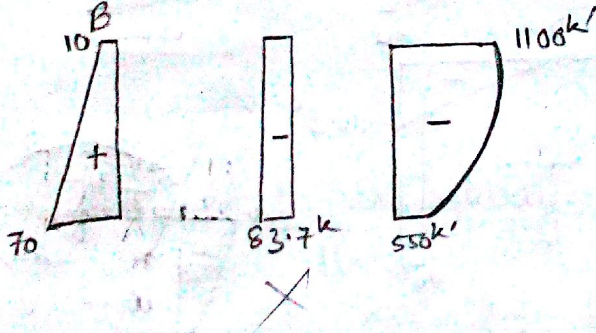
Free body



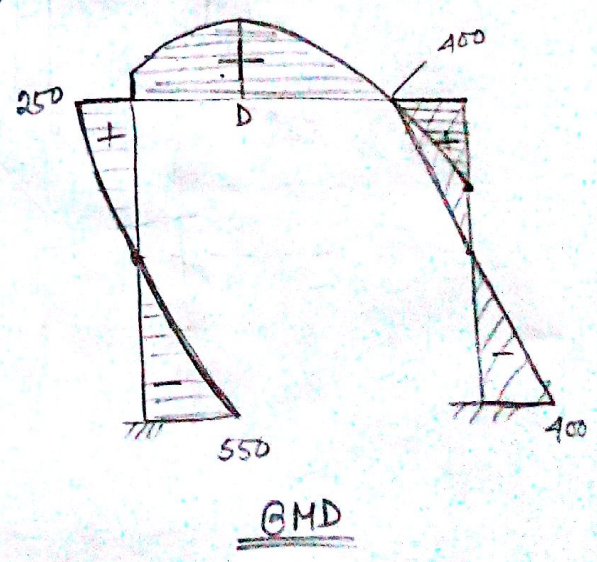
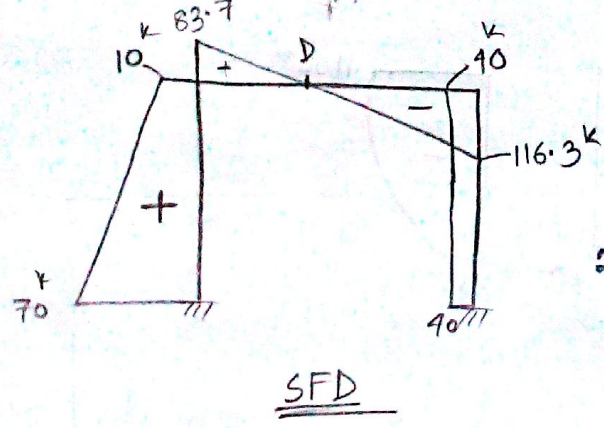
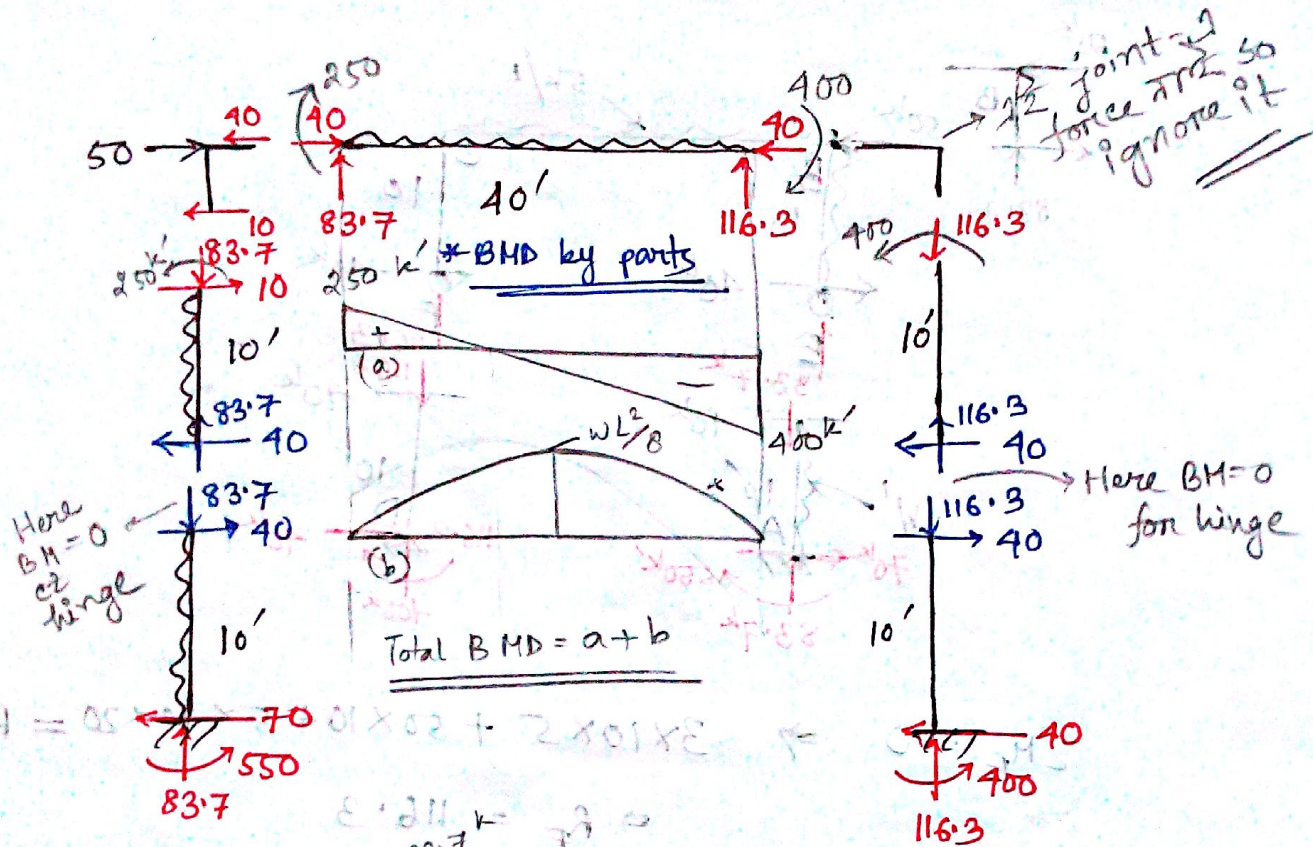
$$\sum M_E = 0 \Rightarrow 3 \times 10 \times 5 + 50 \times 10 + 5 \times 40 \times 20 = R_F \times 40$$

AE

$$\Rightarrow R_F = 116.3$$

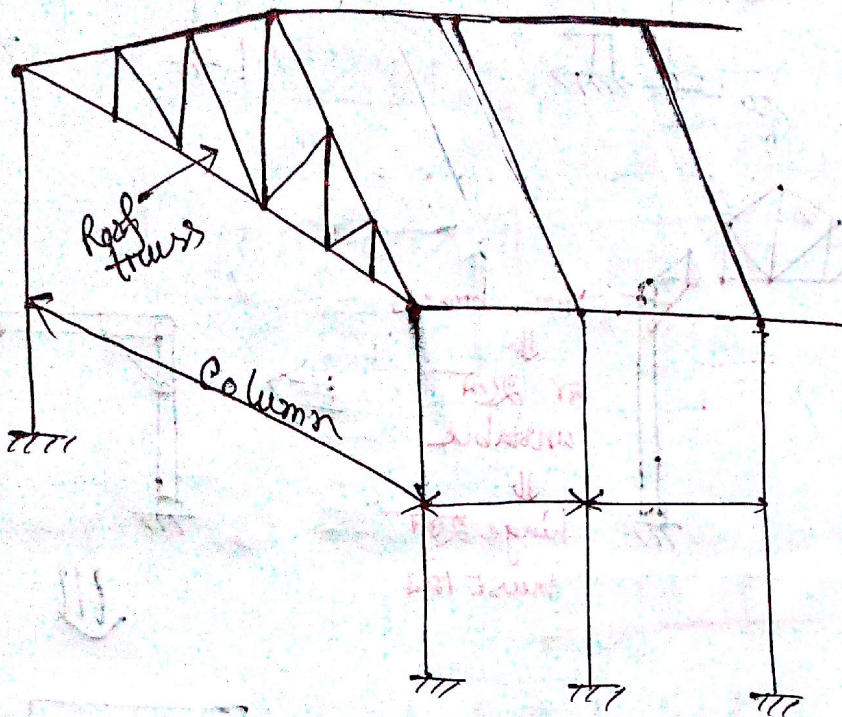


P.T.O



16.11.15

• Mill Bents —



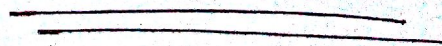
\* Mill bent is always a truss or steel frame.



cz RCC इल beam depth कोलका बनि

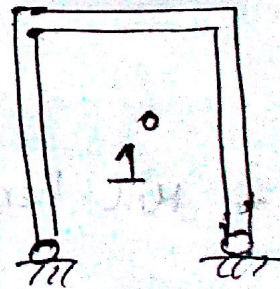
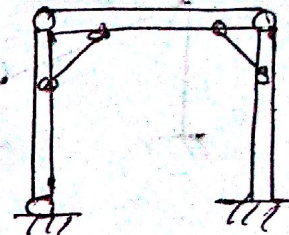
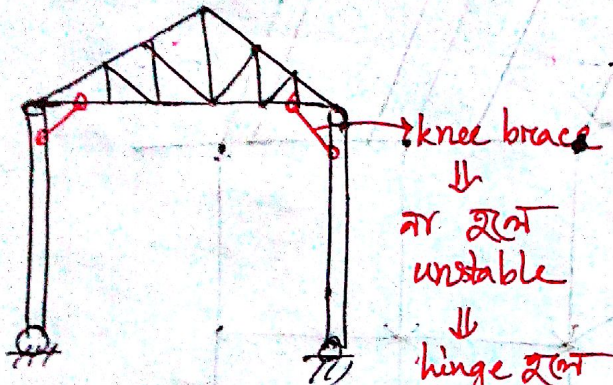
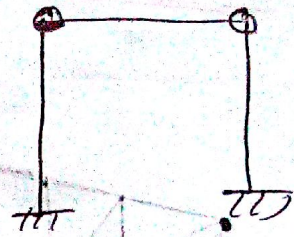
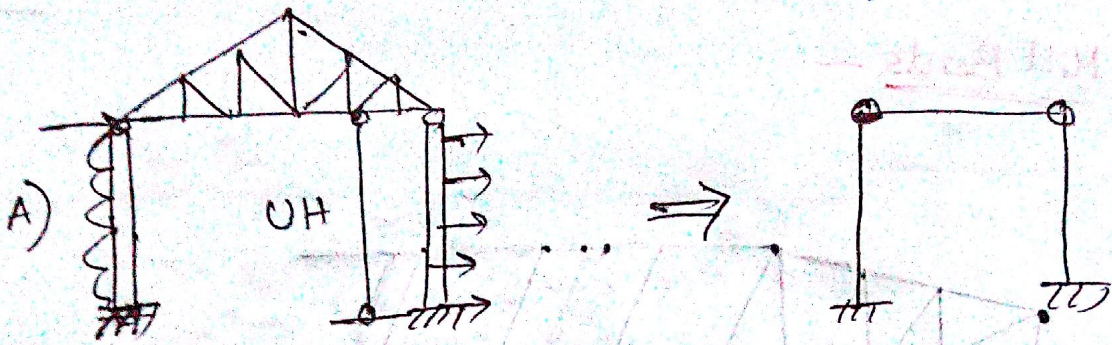
इउ। cz mill - a column देहा

interrupting cz mill - a कोलका long  
space माइल without interruption



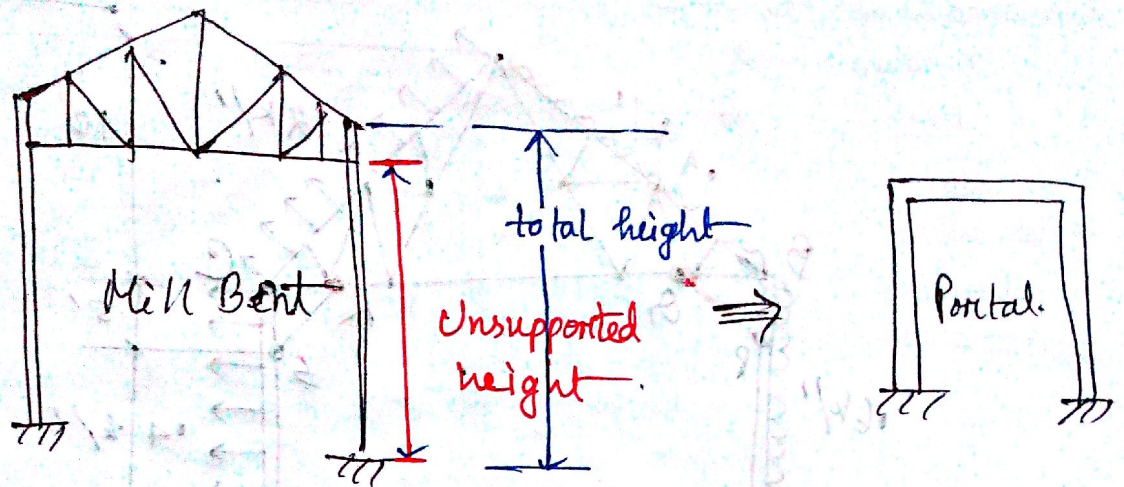
# Mill bents

# Equivalent Portals.



*Faint handwritten notes in Hindi, likely describing the structural analysis process.*

• Unsupported Height of each column —



Assumption ①  $\Rightarrow$  Hinge will be in the middle of unsupported height of column.

② Total lateral load is equally divided among two columns  $\Rightarrow$  1 eqn. at the level of hinges.

$(+)$   $\dots$

$\dots$

$\dots$

$\dots$

$\dots$

$\dots$



$\dots$

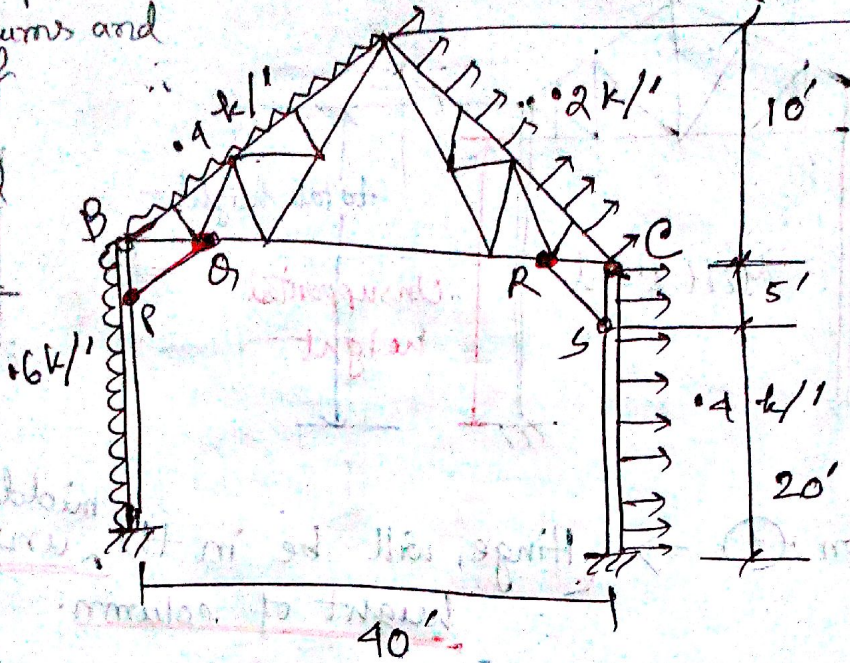
• Mill Bent #2

Draw SF, BM for col<sup>ms</sup> & bar force in

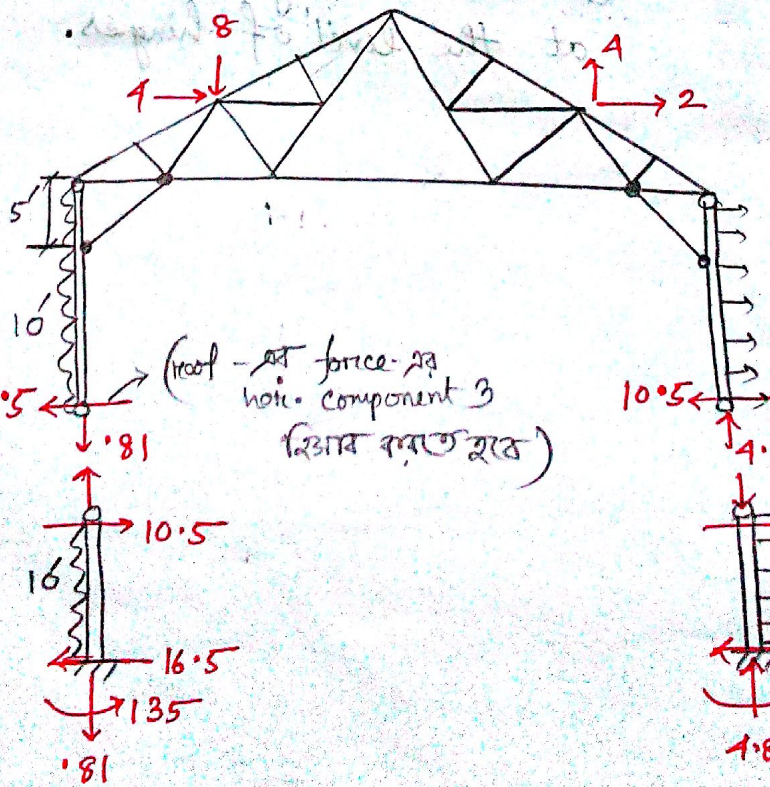
**KNEE BRACES**

downward & upward UDL  
in columns and  
roof

for wind  
load



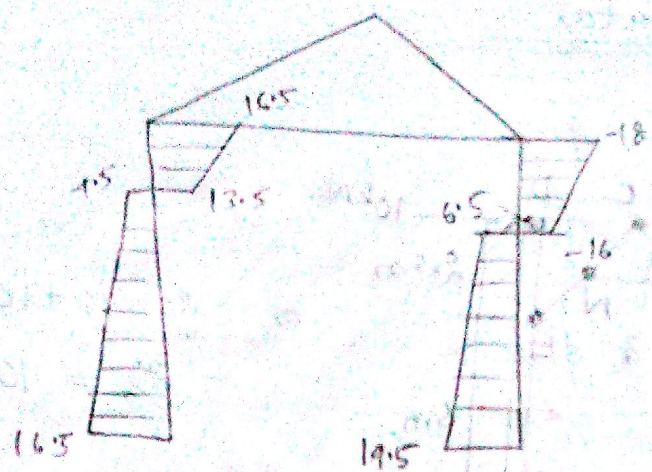
analysis of load transfer to col<sup>ms</sup>



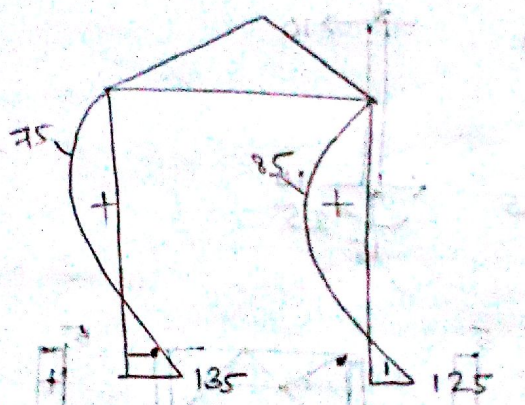
$F_{PQ} = 22.5 \text{ k (+)}$   
 $(-)$   $v = 13.5 \text{ k}$   
 $(\rightarrow)$   $H = 18 \text{ k}$   
 Vert & Hon. comp. of  $F_{PQ}$   
 $F_{RS} = 28.13 \text{ k}$   
 $( ) H = 22.5$   
 $( \searrow ) v = 16.88$

∴ unsupported  
col<sup>m</sup> height  
= 20'  
hinge is  
at 10'

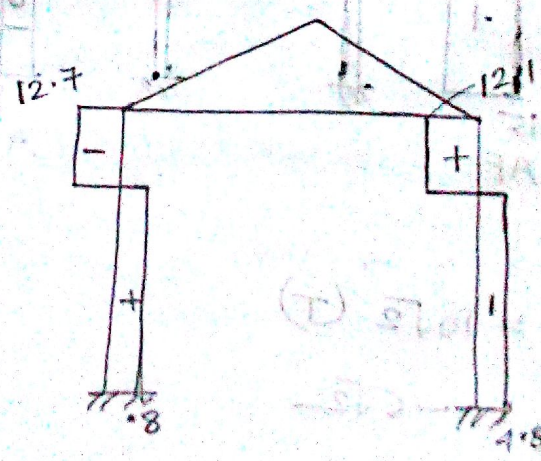
Draw SF, BM, AF dia for col<sup>s</sup>.



SFD

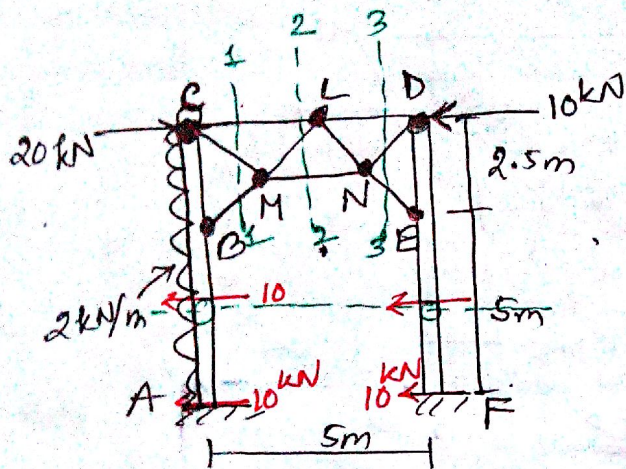


BMD

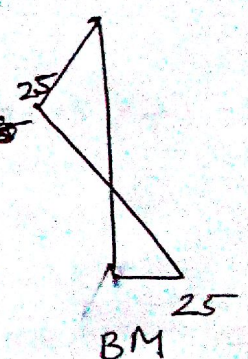
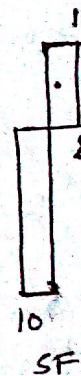
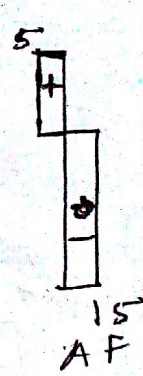
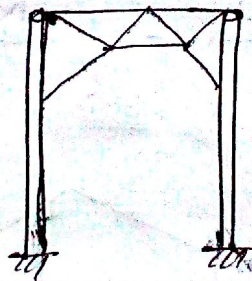
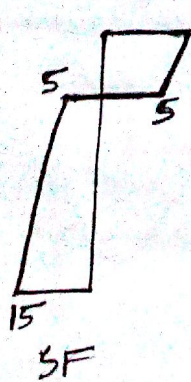
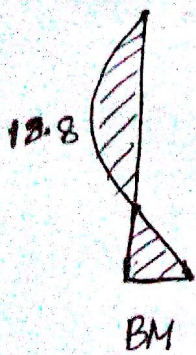
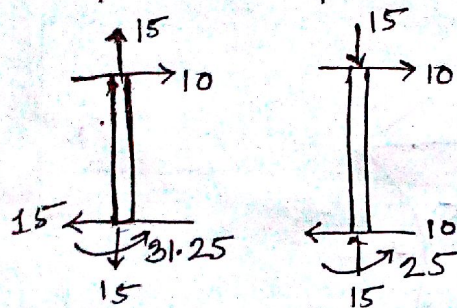


AFD

# Bridge Portal



$$\Rightarrow 20 - 10 + 10 = 20/2 = 10$$



Sec 1-1

$$\Sigma M_c = 0 \rightarrow F_{BM} = 10\sqrt{2} \text{ (T)}$$

$$\Sigma V = 0 \rightarrow F_{CM} = -5\sqrt{2}$$

$$\Sigma H = 0 \rightarrow F_{cL} = -25$$

Sec 2-2

$$\sum M_L = 0, F_{MN} = -10$$

Sec 3-3

$$\sum M_N = 0 \Rightarrow F_{LD} = 5 (+)$$

$$\sum M_D = 0 \Rightarrow F_{NE} = 20\sqrt{2}$$

$$\sum V = 0 \Rightarrow F_{ND} = -5\sqrt{2}$$

JTM

$$\sum H = 0 \Rightarrow F_{ML} = +15\sqrt{2}$$

JTL

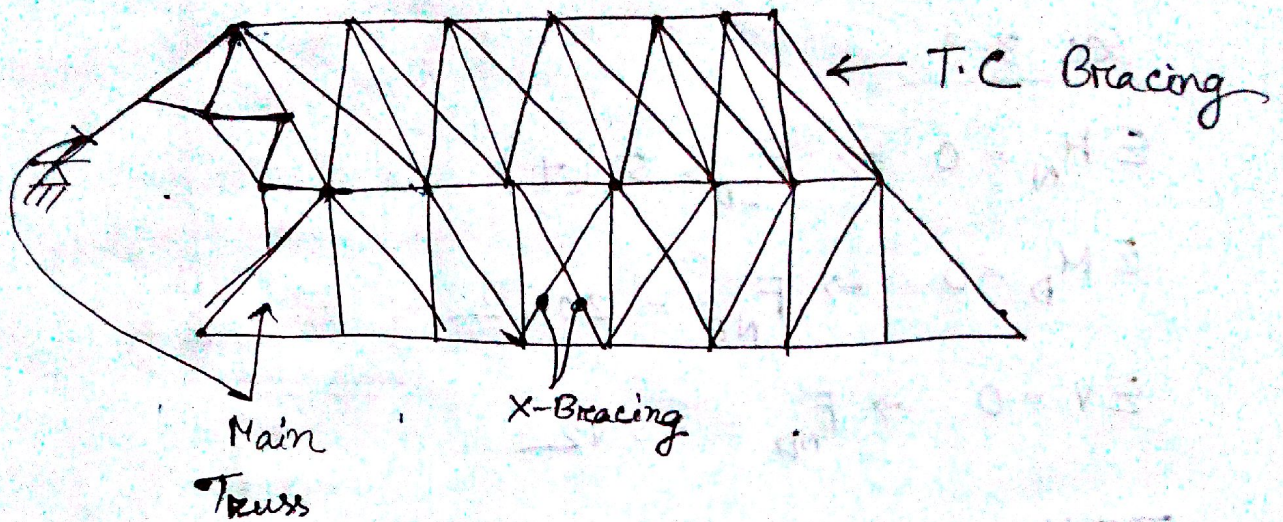
$$\sum V = 0 \Rightarrow F_{LN} = -15\sqrt{2}$$

$$\sum H = 0 \Rightarrow F_{LD} = +5$$

**BAR FORCES**

present in joints is also as per the table

## Approx. Analysis of X-braced Structures



### Assumptions

1) Diagonals (Bracings) can carry tension only (i.e. diagonals are made of rope)

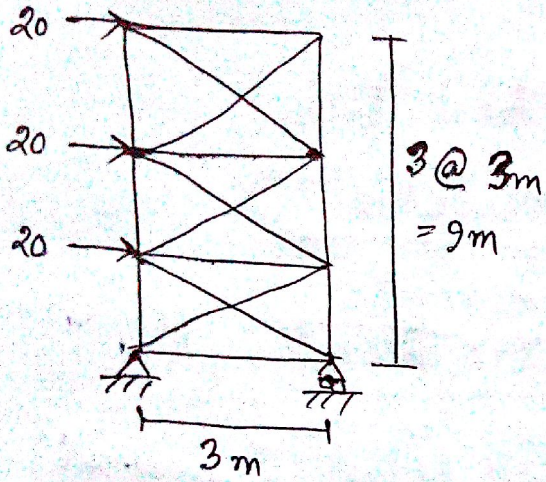
or

i) Diagonals can carry both the tension and compression and shares the panel shear equally.

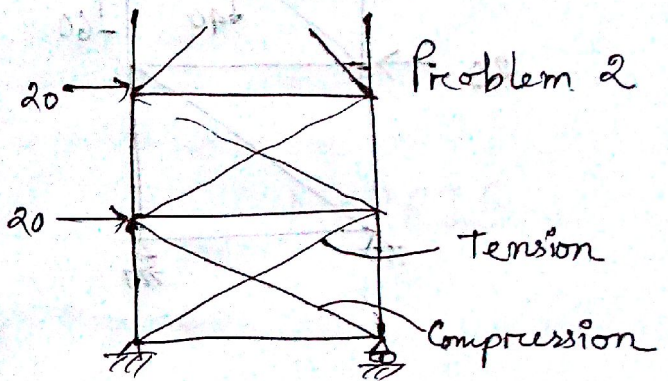
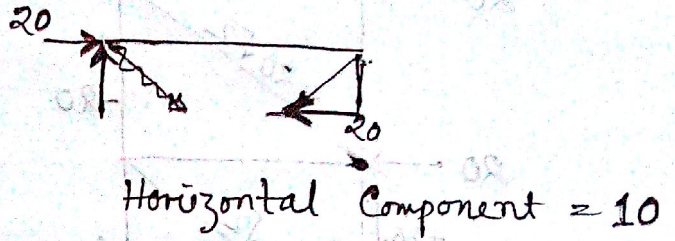
(i.e. Diagonals are made of rigid members)

$\text{Indet.} = 1^\circ$  for each additional x-bracing

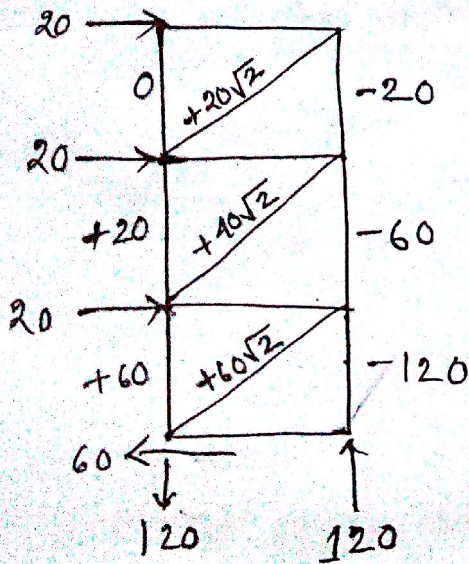
(A) Tension Only



Problem = (3° Indet.)

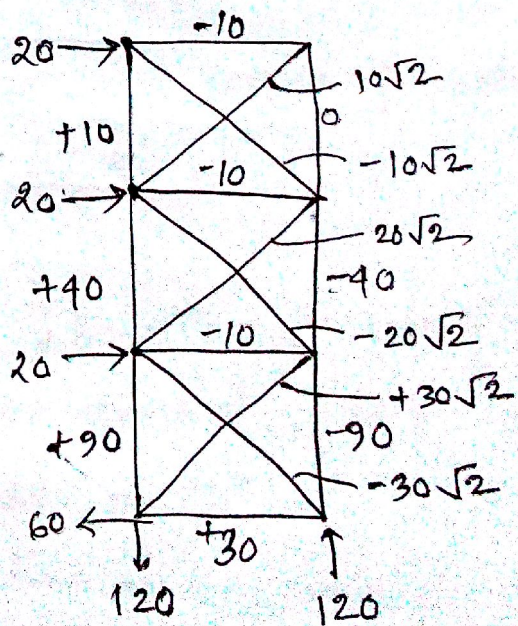


(a) Tension only



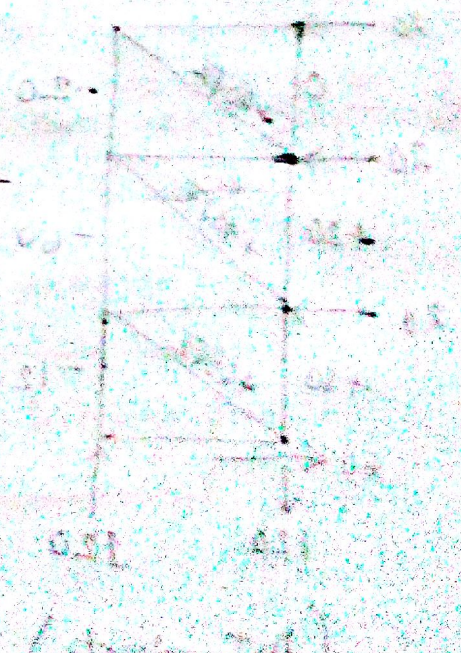
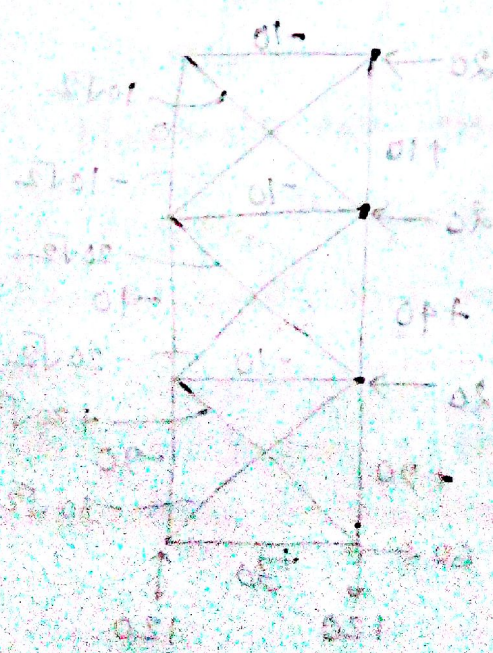
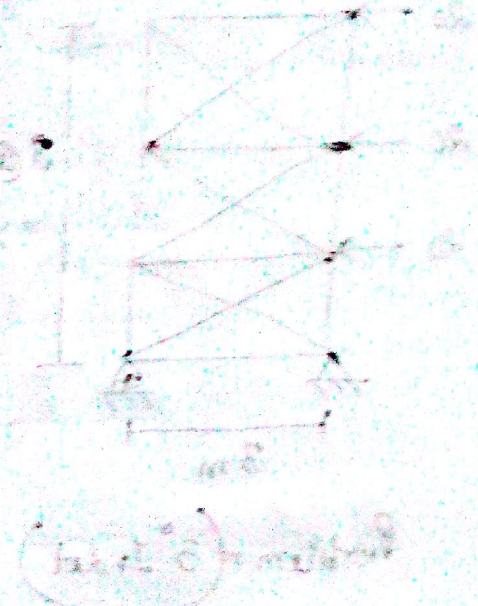
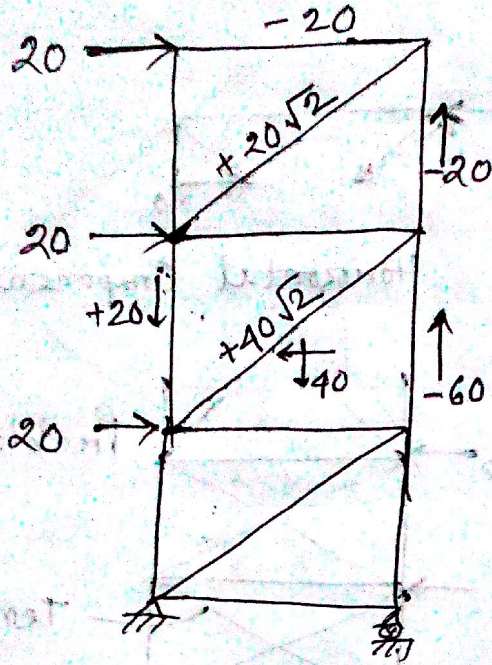
(Determinate)

(b) Can carry Tension & Comp.



(Indet.)

- How to solve trusses easily?



(Answer)

(Answer)

## Multistoried Building Frames under Lateral loads

### Lateral loads:

- 1) Wind load
- 2) Earthquake load

### Approximate analysis Method

- 1) Portal Method
- 2) Cantilever Method
- 3) Factor Method

⇒ NORRIS

### Portal Method

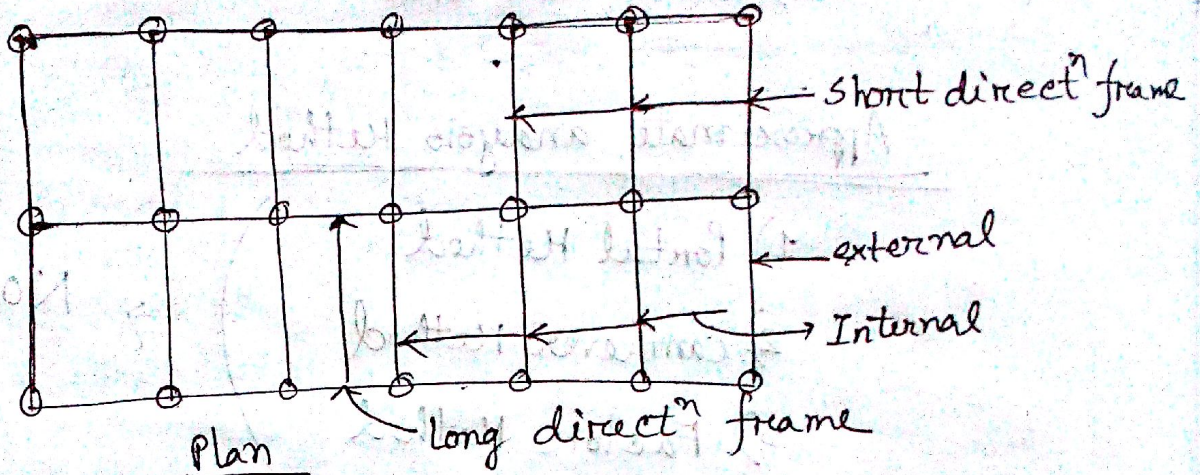
#### Assumptions

- 1) There is a hinge at the center of each beam
- 2) " " " " " " " " " " " " column.
- 3) Total lateral load above any storey (at the level of hinges) shall be divided among the columns of that storey so that an internal column takes twice the shear force taken by an external column.

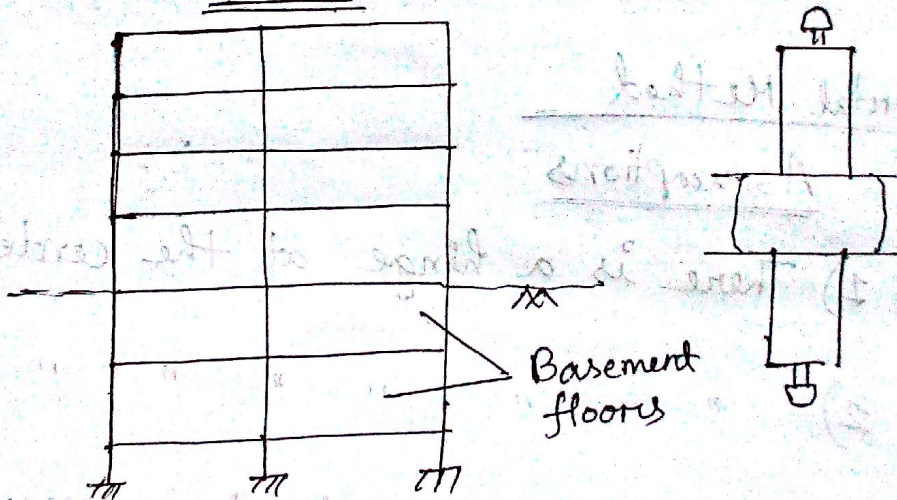
# Analysis

To be done by Freebody.

## Building Frame



Plan



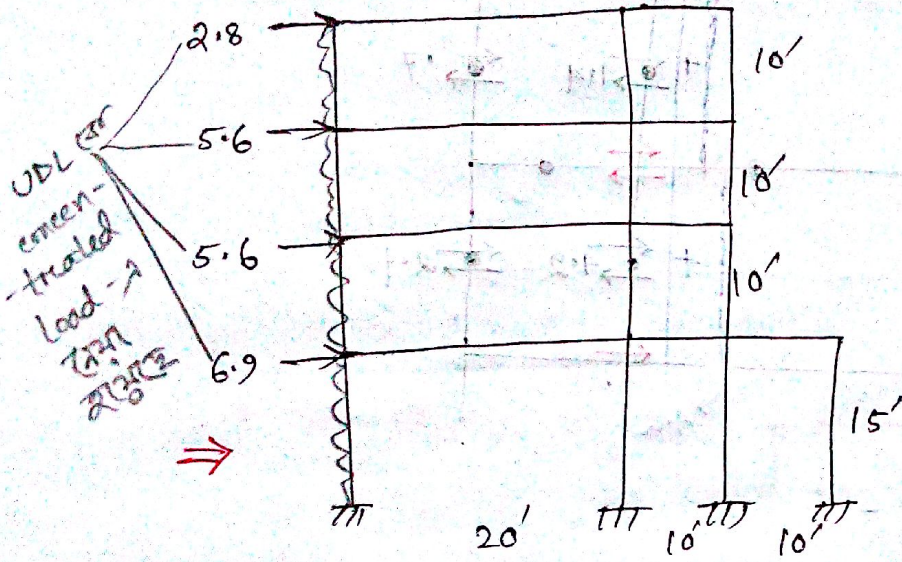
Basement floors

Problem-1

# PORTAL METHOD

30.11.15

Approximate analysis of building frames under lateral loads  
(Portal Method)



Frame spacing @ 15' c/c

Problem

## Wind load

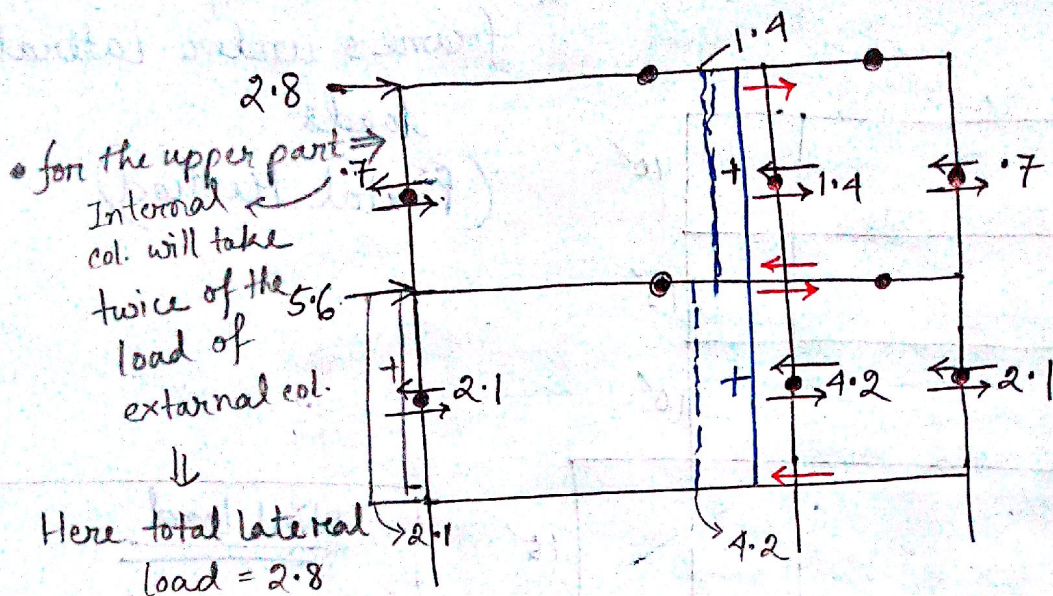
$V = 120 \text{ mph}$   
 $P = 0.00256 V^2$   
 $= 37 \text{ psf}$   
 $W = 0.37 \times 15$   
 $= 5.55 \text{ k/ft of Height}$

*in psf*  
*This is the crude formula for wind load calculation.*

- Analysis steps and order of calculation of =

- 1) Col. SF → By assumptions
  - 2) Beam AF
  - 3) Col. BM
  - 4) Beam BM
  - 5) Beam SF
  - 6) Col. AF
- by Freebody (Top to Bottom)

• Col SF - By Assumption



• for the upper part  
Internal col. will take twice of the load of external col.

Here total lateral load = 2.8

External = 2 & Internal = 1

1 + 1 = 2 & 1 x 2 = 2

$\frac{2.8}{4} = 0.7 \Rightarrow$  This will be taken by external cols  $\frac{2.8}{2} = 1.4$

This is taken by internal cols.

• for 2nd upper part  $\rightarrow$  Total load = 2.8 + 5.6 = 8.4

$\therefore$  External =  $\frac{8.4}{4} = 2.1$  & Internal =  $\frac{8.4}{2} = 4.2$

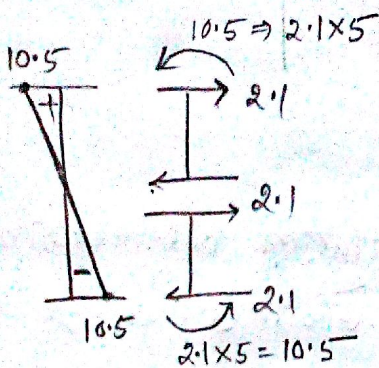
• for the lowest part  $\Rightarrow$  1 col, 2 ex. & 2 int.

$\therefore$  (total load) / 6  $\Rightarrow$

• Beam AF (by freebody) —

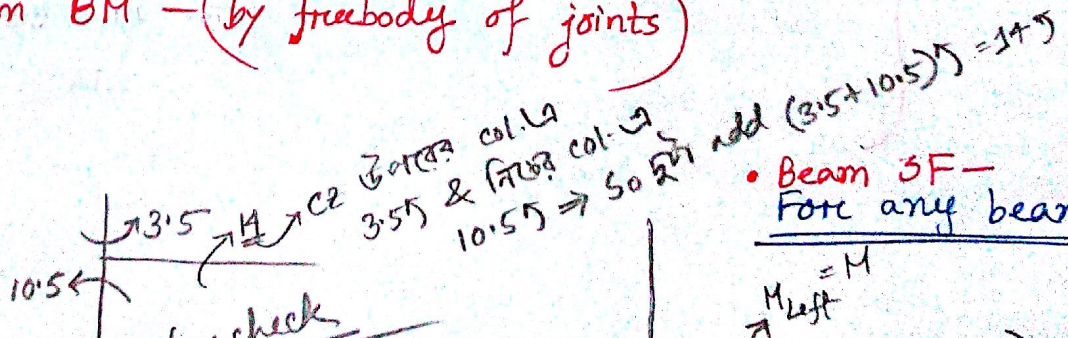
Wind load  
 left → 2.1E 1.2E  
 right → 2.1E then  
 direct<sup>n</sup> & by 2.1E  
 ↑  
 This depends on  
 wind load direct<sup>n</sup>

• Col. BM —

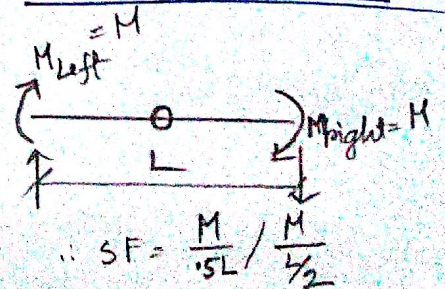


\* for both col & beam only joint →  
 load & Hinge  
 ↓  
 So middle → 0 &  
 upper & lower coordinate  
 equal of the BMD

• Beam BM — (by freebody of joints)

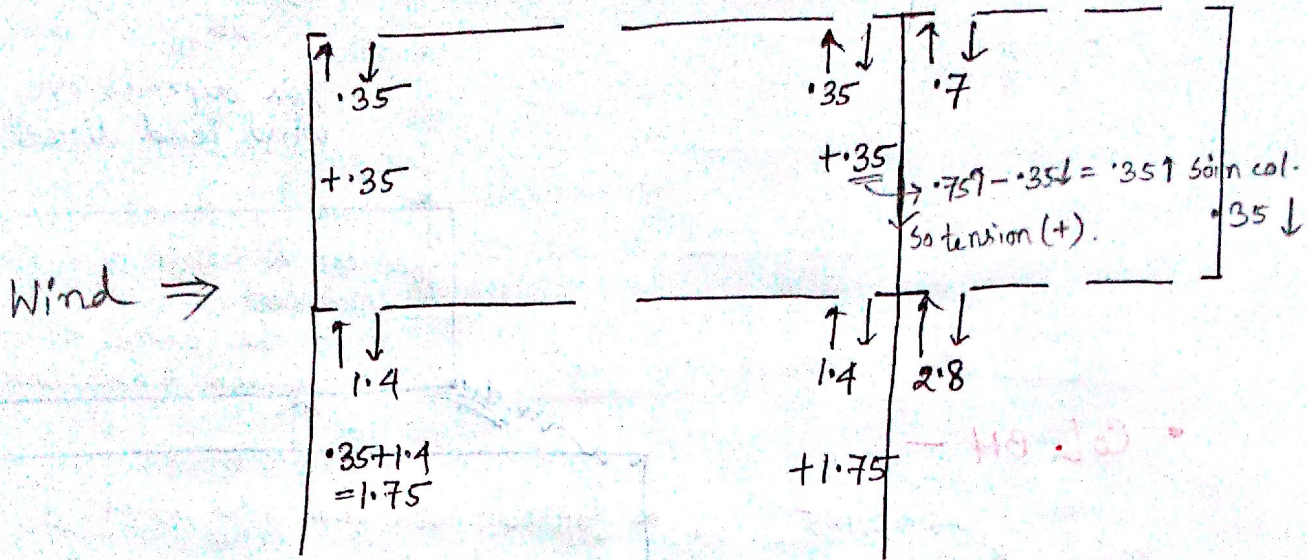


• Beam SF —  
 For any beam



$14 + 14 = 28$  & in col.  $7 + 21 = 28$   
 ↓  
 So BMs were right

• Col AF — (By Freebody)



Pattern  
Imp:

Wind (সম্মুখ) থেকে আসলে  $\Rightarrow$  ডানদিকের col. tension & বাঁদিকের col. এ compression

$\downarrow$

and মত building - এর বিভিন্ন দিকের মাল তে AF বাজত শাসন।

## CANTILEVER METHOD

### • Assumptions —

1. There is a hinge at center of each beam.
2. " " " " " " " " col.
3. Axial stress in a col. of a storey is  $\propto$  to Hon. distance of that col. from c.g. of cols of the storey.

i.e. Ax. stress  $\propto Y$

$$\Rightarrow \quad \quad = KY$$

$$\therefore \text{Ax. Stress} = \frac{MY}{I}$$

as Bending stress  
in col.