

Steel Structure

1	30 Aug	13	1 Nov	25	7 Dec
2	31 Aug	14	2 Nov	26	10 Dec
3	2 Sep	15	4 Nov	27	13 Dec
4	6 Sep	16	8 Nov		
5	7 Sep	17	9 Nov		
6	9 Sep	18	11 Nov		
7	14 Sep	19	15 Nov		
8	4 Oct	20	16 Nov		
9	5 Oct	21	25 Nov		
10	7 Oct	22	29 Nov		
11	11 Oct	23	30 Nov		
12	18 Oct	24	6 Dec		

CE 319 (3.00 Cr. Hr.)

Design of steel structures

Dr. Khan Mahmud Amanat

Main Text Book:

- 1) Steel structures: Design & Behaviour  
5th Edition (2009)  
Salmon/Johnson/Malhas  
(Pearson-Prentice Hall)

We refer to AISC (American Institute of Steel Construction) for steel construction.  
\* BNBC follow 2005 4th edition

Design Philosophy:

$$\text{Resistancy/Strength/Capacity} \geq \text{Effect of applied loads}$$

Limit State

→ failure limit

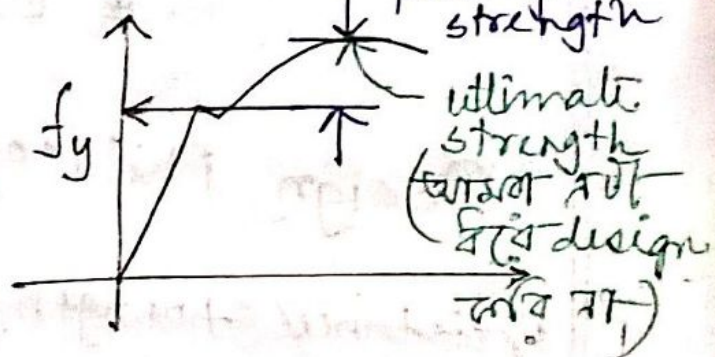
এখন একটি condition যখন তাই হলে  
কাজে unacceptable

deflection এর limit, span length  
এর 360 ডিগ্রি 1 ডিগ্রি এর 25,

- Strength Limit states
- serviceability " "
- special " "

Reserve Strength : unknown additional strength

unutilized part of strength



A5D

- \* load কে load এর  
সহ রাখা হয়,
- \* capacity কে safety  
factor দিয়ে-এক করে  
allowable capacity  
করে দেয়া হয়,



Material

Iron ore → Pig Iron  
(Ferrite)

କୋକା ଲାଡ଼େଇ-  
ବ୍ୟବହାର କରା  
ଗଞ୍ଜା ନା,

Steel is  
iron in  
usable form

Pig Iron → Wrought Iron

Pig Iron → Cast Iron

\* Steel ଏବଂ composition ନିର୍ଦ୍ଧାରିତ ହୁଏ ନା

Steel shapes

cold form shapes (eg. ଡେଇଁକିନ)

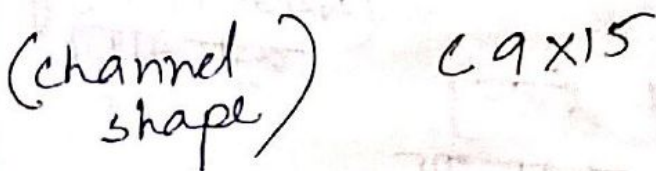
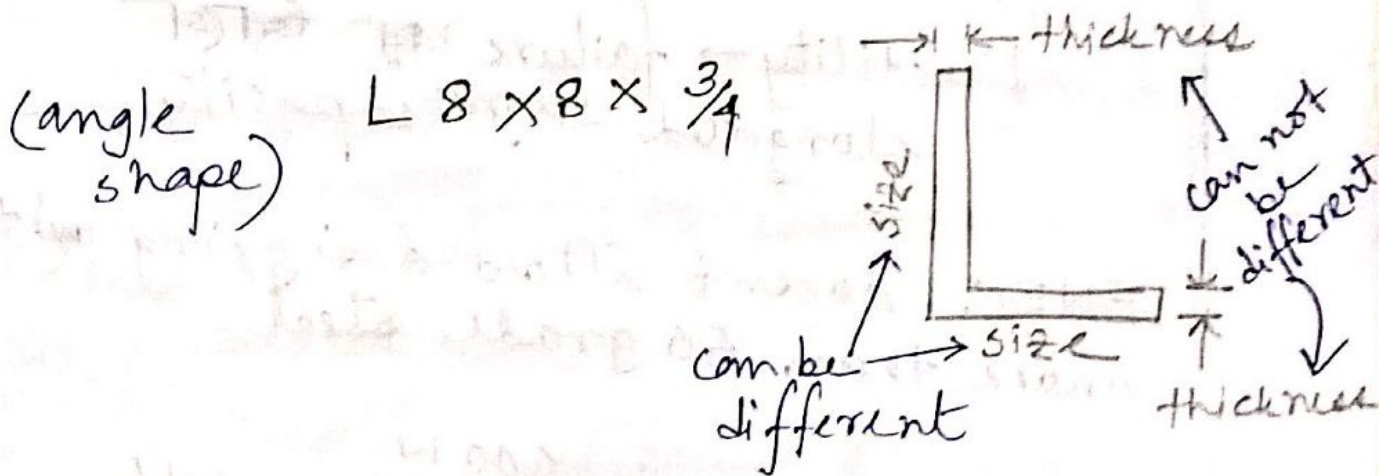
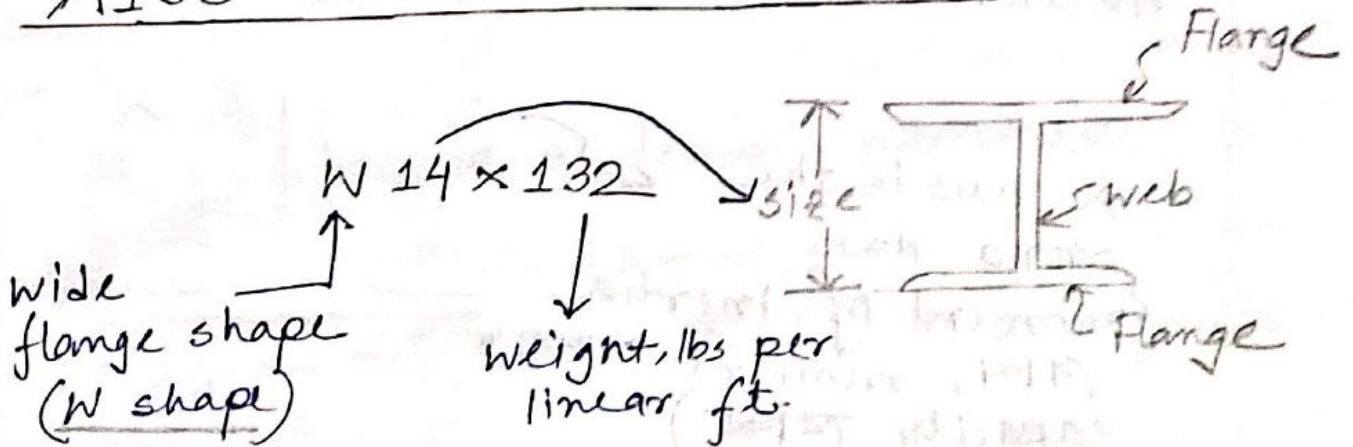
hot rolled shapes

→ red-hot ନିୟମ semi-solid  
condition - ଏ ନିୟମ କରା  
ହୁଏ ।

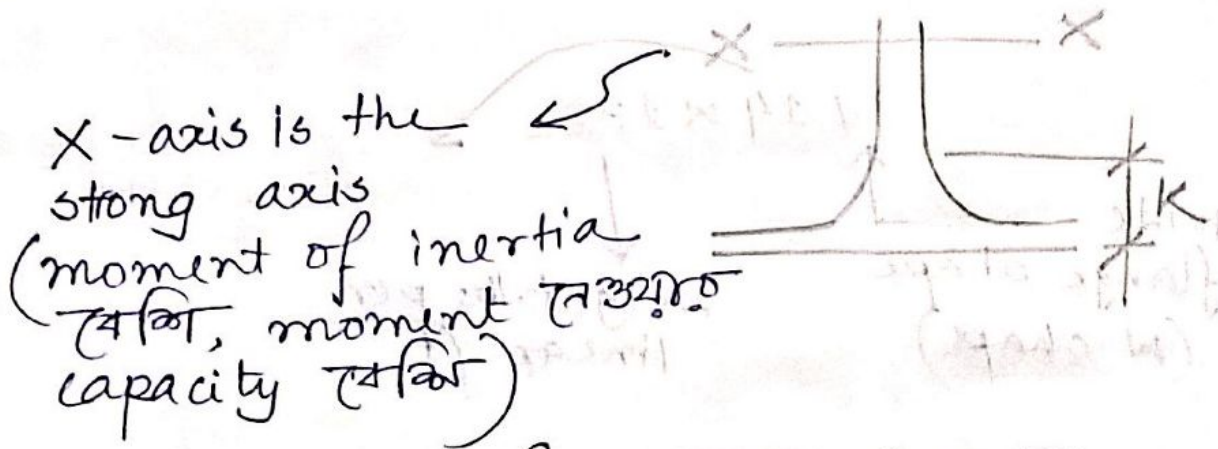
କୃତ୍ରିମ steel ଏବଂ sheet.  
normal temperature  
କରା ହୁଏ ।

Billet → structural steel ଏବଂ - solid  
ଫର୍ମାଟା ।

AISC Steel Construction Manual



\*  $\Delta$  depth of decimal fraction - এ নিশ্চিত হবে,



Ductility  $\rightarrow$  failure  $\Delta$  elongated capacity

BNBC doesn't allow designing with more than 50 grade steel.

500 W  $\rightarrow$  weldable (যদিও welding করা যায়)

\* prestressing এর ক্ষমতা 250 ksi steel use করা হয়,

$f_y \rightarrow$  yield strength

$f_u \rightarrow$  ultimate strength

For earthquake resistant design,

few thousand cycles at least

$$\frac{F_u}{F_y} \geq 1.25$$

\* fatigue is a design parameter for steel structures (specifically, railway bridges) only. RC design is not a design parameter.

Temperature  $\uparrow$ , yield strength  $\downarrow$

Steel toughness & temperature related.

### Elements of a steel building

column to column beams  $\rightarrow$  girder  
(vertical load, wind, earthquake load)

Tension Members - 1

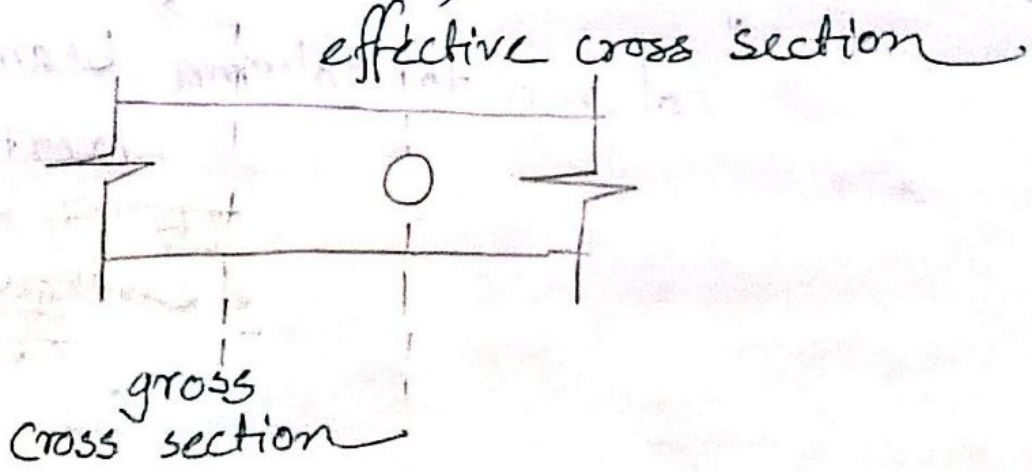
Suspension bridge এর hanger, cable tension member.

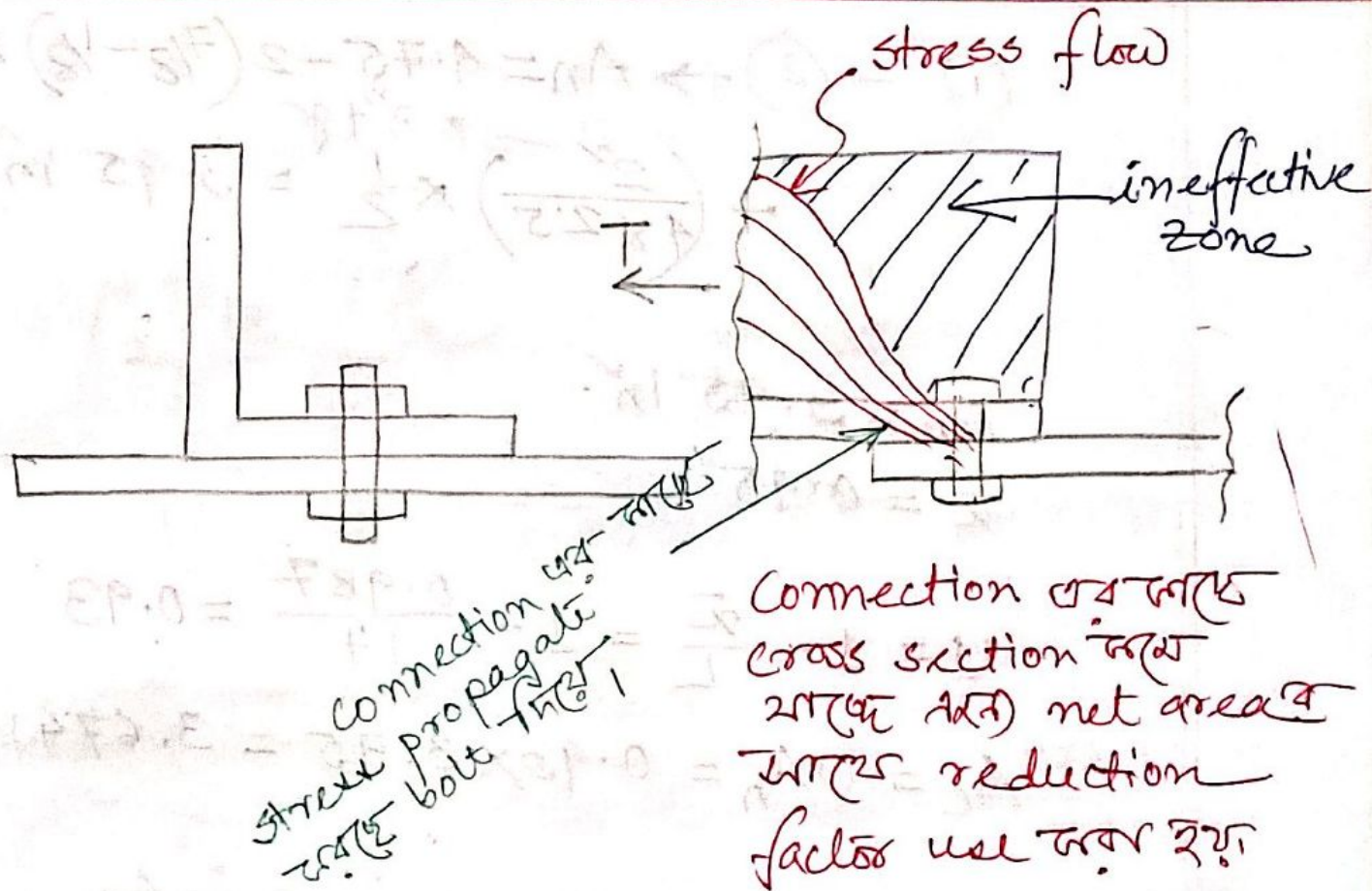
flexural member → x-section  
 এখানে ২টি হলে  
 ২টি I বক্স  
 হয়।

Bolt এর জন্য area লক্ষ্য গিয়ে ২টি  
 থাকবে মোট net area ( $A_n$ )

gross cross section

↪ away from section  
 (joint এর কাছে গিয়ে effective  
 cross section হিসেবে  
 লক্ষ্য হয়।)





$T_n \rightarrow$  Nominal strength

Grade 50 steel

$$F_y = 50 \text{ ksi}$$

$$F_u = 65 \text{ ksi}$$

$$A_g = 4.75 \text{ in}^2$$

Find net area

$$\text{Along } \textcircled{1} - \textcircled{1} \rightarrow A_n = 4.75 - \left(\frac{7}{8} + \frac{1}{8}\right) \frac{1}{2} = 4.25 \text{ in}^2$$

$$\textcircled{1} - \textcircled{2} \rightarrow A_n = 4.75 - 2 \left( 7/8 - 1/8 \right) \times \frac{1}{2} + \left( \frac{2^2}{4 \times 2.5} \right) \times \frac{1}{2} = 3.95 \text{ in}^2$$

$$A_n = 3.95 \text{ in}^2$$

$$\bar{x} = 0.95'$$

$$U = 1 - \frac{\bar{x}}{L} = 1 - \frac{0.987}{14} = 0.93$$

$$A_e = U A_n = 0.93 \times 3.95 = 3.674 \text{ in}^2$$

Yield on gross area,  $T_n = F_y A_g$

$$= 237.5$$

$$\text{ASD capacity} = \frac{T_n}{\Omega} = \frac{237.5}{1.67} = 142.2$$

Net area fracture,  $T_n = F_u A_e$

$$= 65 \times 3.674$$

$$= 238.81$$

$$\text{ASD capacity} = \frac{T_n}{\Omega} = \frac{238.81}{2.0} = 119.4$$

↓  
governs

$$P_u = 1.2D + 1.6L = 1.2 \times 35 + 1.6 \times 70 = 154 \text{ kips}$$

$$P_u \leq \phi P_n$$

$$P_n \geq \frac{P_u}{\phi} = \frac{154}{0.90} = 171.1 \text{ (gross area yielding)}$$

Yielding on gross area:

$$P_n = F_y A_g \Rightarrow A_g = \frac{P_n}{F_y} = \frac{171.1}{36} = 4.75 \text{ in}^2$$

$$\text{on net area, } P_n = \frac{P_u}{\phi} = \frac{154}{0.75} = 205.3$$

$$P_n = A_e F_u \Rightarrow A_e = \frac{P_n}{F_u} = \frac{205.3}{58} = 3.54$$

Assume,  $u = 0.8$  (to be conservative design)

$$A_e = u A_n \Rightarrow A_n = A_e / u = 4.43 \text{ in}^2$$

$$A_g = A_n + \text{bolt hole}$$

$$= 4.43 + 2 \left( \frac{3}{4} + \frac{1}{8} \right) t$$

$\rightarrow \frac{1}{2}$  " assume

choose, L6x4x  $\frac{9}{16}$

(no comp)  $\frac{1.171}{25} = \frac{1.01}{0.45} = \frac{2.24}{0.45} = 5.0$

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Block Shear Failure

$$F_y = 36 \text{ ksi}$$

$$F_u = 58 \text{ ksi}$$

$$A_g = 8 \times \frac{5}{8} = 5 \text{ in}^2$$

1) Yielding on gross area,  $T_n = F_y A_g$   
 $= 36 \times 5$   
 $= 180 \text{ k}$

LRFD  $\phi T_n = 0.9 \times 180 = 162 \text{ k}$

ASD  $\frac{T_n}{\Omega} = \frac{180}{1.67} = 107.8 \text{ k}$

2) Fracture on Net Area

$$A_n = 5 - 2 \left( \frac{3}{4} + \frac{1}{8} \right) \cdot \frac{5}{8}$$

$$= 3.91 \text{ in}^2, \quad (U = 1)$$

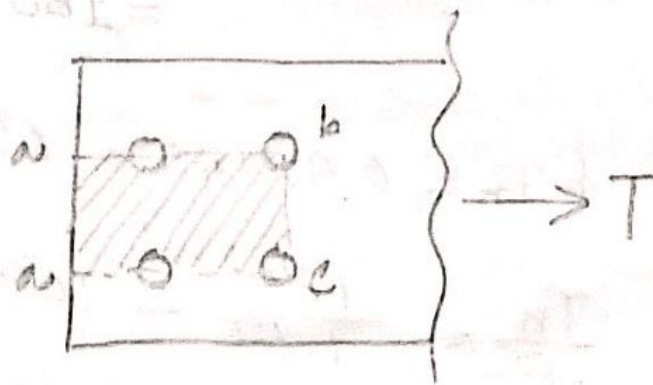
$$A_e = U A_n = 3.91$$

$$T_n = F_u A_e = 58 \times 3.91 = 226.8 \text{ k}$$

$$\text{LRFD} \quad \phi T_n = 0.75 \times 226.8 = 170.1 \text{ k}$$

$$\text{ASD} \quad \frac{T_n}{\Omega} = \frac{226.8}{2.0} = 113.4 \text{ k}$$

### Block Shear Mode - 1



$$\text{Tension, } A_n = \text{Area } bc = \left\{ 5 - 2 \times \frac{1}{2} \left( \frac{3}{4} + \frac{1}{8} \right) \right\} \frac{5}{8}$$

$$= 2.578 \text{ in}^2$$

$$\text{Shear, } A_{gv} = \text{Area } ab + de$$

$$= \left( 1 \frac{1}{2} + 2 \frac{1}{2} \right) \times 2 \times \frac{5}{8} = 5 \text{ in}^2$$

$$\text{Shear, } A_{nv} = A_{gv} - \text{bolts}$$

$$= 5 \left( 1 + \frac{1}{2} \right) \left( \frac{3}{4} + \frac{1}{8} \right) \frac{5}{8} \times 2$$

$$= 3.36 \text{ in}^2$$

$$\text{Tension rupture} = F_u U_{bs} A_{nt}$$

$$= 58 \times 1 \times 2.578$$

shear

$$\text{Tension yielding} = 0.6 F_y A_{gv}$$

$$= 0.6 \times 36 \times 5 = 108$$

$$\text{Shear rupture} = 0.6 F_u A_{nv}$$

$$= 0.6 \times 58 \times 3.36$$

$$= 116.93$$

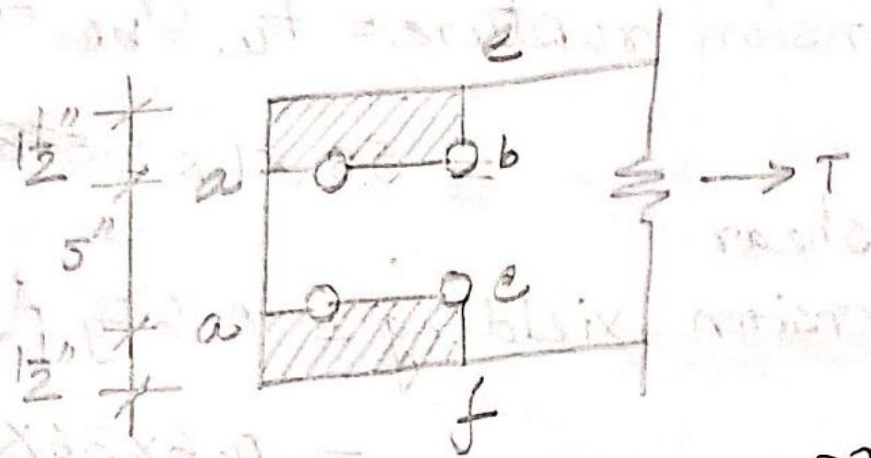
108<sup>k</sup>

$$T_n = 149.5 + 108 = 257.5 \text{ k}$$

$$F_y = 36 \text{ ksi}$$

$$F_u = 58 \text{ ksi}$$

## Block Shear Mode - 2



$$A_{nt} = \left\{ \left( 1\frac{1}{2} + 1\frac{1}{2} \right) - 2 \times \frac{1}{2} \times \left( \frac{3}{4} + \frac{1}{8} \right) \right\} \frac{5}{8}$$

$$= 1.328 \text{ in}^2$$

$$\text{Tension rupture} = F_u U_{bs} A_{nt}$$

$$= 58 \times 1 \times 1.328 = 77.02 \text{ k}$$

$$\left. \begin{array}{l} \text{Shear Yield} \\ \text{Shear rupture} \end{array} \right\} \rightarrow 108 \left. \vphantom{\begin{array}{l} \text{Shear Yield} \\ \text{Shear rupture} \end{array}} \right\} T_n = 77.02 + 108$$

$$= 185 \text{ k} \rightarrow \text{Governing Block Shear}$$

Final

yielding on gross : LRFD : 162  
ASD : 107.8

Fracture on net : LRFD : 170.1  
ASD : 113.4

Block Shear : LRFD: 138.75  
ASD : 92.5

→ Final Ans.



Steel Structure-7

14 September 2015

*[Faint, illegible handwritten notes covering the majority of the page, likely bleed-through from the reverse side.]*

$$F_{nv} = 0.4 F_u^b \text{ (shear plane in the thread)}$$

$$= 0.5 F_u^b \text{ (shear plane outside the thread)}$$

\* \* \*  $F_{nv}$  ବଳ ନା ଅନୁମତ assume the more conservative case.

Bearing failure ଏବଂ ଥିଏଟ୍ ପ୍ଲେଟ୍ ଏବଂ material property  $(F_u, F_y)$  ଥିବାର ଧ୍ୟାନ ରଖିବାକୁ ହେବ।

\*  $L_c$  has a minimum and a maximum value.  
 (to prevent failure by shear) 2.5d  
(ଏବଂ ଥିଏଟ୍-ସାପୋର୍ଟ୍ ଫାଏଲ୍ୟୁର ଏବଂ-ସମ୍ପର୍କିତ ଅନୁମତ)

Pg. 8-15  
Example

Bolt ଏବଂ limit state  
 (ଏବଂ ଥିଏଟ୍- chapter ଏବଂ ଥିଏଟ୍ ପ୍ଲେଟ୍ ଏବଂ limit state ଏବଂ ଥିଏଟ୍),

Bolt A325,  $\rightarrow F_u^b = 120 \text{ ksi}$   
 plate A572 Gr. 50  $\rightarrow F_y = 50 \text{ ksi}$   
 $F_u = 65 \text{ ksi}$

## Bolt shear

Threads are excluded

one bolt,  $R_n = m A_b F_{nv}$

$$= 1 \times \left\{ \frac{\pi}{4} \left( \frac{7}{8} \right)^2 \right\} \times 0.5 F_u^b$$

$$= \frac{\pi}{4} \times \left( \frac{7}{8} \right)^2 \times 0.5 \times 120$$

$$R_n = 36 \text{ kip}$$

shear capacity governs

## Bolt Bearing

Formula  $1.2 L_e t F_u < 2.4 d t F_u$  ← plate material

$$2.4 d t F_u = 2.4 \left\{ \frac{7}{8} + \frac{1}{8} \right\} \times \frac{5}{8} \times 65$$

$$= 91.41 \text{ kip}$$

Exterior bolt

\*\*  $L_e$  এখানে bolt এর জন্য এখানে বক্রতা হতে পারবে

$$1.2 L_e t F_u = 1.2 \left\{ \frac{1}{2} - \frac{\frac{7}{8} + \frac{1}{16}}{2} \right\} \times \frac{5}{8} \times 65$$

এখানে ইঞ্চি থেকে half of the hole diameter বাদ দেব

$$= 50.27 \text{ kips} < 2.4 d t F_u$$

ok

## Interior bolts

$$1.2 L_c t F_u = 1.2 \left\{ 3 - \left( \frac{7}{8} + \frac{1}{6} \right) \right\} \times \frac{5}{8} \times 65$$
$$= 100.5 \text{ k} > 2.4 d t F_u$$

not ok

For interior bolts, bearing capacity is 91.41 kips.

## Considering Bolt shear

$$\text{LRFD Capacity} = 4 \times 36 \times \phi$$
$$= 108 \text{ k} \quad \phi \downarrow 0.75$$

$$\text{ASD Capacity} = \frac{4 \times 36}{\Omega \rightarrow 2}$$
$$= 72 \text{ k}$$

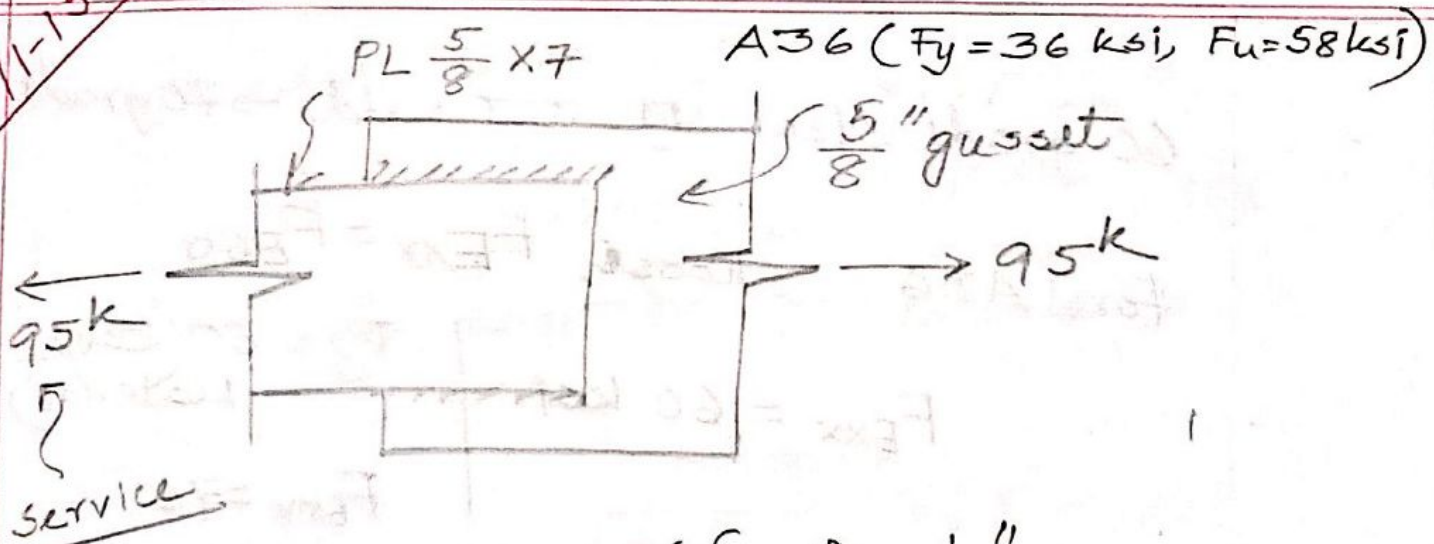
Steel Structure-9 5 October 2015

Shielded Metal Arc Welding (SMAW)

Electrode ଏବଂ ସାହିତ୍ୟିକ element ଏବଂ  
କିମ୍ବା gas ଏବଂ, ତାହା metal ଏବଂ oxide ତିଆରି  
ଏବଂ ନା,

- \* cast iron ଏବଂ ବିମ୍ବ welding ଏବଂ ନା,
- \* types of welded joints and types of welds are different.

11-15



Min<sup>m</sup> weld size (fillet) =  $\frac{1}{4}$ "

Max<sup>m</sup> " " =  $\frac{5}{8} - \frac{1}{16}$

(since plate thickness  $> \frac{1}{4}$ " )

=  $\frac{9}{16}$ "

chosen size =  $\frac{1}{2}$ "

Effective throat,  $t_e = 0.707 a$   
 $= 0.354"$

Weld:  $\frac{R_n}{\Omega} = \frac{6 \times F_{EXX} t_e}{2.0}$

ଅନୁସାରେ ସିଲି  
 ବନ୍ତା କେବଳ, ତାହା -  
 always ବାବଦ,  $F_{EXX} = F_{E60}$

50 grade steel କା- କ୍ରମ weld  $\rightarrow$  60 grade

60 grade steel 4th class weld  $\rightarrow$  70 grade

For, A36, choose,  $F_{EXX} = F_{E60}$

$$F_{EXX} = 60 \text{ ksi}$$

$$F_y = 50 \text{ (Base Material)}$$

$$F_{EXX} = 70$$

weld:

$$\frac{R_n}{\Omega} = \frac{0.6 \times 60 \times 0.354}{2}$$

$$= 6.4 \text{ k/"}$$

Base Material

(plate)

$$\text{yield: } 0.6t \frac{F_y}{\Omega} = 0.6 \times \frac{5}{8} \times \frac{36}{1.5}$$

$$= 9 \text{ k/"}$$

$$\text{Rupture: } 0.6t \frac{f_u}{\Omega} = 0.6 \times \frac{5}{8} \times \frac{58}{2}$$

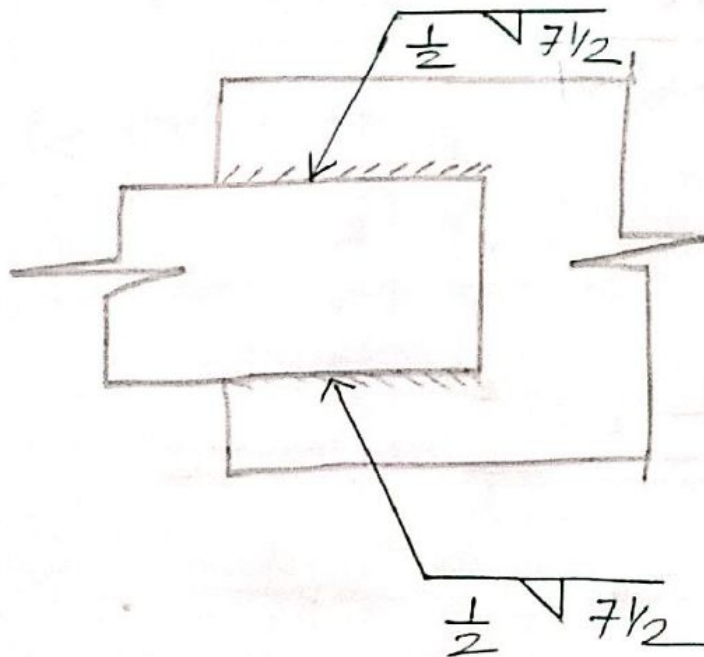
$$= 10.9 \text{ k/"}$$

So, weld length =  $\frac{9.5}{6.4} = 14.84"$

ଅର୍ଥାତ୍ plate thickness different ଅଟେ

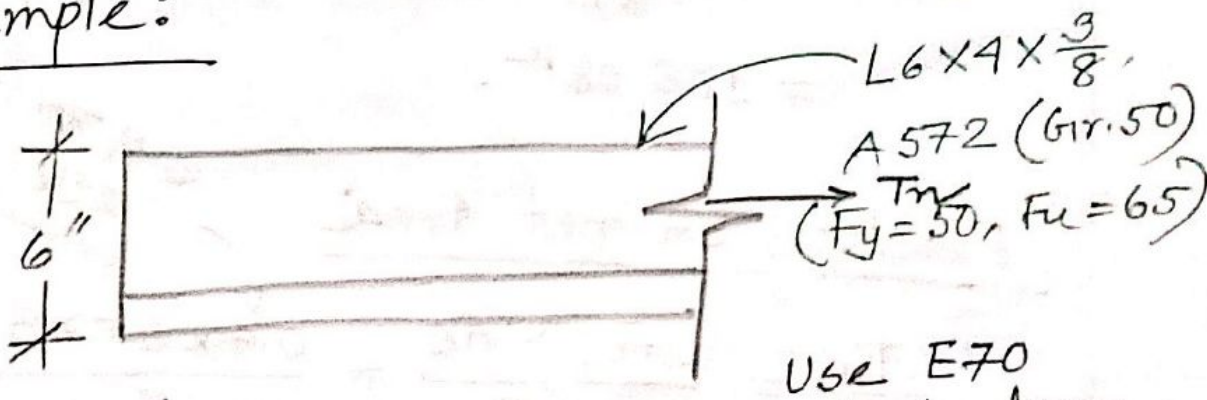
ତାହାଙ୍କ minimum ଉପର-କ୍ରମ calculation

ହେବ,



~~11-17~~

Example:



Use E70 electrodes.  
Follow ASD.

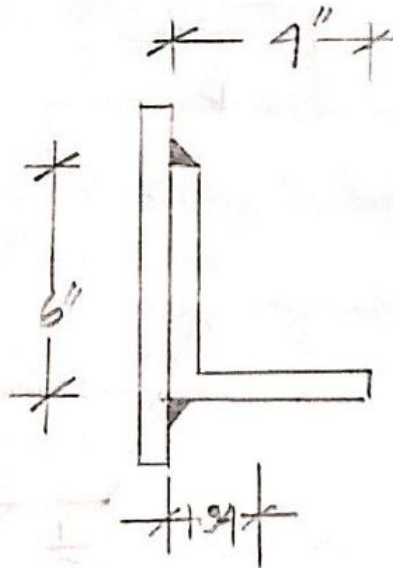
Angle Length:

$$\begin{aligned} \text{yield on gross area: } & \frac{T_n}{\Omega} \\ & = \frac{F_y A_g}{\Omega} \end{aligned}$$

From chart

$$A_y = 3.61$$

Yield on  
Gross area



$$\begin{aligned} \frac{F_y A_g}{\Omega} & = \frac{50 \times 3.61}{1.67} \\ & = 108.08 \text{ k} \end{aligned}$$

Fracture on net Area

$$\begin{aligned} \frac{T_n}{\Omega} & = \frac{U_{f_u} \times A_e}{\Omega} = \frac{0.9 \times 65 \times 3.61}{2} \\ & = 105.6 \text{ ksi} \end{aligned}$$

0.9, conservative  
2 (U<sub>f<sub>u</sub></sub>)  
conservative

So, 105.6k governs

Weld size:

$a = \frac{1}{4}$ " based on max<sup>m</sup> and  
min<sup>m</sup> criteria

$$t_e = 0.707 \times \frac{1}{4} = 0.177"$$

$$\begin{aligned} \text{weld capacity} &= \frac{R_n}{\Omega} = \frac{0.6 t_e F_{Exx}}{2} \\ &= \frac{0.6 \times 0.177 \times 70}{2} \\ &= 3.72 \text{ k/"} \end{aligned}$$

$$\text{Total weld length} = \frac{105.6}{3.72} = 28.4$$

12-7

welding एव size बना ता थापलन आकार  
force per unit length एव करि,

1) c.g. of weld

$$L\bar{x} = L_1x_1 + L_2x_2 + L_3x_3$$

$$\Rightarrow (8 + 2 \times 6)\bar{x} = 6 \times 3 + 8 \times 0 + 6 \times 3$$

$$\Rightarrow \bar{x} = 1.8''$$

$$P = 15k$$

$$M = Pe = 15(8 + 6 - 1.8)$$
$$= 183k''$$

2)  $I_p$  about c.g.

$$I_p = \left[ \frac{6 \times 1^3}{12} + 6 \times 1 \times 4^2 \right] \times 2 + \frac{1 \times 8^3}{12}$$

$$+ \left[ \frac{1 \times 6^3}{12} + 1 \times 6 \times 1.2^2 \right] \times 2 +$$

$$\left[ \frac{8 \times 1^3}{12} + 8 \times 1 \times 8^2 \right]$$

$$= 23.56 + 79.87$$

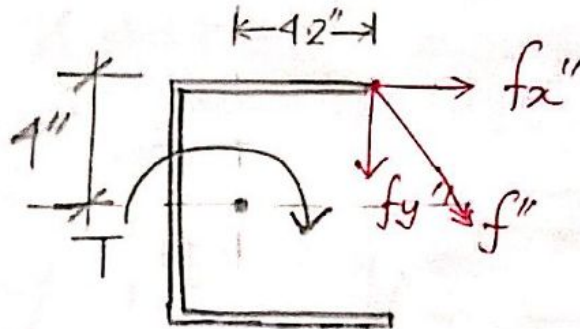
$$= 315.5$$

3) Stress

$$\text{Direct stress} = \frac{P}{L} = \frac{15}{6 \times 2 + 8} = 0.75 \text{ k/in} \downarrow$$

Due to moment

Maximum stress shall occur at a point furthest from c.g. i.e. points A or B.



$$f_x'' = \frac{T_y}{I_p} = \frac{183 \times 4}{315.5} = 2.32 \rightarrow$$

$$f_y'' = \frac{T_x}{I_p} = \frac{183 \times 4.2}{315.5} = 2.44 \downarrow$$

$$\begin{aligned} \text{Resultant} &= \sqrt{(2.44 + 0.75)^2 + 2.32^2} \text{ k/in} \\ &= 3.96 \text{ k/in} \end{aligned}$$

Compression Member - 1

Column: (primarily compression member, assisted by small bending)

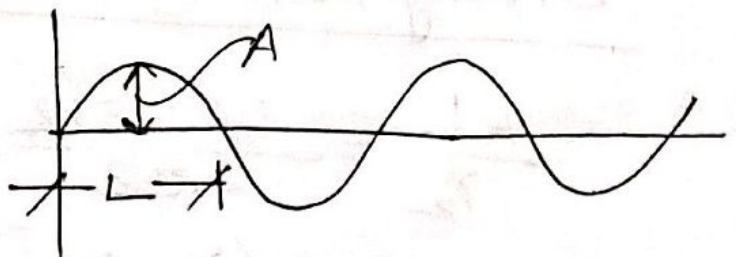
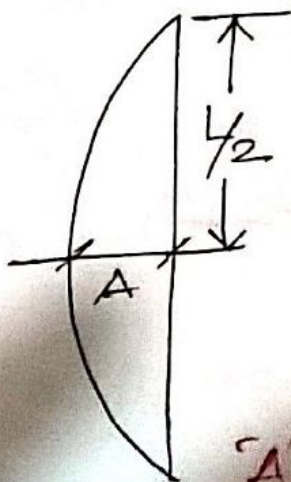
\* Euler's Buckling Theory is applicable to elastic buckling only.

General Eq<sup>n</sup> of curvature:

$$\frac{d^2 y}{dx^2} \left[ \sqrt{1 + \left( \frac{dy}{dx} \right)^2} \right]$$

↳ for small deflection zero.

$$y = A \sin kx$$

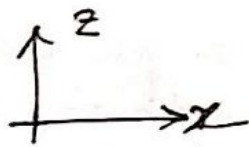


sine curve

A can be figured out using non-linear higher order equations.

Weak Direction

Angle and wide flange  
connection design  
easier.



normal to the plane  $\rightarrow$  Y axis

General tendency is to buckle along the  
weak axis.

axis - 4, moment of inertia  $I_{yy}$ ,

AISC 2005 / BNBC 2015 -  $\phi_c P_n$

$\frac{ksi, inch, k}{\text{for stress for length for force}}$

remember { concrete  $\phi_c$  formula psi  $\phi$   
steel  $\phi_c$  formula ksi  $\phi$

Strong Axis (x-axis)

$$\frac{KL}{r_x} = \frac{1.0 \times 312}{4.32} = 72.2$$

Weak Axis (Y-axis)

$$k = 0.5$$

$$\frac{KL}{r_y} = \frac{0.5 \times 312}{2.01} = 77.61 \quad \left( \begin{array}{l} \text{Governs, to} \\ \text{stay on the} \\ \text{safe side} \end{array} \right)$$

$\frac{KL}{r}$  ଅଠାଏକା columnର  
capacity ଓଠାକା,

$$4.71 \sqrt{\frac{E}{F_y}}$$

$$= 4.71 \sqrt{\frac{29000}{50}}$$

$$= 113.4 > \frac{KL}{r_y}$$

$$F_c = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2} = \frac{\pi^2 \times 29000}{(77.61)^2} = 47.52$$

$$F_{cr} = \left[ 0.658 \frac{F_y}{F_c} \right] F_y$$

$$= 32.17 \text{ ksi}$$

$$P_{cr} = A_g F_{cr} = 13.3 \times 32.2 = 428.11 \text{ k}$$

$$\text{ASD} \Rightarrow \frac{P_{cr}}{\Omega} = \frac{428.11}{1.67} = 256.4 \text{ k}$$

Design का सबसे हल steel section लेना सबसे lightest therefore more economic.

\* सबसे जरूरी elasto-plastic buckling prefer करिए, यहाँ  $\frac{KL}{r} < 113$  लेना है।

W section का  $r_y$  is minimum.

\* use three trials in exam.

Steel Structure - 15 4 November 2015

\* Bracings prevents sway.

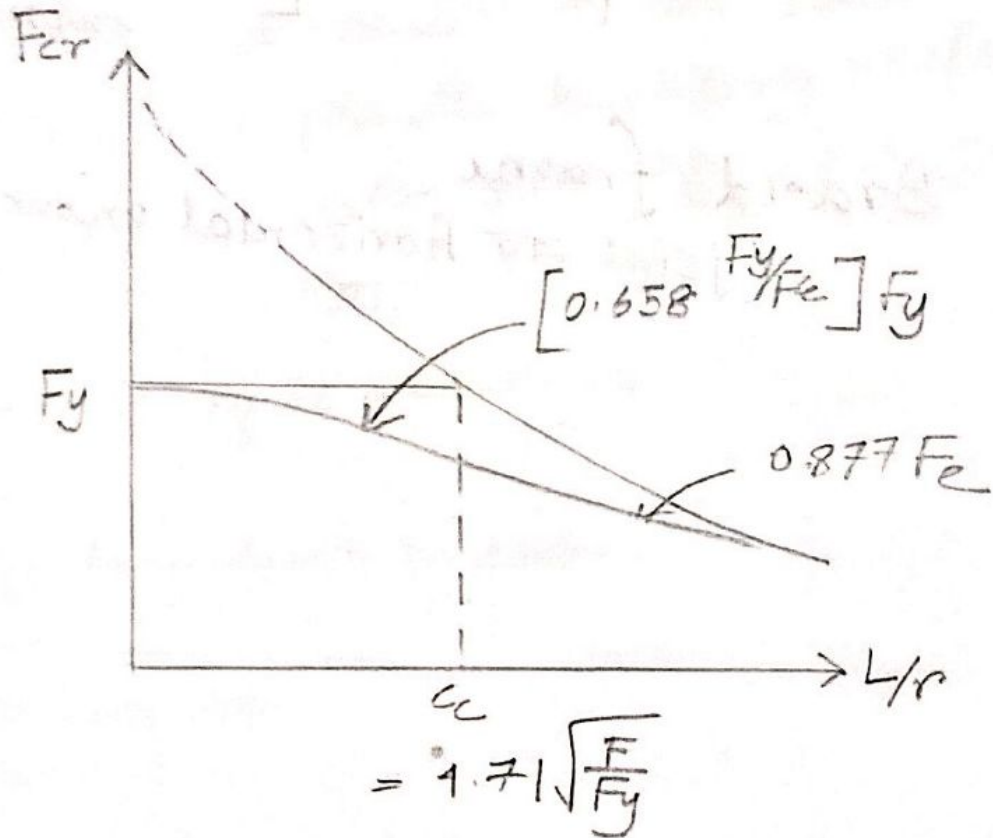
$$\text{Stiffness} = \frac{EI}{L} \rightarrow \text{axis of bending}$$

about -  
moment of inertia

Braced frame

(Joint is horizontal movement zero)

Effect of Residual Stresses in  
Compression Members



## Flexural Members

Beam connected to column  $\rightarrow$  girder

spandrel  
beam

↑  
column

very short  
depth and  
length is  
very same

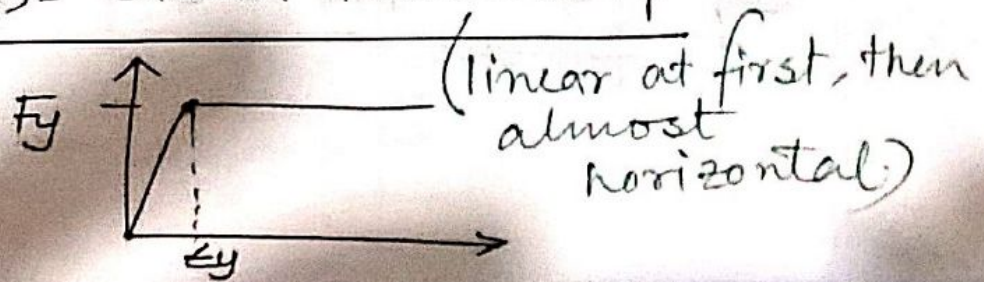
Truss acts as a beam. top chord  
compression, bottom chord tension

$$\sigma = \frac{Mc}{I} = \frac{M}{I/c} = \frac{M}{S}$$

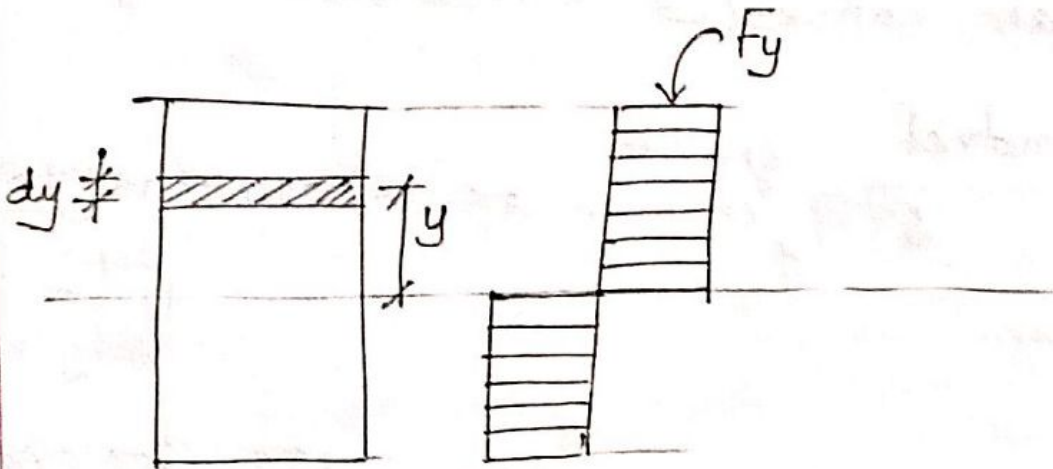
$$\Rightarrow M = \underbrace{\sigma}_{F_y} S \rightarrow \text{section modulus}$$

Maximum moment capacity,  $M_n = M_y$   
 $= S_x F_y$

## Basic Stress-Strain Relationship



जब तक fiber yield नहीं होत तब plastic moment प्राप्त नहीं करता,



$$dM = F_y dA y$$

$$M_p = \int F_y dA y$$

$$= F_y \int y dA$$

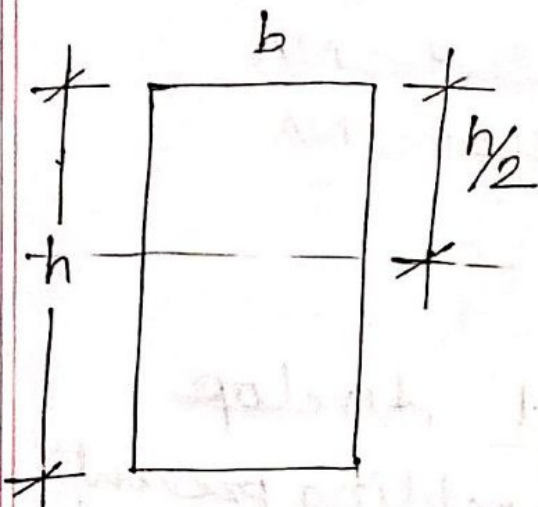
moment of area about the neutral axis

Plastic Section Modulus:

$$Z = \int y dA$$

(Chi)  $\frac{I_p}{I} = \frac{M_p}{M_y} = \frac{Z}{S}$  shape factor

## Shape Factor, $\xi$



$$S_x = \frac{I}{c} = \frac{bh^3}{12} / \frac{h}{2} = \frac{bh^2}{6}$$

$$Z = \int y dA = (b \times \frac{h}{2}) \times \frac{h}{4} \times 2 = \frac{bh^2}{4}$$

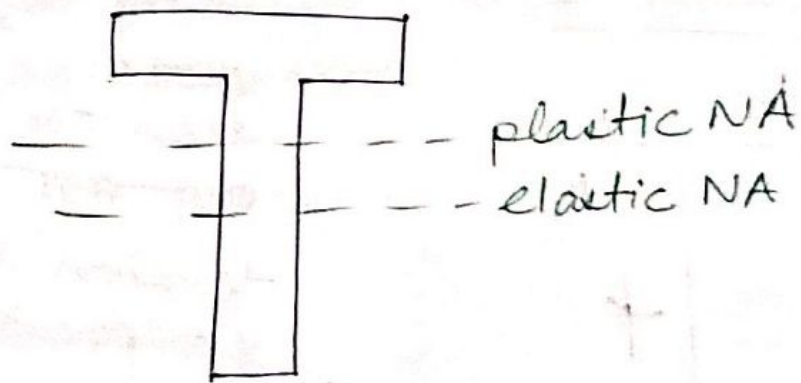
$$\xi = \frac{Z}{S} = \frac{6}{4} = 1.5$$

Unsymmetric section - Elastic NA  
is; Plastic NA same as NA,

Plastic conditions  
neutral axis एवं  
वक्र विस्थापन ३ मं.

cross sectional  
area एक छेदक  
भाग में, एकरे,

tension area =  
compression area

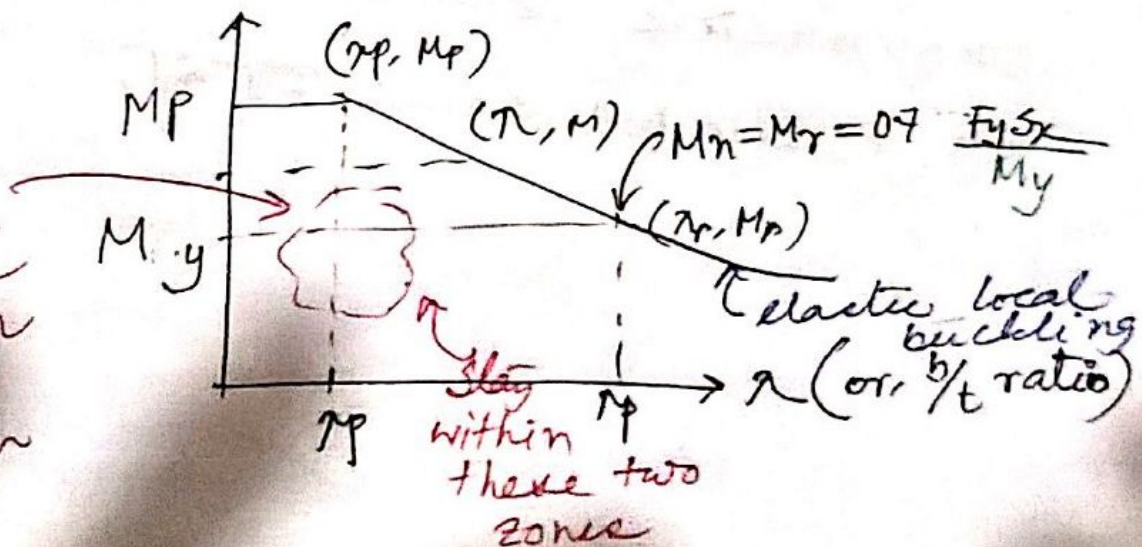


\* plastic moment develop  
 flange  $\uparrow$  local buckling prevent  
 କ୍ଷମକ ସମ୍ପାଦନ  
 କ୍ଷମକ ହେବ।

# ଫର୍ମାଟିଭ୍-limit  $\uparrow$  state plastic moment capacity

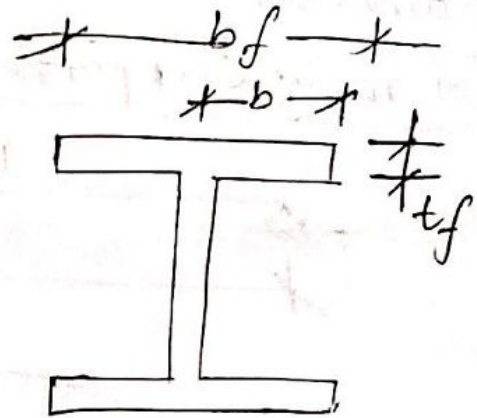
$$M_n = M_p = F_y Z_x$$

for seismic design stay on this design

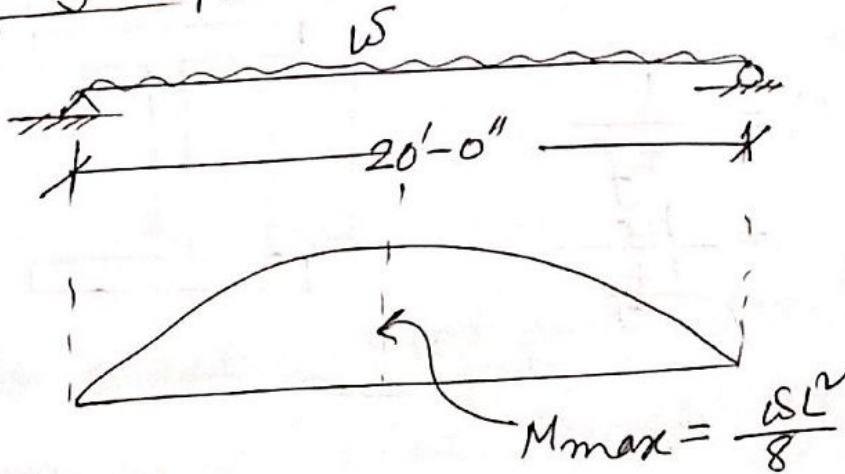


# Design capacity

$$\lambda = \frac{b}{t_f}$$
$$= \frac{b_f}{2t_f}$$



Nominal Moment Capacity of a Beam Section  
Laterally Supported Beams: ASD Design



$$DL = 0.2 \text{ k/ft}$$

$$LL = 0.8 \text{ k/ft}$$

$$w = 0.2 + 0.8 = 1.0 \text{ k/ft} = \frac{1}{12} \text{ k/in}$$

$$L = 20' = 240''$$

Distance  $\frac{1}{12}$  unit  
240 inches  $\times \frac{1}{12}$

$$M = \frac{wL^2}{8} = \frac{(1.0) \times 240^2}{8} = 600 \text{ k''}$$

↑  
Required ASD  
capacity

$$\Omega = 1.67$$

Required Nominal capacity

$$= 600 \times 1.67 = 1002 \text{ k''}$$

For compact section,  $M_n = M_p = F_y Z_x$

$$\Rightarrow Z_x = \frac{M_p}{F_y} = \frac{1002}{36} = 27.83 \text{ in}^3$$

Choose a section where  $Z_x \geq 27.83 \text{ in}^3$

A36,  $F_y = 36 \text{ ksi}$

From AISC manual table, we choose W8X31

$$Z_x = 30.4$$

$$\text{check, } M_n = M_p = F_y Z_x = 36 \times 30.4 = 1094 \text{ k''}$$

$$M = \frac{M_u}{\Omega} = \frac{1094}{1.67} = 655.3 \text{ k''}$$

$$\omega = \frac{8M}{L^2} = 0.091 \text{ k/''} = 1.092 \text{ k/'} \quad (\text{DL})$$

$$\frac{1}{0.092 \text{ k}} = 92^\# > 31^\# \quad \text{ok}$$

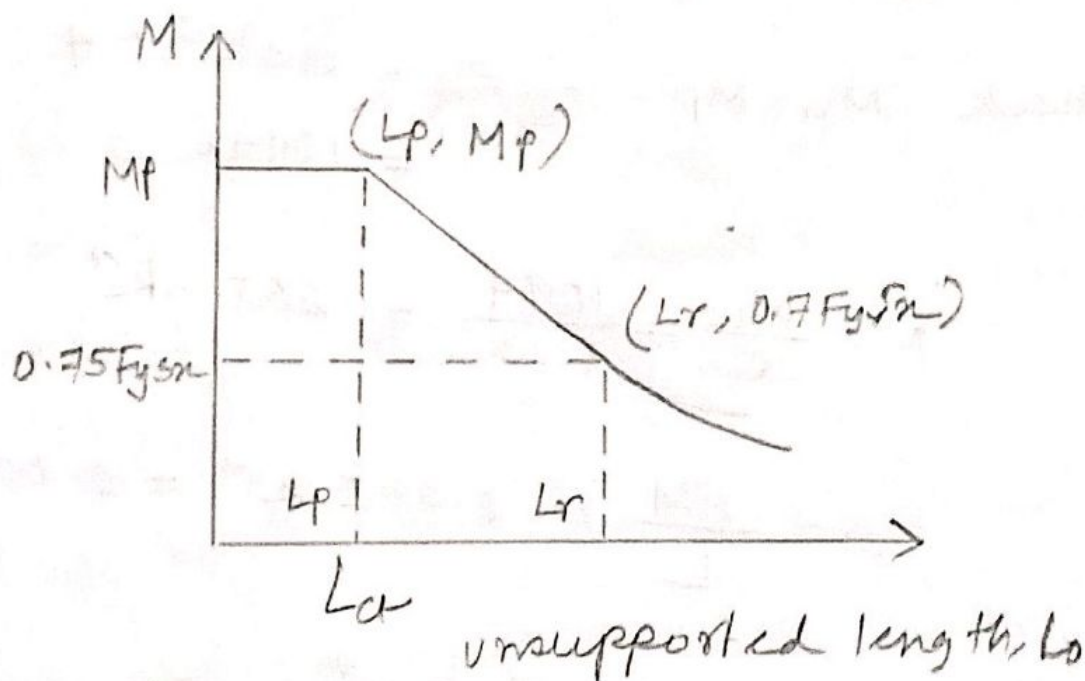
- Lateral torsional buckling depends on the unsupported length.

(Derivation → Not important, দরকার নেই)

$M_{cr} > M_{PM}$  plastic moment & fail করে,

অন্যভাবে, ok

\* Formula গুলো দ্বারা হবে না,



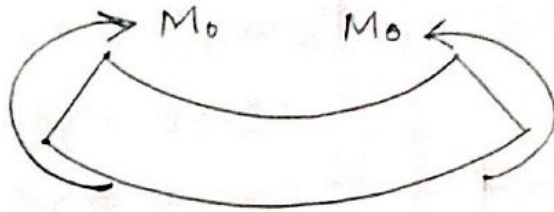
$$\frac{L_P}{r_y} = 1.76 \sqrt{\frac{E}{F_y}} = \frac{300}{\sqrt{F_y} \rightarrow \text{ksi}}$$

$$L_r = 1.95 r_{ts} \frac{E}{0.7 F_y} \sqrt{\frac{J_c}{S_x h_o} \sqrt{1 + \sqrt{1 + 6.76 \left( \frac{0.7 F_y S_x h_o^2}{E J_c} \right)^2}}}$$

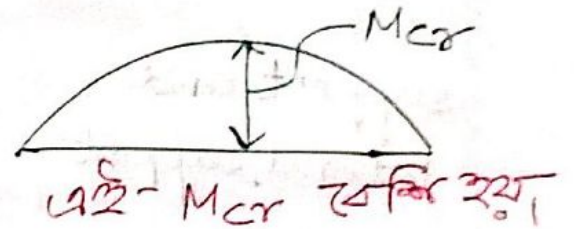
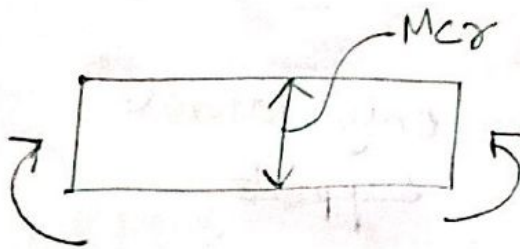
no need to memorize

Symmetric shape,  $c=1$

$C_b$  = modification factor.

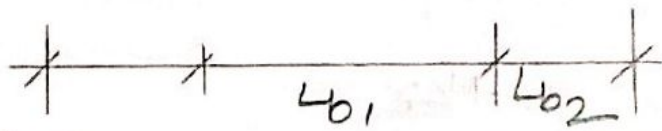
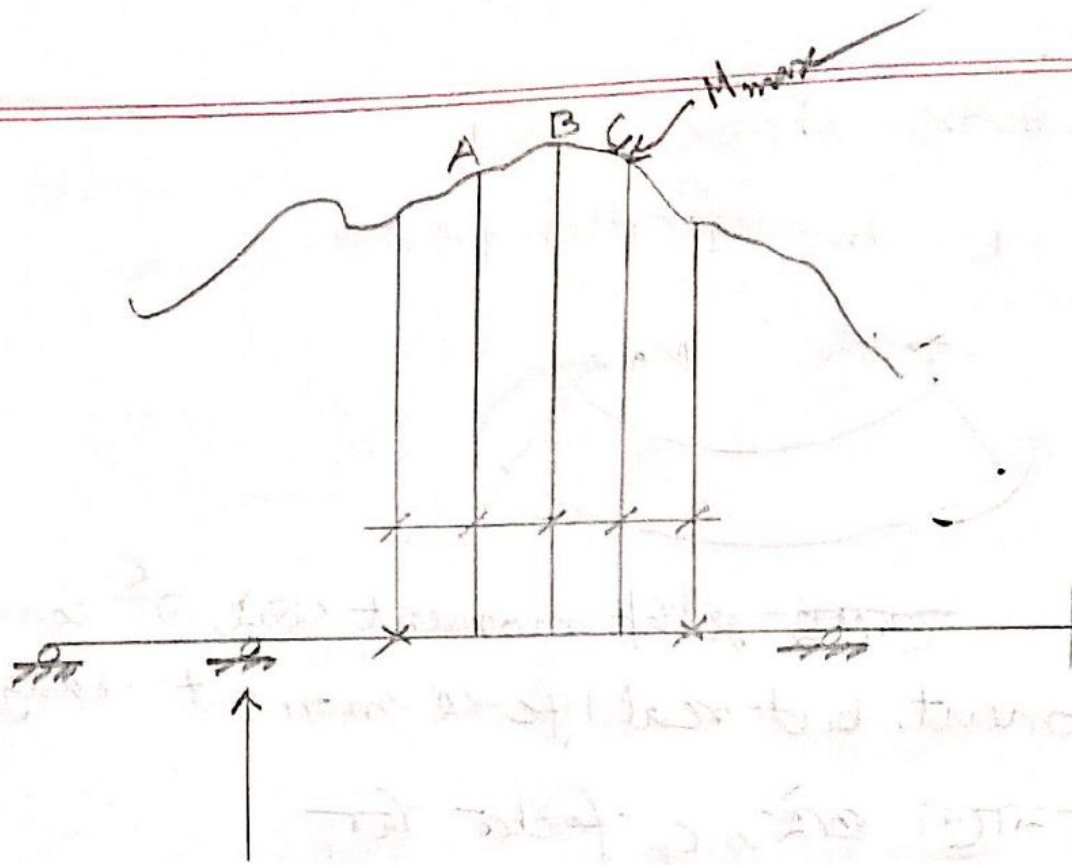


এখানে - ধ্রুৱী moment আছে, তাই constant moment, but real life-এ moment change হলে-যাবে তাই,  $C_b$  factor দিবে,



তাই,  $C_b > 1$ ,

$C_b \sim 1-3$



support and lateral support

only lateral support

W14x74  $F_y = 50 \text{ ksi}$

From, AISC chart  $\rightarrow$  compact section

$$\begin{array}{l|l|l} A = 21.8 \text{ in}^2 & b_x = 112 & r_{ts} = 2.82 \\ b_f = 10.1'' & r_y = 2.48 & h_o = 13.4 \\ t_f = 0.785'' & Z_x = 126 & J = 3.87 \\ & & C_w = 5990 \end{array}$$

1) Continuous lateral support  $\rightarrow$  No Lateral Bracing  
Compact section

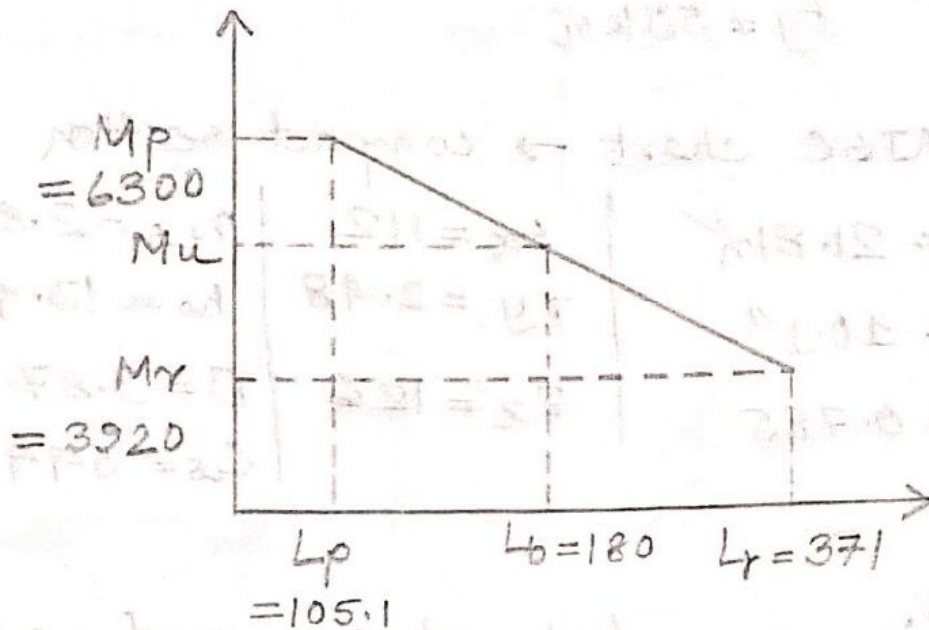
$$M_n = M_p = F_y Z_x = 50 \times 126 = 6300 \text{ k''}$$

LRFD:  $\phi M_n = 0.9 \times 6300 = 5670 \text{ k''}$

2) Unbraced Length = 15',  $C_b = 1$

Find  $L_p$ :

$$\frac{L_p}{r_y} = 1.76 \sqrt{\frac{E}{F_y}} \Rightarrow L_p = 1.76 \sqrt{\frac{29000}{50}} \times 2.48 = 105.1''$$



$$L_r = 1.95 \times 2.82 \times \frac{29000}{0.7 \times 50} \sqrt{\frac{3.87 \times 1}{112 \times 13.4}}$$

$$\times \sqrt{1 + \sqrt{1 + 6.76 \left( \frac{0.7 \times 50}{29000} \times \frac{112 \times 13.4}{3.87 \times 1} \right)^2}}$$

$$= 371.3''$$

$$L_r = 371''$$

$$M_r = 0.7 \times F_y \times S_x$$

$$= 0.7 \times 50 \times 112$$

$$= 3920 \text{ k}''$$

$$M_n = C_b [M_p - (M_p - 0.7 F_y S_x) \left( \frac{L_b - L_p}{L_b - L_p} \right)] \leq M_p$$

$$= 1 \times 5630$$

$$= 5630 \text{ k}''$$

\*  $\frac{1}{4} \text{ inch} \rightarrow$  convert to feet,

LRFD:  $M_n = 0.9 \times 5630$

$$= 5067 \text{ k}''$$

W10x12  $F_y = 50 \text{ ksi}$

From AISC chart  $\rightarrow$  compact section

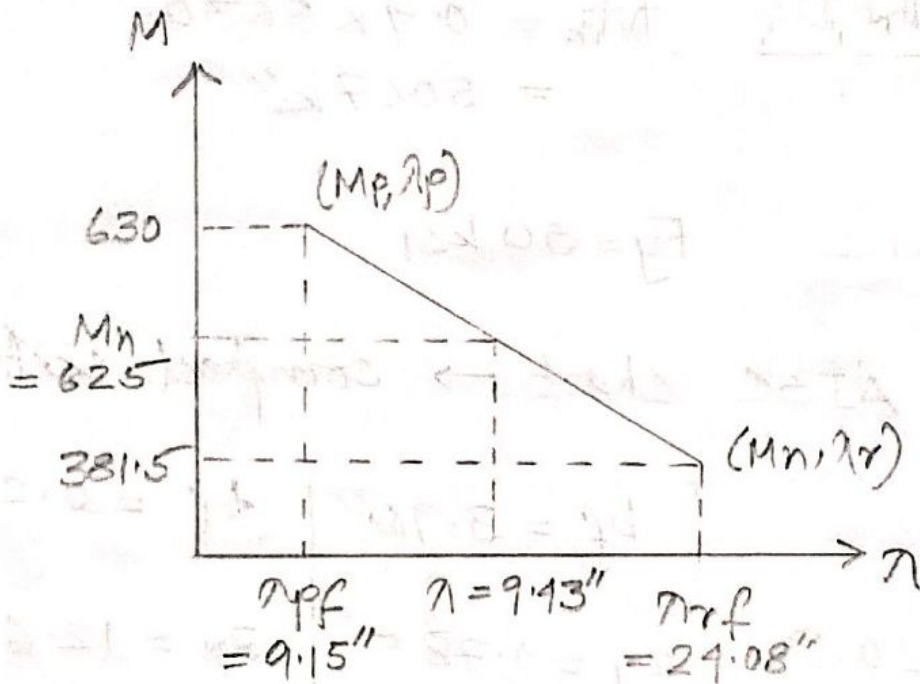
$A =$	$b_f = 3.96''$	$t_f = 0.21''$
$S_x = 10.9$	$r_y = 0.785$	$Z_x = 12.6$
$r_{tf} = 0.983$	$h_o = 9.66$	$J = 0.0547$

continuous lateral support  $\rightarrow$  no lateral bracing

$$\lambda = \frac{bf}{2tf} = \frac{3.96}{2 \times 0.21} = 9.43$$

For compact section,  $\lambda_{pf} = 0.38 \sqrt{\frac{E}{F_y}}$   
 $= 9.15'' < \lambda$

$$M_p = F_y Z_k = 50 \times 12.6 = 630 \text{ K}'$$



$$M_n = 0.7 F_y S_x = 0.7 \times 50 \times 10.9$$

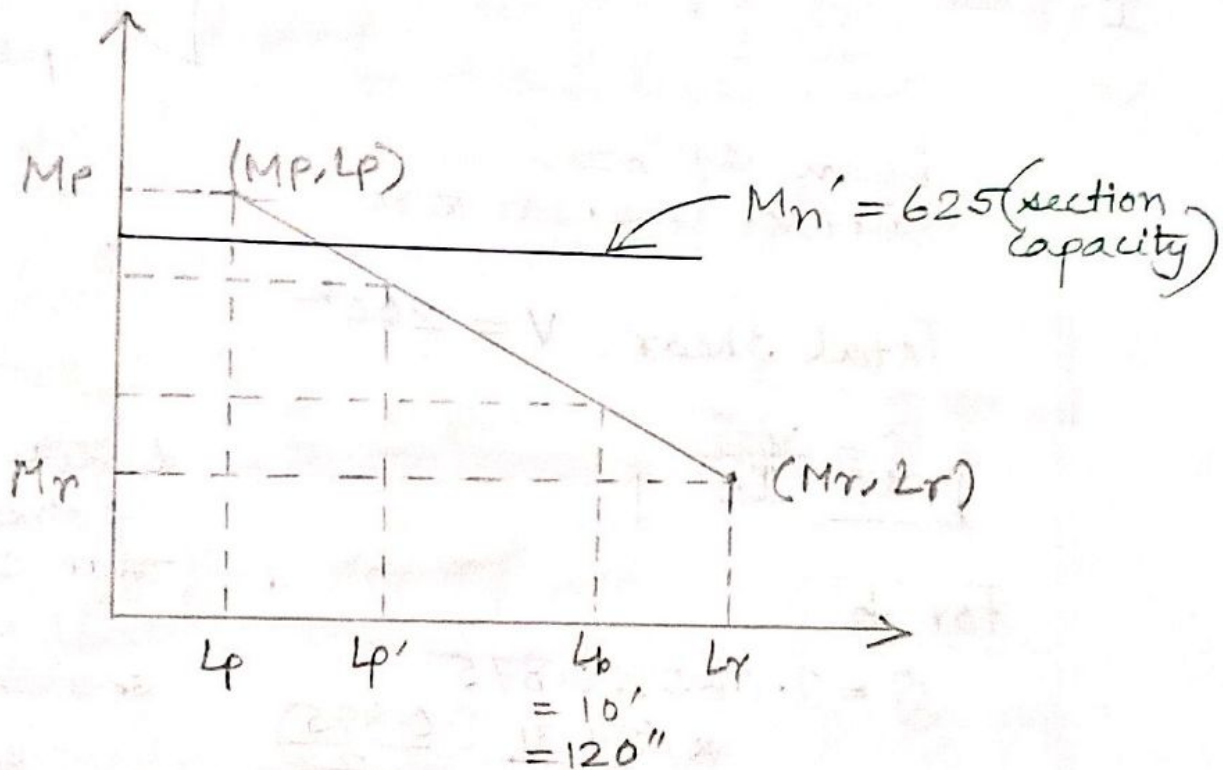
$$= 381.5$$

$$\lambda_{rf} = 1.0 \sqrt{\frac{E}{F_y}} = 24.08$$

Cross-section capacity,

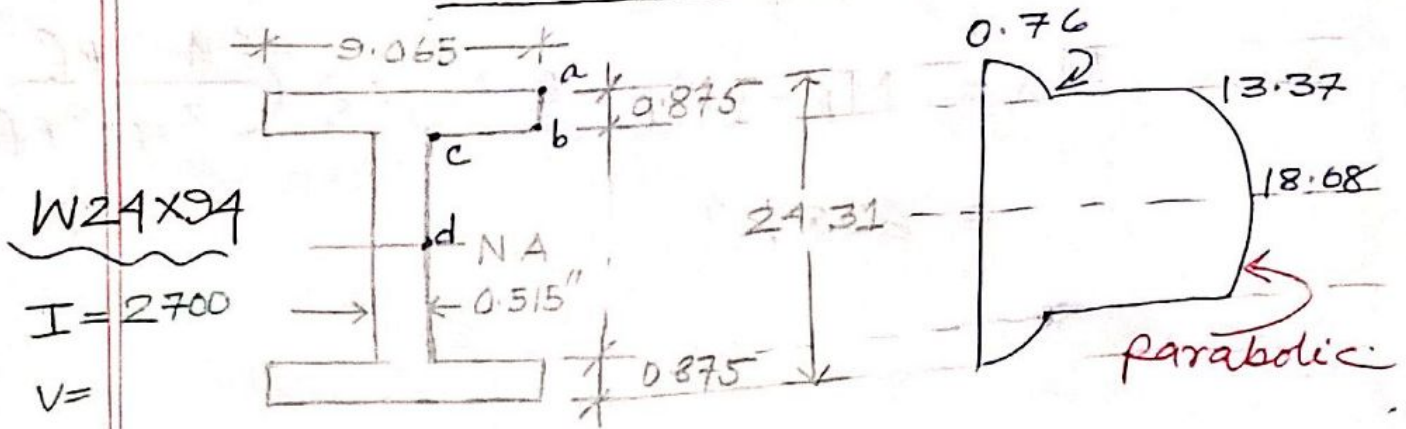
$$M_n = M_p - (M_p - 0.7 F_y S_x) \left( \frac{\lambda - \lambda_{pf}}{\lambda_{rf} - \lambda_{pf}} \right)$$

$$= 625 \text{ k"}$$



Theoretically possible  $M_n$  practically  
 $M_p$  এর জন্য যেহেতু সীমিত থাকবে না, তাই  $M_n$  এর  
 maximum limit বিবেচিত হবে,

Shear in Beam



Beam is open  
 surface is shear zero

Total shear,  $V = 200k$

$$\tau = \frac{VQ}{It}$$

t is the point  
 is the level  
 level is cross  
 sectional area

For 'b'

$$Q = 9.065 \times 0.875 \times \left( \frac{24.31}{2} - \frac{0.875}{2} \right)$$

$$= 92.94$$

$$\tau_b = \frac{200 \times 92.94}{2700 \times 9.065}$$

$$= 0.76$$

At c

$Q$  same as before = 92.94

$$t = 0.515$$

$$\tau_c = \frac{200 \times 92.94}{2700 \times 0.515} = 13.37$$

ଏ ଥିବା ସମୟରେ  
stress ଟରସ  
ଅଧିକ ଥାଏ ତେଣୁ  
ଫିଙ୍ଗର area-  
moment.

At d

$$Q = 92.94 + 0.515 \times 11.28 \times \frac{11.28}{2} = 125.7$$

$$\tau_d = \frac{VQ}{It} = \frac{200 \times 125.7}{2700 \times 0.515} = 18.08 \text{ kg/cm}^2$$

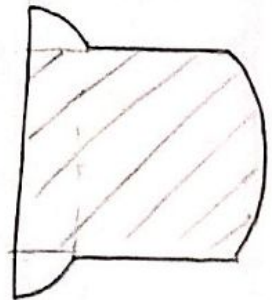
Shear contribution of web

$V_{web} = \text{Area of stress diagram in web} \times t_w$

$$= \frac{13.37 + 4 \times 18.08 + 13.37}{6}$$

$$\times (11.28 \times 2) \times 0.515$$

$$= 191.8 \text{ kg}$$



$$\% \text{ contribution} = \frac{191.8}{200} \times 100 = 96\%$$

total shear  
କିମ୍ବା 96% web ଏ ଥିବା ଅଂଶ

\*\* For 'I' cross section a basic design consideration is: web takes the shear but the flexural stress goes to the flange.

$$\begin{aligned} \text{Average stress} &= \frac{V}{A_w} \leftarrow \text{area of web} \\ &= \frac{200}{11.28 \times 2 \times 0.515} \\ &= 17.21 \end{aligned}$$

(flange area has been ignored due to its limited contribution to shear)

$$\frac{\text{Avg}}{\text{max}} = \frac{17.21}{18.08} = 95\%$$

(Avg. stress is 95% of max<sup>m</sup> stress, so detailed calculation is avoided and avg. stress is considered in design)

Point Load on Beams

→ target is to avoid web crippling, web buckling

\* point load is a shear transfer through the web.

→ this transfer has to be handled precisely

if concentrated load surpasses the capacity the beam has to be strengthened at that point:

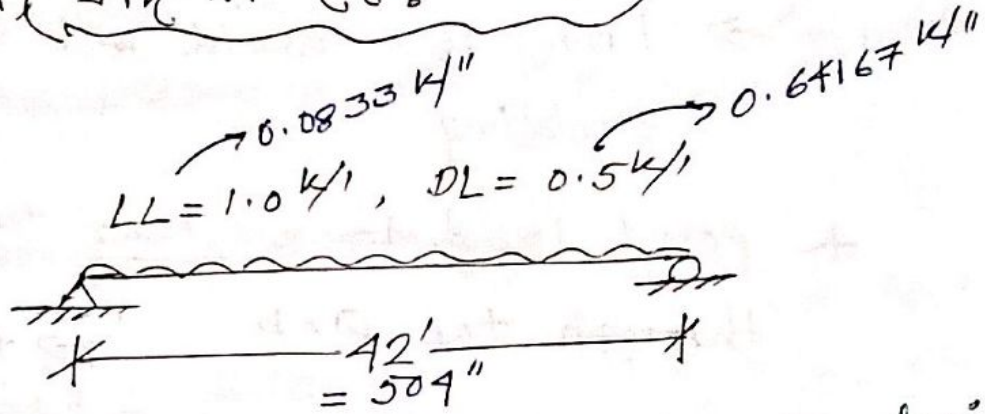
→ using stiffeners (at the supports of I-sections)

stiffener → bearing stiffener (बिन्दु पर point load का बिन्दु पर स्थिति)  
→ intermediate stiffener (web की shear capacity बढ़ाने के लिए)

Concrete की बिन्दु पर structure की बिन्दु पर -  
deflection की limit  $\frac{L}{240}$ , floor is sensitive  
fitting की limit  $\frac{L}{480}$

\* steel structure - L အားသာ  $\frac{L}{360}$  follow  
 လုပ်ရာ (အင်္ဂါ ဘာ ပုံစံ 21-11)

21-11



$$\Delta L \leq \frac{L}{360}$$

$$F_y = 50 \text{ ksi, ASD}$$

Strength design  
 & serviceability  
 design နှစ်ခုလုံး  
 အတွက်

\* 100 lb/ft unit weight လက်  
 heavy section beam - L use လုပ်  
 အင်္ဂါ 50 lb/ft - 70 lb/ft typically  
 use လုပ် အင်္ဂါ

\* convert all  
 lengths to inch

$$\begin{aligned} \text{Assumed self wt} &= 100 \#/\text{ft} \\ &= 8.33 \#/\text{ft} \\ &= 0.00833 \text{ k}/\text{ft} \end{aligned}$$

$$\begin{aligned} w &= (0.00833 + 0.04167 + 0.00833) \text{ k}/\text{ft} \\ &= 0.133 \text{ k}/\text{ft} \end{aligned}$$

$$M = \frac{WL^2}{8} = \frac{0.133 \times 50^2}{8} = 4232.5 \text{ k}''$$

$$M_n = \phi M = 1.67 \times 4232.5 = 7068 \text{ k}''$$

Assuming compact section,

$$M_n = F_y Z_x$$

$$\Rightarrow Z_x = \frac{M_n}{F_y} = \frac{7068}{50} = 141.4 \text{ in}^3$$

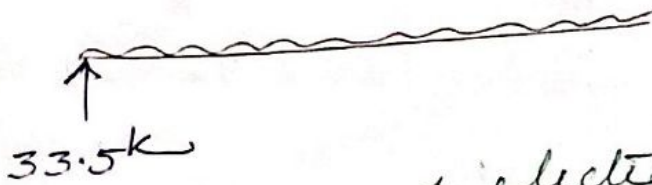
$$\text{Max}^m \text{ deflection} = L/360 = 1.4''$$

$$\frac{5}{384} \frac{WL^4}{EI} = 1.4 \text{ (for LL)}$$

$$\Rightarrow \frac{5}{384} \times \frac{0.083 \times 50^4}{29000 I} = 1.4$$

$$\therefore I = 1.723 \text{ in}^4$$

deflection is a real phenomenon, so it is always figured out using service/unfactored load.



check for shear

selected W21x93

shear = 33.5k

Connections (inherently steel structure  
is weakest part)

Concrete structure → cast in site  
steel structure → assembled in site

steel sections maximum 20' length

(Beam flange to column flange  
connect → Moment transfer flange  
Shear transfer web)

All Bolted Double Angles

Design parameters { Angle 45 size, thickness  
Bolt 25 dia, number of bolts

thread condition { 'N' → threads included in the shear plane  
'X' → threads excluded in the shear plane  
'sc' → slip critical condition

\* Girder is never coped

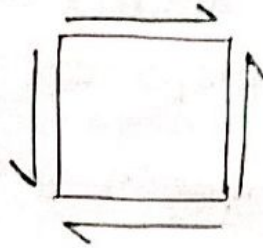
Continuous Beam-to-Column connection

(सूत्र web लिए connection लिए  
अन्य-अन्य न, flange to flange  
connection चाहिए)

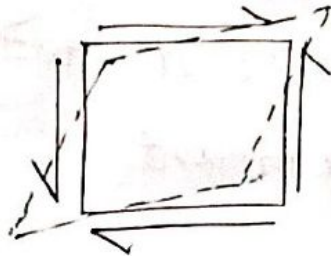
(c.g. is a fictitious point.)

\* column is axial force transmit 25 flange  
लिए।

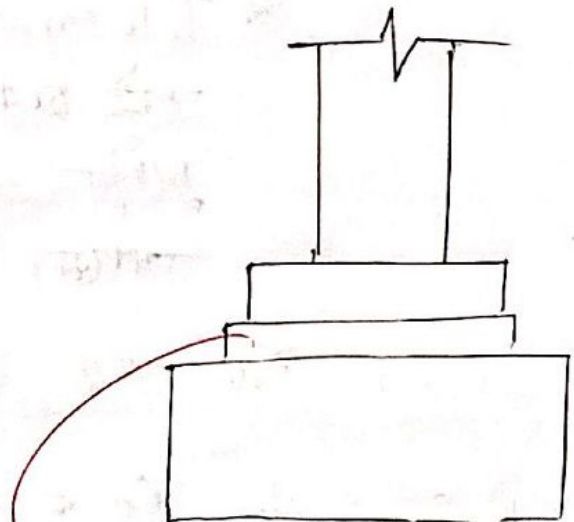
since it's 90%  
of the x-sec area.



Pure shear state



Column Base:



→ Grout → 1:3 1:4 25  
Load uniformly distributed  
कराया गया.

Moment ~~আমলেই~~ Anchor Rod এর design  
consideration - এ ~~খাতিয়া~~ হবে।

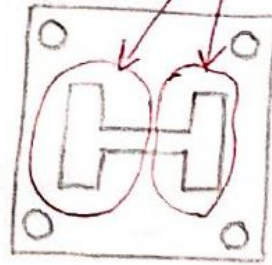
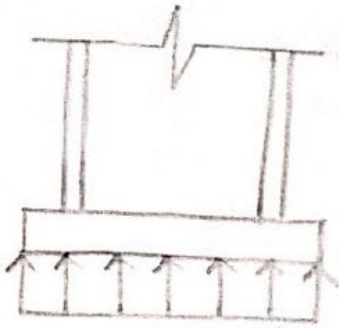
Theoretically compression এ anchor rod  
এর ~~দুর্বলতা~~ হয় না।

column base. ~~চিহ্ন~~ না থাকলে beam কে  
connect করতে পারবে না, তাই column  
base অনেক important.

\* Telecommunication Tower - এ  
এই চরিত্র leg থাকে, ~~একদিকে~~  
দুটোয় compression, দুটোয় tension  
আসে।

Column Base এর Design Consideration:

- 1) Concentric Compressive Axial Load
- 2) Tensile Axial Load
- 3) Base Plates with small Moments
- 4) Base Plates with Large Moments
- 5) Design for shear



इसकी portion AS एजेंटों के  
Canti lever से  
मिलेगा.

24-11 / Example

Beam - ColumnColumn

$$\phi_c P_n > P_u$$

$$\frac{P_u}{\phi_c P_n} \leq 1.0 \quad \boxed{\text{LRFD}}$$

$$\frac{P_u}{P_n / \Omega} \leq 1.0 \quad \boxed{\text{ASD}}$$

In general,  $\frac{\text{Required strength}}{\text{Available strength}} \leq 1.0$

$$\frac{P_r}{P_c} \leq 1.0$$

Including flexural Reinforcement

$$\frac{P_r}{P_c} + \frac{M_r}{M_c} \leq 1.0$$

$$M_r \rightarrow M_u \text{ (LRFD)} \quad \text{or} \quad M_a \text{ (ASD)}$$

$$M_c \rightarrow \phi M_n \text{ (LRFD)} \quad \text{or} \quad \frac{M_n}{\Omega} \text{ (ASD)}$$

Interaction formula considering two axes bending,

$$\frac{P_r}{P_c} + \frac{M_{rx}}{M_{cx}} + \frac{M_{ry}}{M_{cy}} \leq 1.0$$

AISC Design Equations

$$\frac{P_r}{P_c} + \frac{8}{9} \left( \frac{M_{rx}}{M_{cx}} + \frac{M_{ry}}{M_{cy}} \right) \leq 1.0 \text{ when } \frac{P_r}{P_c} > 0.2$$

$$\frac{P_r}{2P_c} + \left( \frac{M_{rx}}{M_{cx}} + \frac{M_{ry}}{M_{cy}} \right) \leq 1.0 \text{ when } \frac{P_r}{P_c} \leq 0.2$$

LRFD Interaction Equations

$$\frac{P_u}{\phi_c P_n} + \frac{8}{9} \left( \frac{M_{ux}}{\phi_b M_{nx}} + \frac{M_{uy}}{\phi_b M_{ny}} \right) \leq 1.0$$

when,  $\frac{P_u}{\phi_c P_n} > 0.2$

$$\frac{P_u}{2\phi_c P_n} + \left( \frac{M_{ux}}{\phi_b M_{nx}} + \frac{M_{uy}}{\phi_b M_{ny}} \right) \leq 1.0$$

when  $\frac{P_u}{\phi_c P_n} \leq 0.2$

## ASD Interaction Equations:

$$\frac{P_a}{P_n/\Omega_c} + \frac{8}{9} \left( \frac{M_{ax}}{M_{nx}/\Omega_b} + \frac{M_{ay}}{M_{ny}/\Omega_b} \right) \leq 1.0$$

When,  $\frac{P_a}{P_n/\Omega} > 0.2$

$$\frac{P_a}{2P_n/\Omega_c} + \left[ \frac{M_{ax}}{M_{nx}/\Omega_b} + \frac{M_{ay}}{M_{ny}/\Omega_b} \right] \leq 1.0$$

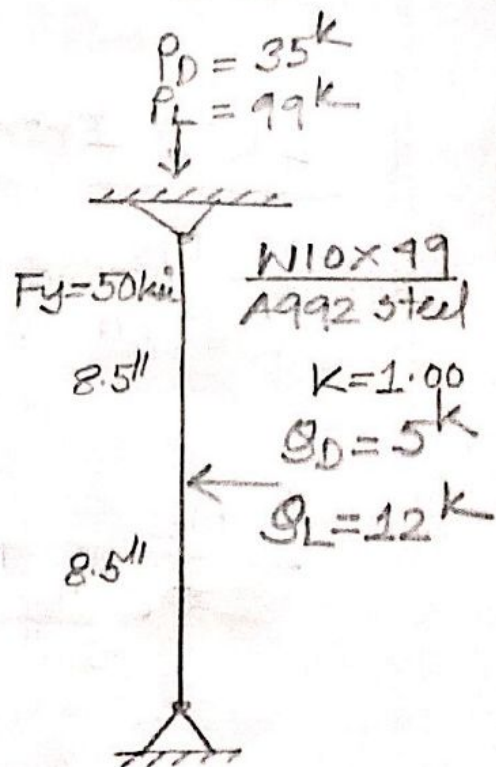
When,  $\frac{P_a}{P_n/\Omega} \leq 0.2$

## Example (LRFD)

Strong axis bending  
due to lateral load.

9.

(Neglect moment  
amplification)



Soln:

Properties of W10x49,

$$A_g = 144$$

$$r_x = 4.35$$

$$r_{ts} = 2.84$$

$$d = 10''$$

$$Z_x = 60.4$$

$$h_o = 9.42$$

$$S_x = 546$$

$$r_y = 2.59$$

$$J = 1.39, C = 1.0$$

$$\text{Given, } [C_b = 1.32]$$

Calculation ~~का~~  
~~का~~ ~~का~~ ~~का~~

$$b_f = 10$$

$$t_f = 0.56$$

$$h/t_w = 23.1$$

$C_b = 1.32$

$$\text{As a column, } L = 17' = 204''$$

$$\text{" " beam, } L_b = 17' = 204''$$

$$P_u = 1.2 \times 35 \times 1.6 \times 99 = 200.7 \text{ k}$$

Compact section

Determine col<sup>m</sup> capacity,

$$1.71 \sqrt{E/F_y} = 1.71 \sqrt{\frac{29000}{50}} = 113.4$$

$$\frac{KL}{r_y} = \frac{1 \times 204}{2.54} = 80.31 < 4.71 \sqrt{E/F_y}$$

$$F_c = \frac{\pi^2 E}{\left(\frac{KL}{r_y}\right)^2} = 44.33 \text{ ksi}$$

$$F_{cr} = \left[ 0.658^{F_y/F_c} \right] F_y = 31.18 \text{ ksi}$$

$$\phi_c P_n = 0.9 A_g F_{cr}$$

$$= 0.9 \times 14.4 \times 31.18$$

$$= 404.2 \text{ k}$$

$$\frac{P_u}{\phi_c P_n} = \frac{200.4}{404.2} = 0.495 > 0.2$$

only strong axis bending,

$$M_u = 0$$

Flexural Capacity as a beam considering

LTB

$$\text{span, } L_b = 204''$$

$$L_p = 1.76 r_y \sqrt{E/F_y}$$

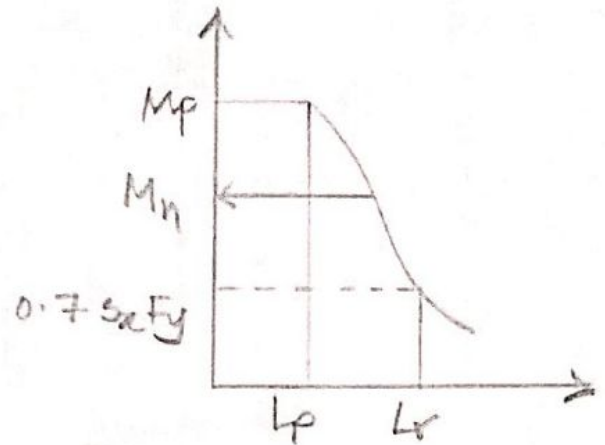
$$= 107.6'' < L_b$$

$$L_r = 1.95 r_{ts} \frac{E}{0.7 F_y} \sqrt{\frac{J_c}{S_x h_o}}$$

$$\sqrt{1 + \sqrt{1 + 6.76 \left( \frac{0.7 F_y S_x h_o}{E J_c} \right)^2}}$$

$$= 379.6''$$

$$M_p = F_y Z_x = 3020 \text{ k''}$$

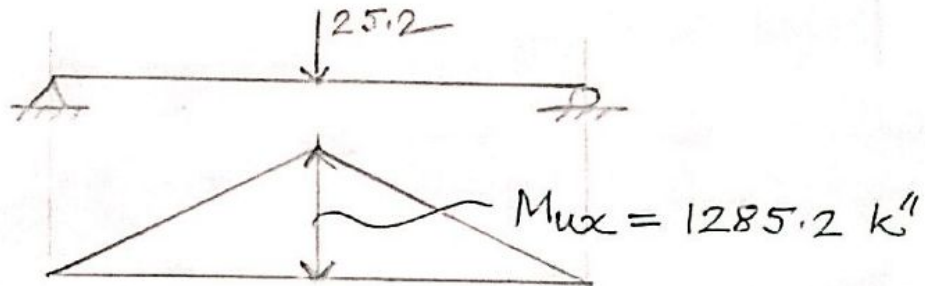


$$M_n = C_b \left[ M_p - (M_p - 0.75 S_x F_y) \frac{L_b - L_p}{L_r - L_p} \right] \leq M_p$$

$$= 3467.6 \text{ k''} > M_p$$

$$M_n = M_{n\alpha} = M_p = 3020$$

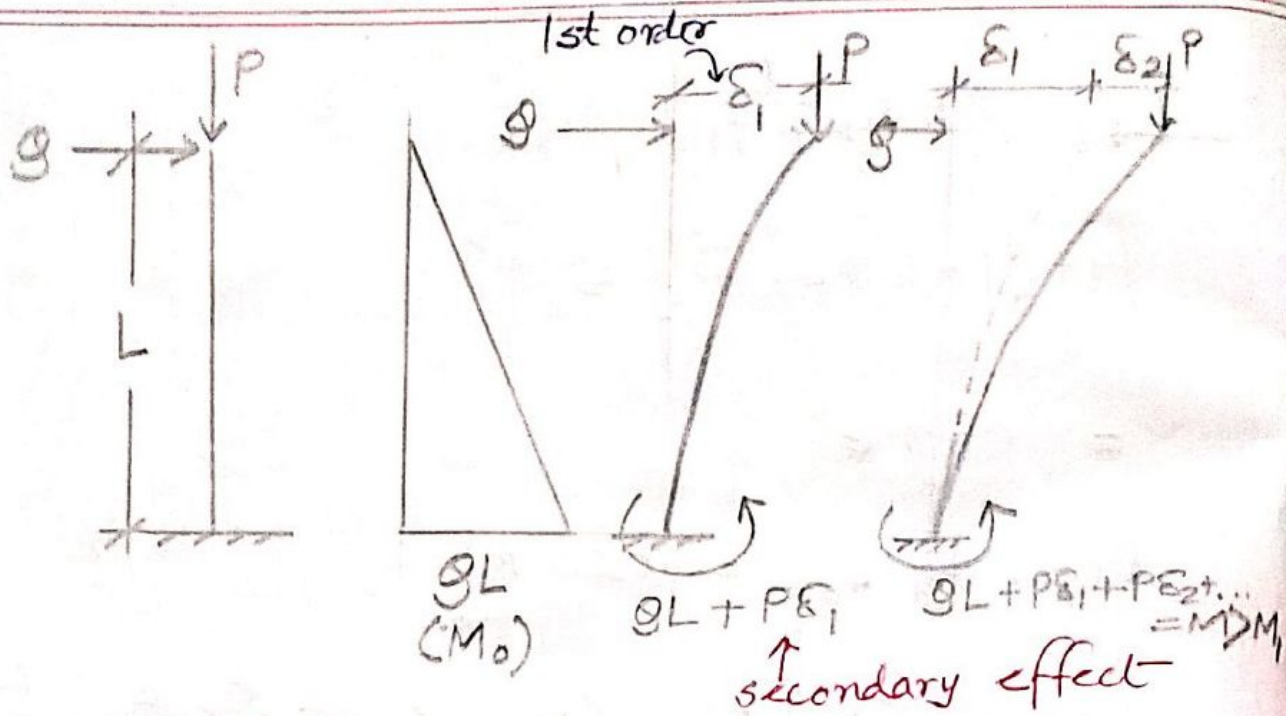
$$S_u = 1.2 \times 5 + 1.6 \times 12 = 25.2 \text{ k}$$



Interaction Equation

$$0.495 + \frac{8}{9} \times \frac{1285.2}{0.9 \times 3020} = 0.915 < 1.0$$

OK

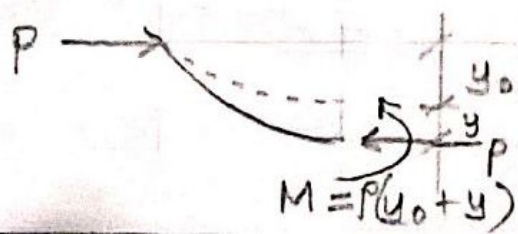
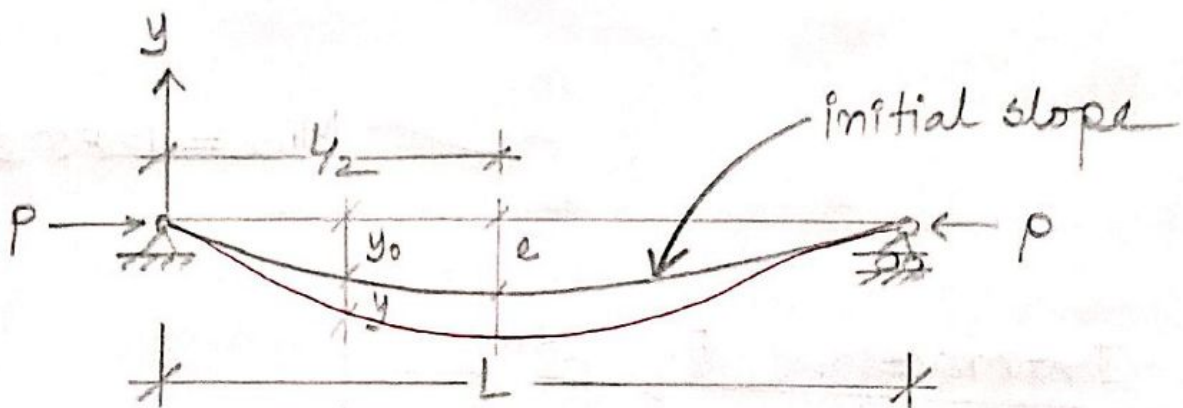


Load first stress develop  $\theta$   $\theta_1$   $\theta_2$   
 जाते जाते  $\theta$   $\theta_1$   $\theta_2$  deflection  $\theta$   $\theta_1$   $\theta_2$

Moment including secondary effect,

$$M = BM_0$$

↓  
?



$$y_0 = e \sin \frac{\pi x}{L}, \text{ initial slope}$$

$$\frac{d^2 y}{dx^2} = -\frac{M}{EI}$$

$$\Rightarrow \frac{d^2 y}{dx^2} = -\frac{P}{EI} (y + y_0)$$

$$\Rightarrow \frac{d^2 y}{dx^2} + \frac{P y}{EI} = -\frac{P y_0}{EI}$$

Boundary condition  $\leftarrow$

A function that satisfies both BC ( $y=0, x=0$  and  $y=0, x=L$ ) and the differential equation,

$$y = B \sin \frac{\pi x}{L}$$

Putting  $y$  in the differential equation,

$$-\frac{\pi^2}{L^2} B \sin \frac{\pi x}{L} + \frac{P}{EI} B \sin \frac{\pi x}{L} = -\frac{P e \sin \frac{\pi x}{L}}{EI}$$

$$\Rightarrow B = \frac{-\frac{P e}{EI}}{\frac{P}{EI} - \frac{\pi^2}{L^2}} = \frac{-e}{1 - \frac{\pi^2 EI}{P L^2}} = \frac{e}{\frac{P}{EI} - 1}$$

$$\text{where, } P_e = \frac{\pi^2 EI}{L^2}$$

= Euler Load

$$\text{Now, } y = B \sin \frac{\pi x}{L} = \frac{e}{\frac{P_e}{P} - 1} \sin \frac{\pi x}{L}$$

$$M = P(y_0 + y)$$

$$M = P \left\{ e \sin \frac{\pi x}{L} + \left[ \frac{e}{\left(\frac{P_e}{P}\right) - 1} \right] \sin \frac{\pi x}{L} \right\}$$

$M$  is maximum when,  $x = L/2$ .

$$M_{\max} = P \left[ e + \frac{e}{\left(\frac{P_e}{P}\right) - 1} \right]$$

$$= P e \left[ \frac{1}{1 - \left(\frac{P}{P_e}\right)} \right]$$

$$= M_0 \left[ \frac{1}{1 - \left(\frac{P}{P_e}\right)} \right]$$

$$= B M_0$$

Moment amplification or magnification factor,  $B = \frac{1}{1 - P/P_e}$



Steel structure - 27

13 December 2024