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Dr. Raquib Ahsan

Lec-1

Flexibility Method:

Approximate analysis  $\rightarrow$  1) portal  
2) cantilever

Exact  $\rightarrow$  1) stiffness (সব কিছু force এ)  
2) moment distribution  
3) flexibility (সব কিছু deformation এ)

2D  $\rightarrow \sum F_x, \sum F_y, \sum M = 0$  ( 3টি linearly independent Eqn )

3D  $\rightarrow \sum F_x, \sum F_y, \sum F_z, \sum M_x, \sum M_y, \sum M_z = 0$  ( 6টি " " " )

- approximate  $\rightarrow$  carelessly অবহেলা condition এনে eqn আনা হয়।
- exact  $\rightarrow$  original deformation থেকে eqn develop করা হয়।

Equilibrium এ  $\rightarrow$  সব Term হল force

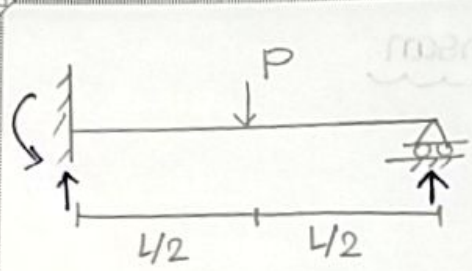
indeterminate এ  $\rightarrow$  1) 1 set of eqn  $\rightarrow$  in terms of force  
2) new set " "  $\rightarrow$  " " " deformation

compatibility eqns or conditions.

Exact method এ  $\rightarrow$  হয় সব force or সব deformation based eqn এ convert করা হয়।

Stiffness method more user friendly.

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axial force ignore করুন,

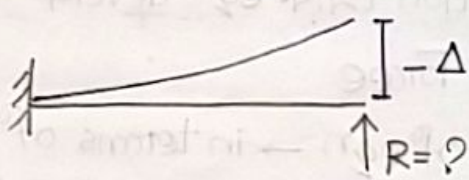
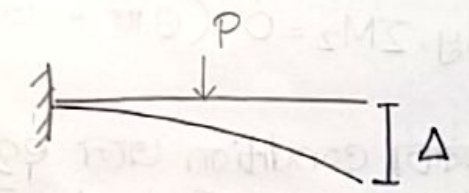
$UK = 3$

Equilibrium eq<sup>n</sup> = 2

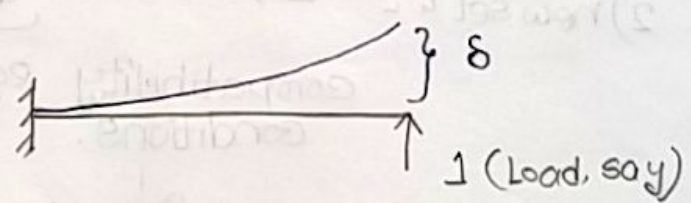
Degree of Statical Indeterminacy

$DOSI = (3 - 2) = 1$

- Static = related to force equilibrium
- Kinematic = something related to deformation
- Kinetic = " " " " speed / velocity
- Dynamic = " " " " acceleration



$R = ?$  যাতে  $\Delta$  deformation opposite দিকে হয়।



$\delta = \text{Flexibility}$

1 load এর জন্য  $\delta$  deformation

$\therefore R \propto \delta$

$\therefore R\delta = -\Delta$

$\therefore R = -\frac{\Delta}{\delta}$  ← Deformation per unit load

শিউরী বুক বাইন্ডিং



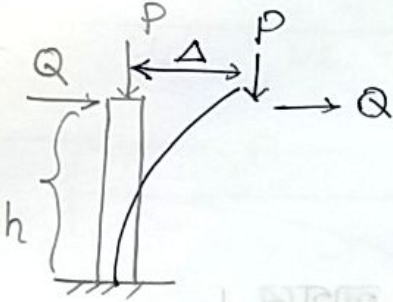


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• So, যেখানে deformed shape ও Equilibrium condition apply হয়, সেখানে principle of superposition applicable না।



Moment Amplification

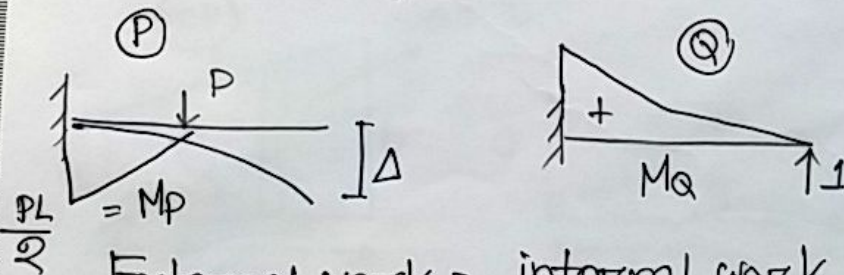
P-Δ effect আসলে deformation Moment চিত্রকরে equilibrium eq<sup>n</sup> বের করতে হয়। তাই এক্ষেত্রে principle of superposition applicable না।

→ Virtual work (external work)

P Force strain & Q Force stress → internal, <sup>virtual</sup> work

→ σ-E diagram এর area under the curve → per unit vol<sup>m</sup> এর energy

P (actual load) ; Q (virtual load)



External work = internal work

$$1 \times \Delta = \int \epsilon_p \sigma_Q dV$$

$$E = \sigma / \epsilon$$

$$= \int_L \int_A \frac{M_p y}{EI} \frac{M_Q y}{I} dA dx$$

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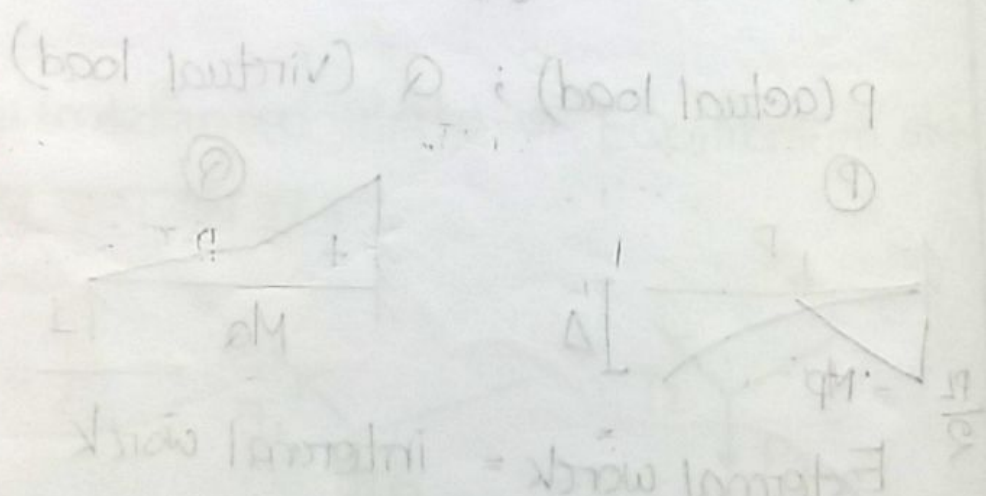
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$$= \int_L \frac{M_P M_Q}{EI^2} \int_A y^2 dA dx$$

$$\Delta = \int_L \frac{M_P M_Q}{EI} dx$$

∴ P বিবেকে এর BMD } লাগবে  
and Q - - - BMD }

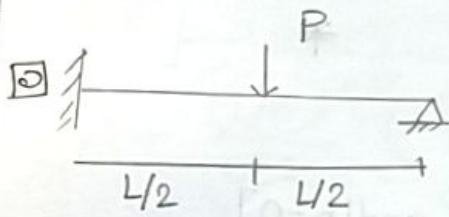
$M_P$  &  $M_Q$  কে  $x$  এ convertt বন্ধা লাগবে।



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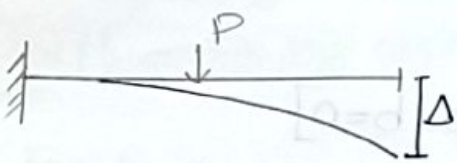
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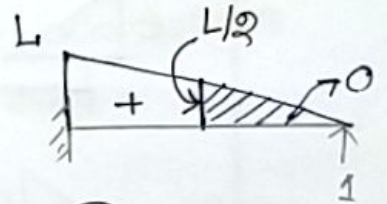
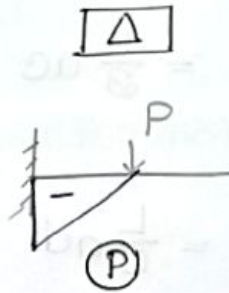


$$R\delta = -\Delta$$

$$\therefore R = -\frac{\Delta}{\delta}$$



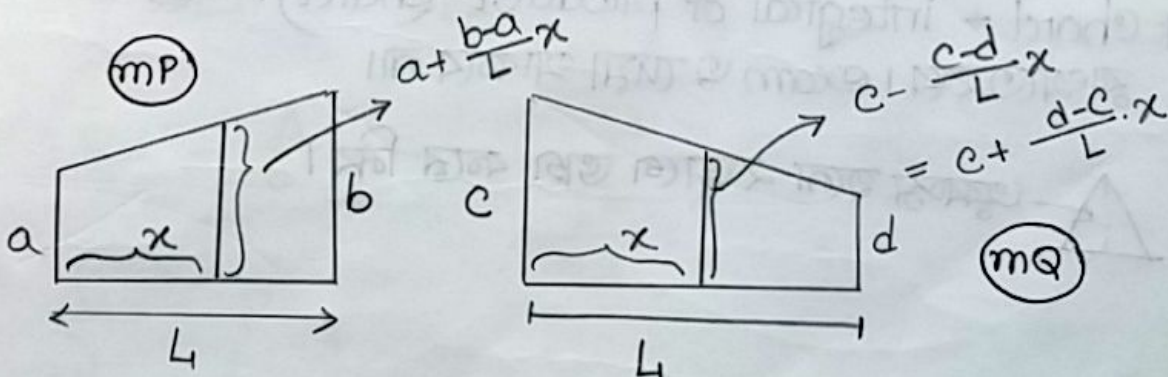
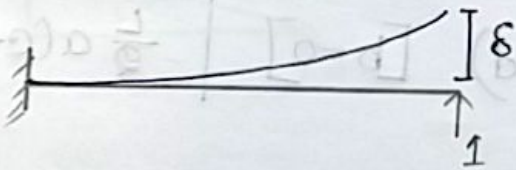
$$\frac{PL}{2}$$



২) অথান defination অথান Q force দিব



$$\Delta = \int_L \frac{m_p m_q}{EI} dx$$

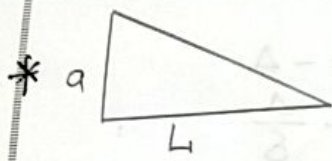


$$I = \int_L m_p m_q dx = \int_0^L \left[ a + (b-a)\frac{x}{L} \right] \left[ c + (d-c)\frac{x}{L} \right] dx$$

$$= \frac{L}{6} \{ c(2a+b) + d(a+2b) \} = \frac{L}{6} \{ a(2c+d) + b(c+2d) \}$$

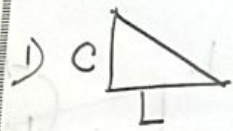
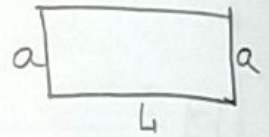
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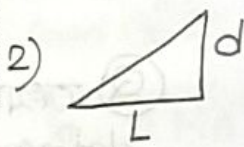


$b=0$

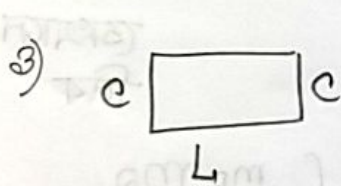
\*



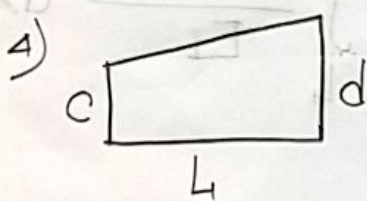
$= \frac{L}{2} ac \quad [d=0] \text{ and } [b=0]$



$= \frac{L}{2} ad \quad [c=0, b=0]$



$= \frac{L}{2} ac \quad [b=0, c=d] \quad | \quad Lac$



$= \frac{L}{2} a(c+d) \quad [b=0] \quad | \quad \frac{L}{2} a(c+d)$

\* Chart  $\rightarrow$  integral of product (chart). অর্থাৎ সনে  
 রাখতে হয়। exam এ দেয়া থাকবে না।

অক্ষয় হলে ২ ভাগে ভাগ করে নিব।



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..... - অধুনা,

$$\Delta x \quad \Delta = \frac{1}{6} a (2c+d)$$

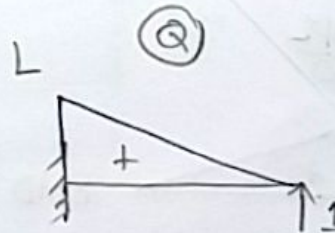
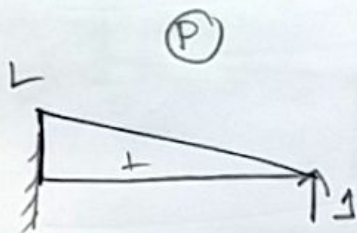
$$\Delta = \int_L^0 \frac{m p m a}{EI} dx$$

$$\Delta = \frac{\left(\frac{L}{2}\right)}{6EI} \left(-\frac{PL}{2}\right) \times \left(2L + \frac{L}{2}\right) = -\frac{PL^2}{24EI} \cdot \frac{5L}{2} = -\frac{5PL^3}{48EI}$$

-ve sign indicate করে, আমরা 1 unit load এদিকে apply করেছি, তার opposite এ deformation হবে।

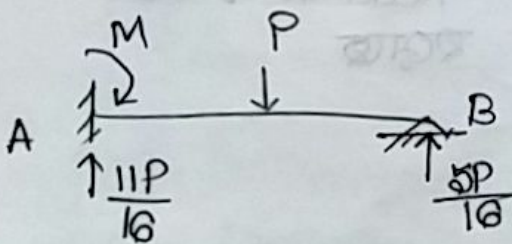
⊗ ফাট δ

diagram,

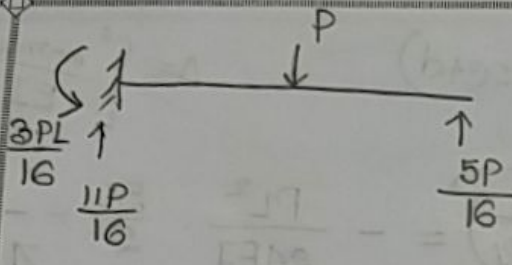


$$\delta = \frac{L}{3EI} L^2 = \frac{L^3}{3EI}$$

$$\therefore R = -\frac{\Delta}{\delta} = -\frac{-\frac{5PL^3}{48EI}}{\frac{L^3}{3EI}} = \frac{5P}{16}$$



$$\begin{aligned} \sum M_A = 0 \\ \Rightarrow P \times \frac{L}{2} - \frac{5P}{16} \times L + M = 0 \\ \Rightarrow M = \frac{5PL}{16} - \frac{PL}{2} = -\frac{3}{16} PL \\ \therefore M = \frac{3}{16} PL (\curvearrowright) \end{aligned}$$

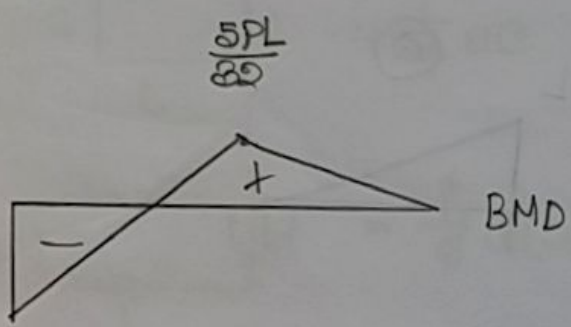
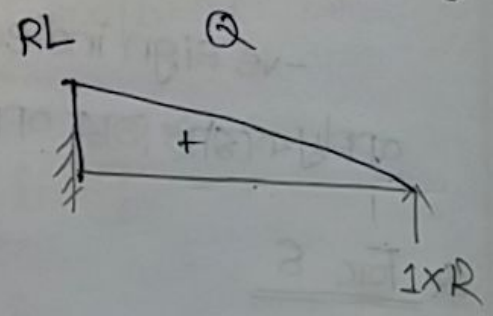
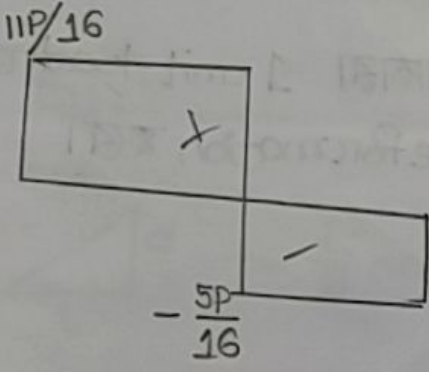


super position

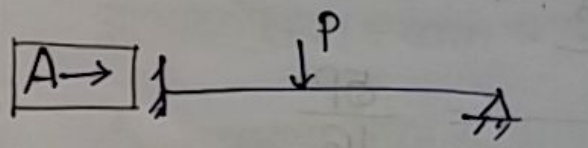
$$M = m_0 + m_1 R$$

Redundent

unit load  
একক লোড



$$\frac{3PL}{16}$$

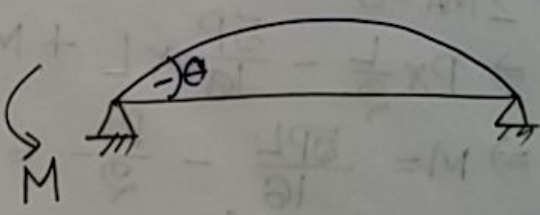
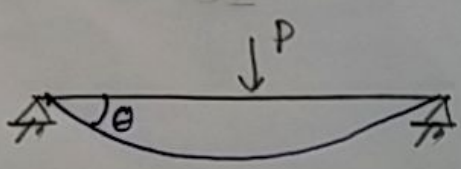


আগের Math,

fixed support কে

redundent করা

হচ্ছে



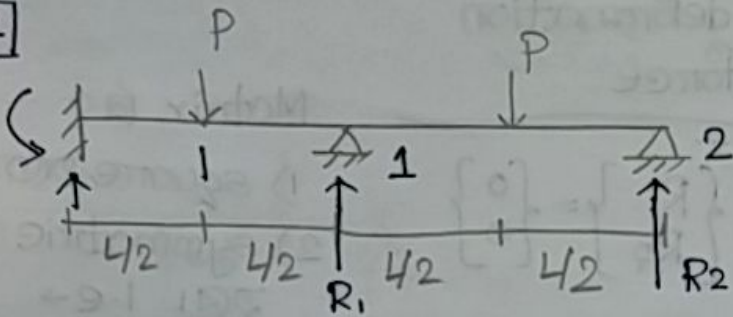
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[2]



D.O.F = 2 (No axial load)

$$\Delta_{01} + \Delta_{11} + \Delta_{12} = 0$$

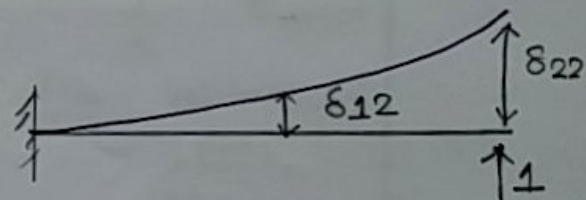
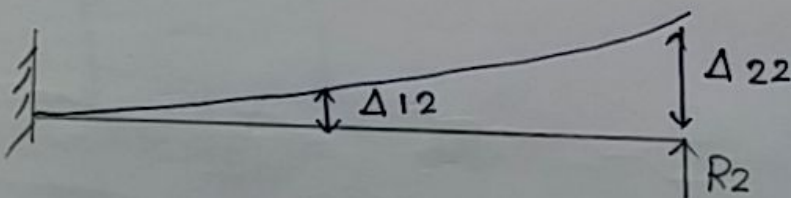
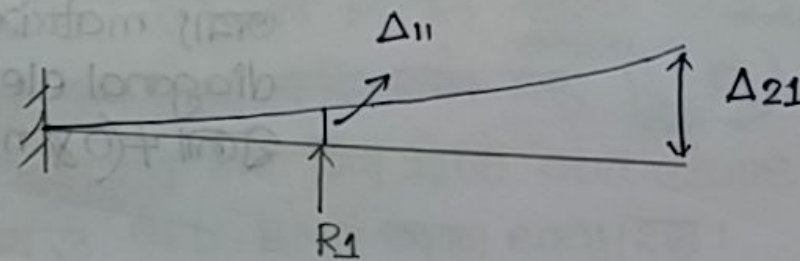
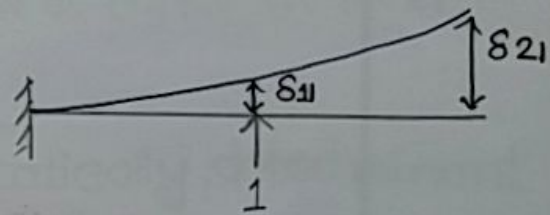
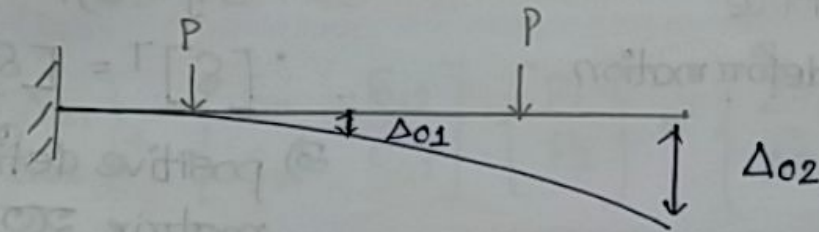
$$\Delta_{02} + \Delta_{21} + \Delta_{22} = 0$$

$$\Delta_{11} = \delta_{11} R_1$$

$$\Delta_{21} = \delta_{21} R_1$$

$$\Delta_{12} = \delta_{12} R_2$$

$$\Delta_{22} = \delta_{22} R_2$$



$$\text{Now, } \Delta_{01} + \delta_{11} R_1 + \delta_{12} R_2 = 0$$

$$\Delta_{02} + \delta_{21} R_1 + \delta_{22} R_2 = 0$$

$$\delta = \frac{\Delta}{R}$$

= Deformation per unit force  
 = flexibility

$$\boxed{\{\Delta_0\} + [\delta]\{R\} = \{\Delta_s\}} \leftarrow \text{Flexibility eqn}$$

$\{\Delta_0\}$  = Deformation vector of the released structure

$[\delta]$  = Flexibility matrix

$\{R\}$  = Redundant vector

$\{\Delta_s\}$  = Support deformation vector

$\delta_{ij} \rightarrow$   $i =$  কোন deformation  
 $j =$  " force

$$\begin{Bmatrix} \Delta_{01} \\ \Delta_{02} \end{Bmatrix} + \begin{bmatrix} \delta_{11} & \delta_{12} \\ \delta_{21} & \delta_{22} \end{bmatrix} \begin{Bmatrix} R_1 \\ R_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

$K_{ij} \rightarrow$   $i \rightarrow$  force  
 stiffness  $j \rightarrow$  deformation

Matrix. চি -

- 1) square হতে হবে
- 2) symmetric হতে হবে। i.e  $\rightarrow$

•  $\delta_{ij} = \delta_{ji}$

•  $[\delta]^T = [\delta]$

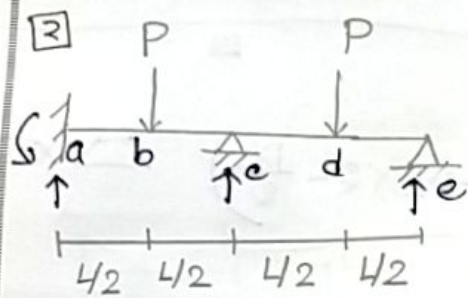
- 3) positive definite matrix হতে হবে।  
 অর্থাৎ matrix এর diagonal element  
 যেনো  $\neq 0$  & neg value.

$$\{\Delta\} = \{R\}[\delta] + \{\Delta_0\}$$

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Lec-3

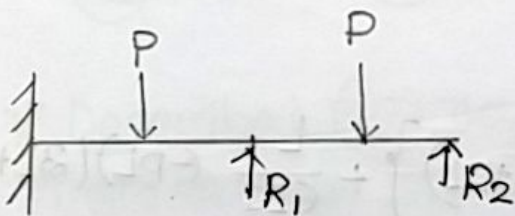


Flexibility eq<sup>n</sup>,  
 $\{\Delta_0\} + [S] \{R\} = \{\Delta_s\}$   
 DOF = 2 = (4-2)  
 order হবে 2.

$$\begin{Bmatrix} \Delta_{01} \\ \Delta_{02} \end{Bmatrix} + \begin{bmatrix} S_{11} & S_{12} \\ S_{21} & S_{22} \end{bmatrix} \begin{Bmatrix} R_1 \\ R_2 \end{Bmatrix} = \begin{Bmatrix} \Delta_{s1} \\ \Delta_{s2} \end{Bmatrix}$$

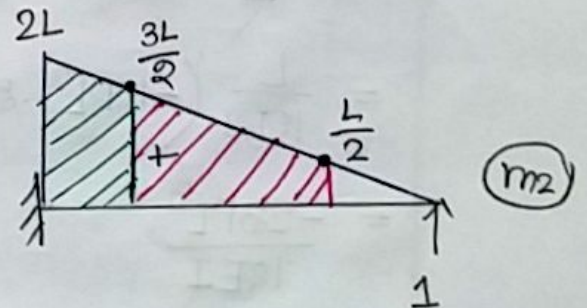
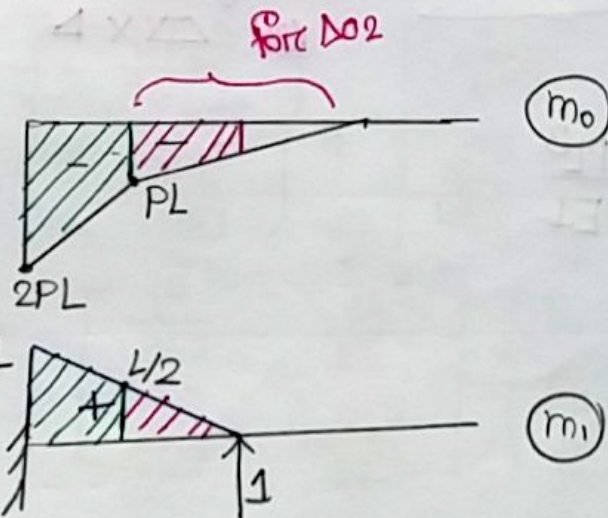
• যেকোন দুইটোকে  $R_1, R_2$  ধরতে পারি। (এটা থেকে যে কোন  
 ২টা বাদ দিলে ~~কমি~~ <sup>ওখানে</sup> ~~২টা~~ হবে  $R_1, R_2$  <sub>ব্যব</sub>)

• একনভাবে বাদ দিব যাতে structure statically determinant  
 হয় ও SFD, BMD অঙ্কন করা যায়।



$$\Delta_{01} = \int_L \frac{m_0 m_1}{EI} dx$$

$m_0$  = released structure এর M.



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$$\Delta_{01} = \int \frac{m_0 m_1}{EI} dx = \frac{L/2}{6EI} \left\{ (-2PL) \left( 2L + \frac{L}{2} \right) - PL \left( 2 \times \frac{L}{2} + L \right) \right\}$$

trapezium x trapezium

$$+ \frac{L/2}{6EI} \times \frac{L}{2} \left( -PL \times 2 - \frac{PL}{2} \right) \quad \triangle \times \triangle$$

$$= \frac{L}{12EI} \left\{ -5PL^2 - 2PL^2 \right\} + \frac{L^2}{24EI} \left( -\frac{5PL}{2} \right)$$

$$= -\frac{7PL^3}{12EI} - \frac{5PL^3}{48EI} = -\frac{33PL^3}{48EI} = -\frac{11PL^3}{16EI}$$

virtual load (1 unit) ↑  
 but deformation ↓  
 অর্থাৎ (-ve) এসেছে।

$$\Delta_{02} = \int \frac{m_0 m_2}{EI} dx$$

$$= \frac{L/2}{6EI} \left\{ (-2PL) \left( 4L + \frac{3L}{2} \right) - PL (3L + 2L) \right\} + \frac{L}{6EI} (-PL) \left( 3L + \frac{L}{2} \right)$$

▣    △ x △
▣    △ x △

$$= \frac{L}{12EI} \left( -11PL^2 - 5PL^2 \right) - \frac{7PL^3}{12EI}$$

$$= -\frac{23PL^3}{12EI}$$

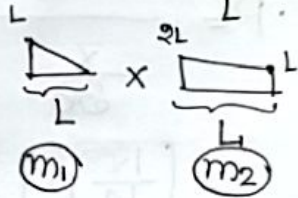
Sub:  $\delta_{11}, \delta_{22} \rightarrow$  must be (ve)

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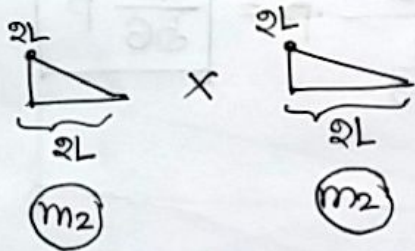
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$$\delta_{11} = \int_L \frac{m_1 m_1}{EI} dx = \frac{L}{3EI} \times L \times L = \frac{L^3}{3EI} \quad (L \text{ length এর ২টা triangle গ্রন})$$

$$\delta_{12} = \delta_{21} = \int_L \frac{m_1 m_2}{EI} dx = \frac{L}{6EI} \times L (2 \times 2L + L) = \frac{5L^3}{6EI}$$



$$\delta_{22} = \int_L \frac{m_2 m_2}{EI} dx = \frac{2L}{3EI} \times 2L \times 2L = \frac{8L^3}{3EI}$$



\* Described (বিনে দেয়া) support deformation (কম্পন) নামের বা (কি) না থাকলে  $\Delta s_1, \Delta s_2 = 0$ .

$$\begin{Bmatrix} -\frac{11PL^3}{16EI} \\ -\frac{23PL^3}{12EI} \end{Bmatrix} + \begin{bmatrix} \frac{L^3}{3EI} & \frac{5L^3}{6EI} \\ \frac{5L^3}{6EI} & \frac{8L^3}{3EI} \end{bmatrix} \begin{Bmatrix} R_1 \\ R_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

$$\Rightarrow \begin{bmatrix} \frac{L^3}{3EI} & \frac{5L^3}{6EI} \\ \frac{5L^3}{6EI} & \frac{8L^3}{3EI} \end{bmatrix} \begin{Bmatrix} R_1 \\ R_2 \end{Bmatrix} = \begin{Bmatrix} \frac{11PL^3}{16EI} \\ \frac{23PL^3}{12EI} \end{Bmatrix}$$

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using Cramer's Rule:

$$R_1 = \frac{\begin{vmatrix} \frac{11PL^3}{16EI} & \frac{5L^3}{6EI} \\ \frac{23PL^3}{12EI} & \frac{8L^3}{3EI} \end{vmatrix}}{\begin{vmatrix} \frac{L^3}{3EI} & \frac{5L^3}{6EI} \\ \frac{5L^3}{6EI} & \frac{8L^3}{3EI} \end{vmatrix}} = \frac{\begin{vmatrix} \frac{11}{16} & \frac{5}{6} \\ \frac{23}{12} & \frac{8}{3} \end{vmatrix}}{\begin{vmatrix} \frac{1}{3} & \frac{5}{6} \\ \frac{5}{6} & \frac{8}{3} \end{vmatrix}} \cdot P = \frac{\frac{17}{72}}{\frac{7}{36}} \cdot P = \boxed{\frac{17}{14} P}$$

$$R_2 = \frac{\begin{vmatrix} \frac{1}{3} & \frac{11}{16} \\ \frac{5}{6} & \frac{23}{12} \end{vmatrix}}{\frac{7}{36}} \cdot P = \frac{\frac{19}{288}}{\frac{7}{36}} \cdot P = \boxed{\frac{19}{56} P}$$

$$M = m_0 + m_1 R_1 + m_2 R_2$$

$$M_a = -2PL + L \times \frac{17}{14} P + 2L \times \frac{19}{56} P$$

$$= -2PL + \frac{17}{14} PL + \frac{19}{28} PL$$

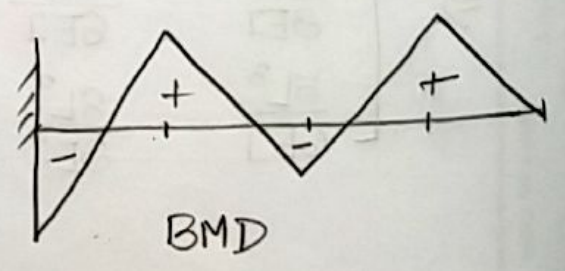
$$M_a = -\frac{3}{28} PL$$

$$M_b = -PL + \frac{L}{2} \times \frac{17}{14} P + \frac{3L}{2} \times \frac{19}{56} P$$

$$M_b = +\frac{13}{112} PL$$

- point load থাকলে at line
- UDL হলে curve হবে।

\* SFD এর জন্য  $M_0, V_1, V_2$  diagram একে math start করা।



শিউলী বুক বাইন্ডিং

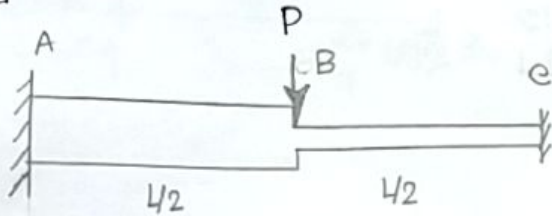


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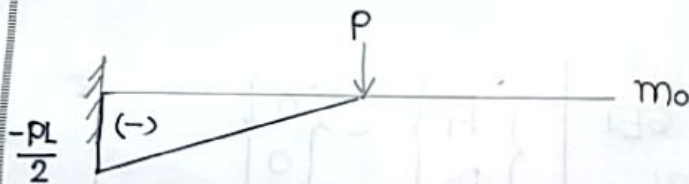
$AB = 2EI$   
 $BC = EI$

$DOF = 4 - 2 = 2$

$\{\Delta_0\} + [\delta] \{R\} = \{\Delta_S\}$

$\begin{Bmatrix} \Delta_{01} \\ \Delta_{02} \end{Bmatrix} + \begin{bmatrix} \delta_{11} & \delta_{12} \\ \delta_{21} & \delta_{22} \end{bmatrix} \begin{Bmatrix} R_1 \\ R_2 \end{Bmatrix} = \begin{Bmatrix} \Delta_{S1} \\ \Delta_{S2} \end{Bmatrix}$

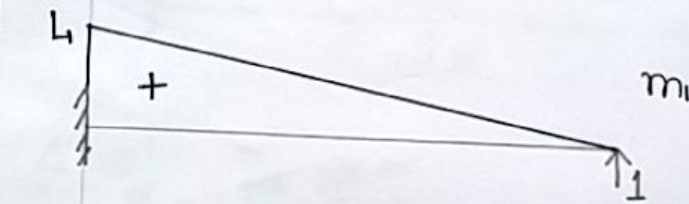
Released str:



$$\Delta_{01} = \int_L \frac{m_0 m_1}{EI} dx$$

$$= \frac{L/2}{2 \times 6} \times \left(-\frac{PL}{2}\right) \left(2L + \frac{L}{2}\right)$$

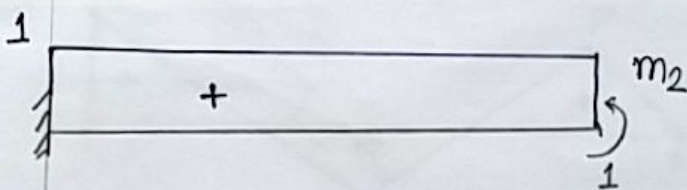
$$= -\frac{5PL^3}{96EI} \quad (\text{only deformation})$$



$$\Delta_{02} = \int_L \frac{m_0 m_2}{EI} dx$$

$$= \frac{L/2}{2 \times 2EI} \times \left(-\frac{PL}{2}\right) \times 1$$

$$= -\frac{PL^2}{16EI} \quad (\text{only rotation})$$



$$\delta_{11} = \int_L \frac{m_1 m_1}{EI} dx$$

$$= \frac{L/2}{6 \cdot 2EI} \left[ L \times (2L + L/2) + \frac{L}{2} (1 + 2 \times \frac{L}{2}) \right] + \frac{L/2}{3EI} \times \frac{L}{2} \times \frac{L}{2}$$

$$= \frac{L}{24EI} \left( \frac{5L^2}{2} + \frac{L^2}{2} + \frac{L^2}{4} \right) = \frac{3L^3}{16EI}$$

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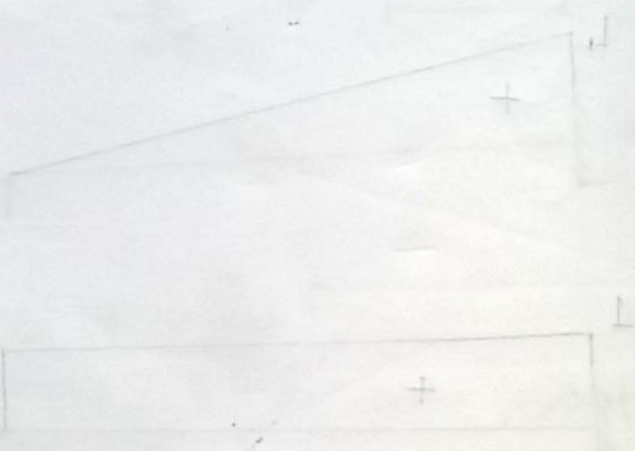
$$\delta_{12} = \delta_{21} = \int_L \frac{m_1 m_2}{EI} dx$$

$$= \frac{L/2}{2 \times 2EI} \times 1 \times (L + \frac{L}{2}) + \frac{L/2}{2EI} \times \frac{L}{2}$$

$$= \frac{5L^2}{16EI}$$

$$\delta_{22} = \int_L \frac{m_2 m_2}{EI} dx = \frac{L/2}{2EI} \times 1 \times 1 + \frac{L/2}{EI} \times 1 \times 1 = \frac{3L}{4EI}$$

$$\begin{Bmatrix} \frac{-5PL^3}{96EI} \\ -\frac{PL^2}{16EI} \end{Bmatrix} + \begin{bmatrix} \frac{9L^3}{48EI} & \frac{5L^2}{6EI} \\ \frac{5L^2}{6EI} & \frac{3L}{4EI} \end{bmatrix} \begin{Bmatrix} R_1 \\ R_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

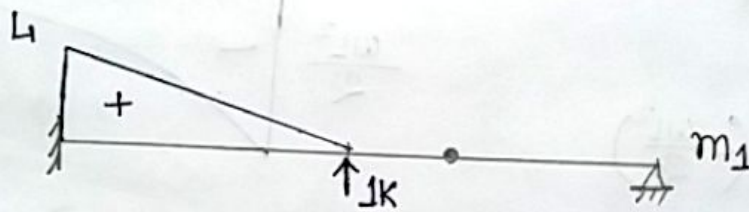
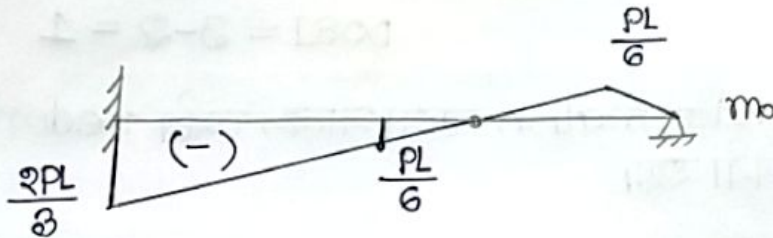
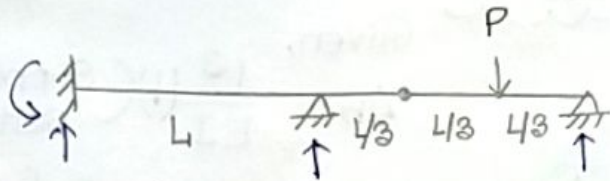


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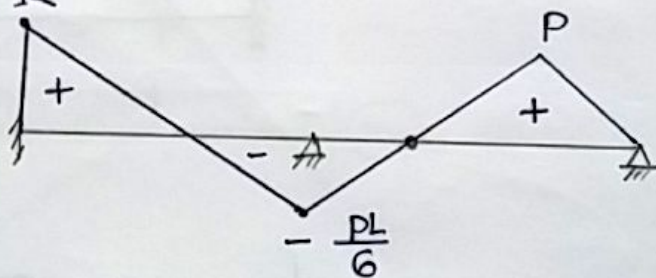
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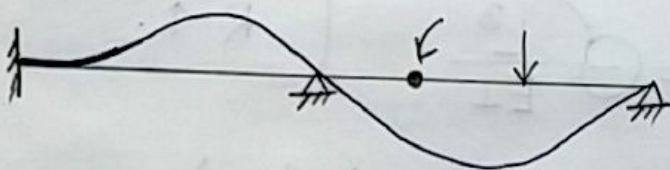
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$$\Delta = \frac{PL}{12} = -\frac{2PL}{3} + \frac{3PL}{4}$$



point of contraflexure



+ve 'M' saggings

$$DOSI = 1$$

[No. of Reaction = 4  
eqn of equilibrium = 2  
from internal hinge  $\rightarrow = 1$   
( $\sum M = 0$  at hinge)]

$$\text{So, } DOSI = 4 - 2 - 1 = 1$$

$$\Delta_0 = \int_L \frac{m_0 m_1}{EI} dx$$

$$= \frac{L}{6EI} \times L \times \left(-\frac{4PL}{3} - \frac{PL}{6}\right)$$

$$= -\frac{PL^3}{4EI}$$

$$\delta = \int_L \frac{m_1 m_1}{EI} dx$$

$$= \frac{L}{3EI} \times L \times L = \frac{L^3}{3EI}$$

$$\therefore -\frac{PL^3}{4EI} + \frac{L^3}{3EI} \times R = 0$$

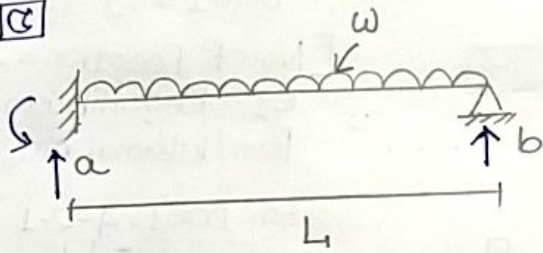
$$\therefore R = \frac{PL^3}{4EI} \times \frac{3EI}{L^3} = \frac{3P}{4}$$

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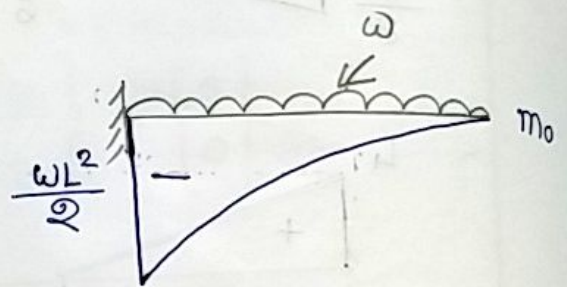
Given,  $\Delta_b = \frac{L^3}{EI} (\downarrow)$  (Support settlement)

$$\{\Delta_0\} + [\delta] \{R\} = \{\Delta_s\}$$

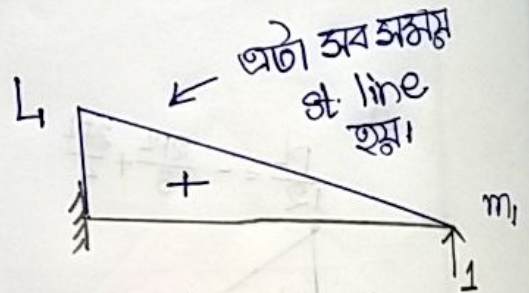
$$DOFI = 3 - 2 = 1$$

• এই support ওয় deflection দিয়া থাকে, হেঁচা redundant বিনলে ওয় এ সুবিধা হয়।

$$\begin{aligned} \Delta_0 &= \int_L \frac{m_0 m_1}{EI} dx \\ &= \frac{L}{8EI} (3L+0) \left(-\frac{wL^2}{2}\right) \\ &= -\frac{wL^4}{8EI} \end{aligned}$$



$$\begin{aligned} \delta &= \int_L \frac{m_1 m_1}{EI} dx \\ &= \frac{L}{3EI} \times L \times L = \frac{L^3}{3EI} \end{aligned}$$



$$\Delta_0 + \delta R = \Delta_s$$

$$\therefore -\frac{wL^4}{8EI} + \frac{L^3}{3EI} \cdot R = -\frac{L^3}{EI}$$

{ -ve, কারণ, support settlement ↓, but R ↑. }

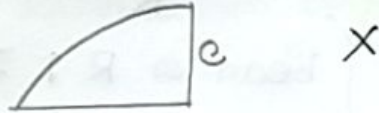
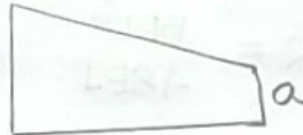
$$\Rightarrow R = \left( \frac{wL^4}{8EI} - \frac{L^3}{EI} \right) \times \frac{3EI}{L^3} = \left( \frac{wL}{8} - 1 \right) \times 3$$

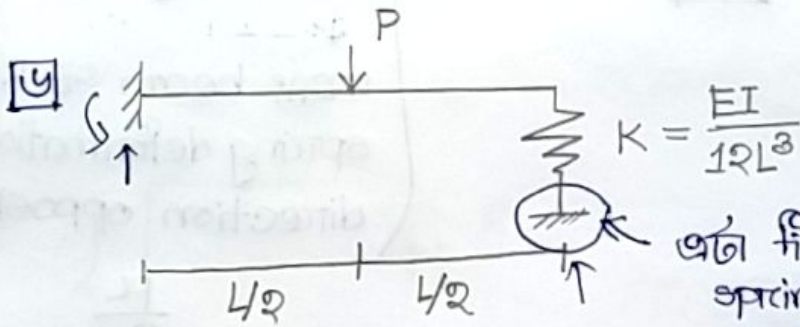
অর্থাৎ, settlement নিচে হওয়াতে Reaction বন্ধে গেছে।

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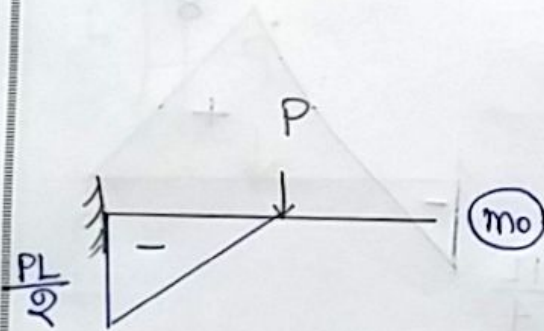
•   $\times$    $= \frac{c}{12} (3a+b)c$

•   $\times$    $= \frac{c}{12} (5a+3b)c$



এটা fixed support না, just  
 spring বুঝা। এটা reaction  
 আছে spring অর্থাৎ  $\uparrow$  বরাবর।

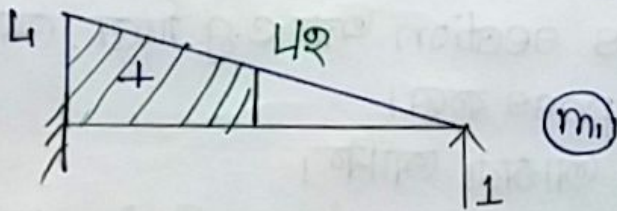
$DOKI = 3 - 2 = 1$   
 $\Delta_0 + \delta R = \Delta_s$



$\Delta_0 = \int_L \frac{m_0 m_1}{EI} dx$

$= \frac{L/2}{6EI} \left( -\frac{PL}{2} \right) \left( 2L + \frac{L}{2} \right)$

$= -\frac{5PL^3}{48EI}$



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$$\delta = \int_0^L \frac{m_i m_j}{EI} dx = \frac{L^3}{8EI}$$

$\Delta_s =$  spring deformation

$$\therefore -\frac{5PL^3}{48EI} + \frac{L^3}{8EI} \cdot R = -\frac{R}{\frac{L^3}{12EI}}$$

deformation =  $\frac{\text{Force}}{\text{stiffness}}$   
 $= \frac{R}{K}$

$$\Rightarrow \left(\frac{1}{3} + \frac{1}{12}\right) \frac{L^3}{EI} \cdot R = \frac{5PL^3}{48EI}$$

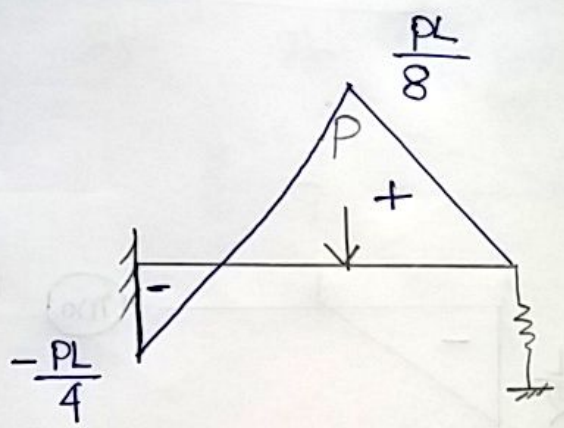
$$\therefore R = \frac{5P}{48} \times \frac{12}{5} = \boxed{\frac{P}{4}}$$

Beam এ R ↑ হলে, spring এ R ↓.  
 So, spring deformation ↓.  
 অর্থাৎ beam এর R ও spring deformation এর direction opposite.

$M = m_0 + m_1 R$

$$M_a = -\frac{PL}{4}$$

- Ordinary DE দিয়ে যেগুলো solve করা যায়, সেগুলো 1D structure.  $f(x)$ .
- partial DE দিয়ে যেগুলো solve করা যায় সেগুলো 2D, 3D structure.
- 1D structure কে skeleton structure ও বলা হয়।



Longitudinal axis → cross section এর c.g দিয়ে যে line pass করে।  
 অর্থাৎ আক্ষর আকি।

• we assume all the (parameters are only  $f(x)$ ). কারণ, and sectional properties, loads, deformation

$$\boxed{EI \frac{d^4 y}{dx^4} = p(x)} \leftarrow \text{ODE}$$

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1D - beam, frame

2D - slab, truss

3D - Dome, shell

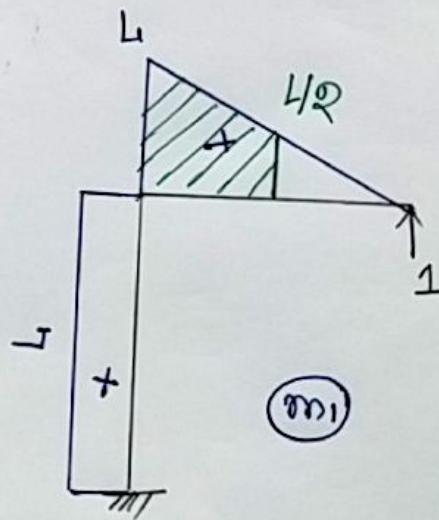
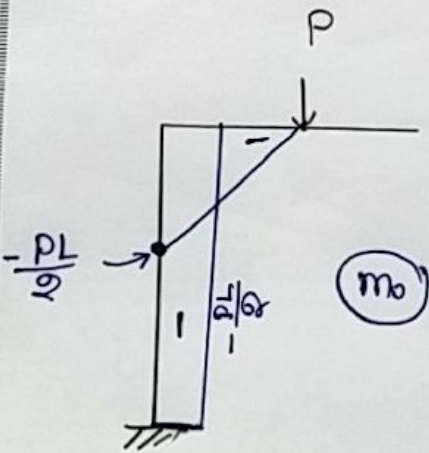
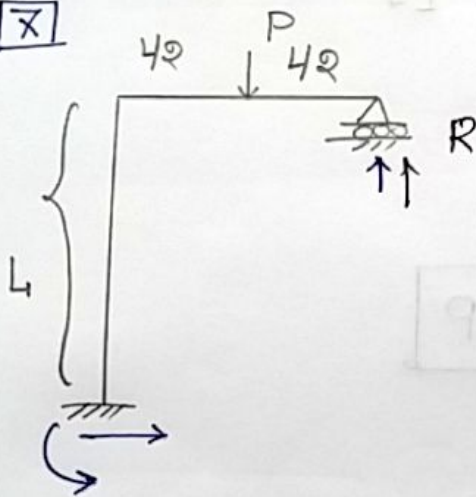
- Idealization: beam 3D বস্তু। But analysis সহজে করার জন্য 1D বস্তুে নিচ্ছি। একে dimensional idealization বলে।
- আরেকটা হল material idealization.

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Frame

$E, I, A$

$$DOF = 3 - 2 = 1$$



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$$\Delta_0 = \int_0^L \frac{m_0 m_1}{EI} dx$$

$$= \frac{L}{EI} \left(-\frac{PL}{2}\right) L + \frac{L/2}{6EI} \left(-\frac{PL}{2}\right) \left(2L + \frac{L}{2}\right)$$

$$= -\frac{PL^3}{2EI} - \frac{5PL^3}{48EI} = -\frac{29PL^3}{48EI}$$

$$\delta = \int_0^L \frac{m_1 m_1}{EI} dx$$

$$= \frac{L}{EI} \times L \times L + \frac{L}{3EI} \times L \times L = \frac{L^3}{EI} \left(1 + \frac{1}{3}\right) = \frac{4L^3}{3EI}$$

$$\therefore -\frac{29PL^3}{48EI} + \frac{4L^3}{3EI} R = 0$$

$$\therefore R = \frac{29}{48} \times \frac{3}{4} \times P = \boxed{\frac{29}{64} P}$$



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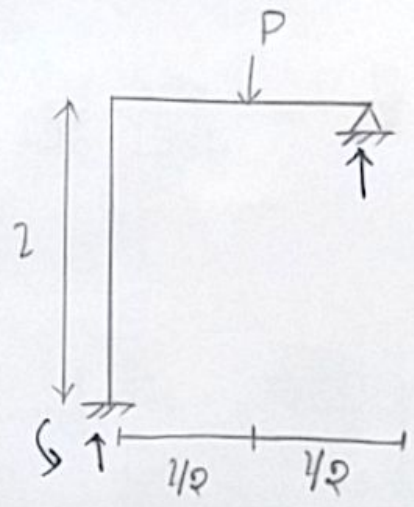
Next week C.T  
Beam

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axial force consider করবে।

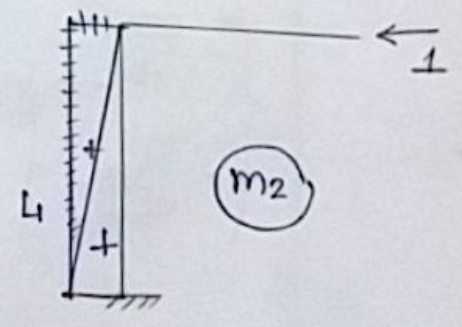
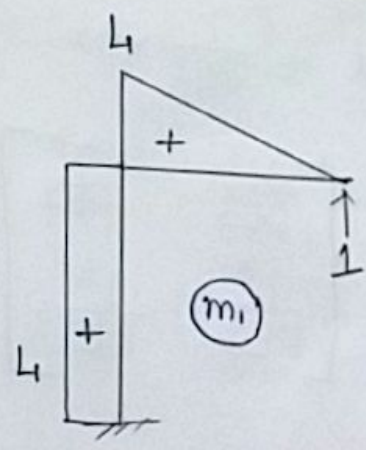
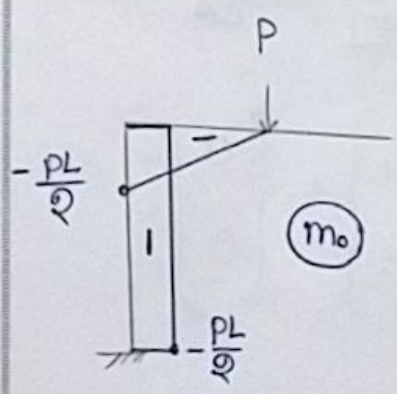
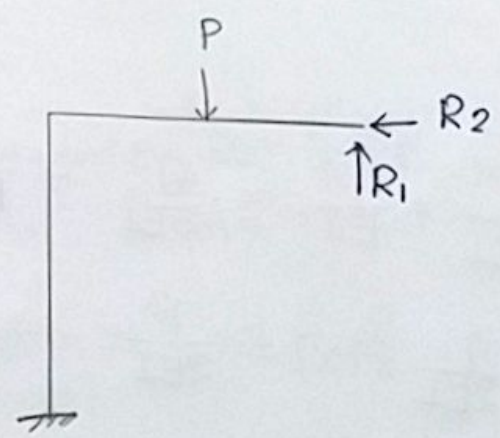
$DOF = 5 - 3 = 2$

$\{\Delta_0\} + [\delta] \{R\} = \{\Delta_s\}$

$\begin{Bmatrix} \Delta_{01} \\ \Delta_{02} \end{Bmatrix} + \begin{bmatrix} \delta_{11} & \delta_{12} \\ \delta_{21} & \delta_{22} \end{bmatrix} \begin{Bmatrix} R_1 \\ R_2 \end{Bmatrix} = \begin{Bmatrix} \Delta_{s1} \\ \Delta_{s2} \end{Bmatrix}$

$\Delta_{s1} = \Delta_{s2} = 0$

• compression side এ diagram আঁকি।



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$$\Delta_{01} = \int_4 \frac{m_0 m_1}{EI} dx$$

$$= \frac{l/2}{6EI} \times \left(-\frac{PL}{2}\right) \left(2l + \frac{l}{2}\right) + \frac{l}{EI} \left(-\frac{PL}{2}\right)^2 = -\frac{5Pl^3}{48EI} - \frac{Pl^3}{2EI} = -\frac{29Pl^3}{48EI} - \frac{PL}{EA}$$

$$\Delta_{02} = \int_4 \frac{m_0 m_2}{EI} dx$$

$$= \frac{l}{2EI} \left(-\frac{PL}{2}\right)^2 = -\frac{Pl^3}{4EI}$$

$$\delta_{11} = \int_4 \frac{m_1 m_1}{EI} dx$$

$$= \frac{l}{3EI} \times l \times l + \frac{l}{EI} \times l \times l = \frac{l^3}{3EI} + \frac{l^3}{EI} = \frac{4l^3}{3EI} + \frac{L}{EA}$$

$$\delta_{12} = \delta_{21} = \int_4 \frac{m_1 m_2}{EI} dx = \frac{l}{2EI} \times l \times l = \frac{l^3}{2EI}$$

$$\delta_{22} = \int_4 \frac{m_2 m_2}{EI} dx = \frac{l}{3EI} \times l \times l = \frac{l^3}{3EI} + \frac{L}{EA}$$

$$\begin{Bmatrix} -\frac{29Pl^3}{48EI} \\ -\frac{Pl^3}{4EI} \end{Bmatrix} + \begin{bmatrix} \frac{4l^3}{3EI} & \frac{l^3}{2EI} \\ \frac{l^3}{2EI} & \frac{l^3}{3EI} \end{bmatrix} \begin{Bmatrix} R_1 \\ R_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

$$R_1 = \frac{\begin{vmatrix} \frac{4l^3}{3EI} & \frac{l^3}{2EI} \\ \frac{29Pl^3}{48EI} & \frac{Pl^3}{4EI} \end{vmatrix}}{\begin{vmatrix} \frac{4l^3}{3EI} & \frac{l^3}{2EI} \\ \frac{l^3}{2EI} & \frac{l^3}{3EI} \end{vmatrix}} = \frac{\frac{29}{48} \times \frac{1}{3} - \frac{1}{12} \times \frac{1}{4}}{\frac{4}{3} \times \frac{1}{3} - \frac{1}{2} - \frac{1}{3}} P = \frac{\frac{11}{144}}{\frac{7}{36}} P$$

$$= \frac{11}{28} P$$

$$= 0.4P$$

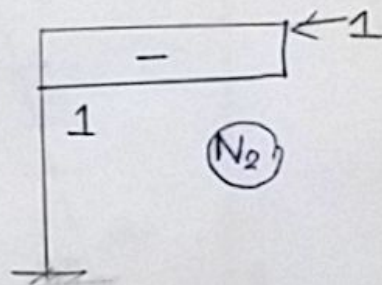
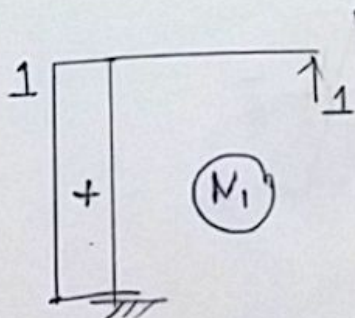
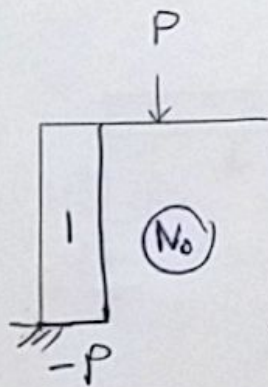
$$R_2 = \frac{\begin{vmatrix} \frac{4l^3}{3EI} & \frac{29Pl^3}{48EI} \\ \frac{l^3}{2EI} & \frac{Pl^3}{4EI} \end{vmatrix}}{7/36}$$

support এ  $R_1 = 0.45P$

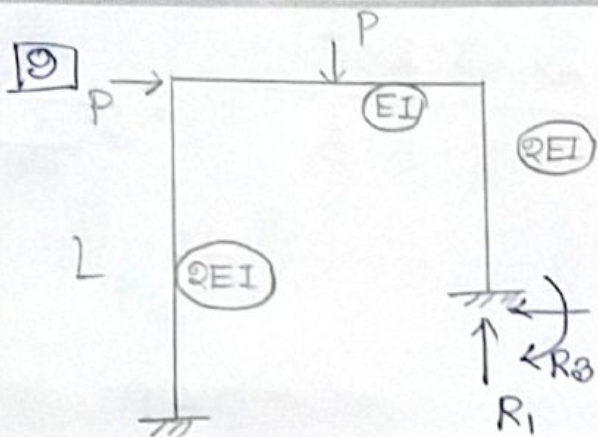
horizontal force  $R_2$  থাকবে বলেছে  $R_1$ .

তাই Beam এ axial force neglect করলেও frame এ neglect করা যাবে না।

$$\Delta_{01} = \int_L \frac{m_0 m_1}{EI} dx + \sum \frac{N_0 N_1}{EA} L$$

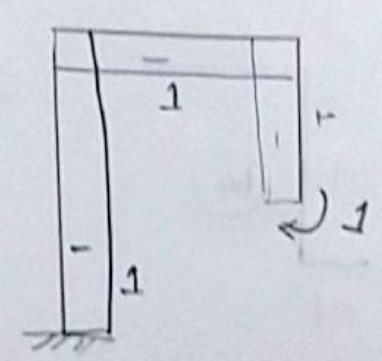
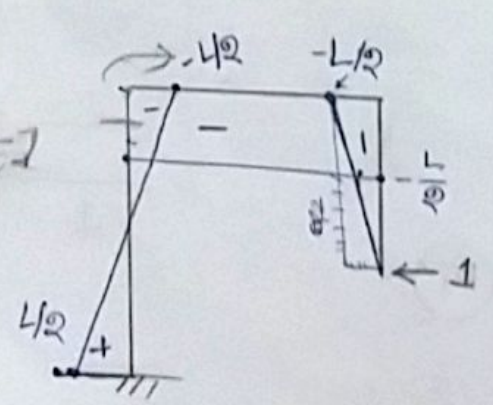
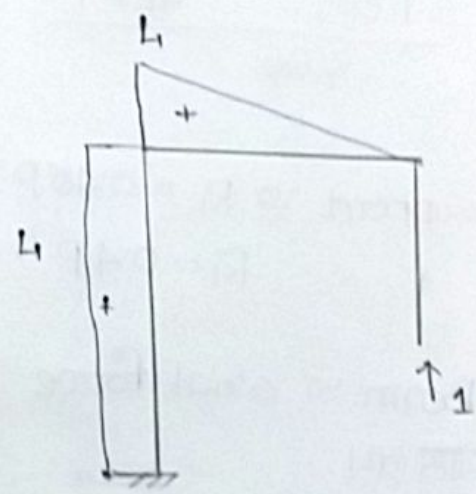
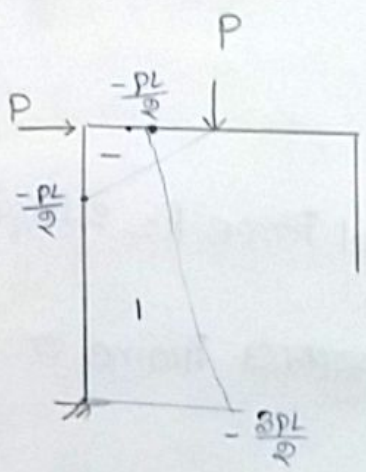


axial force এর জন্য যে deformation,  $\frac{L}{EA}$  হলে  $\frac{l^3}{EI}$  এর তুলনায় অনেক বড়, তাই সেটা neglect করতে পারি। exam এ consider করতে যখন consider (A. Defor.) দেবে।



$DOF = 3$

$$\begin{Bmatrix} \Delta_{01} \\ \Delta_{02} \\ \Delta_{03} \end{Bmatrix} + \begin{bmatrix} \delta_{11} & \delta_{12} & \delta_{13} \\ \delta_{21} & \delta_{22} & \delta_{23} \\ \delta_{31} & \delta_{32} & \delta_{33} \end{bmatrix} \begin{Bmatrix} R_1 \\ R_2 \\ R_3 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \\ 0 \end{Bmatrix}$$



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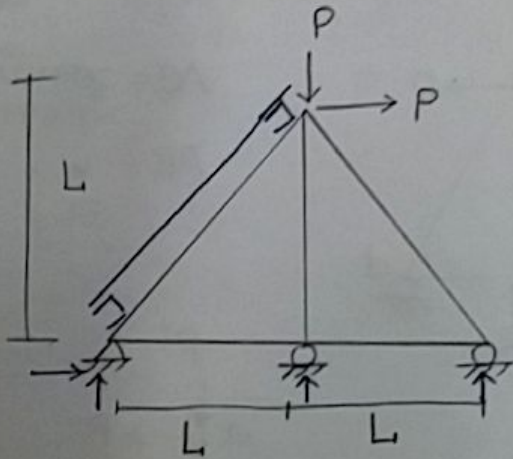
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Truss Characteristics

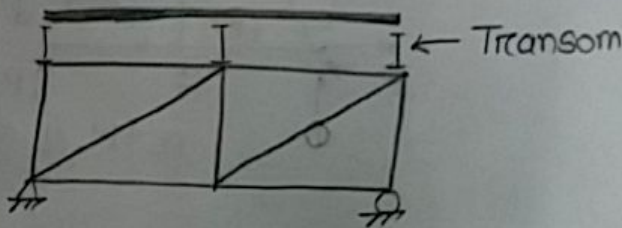
Truss Problem:



• সবগুলো member pin connected.

• সবগুলো ফিলে joint ও কাজ বসাবে।

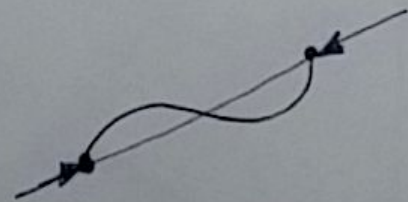
• পেরলিন গুলো joint ও connected থাকে, so load joint ও আসে।



• co-planar con-current হতে হবে Moment যাতে develop বসতে না পারে, সে জন্য। Line of action of resultant ফিলে at joints same হতে হবে।

• Truss এর Member গুলো straight হলে স্কর্প axial ফিলে develop করে।

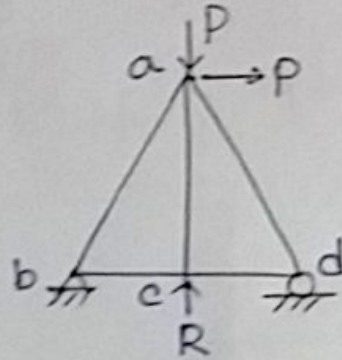
• Curve হলে V, M ও develop করে।



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• Axial force develop করলে material optimization হয়, X-section ও uniform stress develop করে।

যদি V, M develop করলে বেগাও করে, বেগাও বেশি material লাগবে।



AC = zero force member

AB = u u u

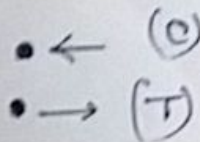
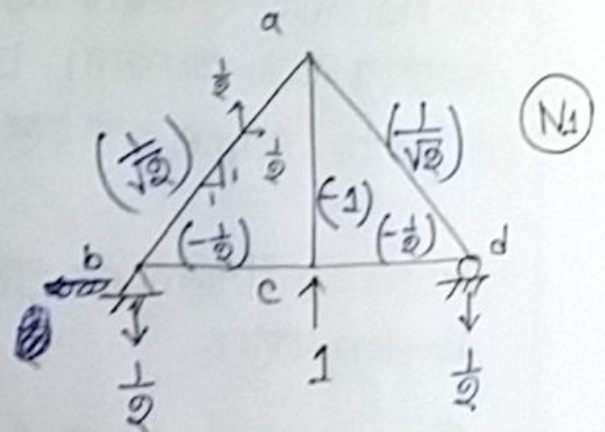
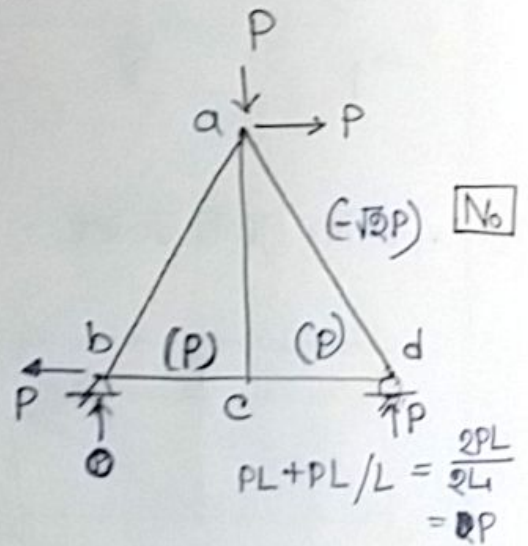
DOF = 4 - 3 = 1

$\Delta_0 + \delta_R = \Delta_s$

$\Delta_0 = \sum \frac{N_0 N_1 L}{AE}$

$\delta_R = \sum \frac{N_1 N_1 L}{AE}$

$\Delta_s = 0$



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	Bars	$L$	$N_0$	$N_1$	$N_0 N_1 L$	$N_1 N_1 L$
diagonal	ab	$\sqrt{2}L$	0	$1/\sqrt{2}$	0	$L/\sqrt{2}$
	ad	$\sqrt{2}L$	$-\sqrt{2}P$	$1/\sqrt{2}$	$-\sqrt{2}PL$	$+L/\sqrt{2}$
bottom chord	bc	L	P	$-1/2$	$-\frac{1}{2}PL$	L/4
	cd	L	P	$-1/2$	$-\frac{1}{2}PL$	L/4
vertical	ae	L	0	-1	0	L
						$\Sigma = -PL(1+\sqrt{2})$

$$R = -\frac{\Delta_0}{\delta} = -\frac{\frac{-PL(1+\sqrt{2})}{EA}}{\frac{L(\frac{3}{2} + \sqrt{2})}{EA}} = \frac{1+\sqrt{2}}{\frac{3}{2} + \sqrt{2}} \cdot P$$

$$N = N_0 + N_1 R$$

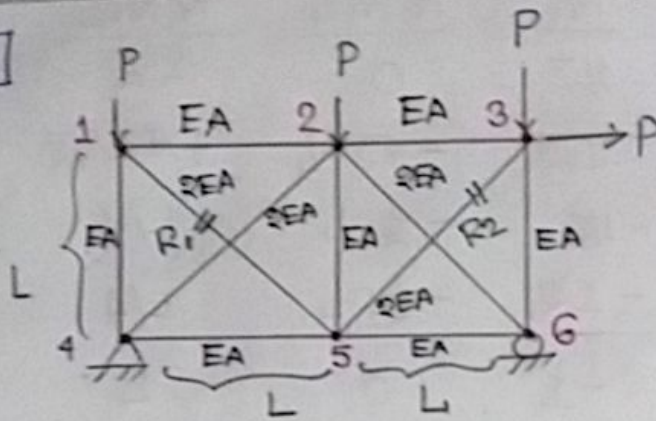
$$N_{ab} = 0 + \frac{1}{\sqrt{2}} \cdot \left( \frac{1+\sqrt{2}}{\frac{3}{2} + \sqrt{2}} \right) \cdot P =$$

$$N_{ad} = -\sqrt{2}P + \frac{1}{\sqrt{2}} \cdot \left( \frac{1+\sqrt{2}}{\frac{3}{2} + \sqrt{2}} \right) \cdot P =$$

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Internal indeterminacy

Unknowns:

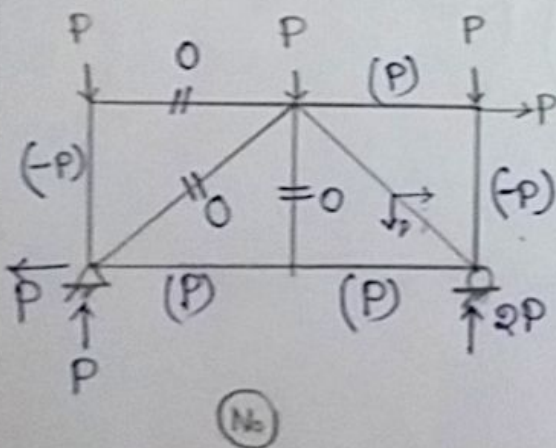
$btr = 11 + 3 = 14$   
 member force      support reaction

concurrent force system এর জন্য equi. eqn = ২টা,  
 Moment zero.  $F_x, F_y$ .

Known =  $2j = 2 \times 6 = 12$   
 ← no. of joint

$\therefore DOBI = 14 - 12 = 2$  (internal)

Support force কে redundant করতে পারবে না, unbalance হওয়ায়।



$$\begin{Bmatrix} \Delta_{01} \\ \Delta_{02} \end{Bmatrix} + \begin{bmatrix} \delta_{11} & \delta_{12} \\ \delta_{21} & \delta_{22} \end{bmatrix} \begin{Bmatrix} R_1 \\ R_2 \end{Bmatrix} = \begin{Bmatrix} \Delta_{j1} \\ \Delta_{j2} \end{Bmatrix}$$

$$\Delta_{01} = \sum \frac{N_0 N_1 L}{EA}$$

Redundant এর misfit এর effect  $\Delta_j$  তে আসে।

$T_p$  বা external force বা অন্য কিছু effect  $\Delta_0$  এ আসে।

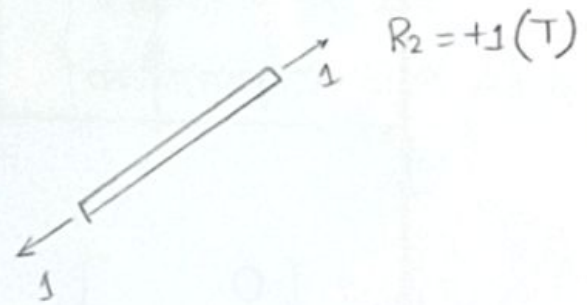
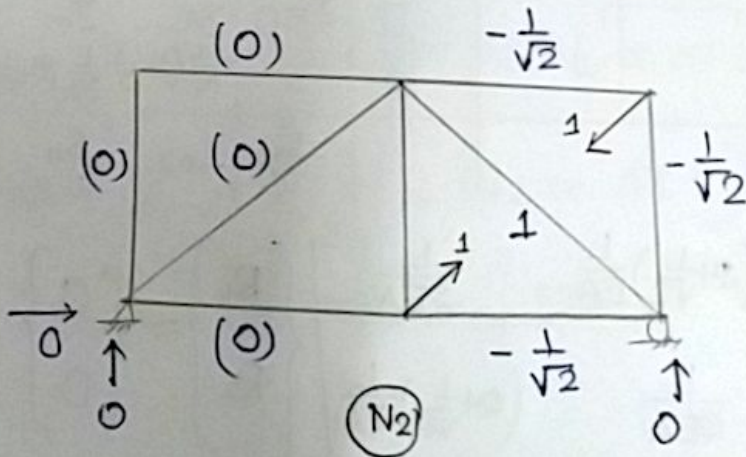
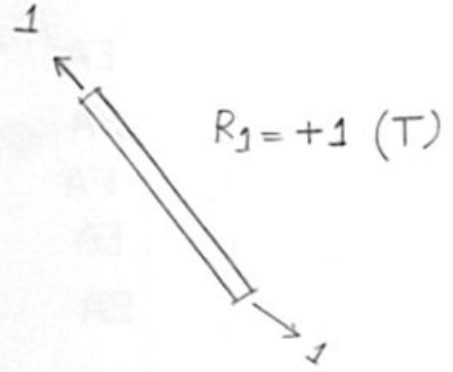
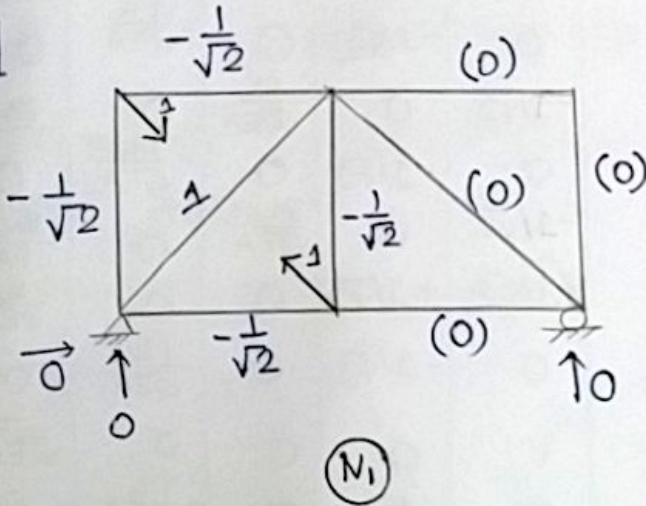


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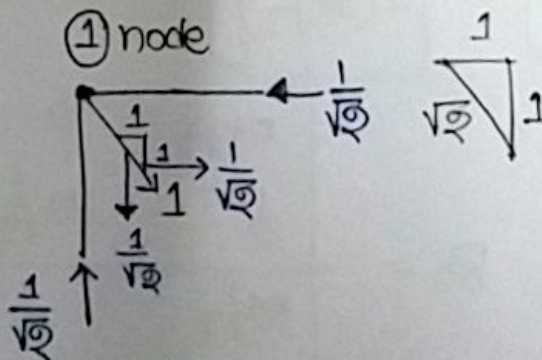
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[11]



Global structure  
 চিহ্না বস্তুনে 1, 1 জাচে  
 ২টি colinear, balance  
 হয় মাঝে। তাই support  
 reaction হুন্না zero.



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member	L	EA	N <sub>0</sub>	N <sub>1</sub>	N <sub>2</sub>	$\frac{N_0 N_1 L}{EA}$	$\frac{N_0 N_2 L}{EA}$	$\frac{N_1^2 L}{EA}$	$\frac{N_2^2 L}{EA}$	$\frac{N_1 N_2 L}{EA}$
12	L	EA	0	-1/√2	0	0	0	L/2EA	0	0
23	L	EA	P	0	-1/√2	0	-PL/√2EA	0	L/2EA	0
45	L	EA	P	-1/√2	0	-PL/√2EA	0	L/2EA	0	0
56	L	EA	P	0	-1/√2	0	-PL/√2EA	0	L/2EA	0
14	L	EA	-P	-1/√2	0	PL/√2EA	0	L/2EA	0	0
25	L	EA	0	-1/√2	-1/√2	0	0	L/2EA	L/2EA	L/2EA
36	L	EA	-P	0	-1/√2	0	PL/√2EA	0	L/2EA	0
24	√2L	2EA	0	1	0	0	0	L/√2EA	0	0
26	√2L	2EA	-√2P	0	1	0	-PL/√2EA	0	L/√2EA	0
						Σ = 0	Σ = -PL/AE (1+1/√2)	Σ = L/EA (2+1/√2)	Σ = L/EA (2+1/√2)	Σ = L/2EA
						Δ <sub>01</sub>	Δ <sub>02</sub>	δ <sub>11</sub>	δ <sub>22</sub>	δ <sub>12</sub>

$$\therefore \begin{Bmatrix} 0 \\ \frac{PL}{AE} (1 + \frac{1}{\sqrt{2}}) \end{Bmatrix} + \begin{bmatrix} (2 + \frac{1}{\sqrt{2}}) \frac{L}{EA} & \frac{L}{2EA} \\ \frac{L}{2EA} & (2 + \frac{1}{\sqrt{2}}) \frac{L}{EA} \end{bmatrix} \begin{Bmatrix} R_1 \\ R_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

$$\begin{bmatrix} 2 + \frac{1}{\sqrt{2}} & \frac{1}{2} \\ \frac{1}{2} & 2 + \frac{1}{\sqrt{2}} \end{bmatrix} \begin{Bmatrix} R_1 \\ R_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ -(1 + \frac{1}{\sqrt{2}}) \end{Bmatrix} P$$

$$R_1 = \frac{\begin{vmatrix} 0 & \frac{1}{2} \\ -(1 + \frac{1}{\sqrt{2}}) & 2 + \frac{1}{\sqrt{2}} \end{vmatrix}}{\begin{vmatrix} 2 + \frac{1}{\sqrt{2}} & \frac{1}{2} \\ \frac{1}{2} & 2 + \frac{1}{\sqrt{2}} \end{vmatrix}}, \quad R_2 = \frac{\begin{vmatrix} 2 + \frac{1}{\sqrt{2}} & 0 \\ \frac{1}{2} & -(1 + \frac{1}{\sqrt{2}}) \end{vmatrix}}{\begin{vmatrix} 2 + \frac{1}{\sqrt{2}} & \frac{1}{2} \\ \frac{1}{2} & 2 + \frac{1}{\sqrt{2}} \end{vmatrix}}$$

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$$N = N_0 + N_1 R_1 + N_2 R_2$$

Redundent অর misfit

- Redundent অর দিকে shortening হলে R.H.S. এ  $\Delta$  (+ve) হিসেবে বসবে।
- " elongated হলে " "  $\Delta$  (-ve) হিসেবে বসবে।

ফিরে  $T_p$  change

$$\Delta_{01} = \sum \left( \frac{N_0 N_1 L}{EA} + N_1 \cdot \overbrace{\alpha \cdot \Delta T \cdot L}^{\text{ফিরে } T_p} \right) \left\{ \begin{array}{l} \text{strain} = \alpha \cdot \Delta T \\ \text{deformation} = \alpha \cdot \Delta T \cdot L \end{array} \right.$$

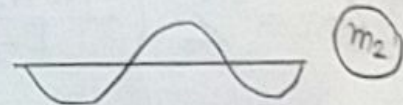
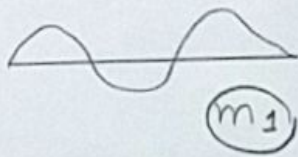
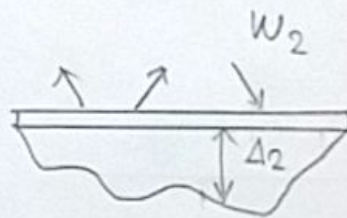
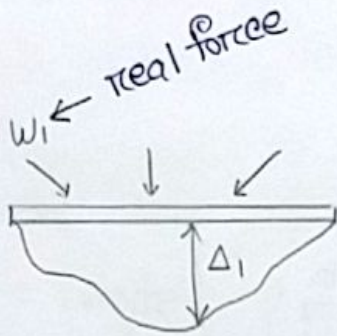
$$\Delta_{02} = \sum \left( \frac{N_0 N_2 L}{EA} + N_2 \cdot \alpha \cdot \Delta T \cdot L \right)$$

$N_0 \leftarrow$  ফিরে external ফোর্সে

অনিবার্ণ চ: ০০ টোম্ব রাফিল অ্যাঙ্কোর class  
 সপ্তাহান্তবার ৯:৩০ এ C.T  $\rightarrow$  Truss

# Influence Lines of Indeterminate Structure

• Betty's Law :



virtual load  $\times$  real deformation = virtual work

Here,  $P = W_1 = \text{real force}$

$Q = W_2 = \text{virtual}$

$$\sum W_1 \Delta_2 = \underbrace{\int \frac{m_1 m_2}{EI} dx}_{\text{internal work done}}$$

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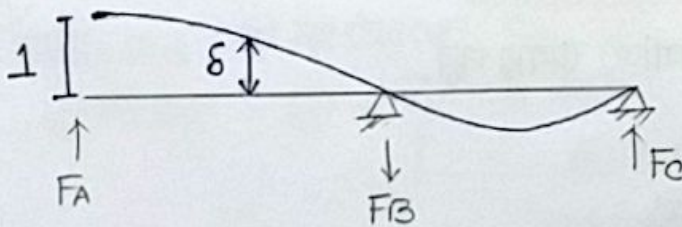
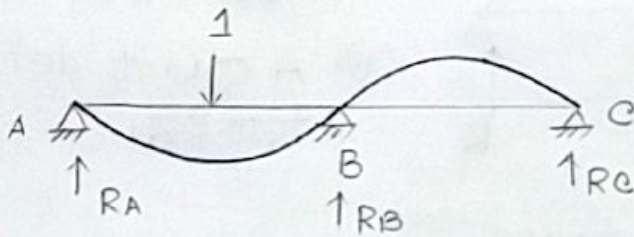
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Again, if,  $p = w_2$ ,  $Q = w_1$

$$\text{then, } \sum W_2 \Delta_1 = \int_L \frac{m_2 m_1}{EI} dx$$

$$\therefore \boxed{\sum W_1 \Delta_2 = \sum W_2 \Delta_1}$$

### Muller-Breslau's Principle



A তে unit deformation  
 দিলে  $F_A$  সঙ্কীর্ণ load  
 দিতে হবে। যখন B তে  
 $F_B$  ও  $F_C$  চিহ্নে develop  
 করবে।

$$R_A \times 1 - 1 \times \delta = 0$$

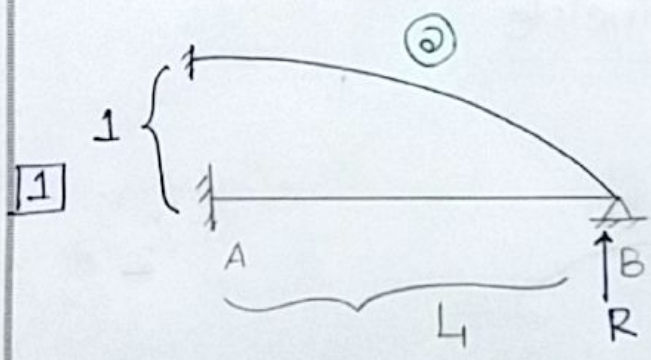
$$\therefore \boxed{R_A = \delta}$$

অন্যভাবে I.L চিহ্নে ওখানে unit deformation দিলে সেই  
 deformed shape দাব হোলেই যে point এ reaction.

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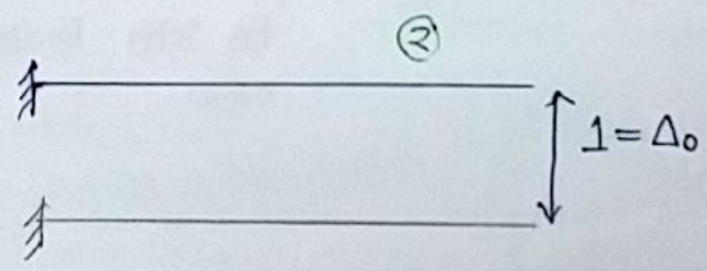
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- determinate structure এ rigid body motion হয়, ফলে strain হয় না। So, I.L সবসময় straight line হবে।
- Indeterminate structure এ rigid body motion হয় না, strain হয়। তাই I.L curve হবে।



- Draw I.L of  $R_A$ .
- (i) A ত unit deformation দিতে হবে।

(ii) • যদি B কে redundant choose করি, তবে released structure হবে, A ত unit deformation দেয়া দর,  $\equiv$



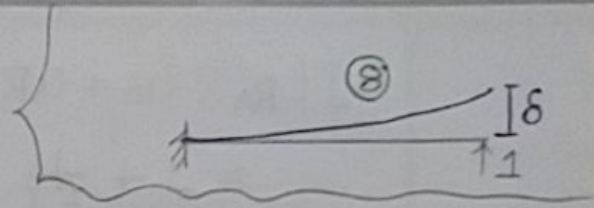
যেহে unit deformation দিচ্ছি, তাই  $\Delta_0 = 1$ .

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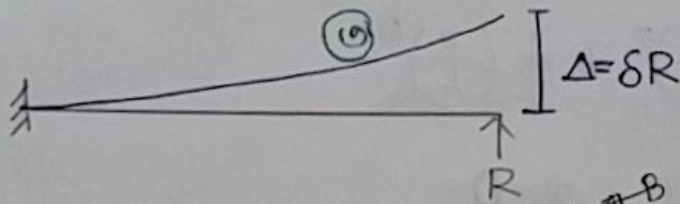
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R এর জন্য deflection  $\Delta$ .



So,  $\Delta_0 + \Delta = \Delta_j$   
 $\Rightarrow \Delta_0 + \delta R = \Delta_j$

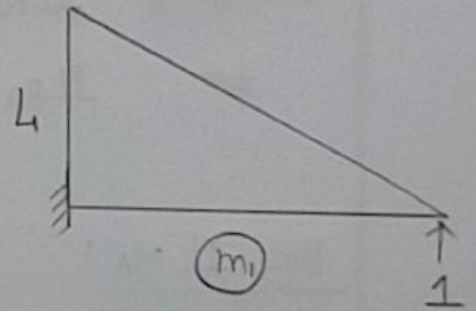
$\Rightarrow 1 + \frac{L^3}{3EI} R = 0$

$\therefore R = -\frac{3EI}{L^3}$

? যদি A বা B point এ support settlement হয় অর্থ কি হবে?

$\Delta_j \rightarrow$  হল ৯ নং Diagram এ

এ point কে redundant choose করছি (যে point এর  $\Delta$  এখানে  $\Delta_j$  এর diagram হল  $\Delta B = 0$ .)



$\delta = \int_0^L \frac{m_1 m_1}{EI} dx$

• আমাদের ৯ নং curve এর  $e_2^n$  লাগবে।

• ৯ নং  $e_2^n = ৯ + ৯$  এর  $e_2^n$

৯ এর  $e_2^n$  নাওয়া যাবে ৯ নং  $e_2^n$  হতে।

$\frac{d^2u}{dx^2} = \frac{M}{EI}$   
 $\Rightarrow \frac{du}{dx} = \frac{1}{EI} \int_0^x M dx = \frac{1}{EI} \int_0^x (L-x) dx = \frac{1}{EI} (Lx - \frac{x^2}{2})$

$\therefore u(x) = \frac{1}{EI} (L \frac{x^2}{2} - \frac{x^3}{6})$

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2 +ve  
5 -ve

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$$IL_{RA} = u_D + u_R$$

$$= 1 + \frac{1}{EI} \left( L \frac{x^2}{2} - \frac{x^3}{6} \right) \left( - \frac{3EI}{L^3} \right)$$

$$= 1 - \frac{3}{2} \left( \frac{x}{L} \right)^2 + \frac{1}{2} \left( \frac{x}{L} \right)^3$$

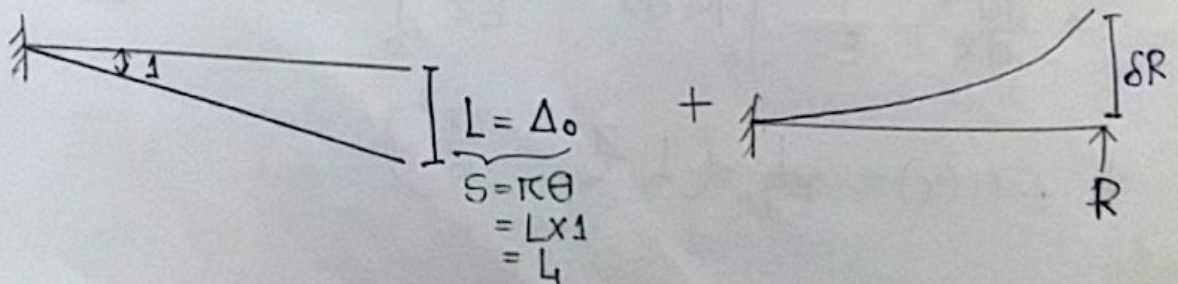
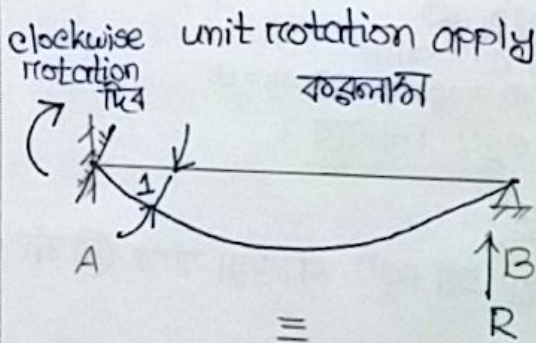
For  $x=0$   $IL_{RA} = 1$ ,  $x=0$

For  $x=L$   $IL_{RA} = 0$ ,  $x=L$

For  $IL_{MA}$ :

• Reaction এর জন্য deformation  $\Delta$ .

• Moment এর জন্য rotation apply করবে।





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$$\Delta_0 + \delta R = \Delta_j$$

$$\Rightarrow -L + \frac{L^3}{3EI} R = 0$$

$$\therefore R = \frac{3EI}{L^2}$$

$$s = \pi \theta$$

$$= \pi \chi \times 1$$

$$= \chi$$

$$ILMA = u_0 + uR$$

$$= -\chi + \frac{1}{EI} \left( L \frac{\chi^2}{2} - \frac{\chi^3}{6} \right) \times \frac{3EI}{L^2}$$

$$= -\chi + \frac{3}{2} \frac{\chi}{L} \chi - \frac{1}{2} \left( \frac{\chi}{L} \right)^2 \cdot \chi$$

$$= -\chi \left\{ 1 - \frac{3}{2} \frac{\chi}{L} + \frac{1}{2} \left( \frac{\chi}{L} \right)^2 \right\}$$

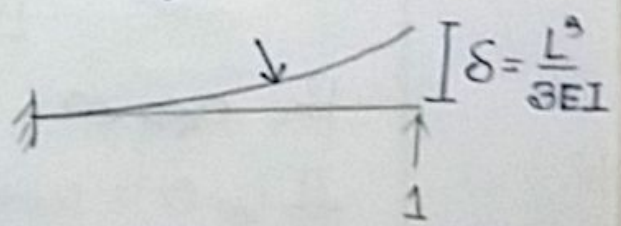
$$\chi = 0, ILMA = 0$$

$$\chi = L, ILMA = 0$$

unit deformation এর জন্য

Load

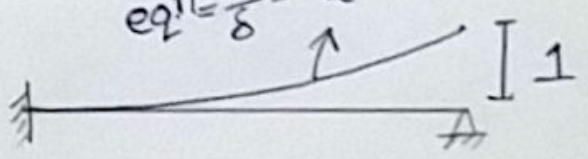
$$\text{eqn, } u(x) = \frac{1}{EI} \left( L \frac{\chi^2}{2} - \frac{\chi^3}{6} \right)$$



$$\delta = 1 \text{ হলে } 1 = \frac{L^3}{3EI}$$

For ILRB :

$$\text{eqn} = \frac{u}{\delta} = \frac{3}{2} \left( \frac{\chi}{L} \right)^2 - \frac{1}{2} \left( \frac{\chi}{L} \right)^3$$



$$\delta \rightarrow u$$

$$\therefore 1 \rightarrow \frac{u}{\delta}$$

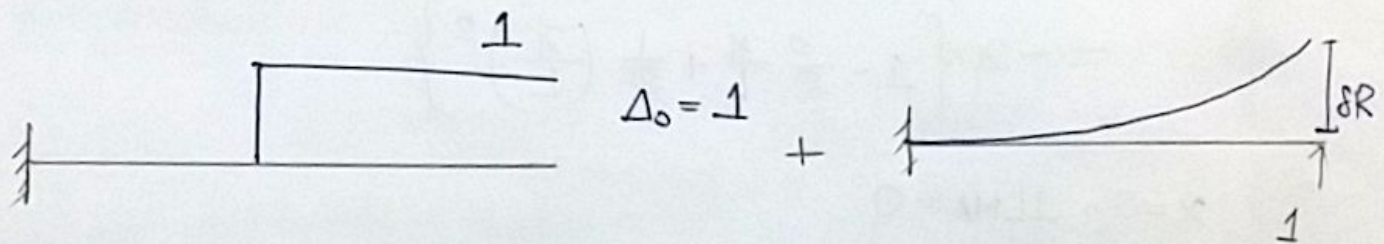
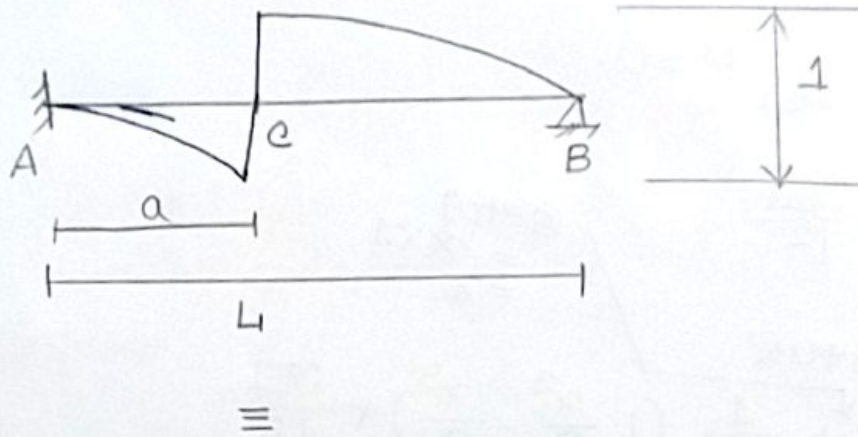
$$\therefore \text{eqn} = \frac{u}{\delta} = \frac{\frac{1}{EI} \left( L \frac{\chi^2}{2} - \frac{\chi^3}{6} \right)}{\frac{L^3}{3EI}} = 3 \left( \frac{\chi^2}{2L^2} - \frac{\chi^3}{6L^2} \right)$$

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~~Δ₀ = δR~~  $\Delta_0 + \delta R = 0$

$$R = - \frac{3EI}{L^3}$$

for  $0 \leq x < a$

$$IL v_e(x) = u_0 + u_R$$

$$= 0 + \frac{1}{EI} \left( L \frac{x^2}{2} - \frac{x^3}{6} \right) \left( - \frac{3EI}{L^3} \right)$$

$$= - \frac{3}{2} \left( \frac{x}{L} \right)^2 + \frac{1}{2} \left( \frac{x}{L} \right)^3$$

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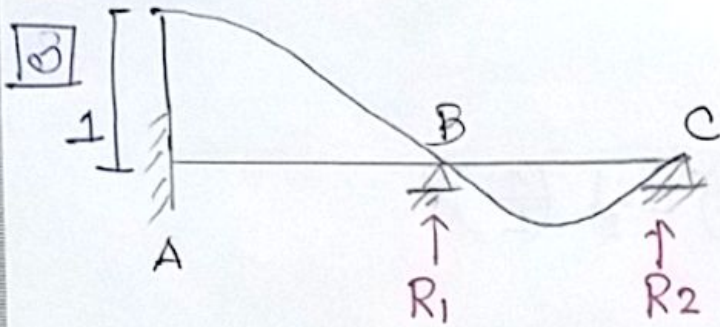
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For  $a < x \leq L$

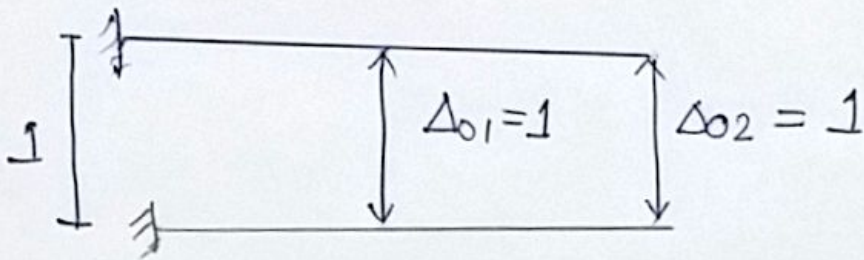
$$ILve(x) = u_0 + uR$$

$$= 1 - \frac{1}{9} \left(\frac{x}{L}\right)^2 + \frac{1}{9} \left(\frac{x}{L}\right)^3$$



ILRA (x) = ?  
 $\{\Delta_0\} + [S] \{R\} = \{\Delta_j\}$

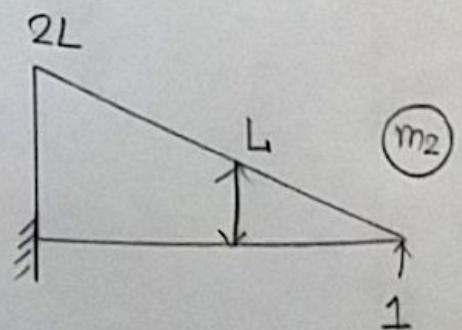
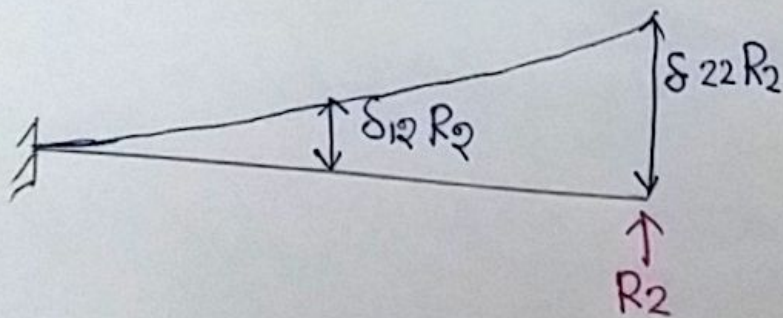
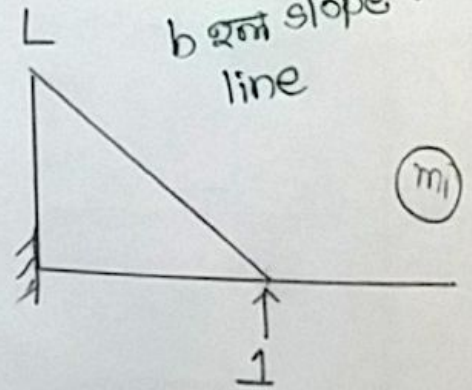
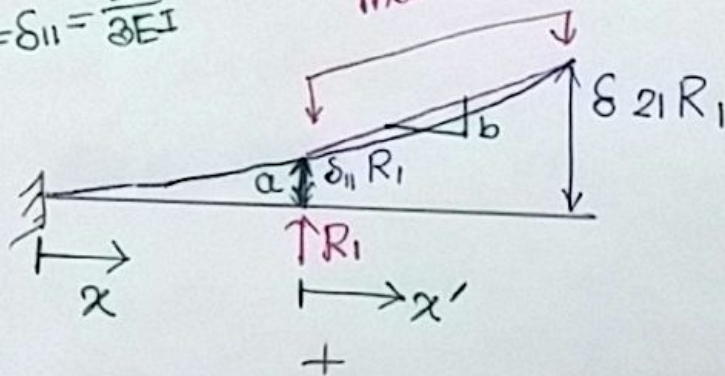
≡



+

$a = \delta_{11} = \frac{L^3}{3EI}$

moment नाहे, bend क्यार ना, अहे st. line शर  
 eqn of this st. line  
 $= a + bx'$   
 b शर slope of line



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$$\delta_{11} = \int_L \frac{m_1 m_1}{EI} dx = \frac{L}{3EI} \times L \times L = \frac{L^3}{3EI}$$

$$\delta_{12} = \delta_{21} = \int_L \frac{m_1 m_2}{EI} dx = \frac{L}{6EI} \times L \times (2 \times 2L + L) = \frac{5L^3}{6EI}$$

$$\delta_{22} = \int_L \frac{m_2 m_2}{EI} dx = \frac{2L}{3EI} \times 2L \times 2L = \frac{8L^3}{3EI}$$

$$\begin{Bmatrix} 1 \\ 1 \end{Bmatrix} + \begin{bmatrix} \frac{L^3}{3EI} & \frac{5L^3}{6EI} \\ \frac{5L^3}{6EI} & \frac{8L^3}{3EI} \end{bmatrix} \begin{Bmatrix} R_1 \\ R_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

$$R_1 = \frac{\begin{vmatrix} -1 & 5/6 \\ -1 & 8/3 \end{vmatrix}}{\begin{vmatrix} 1/3 & 5/6 \\ 5/6 & 8/3 \end{vmatrix}} \frac{EI}{L^3} \quad R_2 = \frac{\begin{vmatrix} 1/3 & -1 \\ 5/6 & -1 \end{vmatrix}}{\begin{vmatrix} 1/3 & 5/6 \\ 5/6 & 8/3 \end{vmatrix}} \frac{EI}{L^3}$$

$$= - \frac{11/6}{7/36} \frac{EI}{L^3} \quad = \frac{1/2}{7/36} \frac{EI}{L^3}$$

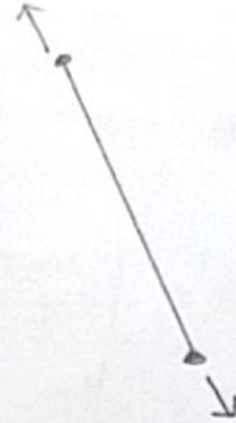
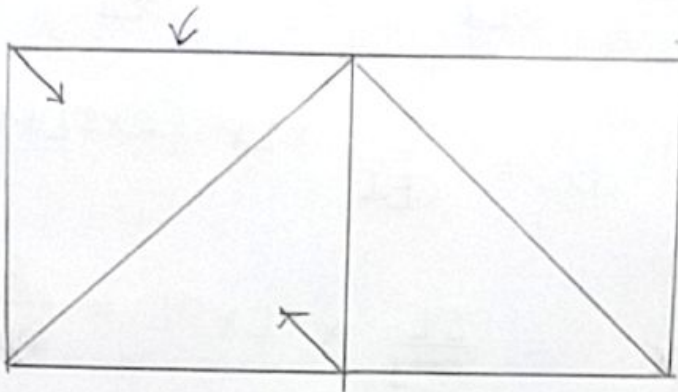
$$= - \frac{66}{7} \frac{EI}{L^3} \quad = \frac{18}{7} \frac{EI}{L^3}$$

$$IL R_A(x) = u_0 + u_1 R_1 + u_2 R_2$$

cont..... (A)

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$$-\frac{N_1^2}{EA} L \cdot R_1 = \Delta_j$$

$$\{\Delta_0\} + [S] \{R\} = \{\Delta_j\}$$

আগের math  
 বুঝাতো এ  
 এর add  
 হবে।

~~but deformation~~

but deformation  
 চিন্তে এর ঠিকতা,  
 অর্থাৎ (-ve)

- elongation বা shortening Redundent এর জন্যই  
 নিবা অর্থাৎ, কোন member এর elongation বা shortening  
 দেয়া থাকলে তাই Redundent choose করব।
- But কোন member এর  $T_p$  এর effect দেয়া থাকলে  
 তাই Redundent choose করব না।

৩-৩ উত্তর.

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(A) cont.....

• u সাই BMD কে ২বার integrate করে।

~~অথবা~~ • u কে differentiate করে slope b পাওয়া যায়।

সি।  $0 \leq x \leq L$

•  $u_0 = 1$

•  $u_1 = \frac{1}{EI} \left( L \frac{x^2}{2} - \frac{x^3}{6} \right)$

সি।  $L \leq x \leq 2L$

•  $u_0 = 1$

•  $u_1 = \frac{L^3}{3EI} + \frac{L^2}{2EI} (x-L)$

$$\left\{ \begin{array}{l} \text{Hence, eqn of line} = a + bx' \\ a = \frac{L^3}{3EI} = \delta_{11} \\ \text{in general, } \frac{d^2u}{dx^2} = \frac{L-x}{EI} \\ \frac{du}{dx} = \frac{1}{EI} \left( Lx - \frac{x^2}{2} \right) \\ \therefore \theta(L) = \frac{L^2}{2EI} \text{ (slope)} = b \\ \therefore b = \frac{L^2}{2EI} \\ x' = (x-L) \end{array} \right.$$

•  $u_2 = \frac{1}{EI} \left( Lx^2 - \frac{x^3}{6} \right)$

$\therefore \text{II } R_A = 1 + \frac{1}{EI} \left( L \frac{x^2}{2} - \frac{x^3}{6} \right)$

$$\begin{aligned} & * \left( -\frac{66}{7} \frac{EI}{L^3} \right) + \frac{1}{EI} \left( Lx^2 - \frac{x^3}{6} \right) x \\ & \left( \frac{18}{7} \frac{EI}{L^3} \right) \end{aligned}$$

সি।  $L \leq x \leq L$

•  $u_2 = \frac{1}{EI} \left( Lx^2 - \frac{x^3}{6} \right)$

সি। ক্রম সংখ্যা

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• স্ল্যাবের IL আকতে চাই সোড়িকে unit deformation দিচ্ছি।

\* যেহেতু এখানে Reaction হলো সমতরময়ই concentrated load বা moment হবে, তাই BMD সমতরময়ই constant বা st. line হবে।