



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

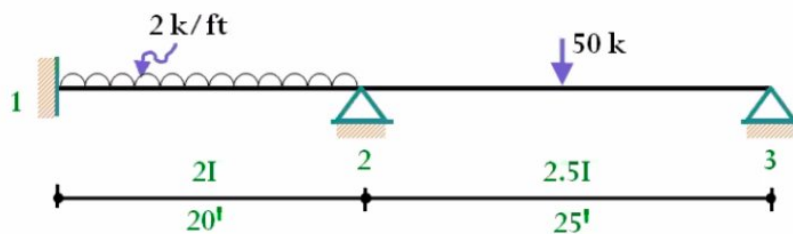
Talking: Dr. Md. Shafiqul ISLAM

## PROBLEM

01

Using Stiffness Matrix Method solve the followings

- Determine the joint deflections
- Determine the joint moments
- Draw Bending Moment Diagram



2



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

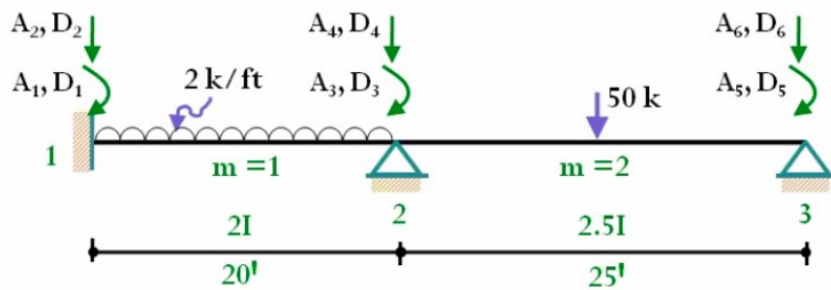
Talking: Dr. Md. Shafiqul ISLAM

## PROBLEM

01

Using Stiffness Matrix Method solve the followings

- Determine the joint deflections
- Determine the joint moments
- Draw Bending Moment Diagram



Total DoF

6

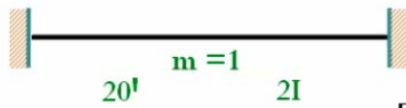
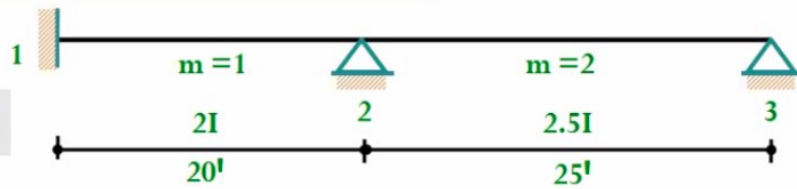


# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

No Applied Loads



Local Stiffness Matrix 1

$$s = \begin{bmatrix} \frac{4EI}{L} & \frac{6EI}{L^2} & \frac{2EI}{L} & -\frac{6EI}{L^2} \\ \frac{6EI}{L^2} & \frac{12EI}{L^3} & \frac{6EI}{L^2} & -\frac{12EI}{L^3} \\ \frac{2EI}{L} & \frac{6EI}{L^2} & \frac{4EI}{L} & -\frac{6EI}{L^2} \\ -\frac{6EI}{L^2} & -\frac{12EI}{L^3} & -\frac{6EI}{L^2} & \frac{12EI}{L^3} \end{bmatrix}$$

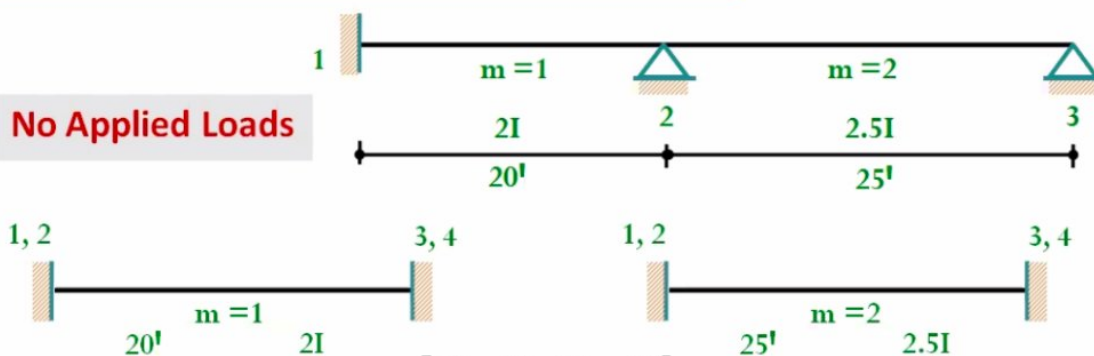


# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

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No Applied Loads



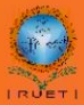
Local Stiffness Matrix 1

$$s = \begin{bmatrix} \frac{4EI}{L} & \frac{6EI}{L^2} & \frac{2EI}{L} & -\frac{6EI}{L^2} \\ \frac{6EI}{L^2} & \frac{12EI}{L^3} & \frac{6EI}{L^2} & -\frac{12EI}{L^3} \\ \frac{2EI}{L} & \frac{6EI}{L^2} & \frac{4EI}{L} & -\frac{6EI}{L^2} \\ -\frac{6EI}{L^2} & -\frac{12EI}{L^3} & -\frac{6EI}{L^2} & \frac{12EI}{L^3} \end{bmatrix}$$

Local Stiffness Matrix 2

$$S_1 = EI \begin{bmatrix} 0.4 & 0.03 & 0.2 & -0.03 \\ 0.03 & 0.003 & 0.03 & -0.003 \\ 0.2 & 0.03 & 0.4 & -0.03 \\ -0.03 & -0.003 & -0.03 & 0.003 \end{bmatrix}$$

$$S_2 = EI \begin{bmatrix} 0.4 & 0.024 & 0.2 & -0.024 \\ 0.024 & 0.00192 & 0.024 & -0.00192 \\ 0.2 & 0.024 & 0.4 & -0.024 \\ -0.024 & -0.00192 & -0.024 & 0.00192 \end{bmatrix}$$



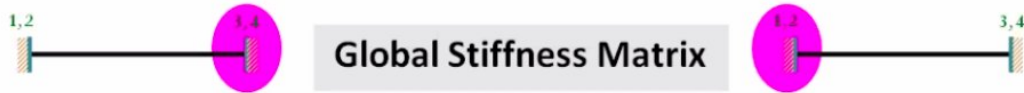
# Stiffness Matrix-BEAM

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$$S_1 = EI \begin{bmatrix} 0.4 & 0.03 & 0.2 & -0.03 \\ 0.03 & 0.003 & 0.03 & -0.003 \\ 0.2 & 0.03 & 0.4 & -0.03 \\ -0.03 & -0.003 & -0.03 & 0.003 \end{bmatrix}$$

$$S_2 = EI \begin{bmatrix} 0.4 & 0.024 & 0.2 & -0.024 \\ 0.024 & 0.00192 & 0.024 & -0.00192 \\ 0.2 & 0.024 & 0.4 & -0.024 \\ -0.024 & -0.00192 & -0.024 & 0.00192 \end{bmatrix}$$



Global Stiffness Matrix

$$S = EI \begin{bmatrix} 0.4 & 0.03 & 0.2 & -0.03 & 0 & 0 \\ 0.03 & 0.003 & 0.03 & -0.03 & 0 & 0 \\ 0.2 & 0.03 & (0.4+0.4) & (-0.03+0.024) & 0.2 & -0.024 \\ -0.03 & -0.003 & (-0.03+0.024) & (0.003+0.00192) & 0.024 & -0.00192 \\ 0 & 0 & 0.2 & 0.024 & 0.4 & -0.024 \\ 0 & 0 & -0.024 & -0.00192 & -0.024 & 0.00192 \end{bmatrix}$$



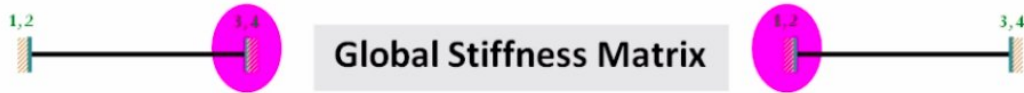
# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

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$$S_1 = EI \begin{bmatrix} 0.4 & 0.03 & 0.2 & -0.03 \\ 0.03 & 0.003 & 0.03 & -0.003 \\ 0.2 & 0.03 & 0.4 & -0.03 \\ -0.03 & -0.003 & -0.03 & 0.003 \end{bmatrix}$$

$$S_2 = EI \begin{bmatrix} 0.4 & 0.024 & 0.2 & -0.024 \\ 0.024 & 0.00192 & 0.024 & -0.00192 \\ 0.2 & 0.024 & 0.4 & -0.024 \\ -0.024 & -0.00192 & -0.024 & 0.00192 \end{bmatrix}$$



Global Stiffness Matrix

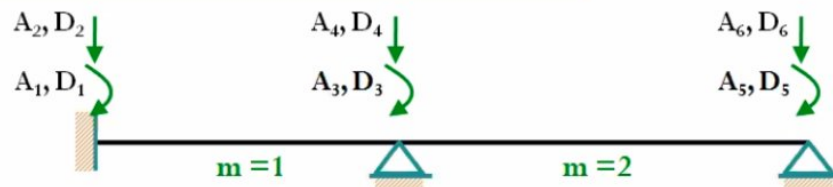
$$S = EI \begin{bmatrix} 0.4 & 0.03 & 0.2 & -0.03 & 0 & 0 \\ 0.03 & 0.003 & 0.03 & -0.03 & 0 & 0 \\ 0.2 & 0.03 & (0.4+0.4) & (-0.03+0.024) & 0.2 & -0.024 \\ -0.03 & -0.003 & (-0.03+0.024) & (0.003+0.00192) & 0.024 & -0.00192 \\ 0 & 0 & 0.2 & 0.024 & 0.4 & -0.024 \\ 0 & 0 & -0.024 & -0.00192 & -0.024 & 0.00192 \end{bmatrix} \begin{matrix} 1 \\ 2 \\ 3+1 \\ 4+2 \\ 3 \\ 4 \end{matrix}$$



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

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Restrained DoF

4

1, 2, 4, 6

$$S = EI \begin{bmatrix} 1 & 2 & 3 & 4 & 5 & 6 \\ 0.4 & 0.03 & 0.2 & -0.03 & 0 & 0 \\ 0.03 & 0.003 & 0.03 & -0.03 & 0 & 0 \\ 0.2 & 0.03 & (0.4+0.4) & (-0.03+0.024) & 0.2 & -0.024 \\ -0.03 & -0.003 & (-0.03+0.024) & (0.003+0.00192) & 0.024 & -0.00192 \\ 0 & 0 & 0.2 & 0.024 & 0.4 & -0.024 \\ 0 & 0 & -0.024 & -0.00192 & -0.024 & 0.00192 \end{bmatrix} \begin{matrix} 1 \\ 2 \\ 3 \\ 4 \\ 5 \\ 6 \end{matrix}$$

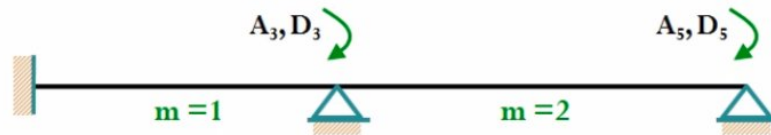
5



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

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Active DoF	2	3, 5
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$$S = EI \begin{bmatrix} 1 & 2 & 3 & 4 & 5 & 6 \\ 0.4 & 0.03 & 0.2 & -0.03 & 0 & 0 \\ 0.03 & 0.003 & 0.03 & -0.03 & 0 & 0 \\ 0.2 & 0.03 & (0.4+0.4) & (-0.03+0.024) & 0.2 & -0.024 \\ -0.03 & -0.003 & (-0.03+0.024) & (0.003+0.00192) & 0.024 & -0.00192 \\ 0 & 0 & 0.2 & 0.024 & 0.4 & -0.024 \\ 0 & 0 & -0.024 & -0.00192 & -0.024 & 0.00192 \end{bmatrix} \begin{matrix} 1 \\ 2 \\ 3 \\ 4 \\ 5 \\ 6 \end{matrix}$$

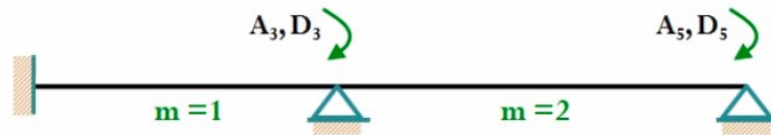
5



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

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Active DoF                      2                      3, 5

$$S = EI \begin{bmatrix} & & & & & \\ & & & & & \\ & & 0.8 & & 0.2 & \\ & & & & & \\ & & 0.2 & & 0.4 & \\ & & & & & \end{bmatrix} \begin{matrix} 1 \\ 2 \\ 3 \\ 4 \\ 5 \\ 6 \end{matrix}$$



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

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Global Stiffness Matrix

$$S = EI \begin{bmatrix} 0.8 & 0.2 \\ 0.2 & 0.4 \end{bmatrix}$$

$$\begin{bmatrix} A_3 \\ A_5 \end{bmatrix} = EI \begin{bmatrix} 0.8 & 0.2 \\ 0.2 & 0.4 \end{bmatrix} \begin{bmatrix} D_3 \\ D_5 \end{bmatrix}$$

Now Load Matrix **[A]** ??????



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

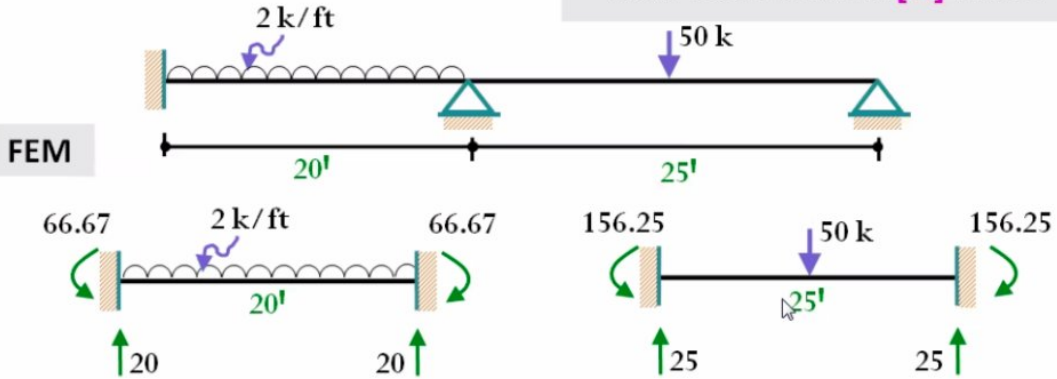
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Now Load Matrix [A] ??????





# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

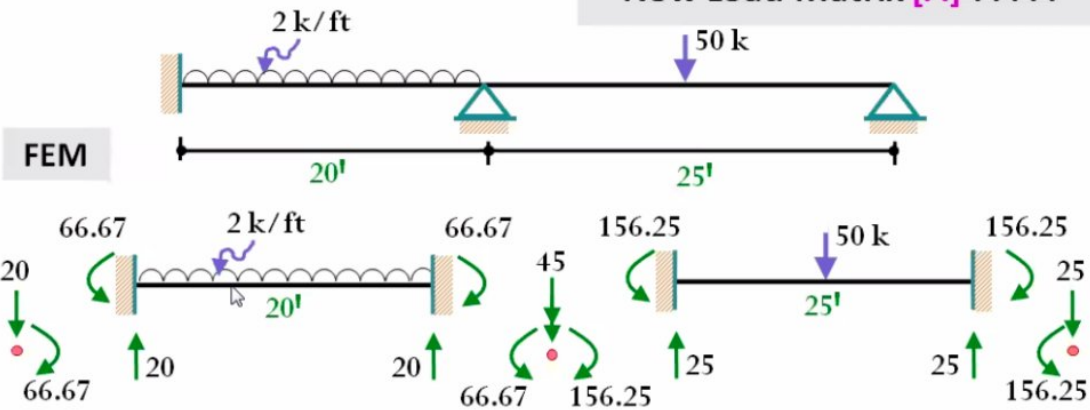
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Global Stiffness Matrix

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Now Load Matrix [A] ??????



EQUIVALENT LOAD TRANSFER



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

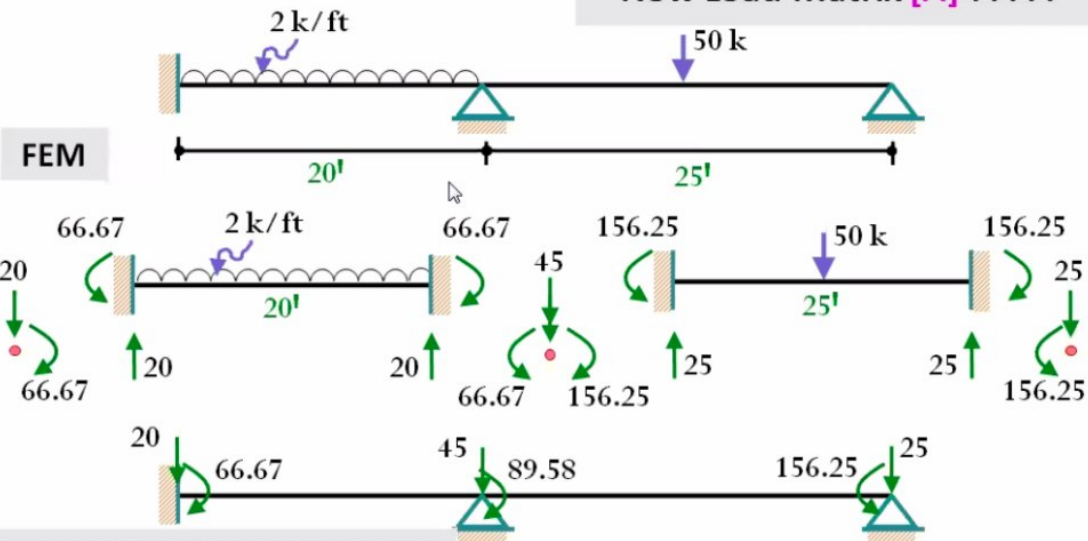
Talking: Dr. Md. Shafiqul ISLAM

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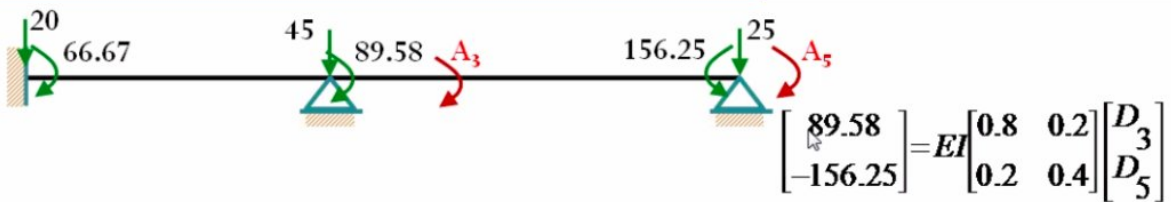
EQUIVALENT LOAD TRANSFER



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

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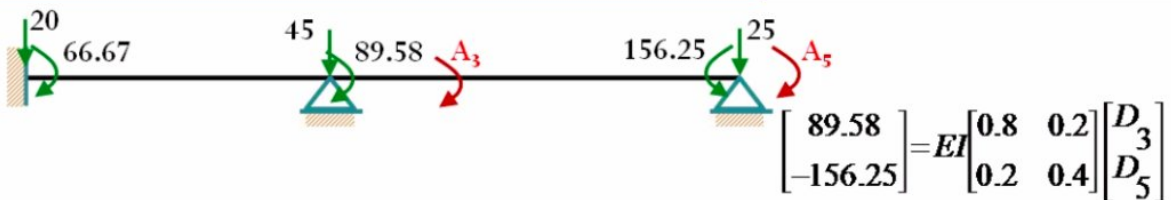




# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking:



$$\begin{bmatrix} D_3 \\ D_5 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} 0.8 & 0.2 \\ 0.2 & 0.4 \end{bmatrix}^{-1} \begin{bmatrix} 89.58 \\ -156.25 \end{bmatrix}$$

Determinant **0.28**

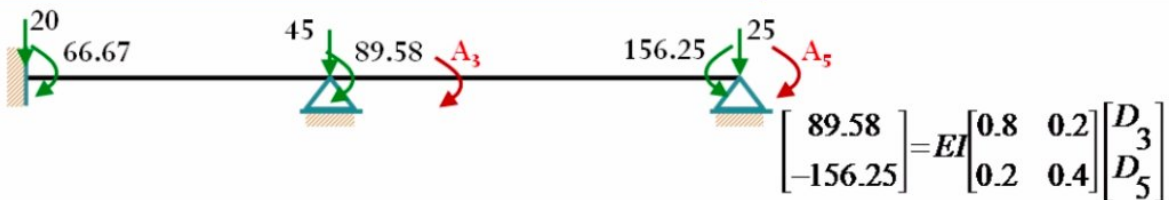
$$\begin{bmatrix} D_3 \\ D_5 \end{bmatrix} = \frac{1}{0.28EI} \begin{bmatrix} 0.4 & -0.2 \\ -0.2 & 0.8 \end{bmatrix} \begin{bmatrix} 89.58 \\ -156.25 \end{bmatrix}$$



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM



$$\begin{bmatrix} 89.58 \\ -156.25 \end{bmatrix} = EI \begin{bmatrix} 0.8 & 0.2 \\ 0.2 & 0.4 \end{bmatrix} \begin{bmatrix} D_3 \\ D_5 \end{bmatrix}$$

$$\begin{bmatrix} D_3 \\ D_5 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} 0.8 & 0.2 \\ 0.2 & 0.4 \end{bmatrix}^{-1} \begin{bmatrix} 89.58 \\ -156.25 \end{bmatrix}$$

Determinant **0.28**

$$\begin{bmatrix} D_3 \\ D_5 \end{bmatrix} = \frac{1}{0.28EI} \begin{bmatrix} 0.4 & -0.2 \\ -0.2 & 0.8 \end{bmatrix} \begin{bmatrix} 89.58 \\ -156.25 \end{bmatrix}$$

$$\begin{bmatrix} D_3 \\ D_5 \end{bmatrix} = \begin{bmatrix} \frac{0.4 \times 89.58 + (-0.2 \times -156.25)}{0.28EI} \\ \frac{-0.2 \times 89.58 + 0.8 \times -156.25}{0.28EI} \end{bmatrix}$$

Recording

You are viewing Dr. Md. Shafiqul ...'s screen

View



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$M = FEM + S \times D$$

8

Unmute Start Video

Participants 100

Chat

Share Screen

Record

Reactions

Leave

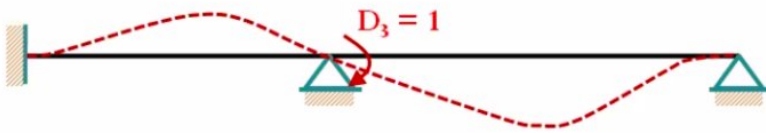


# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$M = FEM + S \times D$$



To Find out S Apply D

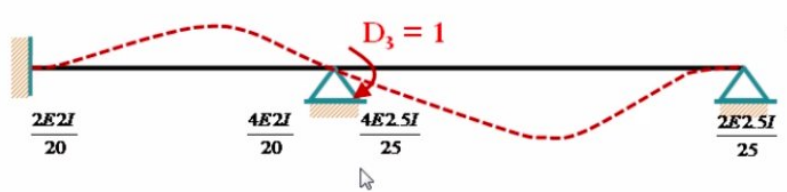


# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

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$$M = FEM + S \times D$$



To Find out S Apply D

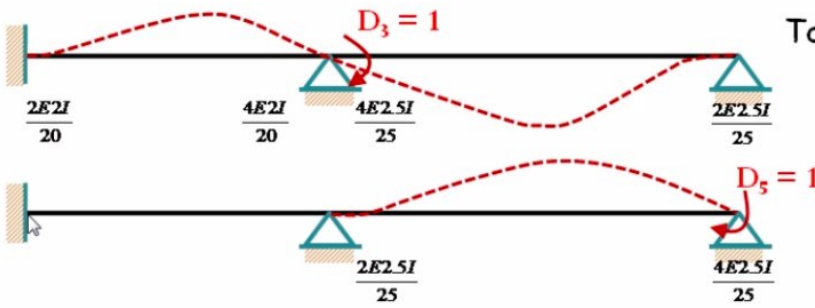


# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$M = FEM + S \times D$$



To Find out **S** Apply **D**

$$\begin{bmatrix} M_{12} \\ M_{21} \\ M_{23} \\ M_{32} \end{bmatrix} = \begin{bmatrix} -66.67 \\ 66.67 \\ -156.25 \\ 156.25 \end{bmatrix} + EI \begin{bmatrix} 0.2 & 0 \\ 0.4 & 0 \\ 0.4 & 0.2 \\ 0.2 & 0.4 \end{bmatrix} \frac{1}{EI} \begin{bmatrix} 239.58 \\ -510.41 \end{bmatrix}$$

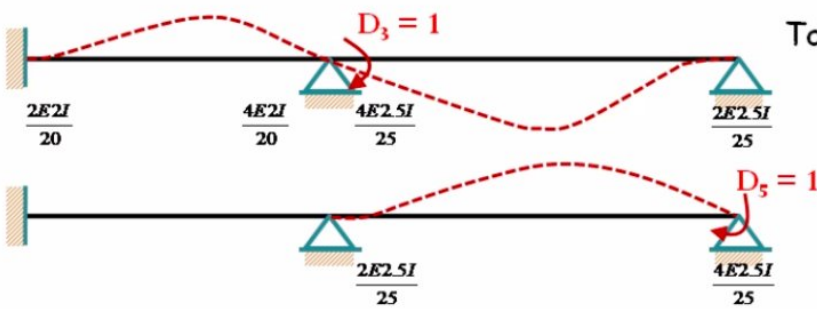


# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

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$$M = FEM + S \times D$$



To Find out **S** Apply **D**

$$\begin{bmatrix} M_{12} \\ M_{21} \\ M_{23} \\ M_{32} \end{bmatrix} = \begin{bmatrix} -66.67 \\ 66.67 \\ -156.25 \\ 156.25 \end{bmatrix} + EI \begin{bmatrix} 0.2 & 0 \\ 0.4 & 0 \\ 0.4 & 0.2 \\ 0.2 & 0.4 \end{bmatrix} \frac{1}{EI} \begin{bmatrix} 239.58 \\ -510.41 \end{bmatrix}$$

$$\begin{bmatrix} M_{12} \\ M_{21} \\ M_{23} \\ M_{32} \end{bmatrix} = \begin{bmatrix} -18.76 \\ 162.50 \\ -162.50 \\ 0 \end{bmatrix}$$

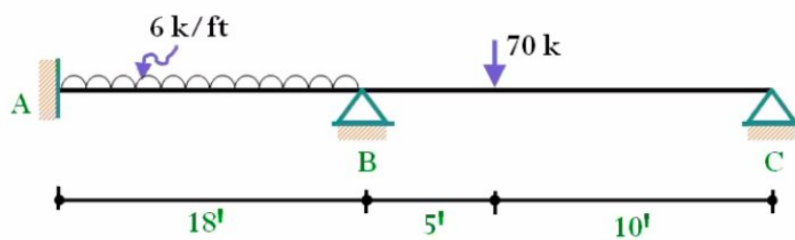


## PROBLEM

02

Using Stiffness Matrix Method solve the followings, EI is constant

- Determine the joint deflections
- Determine the joint moments
- Draw Bending Moment Diagram



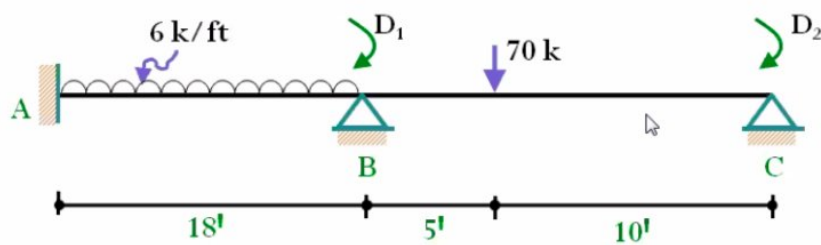


## PROBLEM

02

Using Stiffness Matrix Method solve the followings, EI is constant

- a) Determine the joint deflections
- b) Determine the joint moments
- c) Draw Bending Moment Diagram



Active DoF

2

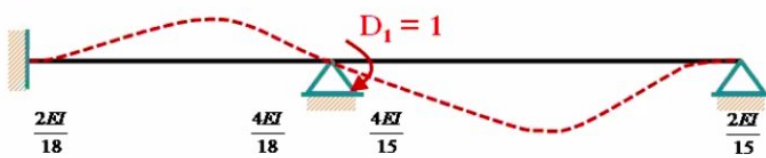


# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$D_1 = 1, D_2 = 0$$



At Position 1 Effect of 1

$$S_{11} = \frac{4EI}{18} + \frac{4EI}{15}$$

$$S_{11} = \frac{20EI + 24EI}{90}$$

$$S_{11} = \frac{22EI}{45}$$

At Position 2 Effect of 1

$$S_{21} = \frac{2EI}{15}$$

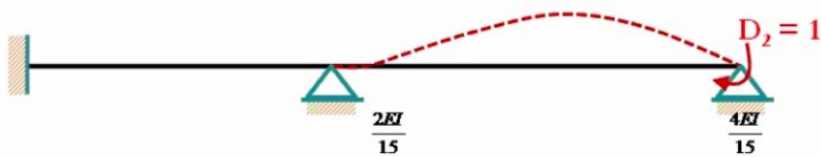


# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

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$$D_2 = 1, D_1 = 0$$



At Position 1 Effect of 2

$$S_{12} = \frac{2EI}{15}$$

At Position 2 Effect of 2

$$S_{22} = \frac{4EI}{15}$$



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

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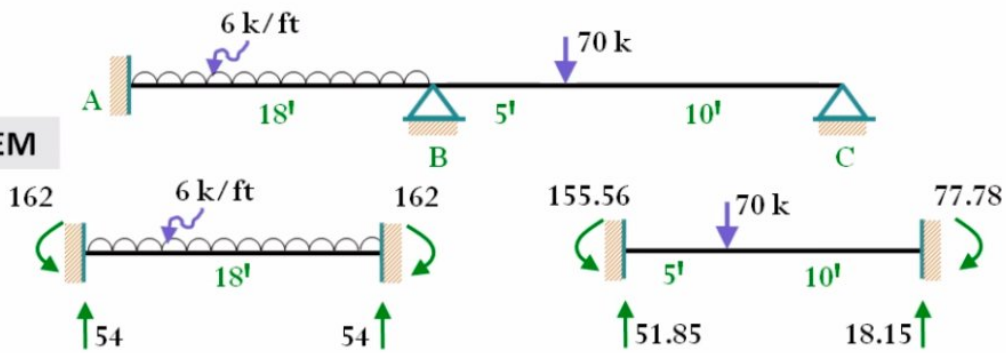
Global Stiffness Matrix

$$S = EI \begin{bmatrix} \frac{22}{15} & \frac{2}{15} \\ \frac{45}{2} & \frac{15}{4} \\ \frac{2}{15} & \frac{15}{15} \end{bmatrix}$$

$$\begin{bmatrix} A_1 \\ A_2 \end{bmatrix} = EI \begin{bmatrix} \frac{22}{15} & \frac{2}{15} \\ \frac{45}{2} & \frac{15}{4} \\ \frac{2}{15} & \frac{15}{15} \end{bmatrix} \begin{bmatrix} D_1 \\ D_2 \end{bmatrix}$$

Now Load Matrix [A] ?????

FEM





# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

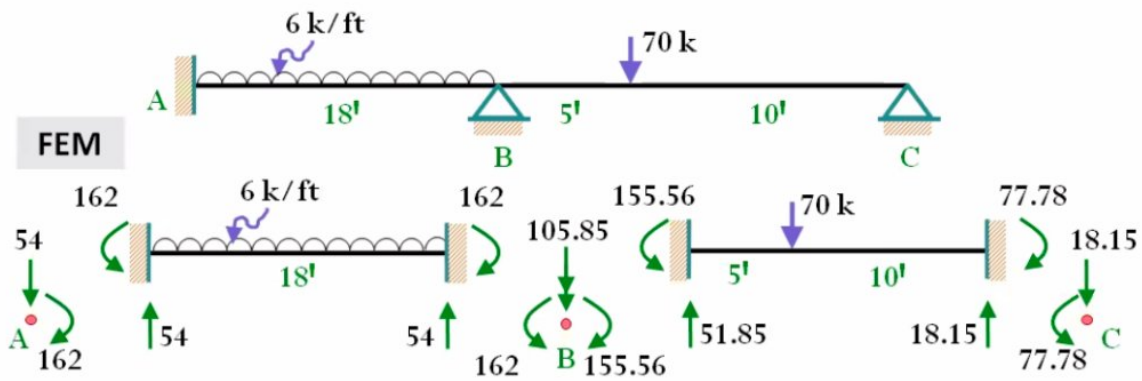
Talking: Dr. Md. Shafiqul ISLAM

Global Stiffness Matrix

$$S = EI \begin{bmatrix} \frac{22}{15} & \frac{2}{15} \\ \frac{45}{2} & \frac{15}{4} \\ \frac{2}{15} & \frac{15}{15} \end{bmatrix}$$

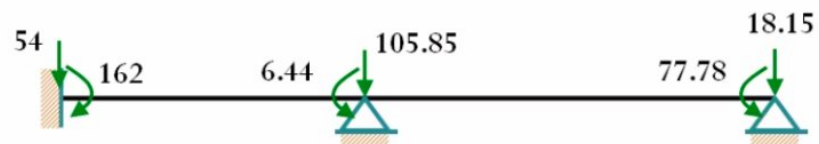
$$\begin{bmatrix} A_1 \\ A_2 \end{bmatrix} = EI \begin{bmatrix} \frac{22}{15} & \frac{2}{15} \\ \frac{45}{2} & \frac{15}{4} \\ \frac{2}{15} & \frac{15}{15} \end{bmatrix} \begin{bmatrix} D_1 \\ D_2 \end{bmatrix}$$

Now Load Matrix [A] ?????





## EQUIVALENT LOAD TRANSFER





## Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

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$$\begin{bmatrix} D_1 \\ D_2 \end{bmatrix} = \frac{45}{EI} \begin{bmatrix} 22 & 6 \\ 6 & 12 \end{bmatrix}^{-1} \begin{bmatrix} -6.44 \\ -77.78 \end{bmatrix}$$

$$\begin{bmatrix} D_1 \\ D_2 \end{bmatrix} = \frac{45}{228EI} \begin{bmatrix} 12 & -6 \\ -6 & 22 \end{bmatrix} \begin{bmatrix} -6.44 \\ -77.78 \end{bmatrix}$$

Determinant **228**

$$\begin{bmatrix} D_1 \\ D_2 \end{bmatrix} = \frac{45}{EI \times 228} \begin{bmatrix} -6.44 \times 12 + 6 \times 77.78 \\ 6.44 \times 6 + 22 \times (-77.78) \end{bmatrix}$$

$$\begin{bmatrix} D_1 \\ D_2 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} 76.86 \\ -330.10 \end{bmatrix}$$



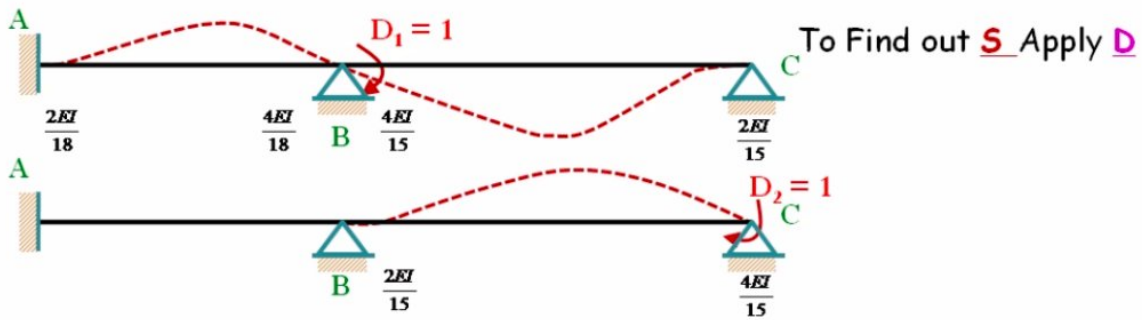


# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$M = FEM + S \times D$$



To Find out **S** Apply **D**

$$\begin{matrix}
 M_{AB} \\
 M_{BA} \\
 M_{BC} \\
 M_{CB}
 \end{matrix}
 =
 \begin{bmatrix}
 -162 \\
 162 \\
 -155.56 \\
 77.78
 \end{bmatrix}
 + EI
 \begin{matrix}
 D_1=1 & D_2=1 \\
 \begin{bmatrix}
 2/18 & 0 \\
 4/18 & 0 \\
 4/15 & 2/15 \\
 2/15 & 4/15
 \end{bmatrix}
 \frac{1}{EI}
 \begin{bmatrix}
 76.86 \\
 -330.10
 \end{bmatrix}
 \end{matrix}
 \qquad
 \begin{matrix}
 M_{AB} \\
 M_{BA} \\
 M_{BC} \\
 M_{CB}
 \end{matrix}
 =
 \begin{bmatrix}
 -153.46 \\
 179.08 \\
 -179.08 \\
 0
 \end{bmatrix}$$

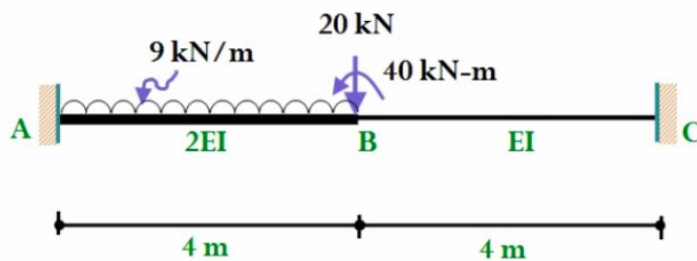


## PROBLEM

03

Using Stiffness Matrix Method solve the followings, EI is constant

- Determine the deflection and rotation at B
- Determine all the joint reactions
- Draw Shear Force and Bending Moment Diagrams

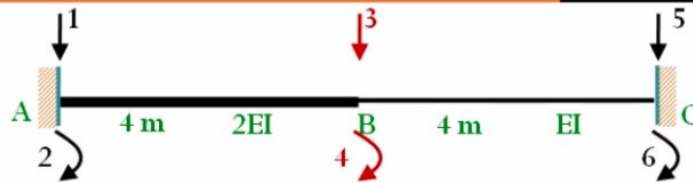




# Stiffness Matrix-BEAM

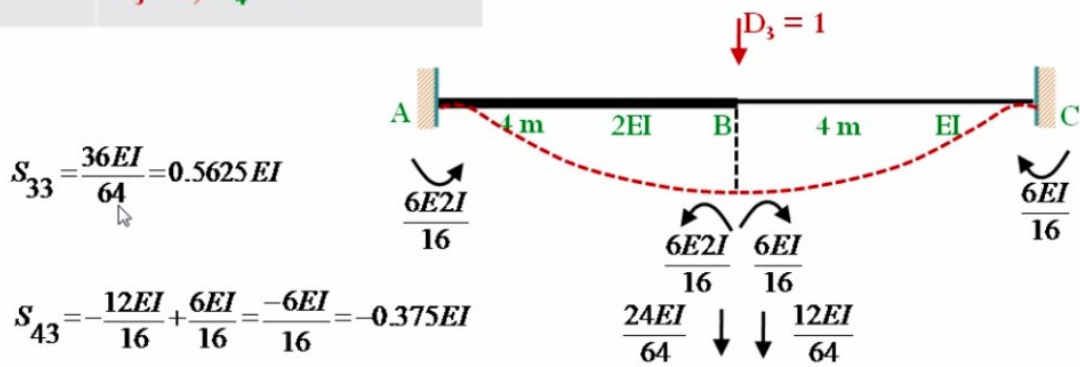
9<sup>th</sup> May, 2020

Talking:



Total DoF	6
Active DoF	2

$D_3 = 1, D_4 = 0$



$$S_{33} = \frac{36EI}{64} = 0.5625EI$$

$$S_{43} = -\frac{12EI}{16} + \frac{6EI}{16} = \frac{-6EI}{16} = -0.375EI$$



# Stiffness Matrix-BEAM

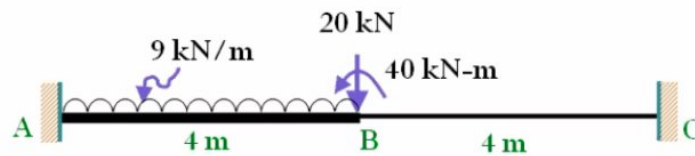
9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

Global Stiffness Matrix

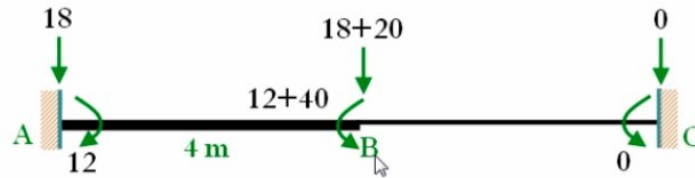
$$S = EI \begin{bmatrix} 0.5625 & -0.375 \\ -0.375 & 3 \end{bmatrix}$$

FEM



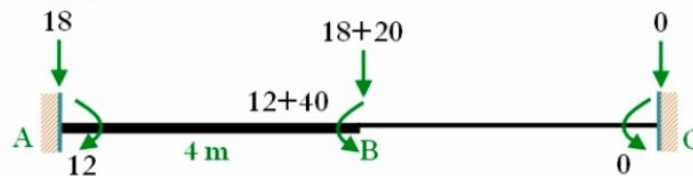


## EQUIVALENT LOAD TRANSFER





## EQUIVALENT LOAD TRANSFER



$$\begin{bmatrix} 38 \\ -52 \end{bmatrix} = EI \begin{bmatrix} 0.5625 & -0.375 \\ -0.375 & 3 \end{bmatrix} \begin{bmatrix} D_3 \\ D_4 \end{bmatrix}$$

$$\begin{bmatrix} D_3 \\ D_4 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} 0.5625 & -0.375 \\ -0.375 & 3 \end{bmatrix}^{-1} \begin{bmatrix} 38 \\ -52 \end{bmatrix}$$

$$\begin{bmatrix} D_3 \\ D_4 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} 61.09 \\ -9.70 \end{bmatrix}$$

$$\begin{bmatrix} D_3 \\ D_4 \end{bmatrix} = \frac{1}{1.547EI} \begin{bmatrix} 3 & 0.375 \\ 0.375 & 0.5625 \end{bmatrix} \begin{bmatrix} 38 \\ -52 \end{bmatrix}$$

Determinant **1.547**



## Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$M = FEM + S \times D$$

$$V = FEV + S \times D$$

Find out **Local**  
**Stiffness Matrix**



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

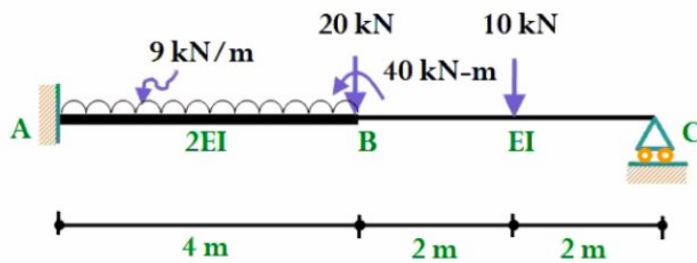
Talking: Dr. Md. Shafiqul ISLAM

## PROBLEM

04

Using Stiffness Matrix Method solve the followings, EI is constant

- Determine the deflection, rotation at B and rotation at C
- Determine all the joint reactions
- Draw Shear Force and Bending Moment Diagrams



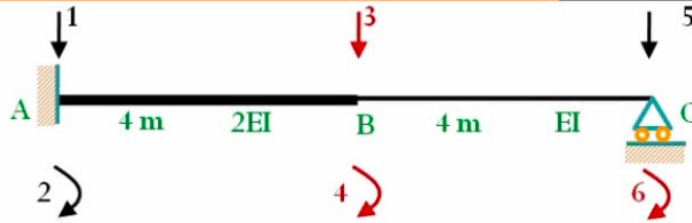
2



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

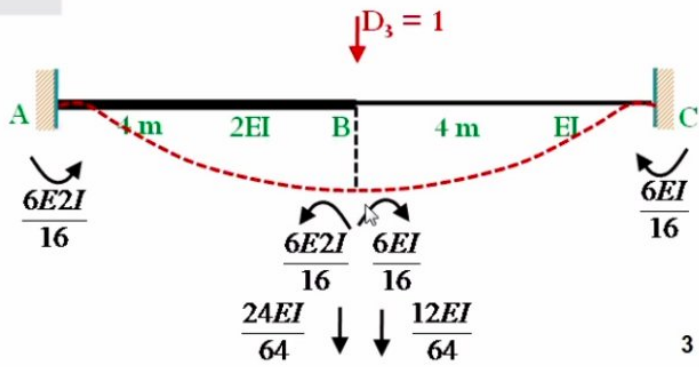
Talking: Dr. Md. Shafiqul ISLAM



Total DoF	6	Active DoF	3
$D_3 = 1, D_4 = 0, D_6 = 0$			

$$S_{33} = \frac{36EI}{64} = 0.5625EI$$

$$S_{43} = -\frac{12EI}{16} + \frac{6EI}{16} = \frac{-6EI}{16} = -0.375EI$$



3

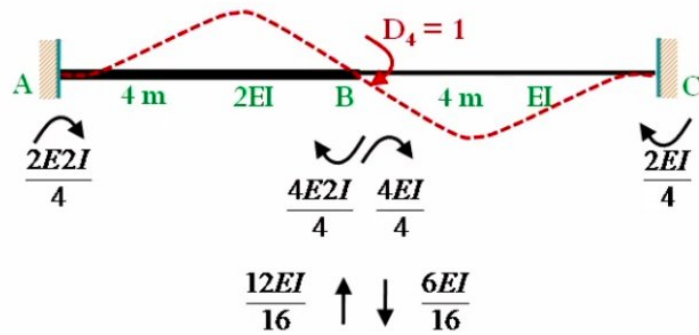


# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$D_4 = 1, D_3 = 0, D_6 = 0$$





# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

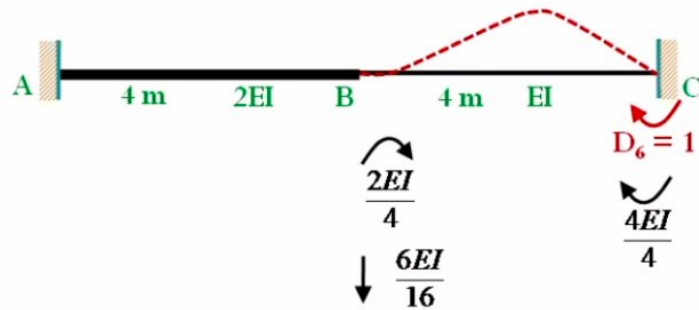
Talking:

$$D_6 = 1, D_3 = 0, D_4 = 0$$

$$S_{36} = 0.375EI$$

$$S_{46} = 0.5EI$$

$$S_{66} = 1EI$$





# Stiffness Matrix-BEAM

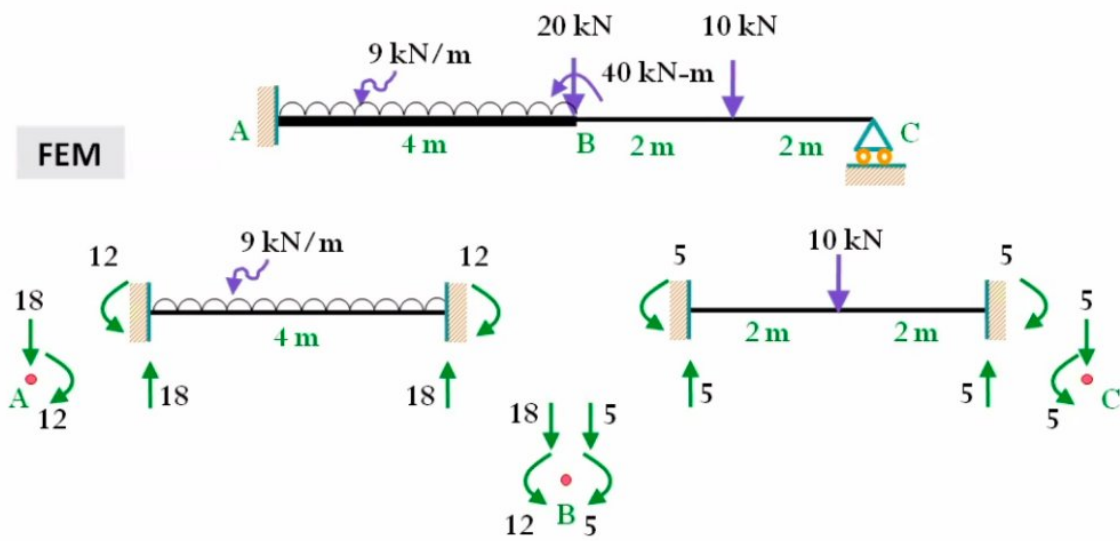
9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

Global Stiffness Matrix

$$S = EI \begin{bmatrix} 0.5625 & -0.375 & 0.375 \\ -0.375 & 3 & 0.5 \\ 0.375 & 0.5 & 1 \end{bmatrix}$$

FEM



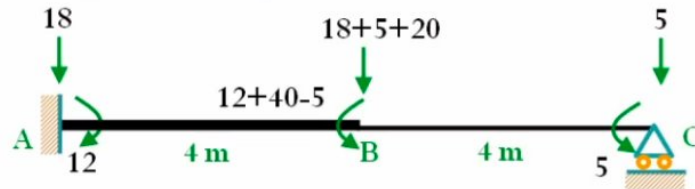


# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## EQUIVALENT LOAD TRANSFER

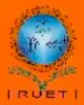


$$\begin{bmatrix} 43 \\ -47 \\ -5 \end{bmatrix} = EI \begin{bmatrix} 0.5625 & -0.375 & 0.375 \\ -0.375 & 3 & 0.5 \\ 0.375 & 0.5 & 1 \end{bmatrix} \begin{bmatrix} D_3 \\ D_4 \\ D_6 \end{bmatrix}$$

$$A^{-1} = \frac{adjA}{|A|}$$

$$\begin{bmatrix} D_3 \\ D_4 \\ D_6 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} 0.5625 & -0.375 & 0.375 \\ -0.375 & 3 & 0.5 \\ 0.375 & 0.5 & 1 \end{bmatrix}^{-1} \begin{bmatrix} 43 \\ -47 \\ -5 \end{bmatrix}$$

$$\begin{bmatrix} D_3 \\ D_4 \\ D_6 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} 116.59 \\ 7.67 \\ -52.56 \end{bmatrix}$$



# Stiffness Matrix-BEAM

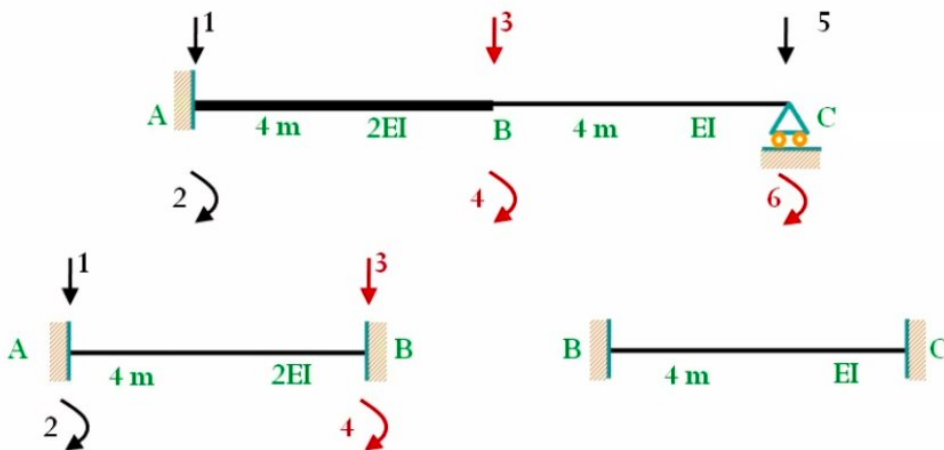
9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$M = FEM + S \times D$$

$$V = FEV + S \times D$$

Find out **Local Stiffness Matrix**





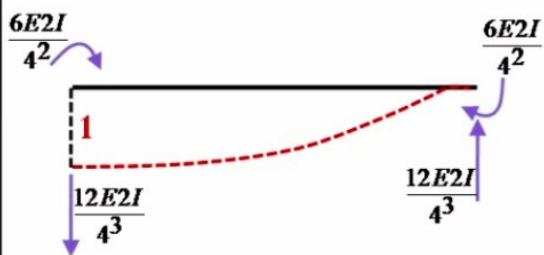
# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Local Stiffness Matrix for member AS

$$D_1 = 1, D_2 = D_3 = D_4 = 0$$



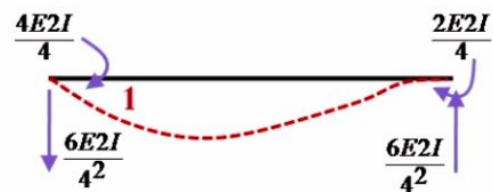
$$S_{11} = -0.375EI$$

$$S_{21} = 0.75EI$$

$$S_{31} = -0.375EI$$

$$S_{41} = 0.75EI$$

$$D_2 = 1, D_1 = D_3 = D_4 = 0$$



$$S_{12} = -0.75EI$$

$$S_{22} = 2EI$$

$$S_{32} = -0.75EI$$

$$S_{42} = 1EI$$



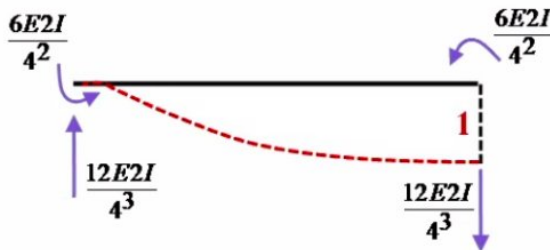
# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Local Stiffness Matrix for member AB

$$D_3 = 1, D_1 = D_2 = D_4 = 0$$



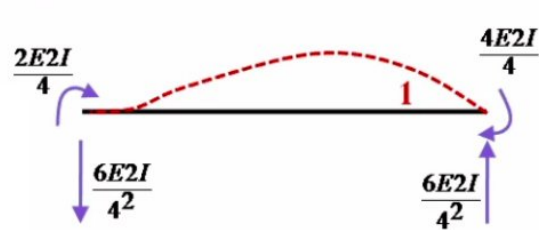
$$S_{13} = -0.375EI$$

$$S_{23} = -0.75EI$$

$$S_{33} = 0.375EI$$

$$S_{43} = -0.75EI$$

$$D_4 = 1, D_1 = D_2 = D_3 = 0$$



$$S_{14} = 0.75EI$$

$$S_{24} = 1EI$$

$$S_{34} = -0.75EI$$

$$S_{44} = 2EI$$

10





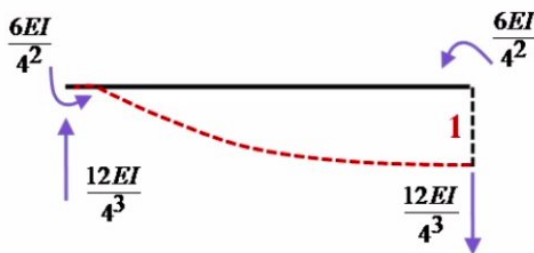
# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Local Stiffness Matrix for member BC

$$D_5 = 1, D_3 = D_4 = D_6 = 0$$



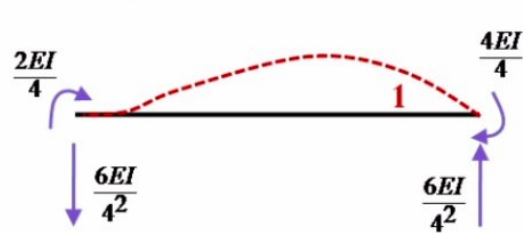
$$S_{35} = -0.1875EI$$

$$S_{45} = -0.375EI$$

$$S_{55} = 0.1875EI$$

$$S_{65} = -0.375EI$$

$$D_6 = 1, D_3 = D_4 = D_5 = 0$$



$$S_{36} = 0.375EI$$

$$S_{46} = 0.5EI$$

$$S_{56} = -0.375EI$$

$$S_{66} = 1EI$$

12



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Local Stiffness Matrix

Member

AB

$$S_{AB} = EI \begin{bmatrix} 0.375 & 0.75 & -0.375 & 0.75 \\ 0.75 & 2 & -0.75 & 1 \\ -0.375 & -0.75 & 0.375 & -0.75 \\ 0.75 & 1 & -0.75 & 2 \end{bmatrix}$$

Member

BC

$$S_{BC} = EI \begin{bmatrix} 0.1875 & 0.375 & -0.1875 & 0.375 \\ 0.375 & 1 & -0.375 & 0.5 \\ -0.1875 & -0.375 & 0.1875 & -0.375 \\ 0.375 & 0.5 & -0.375 & 1 \end{bmatrix}$$

13



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Joint Shear and Moment

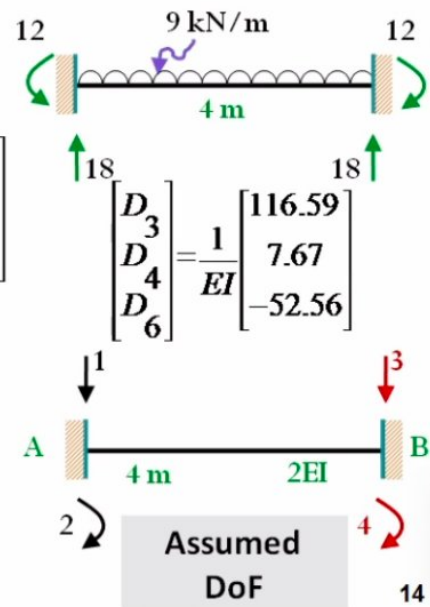
Member

AB

FEM

$$\begin{bmatrix} F_A \\ M_{AB} \\ F_{BL} \\ M_{BA} \end{bmatrix} = \begin{bmatrix} -18 \\ -12 \\ -18 \\ 12 \end{bmatrix} + EI \begin{bmatrix} 0.375 & 0.75 & -0.375 & 0.75 \\ 0.75 & 2 & -0.75 & 1 \\ -0.375 & -0.75 & 0.375 & -0.75 \\ 0.75 & 1 & -0.75 & 2 \end{bmatrix} \frac{1}{EI} \begin{bmatrix} 0 \\ 0 \\ 116.59 \\ 7.67 \end{bmatrix}$$

$$\begin{bmatrix} F_A \\ M_{AB} \\ F_{BL} \\ M_{BA} \end{bmatrix} = \begin{bmatrix} -55.97 \\ -91.77 \\ 19.97 \\ -60.10 \end{bmatrix}$$



14



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

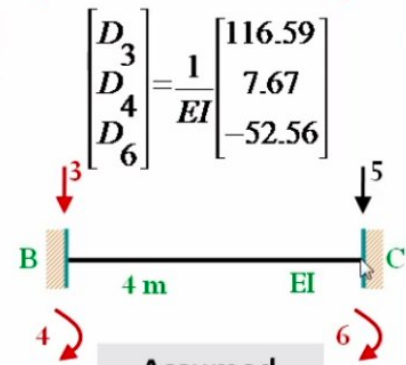
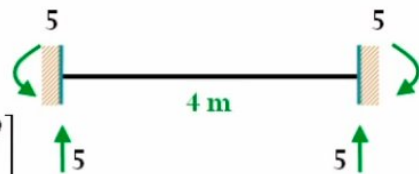
## Joint Shear and Moment

Member

BC

FEM

$$\begin{bmatrix} F_{BR} \\ M_{BC} \\ F_C \\ M_{CB} \end{bmatrix} = \begin{bmatrix} -5 \\ -5 \\ -5 \\ 5 \end{bmatrix} + EI \begin{bmatrix} 0.1875 & 0.375 & -0.1875 & 0.375 \\ 0.375 & 1 & -0.375 & 0.5 \\ -0.1875 & -0.375 & 0.1875 & -0.375 \\ 0.375 & 0.5 & -0.375 & 1 \end{bmatrix} \frac{1}{EI} \begin{bmatrix} 116.59 \\ 7.67 \\ 0 \\ -52.56 \end{bmatrix}$$



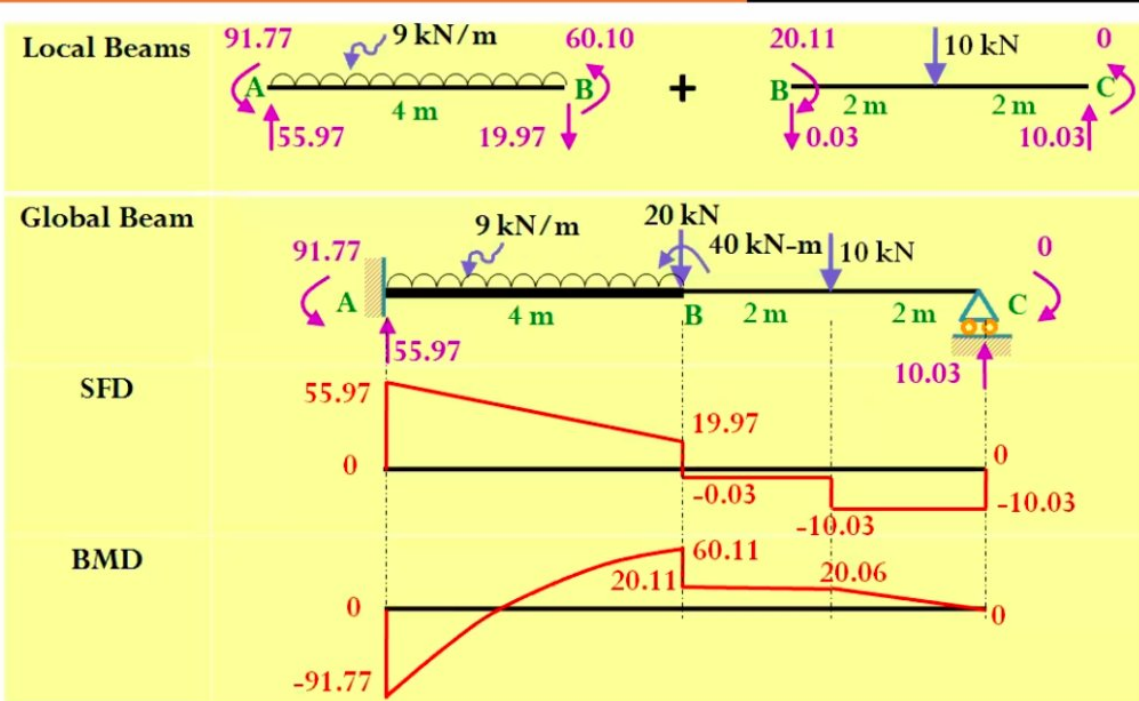
$$\begin{bmatrix} D_3 \\ D_4 \\ D_6 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} 116.59 \\ 7.67 \\ -52.56 \end{bmatrix}$$



# Stiffness Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM





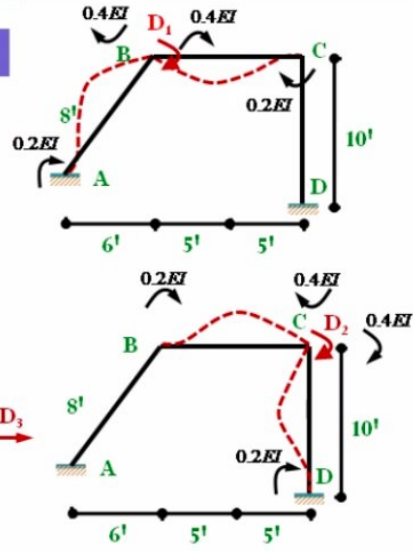
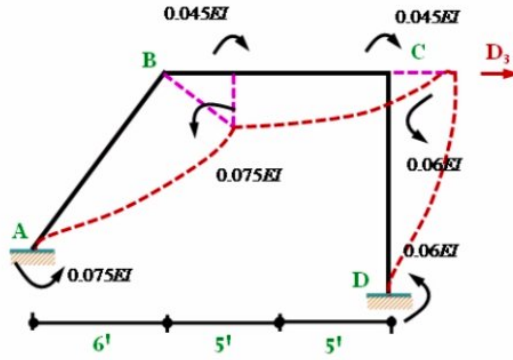
# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$M = FEM + S \times D$$

$$\begin{bmatrix} M_{AB} \\ M_{BA} \\ M_{BC} \\ M_{CB} \\ M_{CD} \\ M_{DC} \end{bmatrix} = \begin{bmatrix} -13.33 \\ 13.33 \\ -18.75 \\ 18.75 \\ 0 \\ 0 \end{bmatrix} + EI \begin{bmatrix} 0.2 & 0 & -0.075 \\ 0.4 & 0 & -0.075 \\ 0.4 & 0.2 & 0.045 \\ 0.2 & 0.4 & 0.045 \\ 0 & 0 & -0.06 \\ 0 & 0.2 & -0.06 \end{bmatrix} \frac{1}{EI} \begin{bmatrix} 28.55 \\ -22.51 \\ 430.64 \end{bmatrix}$$





# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

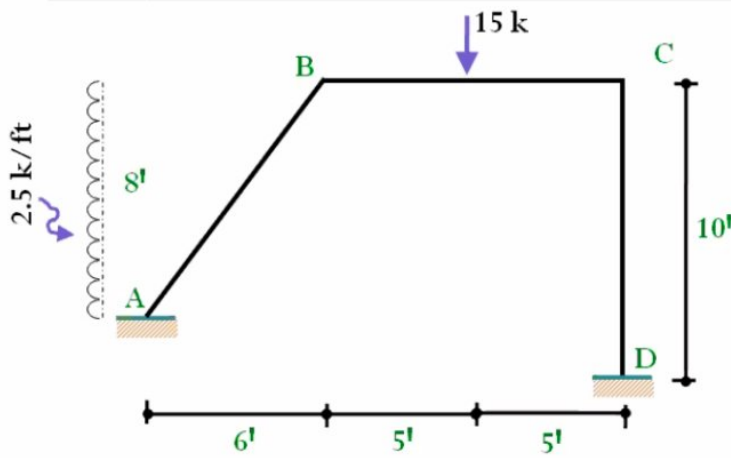
Talking: Dr. Md. Shafiqul ISLAM

## PROBLEM

05

Using Stiffness Matrix Method solve the followings, EI is constant

- Determine the deflection, rotation at joints
- Determine all the joint moments





# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

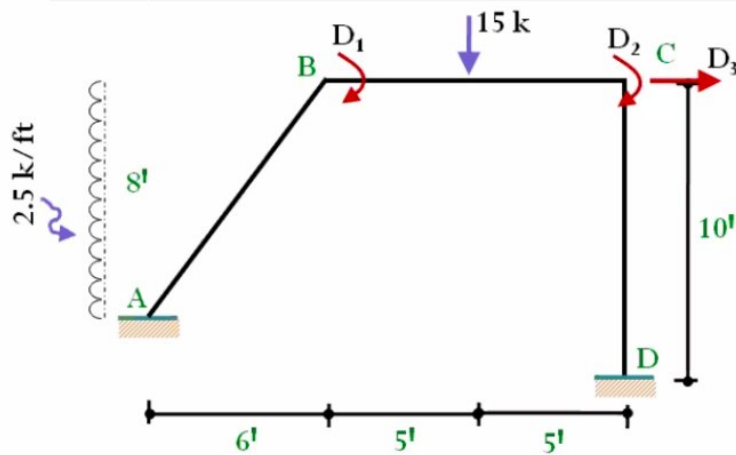
Talking: 1600083 Naim

## PROBLEM

05

Using Stiffness Matrix Method solve the followings, EI is constant

- Determine the deflection, rotation at joints
- Determine all the joint moments



Active DoF 3

2



# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

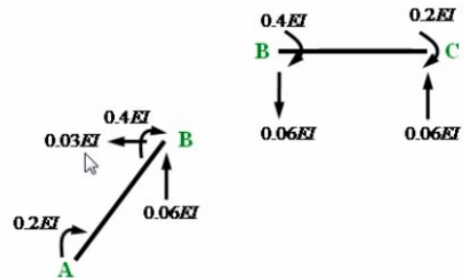
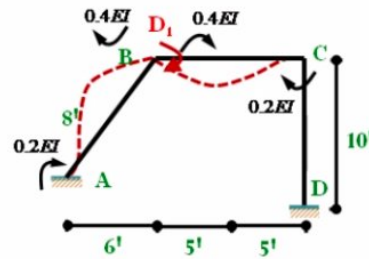
Talking: Dr. Md. Shafiqul ISLAM

$$D_1 = 1, D_2 = 0, D_3 = 0$$

$$S_{11} = 0.4EI + 0.4EI = 0.8EI$$

$$S_{21} = 0.2EI$$

$$S_{31} = -0.03EI$$





# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

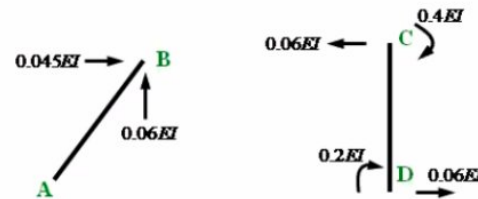
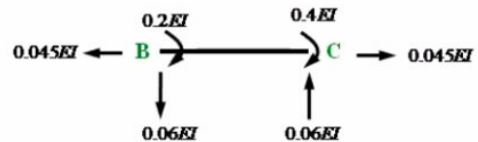
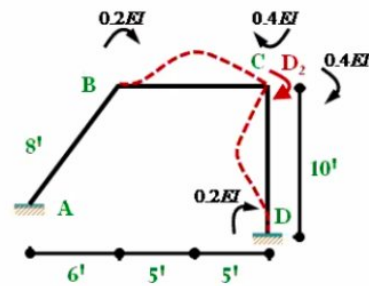
Talking: Dr. Md. Shafiqul ISLAM

$$D_2 = 1, D_1 = 0, D_3 = 0$$

$$S_{12} = 0.2EI$$

$$S_{22} = 0.4EI + 0.4EI = 0.8EI$$

$$S_{32} = 0.045EI - 0.06EI = -0.015EI$$





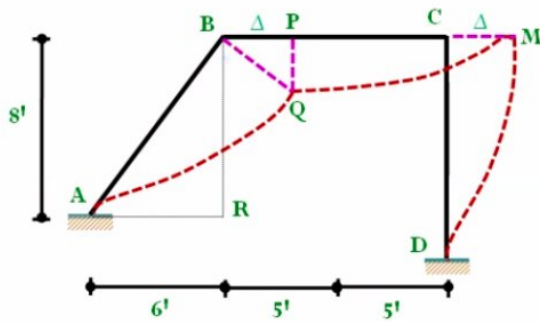
# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$D_3 = 1, D_1 = 0, D_2 = 0$$

Finding equivalent  $\Delta$





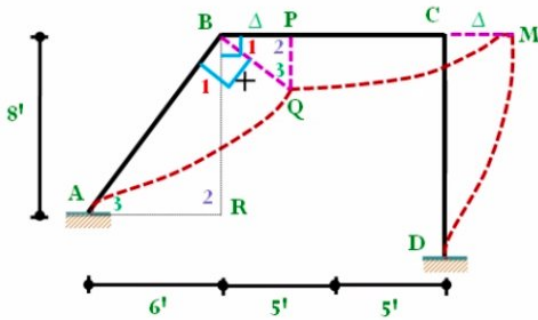
# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$D_3 = 1, D_1 = 0, D_2 = 0$$

Finding equivalent  $\Delta$



Member	Deflection	Deflection
AB	BQ	1.25 $\Delta$
BC	PQ	0.75 $\Delta$
CD	CM	1 $\Delta$

Triangle **BPQ** is similar to triangle **BRA**

$$\frac{BP}{BR} = \frac{\Delta}{8} = \frac{PQ}{RA} = \frac{PQ}{6}$$

$$\frac{PQ}{6} = \frac{\Delta}{8}$$

$$PQ = 0.75\Delta$$

$$\frac{BP}{BR} = \frac{\Delta}{8} = \frac{BQ}{BA} = \frac{BQ}{10}$$

$$\frac{BQ}{10} = \frac{\Delta}{8}$$

$$BQ = 1.25\Delta$$

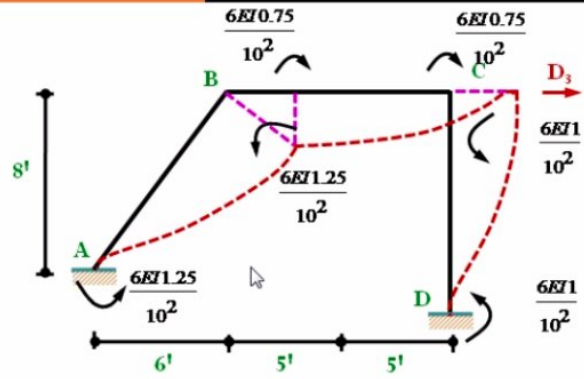


# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: \_\_\_\_\_

$D_3 = 1, D_1 = 0, D_2 = 0$





# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

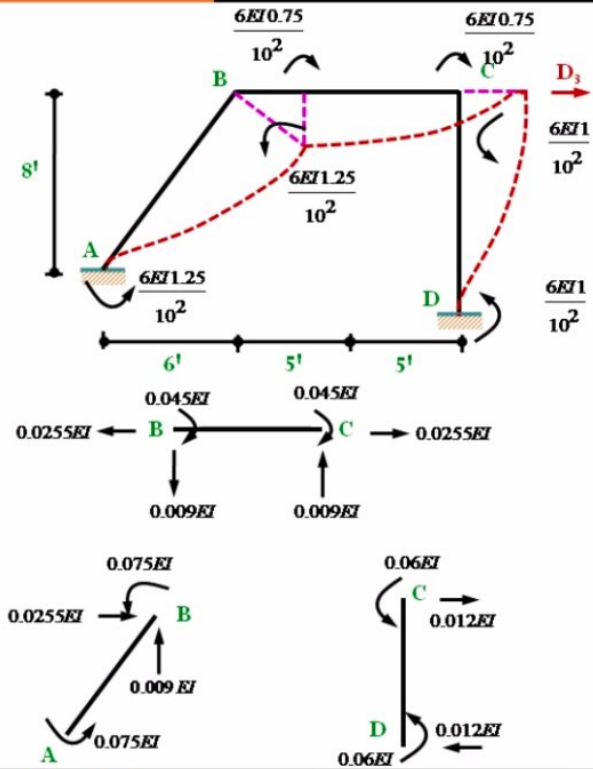
Talking:

$$D_3 = 1, D_1 = 0, D_2 = 0$$

$$S_{13} = 0.045EI - 0.075EI = -0.03EI$$

$$S_{23} = 0.045EI - 0.06EI = -0.015EI$$

$$S_{33} = 0.0255EI + 0.012EI = 0.0375EI$$





# Stiffness Matrix-FRAME

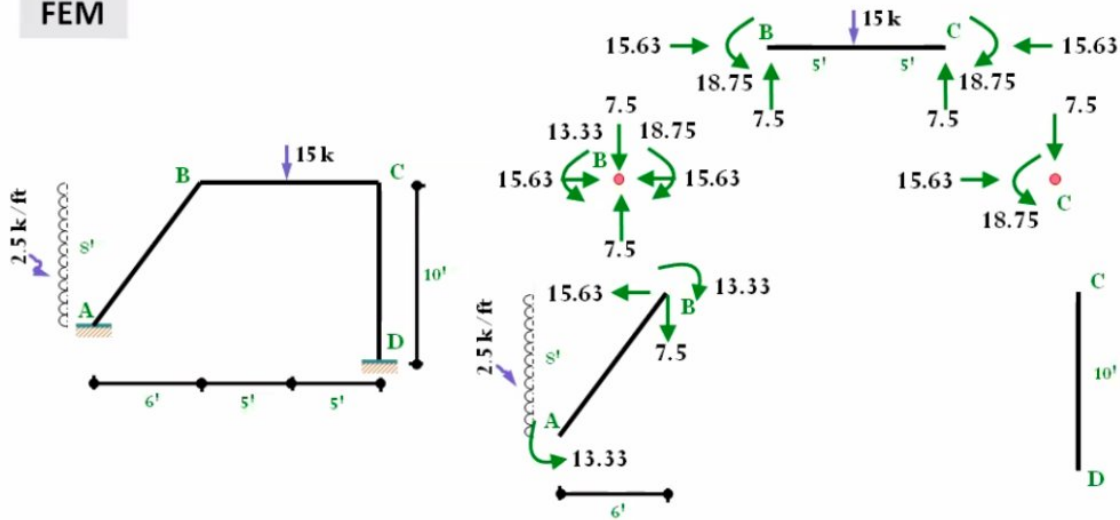
9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

Global Stiffness Matrix

$$S = EI \begin{bmatrix} 0.8 & 0.2 & -0.03 \\ 0.2 & 0.8 & -0.015 \\ -0.03 & -0.015 & 0.0375 \end{bmatrix}$$

FEM





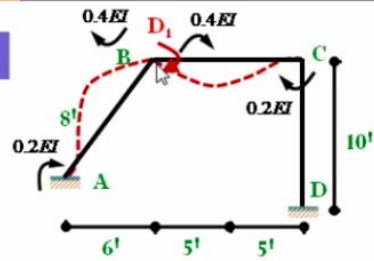
# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$M = FEM + S \times D$$

$$\begin{bmatrix} M_{AB} \\ M_{BA} \\ M_{BC} \\ M_{CB} \\ M_{CD} \\ M_{DC} \end{bmatrix} = \begin{bmatrix} -13.33 \\ 13.33 \\ -18.75 \\ 18.75 \\ 0 \\ 0 \end{bmatrix} + EI \begin{bmatrix} 0.2 \\ 0.4 \\ 0.4 \\ 0.2 \\ 0 \\ 0 \end{bmatrix} \times \frac{1}{EI} \begin{bmatrix} 28.55 \\ -22.51 \\ 430.64 \end{bmatrix}$$





# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

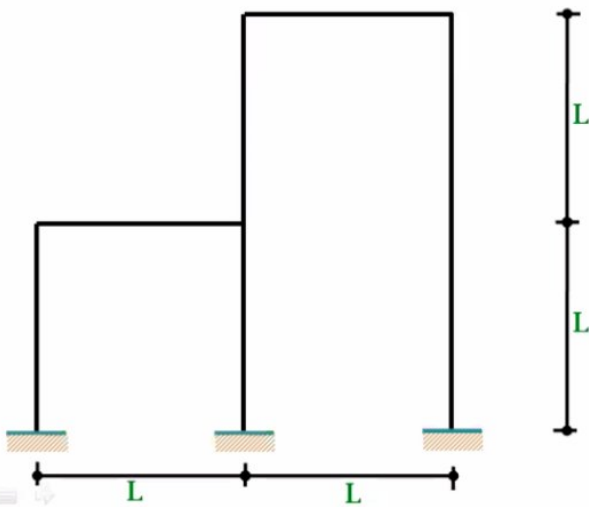
Talking: Dr. Md. Shafiqul ISLAM

## PROBLEM

07

Using Stiffness Matrix Method solve the followings, EI is constant

- a) Determine the **Stiffness matrix**



2



# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

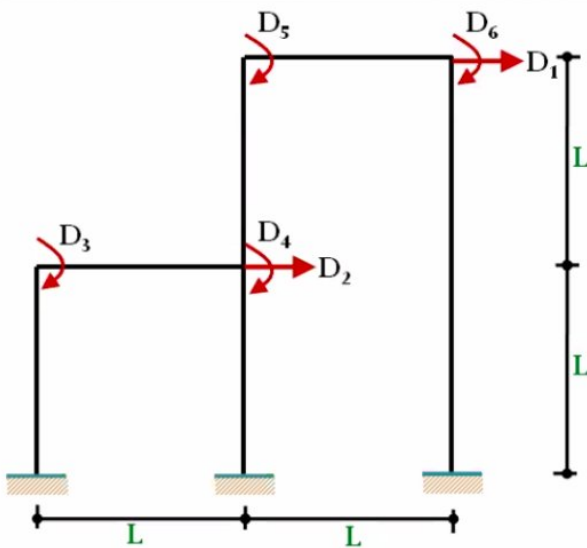
## PROBLEM

07

Using Stiffness Matrix Method solve the followings, EI is constant

a) Determine the **Stiffness matrix**

Active DoF **6**



2

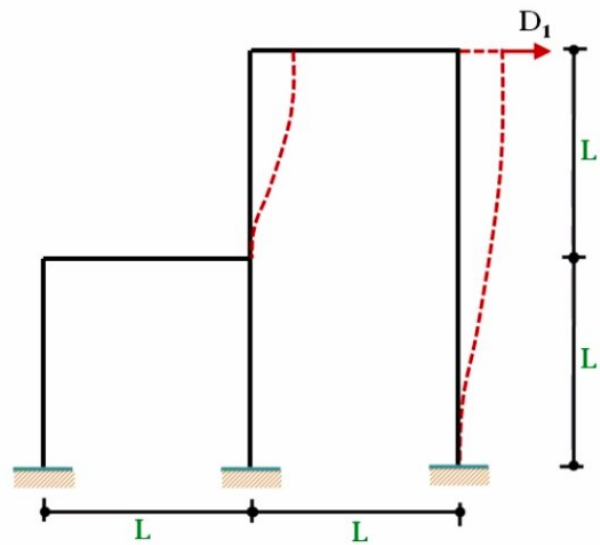


# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$D_1 = 1, D_2 = D_3 = D_4 = D_5 = D_6 = 0$$



3

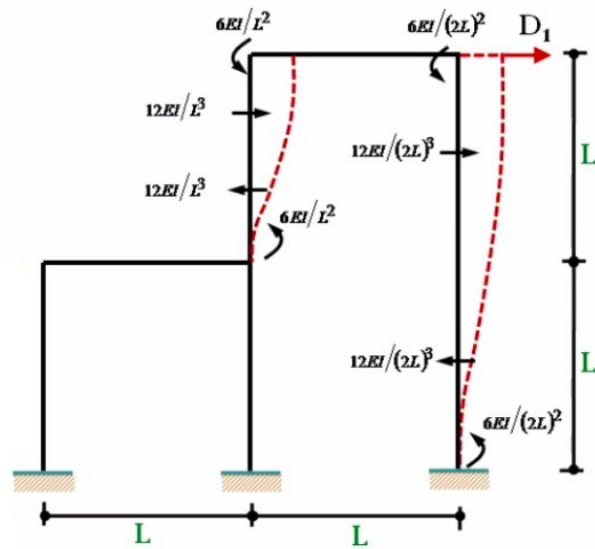


# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$D_1 = 1, D_2 = D_3 = D_4 = D_5 = D_6 = 0$$





# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$D_1 = 1, D_2 = D_3 = D_4 = D_5 = D_6 = 0$$

$$S_{11} = \frac{12EI}{L^3} + \frac{12EI}{(2L)^3} = \frac{13.5EI}{L^3}$$

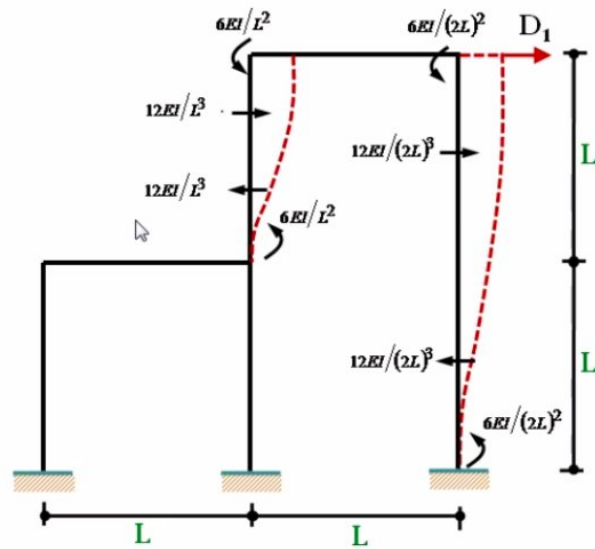
$$S_{21} = -\frac{12EI}{L^3}$$

$$S_{31} = 0$$

$$S_{41} = -\frac{6EI}{L^2}$$

$$S_{51} = -\frac{6EI}{L^2}$$

$$S_{61} = -\frac{6EI}{(2L)^2} = -\frac{1.5EI}{L^2}$$





# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

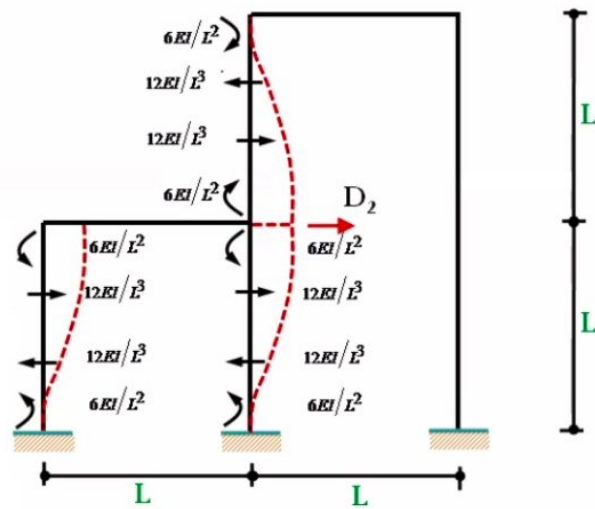
$$D_2 = 1, D_1 = D_3 = D_4 = D_5 = D_6 = 0$$

$$S_{12} = -\frac{12EI}{L^3}$$

$$S_{22} = 3 \times \frac{12EI}{L^3} = \frac{36EI}{L^3}$$

$$S_{32} = -\frac{6EI}{L^2}$$

$$S_{42} = -\frac{6EI}{L^2} + \frac{6EI}{L^2} = 0$$





# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$D_3 = 1, D_1 = D_2 = D_4 = D_5 = D_6 = 0$$

$$S_{13} = 0$$

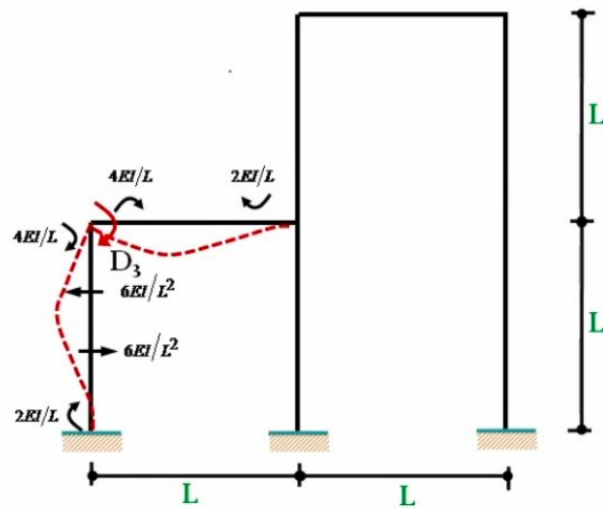
$$S_{23} = -\frac{6EI}{L^2}$$

$$S_{33} = \frac{4EI}{L} + \frac{4EI}{L} = \frac{8EI}{L}$$

$$S_{43} = \frac{2EI}{L}$$

$$S_{53} = 0$$

$$S_{63} = 0$$



5



# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$D_4 = 1, D_1 = D_2 = D_3 = D_5 = D_6 = 0$$

$$S_{14} = -\frac{6EI}{L^2}$$

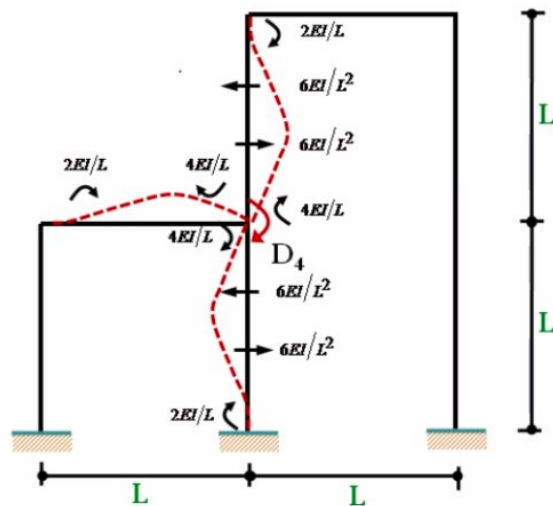
$$S_{24} = \frac{6EI}{L^2} - \frac{6EI}{L^2} = 0$$

$$S_{34} = \frac{2EI}{L}$$

$$S_{44} = 3 \times \frac{4EI}{L} = \frac{12EI}{L}$$

$$S_{54} = \frac{2EI}{L}$$

$$S_{64} = 0$$



6



# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$D_5 = 1, D_1 = D_2 = D_3 = D_4 = D_6 = 0$$

$$S_{15} = -\frac{6EI}{L^2}$$

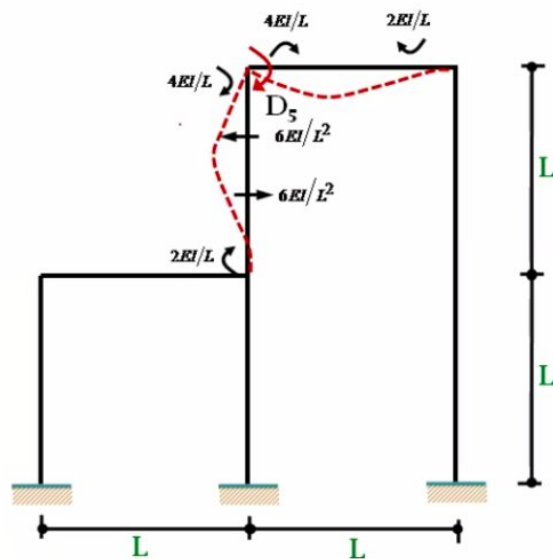
$$S_{25} = \frac{6EI}{L^2}$$

$$S_{35} = 0$$

$$S_{45} = \frac{2EI}{L}$$

$$S_{55} = 2 \times \frac{4EI}{L} = \frac{8EI}{L}$$

$$S_{65} = \frac{2EI}{L}$$



7



# Stiffness Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$D_6 = 1, D_1 = D_2 = D_3 = D_4 = D_5 = 0$$

$$S_{16} = -\frac{6EI}{(2L)^2} = -\frac{1.5EI}{L^2}$$

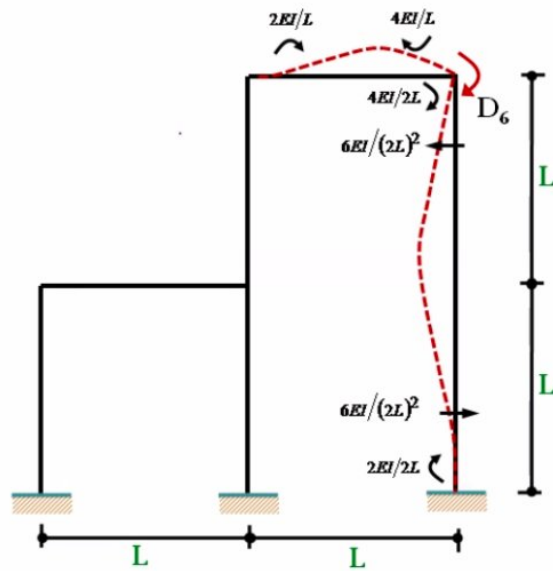
$$S_{26} = 0$$

$$S_{36} = 0$$

$$S_{46} = 0$$

$$S_{56} = \frac{2EI}{L}$$

$$S_{66} = \frac{4EI}{2L} + \frac{4EI}{L} = \frac{6EI}{L}$$





# Stiffness Matrix-FRAME

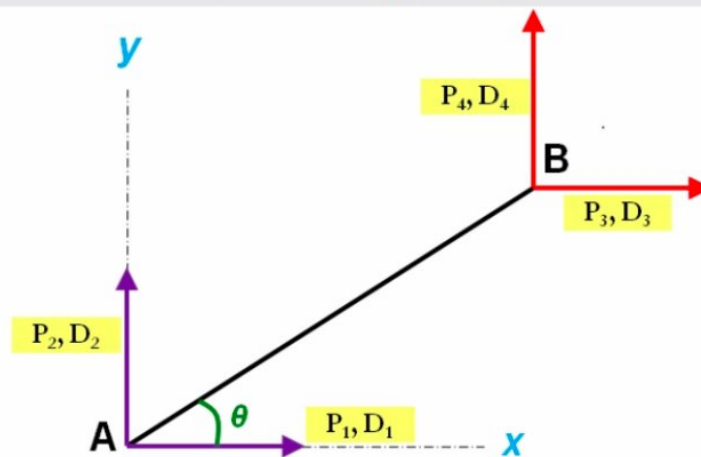
9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$S = EI \begin{bmatrix} 13.5/L^3 & -12/L^3 & 0 & -6/L^2 & -6/L^2 & -1.5/L^2 \\ 12/L^3 & 36/L^3 & -6/L^2 & 0 & 6/L^2 & 0 \\ 0 & -6/L^2 & 8/L & 2/L & 0 & 0 \\ -6/L^2 & 0 & 2/L & 12/L & 2/L & 0 \\ -6/L^2 & 6/L^2 & 0 & 2/L & 8/L & 2/L \\ -1.5/L^2 & 0 & 0 & 0 & 2/L & 6/L \end{bmatrix}$$



## Derivation of **Stiffness Matrix** for **Truss Element** having inclination $\theta$ with $x$ axis



Consider a single element of truss  
AB with Length =  $L$   
Modulus of Elasticity =  $E$   
Cross section area =  $A$

$P_1, D_1$  = Force and Deflection along  
x axis  
 $P_2, D_2$  = Force and Deflection along  
y axis



## Stiffness Matrix Method

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

Case 1  $D_1 = 1, D_2 = D_3 = D_4 = 0$

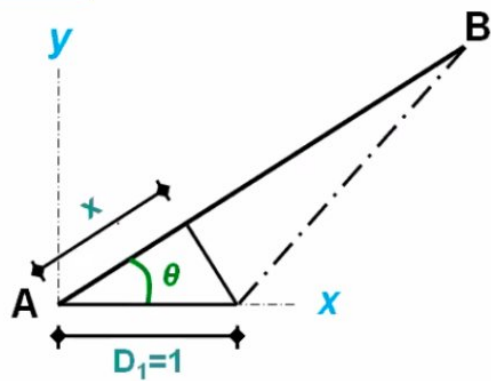
$$\cos\theta = \frac{x}{D_1} \quad \cos\theta = \frac{x}{1} \quad x = \cos\theta$$

Along **AB**

$$\text{Strain} = \frac{x}{L} = \frac{\cos\theta}{L}$$

$$\text{Stress} = \frac{\cos\theta}{L} \times E$$

$$\text{Force} = \frac{\cos\theta}{L} \times EA$$





# Stiffness Matrix Method

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

Case 1  $D_1 = 1, D_2 = D_3 = D_4 = 0$

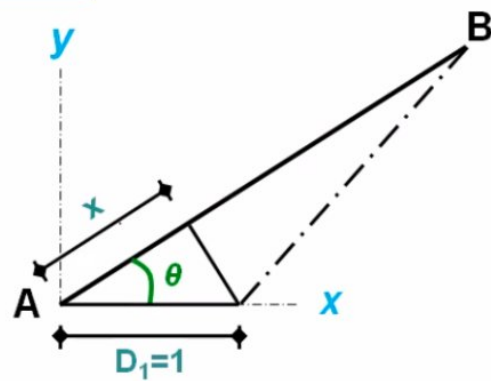
$$\cos\theta = \frac{x}{D_1} \quad \cos\theta = \frac{x}{1} \quad x = \cos\theta$$

Along **AB**

$$\text{Strain} = \frac{x}{L} = \frac{\cos\theta}{L}$$

$$\text{Stress} = \frac{\cos\theta}{L} \times E$$

$$\text{Force} = \frac{\cos\theta}{L} \times EA$$



Force Along **DoF**

$$S_{11} = \frac{\cos\theta}{L} \times EA \times \cos\theta = \frac{EA}{L} \cos^2\theta$$

$$S_{31} = -\frac{EA}{L} \cos^2\theta$$

$$S_{21} = \frac{EA}{L} \cos\theta \sin\theta$$

$$S_{41} = -\frac{EA}{L} \cos\theta \sin\theta$$

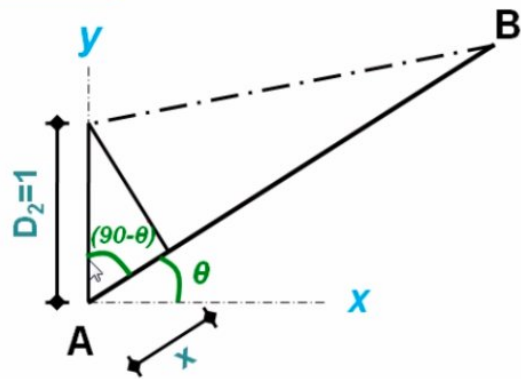


# Stiffness Matrix Method

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

Case 2  $D_2 = 1, D_1 = D_3 = D_4 = 0$





Case 2  $D_2 = 1, D_1 = D_3 = D_4 = 0$

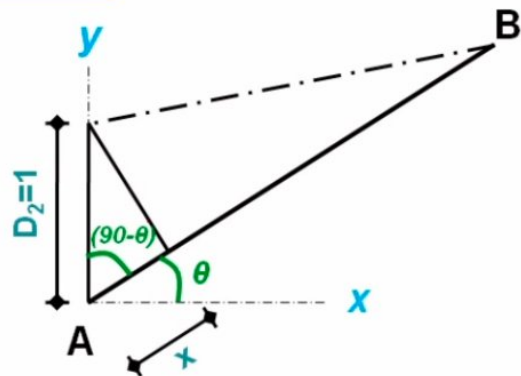
$$\cos(90 - \theta) = \frac{x}{D_2} \quad \sin \theta = \frac{x}{1} \quad x = \sin \theta$$

Along AB

$$\text{Strain} = \frac{x}{L} = \frac{\sin \theta}{L}$$

$$\text{Stress} = \frac{\sin \theta}{L} \times E$$

$$\text{Force} = \frac{\sin \theta}{L} \times EA$$



Force Along DoF

$$S_{12} = \frac{\sin \theta}{L} \times EA \times \cos \theta = \frac{EA}{L} \cos \theta \sin \theta$$

$$S_{32} = -\frac{EA}{L} \cos \theta \sin \theta$$

$$S_{22} = \frac{EA}{L} \sin^2 \theta$$

$$S_{42} = -\frac{EA}{L} \sin^2 \theta$$



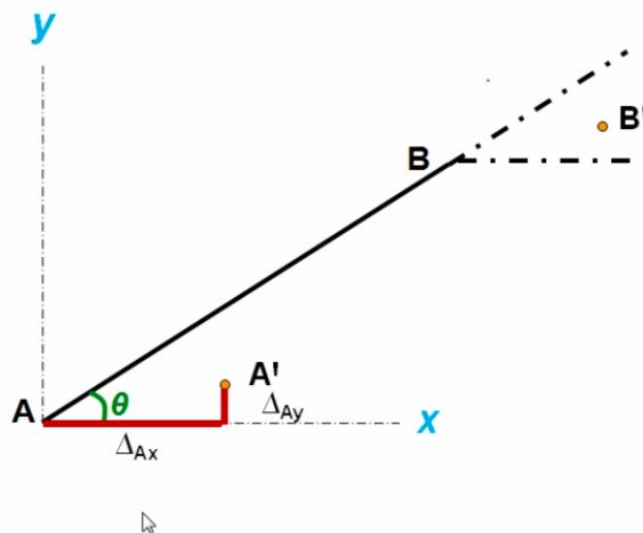
$$P_1 = -P_3$$

$$P_2 = -P_4$$

$$S = \frac{EA}{L} \begin{bmatrix} \cos^2 \theta & \cos \theta \sin \theta & -\cos^2 \theta & -\cos \theta \sin \theta \\ \cos \theta \sin \theta & \sin^2 \theta & -\cos \theta \sin \theta & -\sin^2 \theta \\ -\cos^2 \theta & -\cos \theta \sin \theta & \cos^2 \theta & \cos \theta \sin \theta \\ -\cos \theta \sin \theta & -\sin^2 \theta & \cos \theta \sin \theta & \sin^2 \theta \end{bmatrix}$$

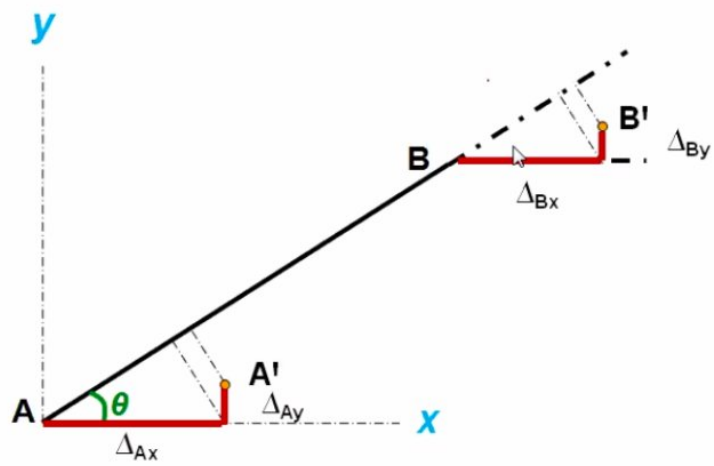


## Force along the member of truss element



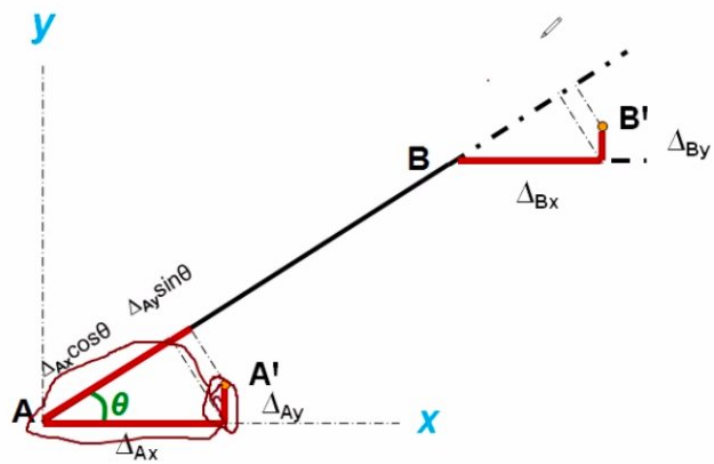


## Force along the member of truss element



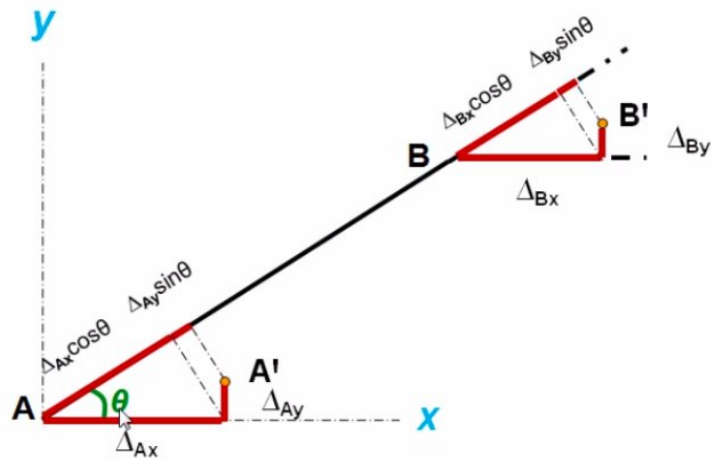


## Force along the member of truss element





## Force along the member of truss element





### Force along the member of truss element

Shortening due to displacement of *joint A*  $= \Delta_{Ax} \cos \theta + \Delta_{Ay} \sin \theta$

Elongation due to displacement of *joint B*  $= \Delta_{Bx} \cos \theta + \Delta_{By} \sin \theta$



## Force along the member of truss element

$$\text{Shortening due to displacement of joint A} = \Delta_{Ax} \cos \theta + \Delta_{Ay} \sin \theta$$

$$\text{Elongation due to displacement of joint B} = \Delta_{Bx} \cos \theta + \Delta_{By} \sin \theta$$

$$\text{Net Shortening} = \left( \Delta_{Ax} - \Delta_{Bx} \right) \cos \theta + \left( \Delta_{Ay} - \Delta_{By} \right) \sin \theta$$



### Force along the member of truss element

Shortening due to displacement of *joint A*  $= \Delta_{Ax} \cos \theta + \Delta_{Ay} \sin \theta$

Elongation due to displacement of *joint B*  $= \Delta_{Bx} \cos \theta + \Delta_{By} \sin \theta$

Net Shortening  $= \left( \Delta_{Ax} - \Delta_{Bx} \right) \cos \theta + \left( \Delta_{Ay} - \Delta_{By} \right) \sin \theta$

$$F_{AB} = -\frac{AE}{L} \left[ \left( \Delta_{Ax} - \Delta_{Bx} \right) \cos \theta + \left( \Delta_{Ay} - \Delta_{By} \right) \sin \theta \right]$$



# Stiffness Matrix-TRUSS

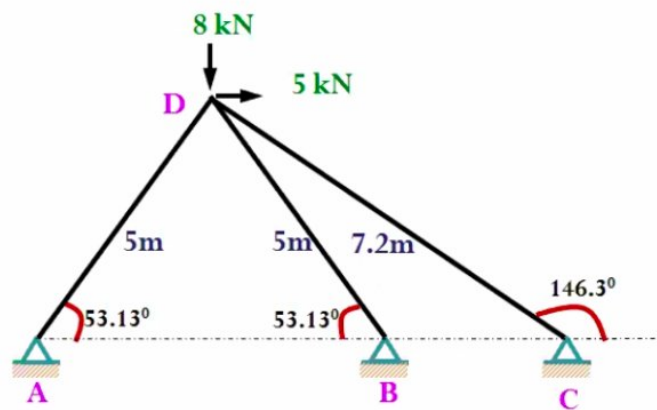
9<sup>th</sup> May, 2020 Talking:

## PROBLEM

10

Using Stiffness Matrix Method solve the followings, EA is constant

- a) Find the **Support Reactions**
- b) Find the **Member Forces**

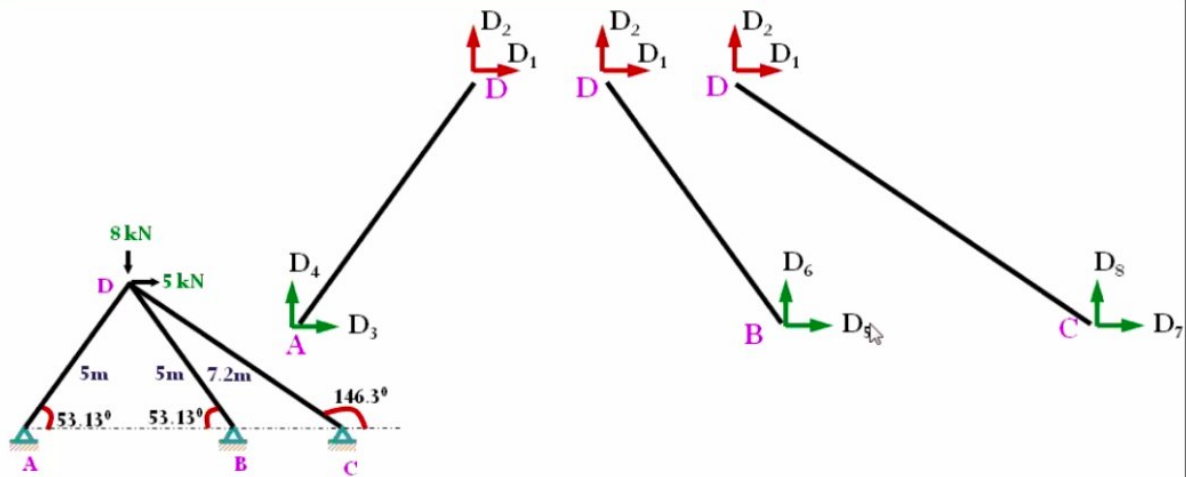




# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020

Talking:Dr. Md. Shafiqul ISLAM



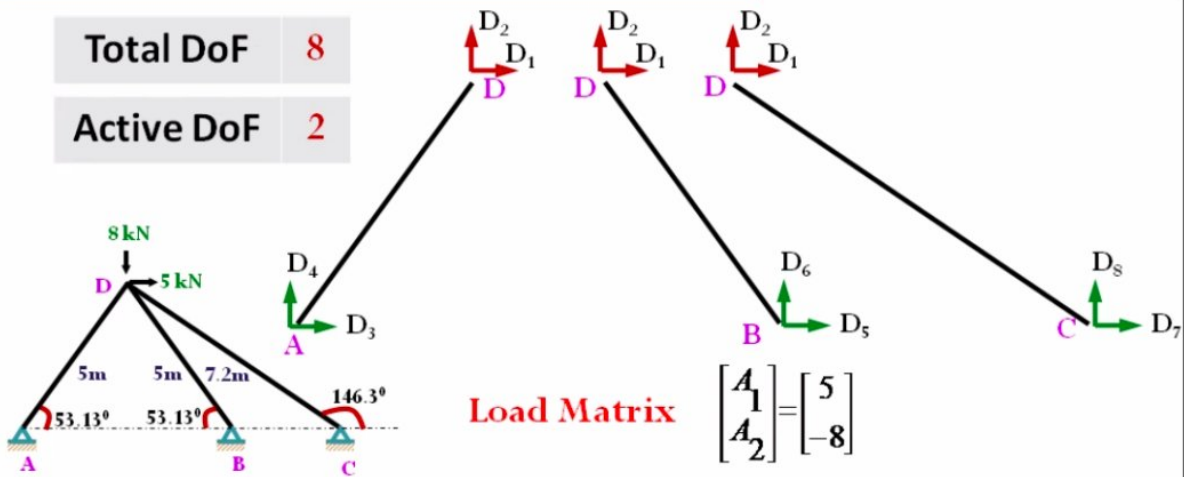


# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020

Talking:1600035 , Dr. Md. Shafiqul

Total DoF	8
Active DoF	2



Load Matrix

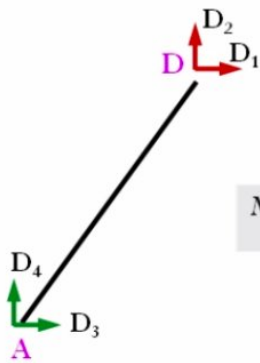
$$\begin{bmatrix} A_1 \\ A_2 \end{bmatrix} = \begin{bmatrix} 5 \\ -8 \end{bmatrix}$$

Member	Inclination ( <sup>o</sup> ) with respect to 'x'	Length (m)
AD	53.13	5
BD	126.87	5
CD	146.3	7.2



# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020 Talking:



$$S = \frac{EA}{L} \begin{bmatrix} \cos^2 \theta & \cos \theta \sin \theta & -\cos^2 \theta & -\cos \theta \sin \theta \\ \cos \theta \sin \theta & \sin^2 \theta & -\cos \theta \sin \theta & -\sin^2 \theta \\ -\cos^2 \theta & -\cos \theta \sin \theta & \cos^2 \theta & \cos \theta \sin \theta \\ -\cos \theta \sin \theta & -\sin^2 \theta & \cos \theta \sin \theta & \sin^2 \theta \end{bmatrix}$$

Member	AD	Length	5	$\theta$	53.13
--------	----	--------	---	----------	-------

W	H	O
D <sub>3</sub> =1	D <sub>4</sub> =1	D <sub>1</sub> =1
D <sub>2</sub> =1		

$$S_{AD} = EA \begin{bmatrix} 0.072 & 0.096 & -0.072 & -0.096 \\ 0.096 & 0.128 & -0.096 & -0.128 \\ -0.072 & -0.096 & 0.072 & 0.096 \\ -0.096 & -0.128 & 0.096 & 0.128 \end{bmatrix} \begin{matrix} D_3=1 \\ D_4=1 \\ D_1=1 \\ D_2=1 \end{matrix} \begin{matrix} W \\ H \\ R \\ E \end{matrix}$$

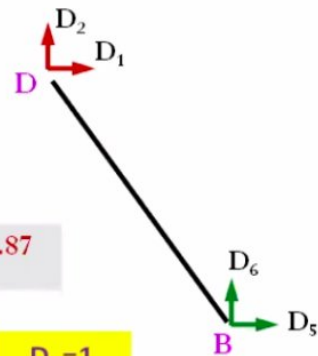


# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$S = \frac{EA}{L} \begin{bmatrix} \cos^2 \theta & \cos \theta \sin \theta & -\cos^2 \theta & -\cos \theta \sin \theta \\ \cos \theta \sin \theta & \sin^2 \theta & -\cos \theta \sin \theta & -\sin^2 \theta \\ -\cos^2 \theta & -\cos \theta \sin \theta & \cos^2 \theta & \cos \theta \sin \theta \\ -\cos \theta \sin \theta & -\sin^2 \theta & \cos \theta \sin \theta & \sin^2 \theta \end{bmatrix}$$



Member	BD	Length	5	$\theta$	126.87
--------	----	--------	---	----------	--------

$D_5=1$	$D_6=1$	$D_1=1$	$D_2=1$
---------	---------	---------	---------

$$S_{BD} = EA \begin{bmatrix} 0.072 & -0.096 & -0.072 & 0.096 \\ -0.096 & 0.128 & 0.096 & -0.128 \\ -0.072 & 0.096 & 0.072 & -0.096 \\ 0.096 & -0.128 & -0.096 & 0.128 \end{bmatrix}$$

$D_5=1$
$D_6=1$
$D_1=1$
$D_2=1$

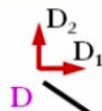


# Stiffness Matrix-TRUSS

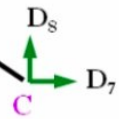
9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

$$S = \frac{EA}{L} \begin{bmatrix} \cos^2 \theta & \cos \theta \sin \theta & -\cos^2 \theta & -\cos \theta \sin \theta \\ \cos \theta \sin \theta & \sin^2 \theta & -\cos \theta \sin \theta & -\sin^2 \theta \\ -\cos^2 \theta & -\cos \theta \sin \theta & \cos^2 \theta & \cos \theta \sin \theta \\ -\cos \theta \sin \theta & -\sin^2 \theta & \cos \theta \sin \theta & \sin^2 \theta \end{bmatrix}$$



Member	CD	Length	7.2	$\theta$	146.3
--------	----	--------	-----	----------	-------



$D_7=1$	$D_8=1$	$D_1=1$	$D_2=1$
---------	---------	---------	---------

$$S_{CD} = EA \begin{bmatrix} 0.096 & -0.063 & -0.096 & 0.063 \\ -0.063 & 0.042 & 0.063 & -0.042 \\ -0.096 & 0.063 & 0.096 & -0.063 \\ 0.063 & -0.042 & -0.063 & 0.042 \end{bmatrix}$$

$D_7=1$
$D_8=1$
$D_1=1$
$D_2=1$



# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020 Talking:

$$S_{AD} = EA \begin{bmatrix} 0.072 & 0.096 & -0.072 & -0.096 \\ 0.096 & 0.128 & -0.096 & -0.128 \\ -0.072 & -0.096 & 0.072 & 0.096 \\ -0.096 & -0.128 & 0.096 & 0.128 \end{bmatrix}$$

Nodes: 3, 4, 1, 2

$$S_{BD} = EA \begin{bmatrix} 0.072 & -0.096 & -0.072 & 0.096 \\ -0.096 & 0.128 & 0.096 & -0.128 \\ -0.072 & 0.096 & 0.072 & -0.096 \\ 0.096 & -0.128 & -0.096 & 0.128 \end{bmatrix}$$

Nodes: 5, 6, 1, 2

$$S_{CD} = EA \begin{bmatrix} 0.096 & -0.063 & -0.096 & 0.063 \\ -0.063 & 0.042 & 0.063 & -0.042 \\ -0.096 & 0.063 & 0.096 & -0.063 \\ 0.063 & -0.042 & -0.063 & 0.042 \end{bmatrix}$$

Nodes: 7, 8, 1, 2



# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020 Talking:

$$S_{AD} = EA \begin{bmatrix} 0.072 & 0.096 & -0.072 & -0.096 \\ 0.096 & 0.128 & -0.096 & -0.128 \\ -0.072 & -0.096 & 0.072 & 0.096 \\ -0.096 & -0.128 & 0.096 & 0.128 \end{bmatrix}$$

Nodes: 3, 4, 1, 2

$$S_{BD} = EA \begin{bmatrix} 0.072 & -0.096 & -0.072 & 0.096 \\ -0.096 & 0.128 & 0.096 & -0.128 \\ -0.072 & 0.096 & 0.072 & -0.096 \\ 0.096 & -0.128 & -0.096 & 0.128 \end{bmatrix}$$

Nodes: 5, 6, 1, 2

$$S_{CD} = EA \begin{bmatrix} 0.096 & -0.063 & -0.096 & 0.063 \\ -0.063 & 0.042 & 0.063 & -0.042 \\ -0.096 & 0.063 & 0.096 & -0.063 \\ 0.063 & -0.042 & -0.063 & 0.042 \end{bmatrix}$$

Nodes: 7, 8, 1, 2

$$S = EA \begin{bmatrix} 0.072+0.072+0.096 & 0.096-0.096-0.063 \\ 0.096-0.096-0.063 & 0.128+0.128+0.042 \end{bmatrix}$$



## Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020 Talking:

$$[A]=[S][D]$$

$$\begin{bmatrix} 5 \\ -8 \end{bmatrix} = EA \begin{bmatrix} 0.24 & -0.063 \\ -0.063 & 0.298 \end{bmatrix} \begin{bmatrix} D_1 \\ D_2 \end{bmatrix}$$

$$\begin{bmatrix} D_1 \\ D_2 \end{bmatrix} = \frac{1}{EA} \begin{bmatrix} 0.24 & -0.063 \\ -0.063 & 0.298 \end{bmatrix}^{-1} \begin{bmatrix} 5 \\ -8 \end{bmatrix}$$

$$\begin{bmatrix} D_1 \\ D_2 \end{bmatrix} = \frac{1}{EA} \begin{bmatrix} 14.59 \\ -23.74 \end{bmatrix}$$

8

Unmute Start Video

Participants 88

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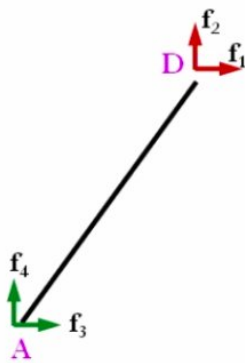
More

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# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020 Talking:



Member	AD	Component Forces
Joint	A	Component Reactions

$$\begin{bmatrix} f_3 \\ f_4 \\ f_1 \\ f_2 \end{bmatrix} = EA \begin{bmatrix} 0.072 & 0.096 & -0.072 & -0.096 \\ 0.096 & 0.128 & -0.096 & -0.128 \\ -0.072 & -0.096 & 0.072 & 0.096 \\ -0.096 & -0.128 & 0.096 & 0.128 \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 14.59 \\ -23.74 \end{bmatrix} \frac{1}{EA}$$

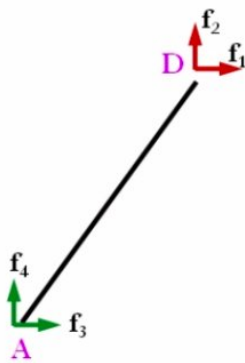
$$\begin{bmatrix} D_1 \\ D_2 \end{bmatrix} = \frac{1}{EA} \begin{bmatrix} 14.59 \\ -23.74 \end{bmatrix}$$

$A = S D$



# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020 Talking:

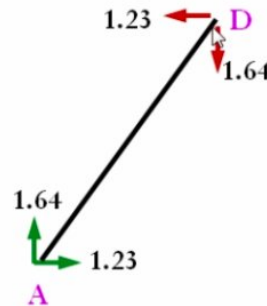


Member	AD	Component Forces
Joint	A	Component Reactions

$$\begin{bmatrix} f_3 \\ f_4 \\ f_1 \\ f_2 \end{bmatrix} = \frac{EA}{L} \begin{bmatrix} 0.072 & 0.096 & -0.072 & -0.096 \\ 0.096 & 0.128 & -0.096 & -0.128 \\ -0.072 & -0.096 & 0.072 & 0.096 \\ -0.096 & -0.128 & 0.096 & 0.128 \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 14.59 \\ -23.74 \end{bmatrix} \frac{1}{EA}$$

$$\begin{bmatrix} D_1 \\ D_2 \end{bmatrix} = \frac{1}{EA} \begin{bmatrix} 14.59 \\ -23.74 \end{bmatrix}$$

$$\begin{bmatrix} f_3 \\ f_4 \\ f_1 \\ f_2 \end{bmatrix} = \begin{bmatrix} 1.23 \\ 1.64 \\ -1.23 \\ -1.64 \end{bmatrix}$$

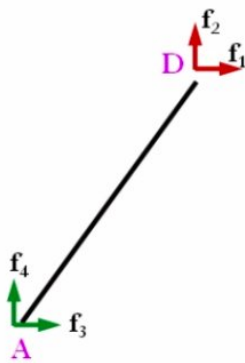




# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020

Talking:1600068 - Rana Hamid, Dr. Md.

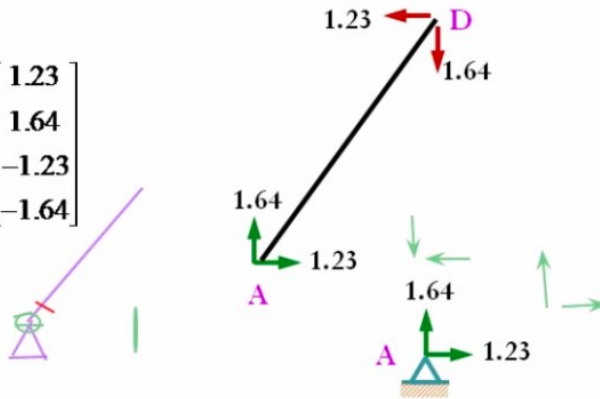


Member	AD	Component Forces
Joint	A	Component Reactions

$$\begin{bmatrix} f_3 \\ f_4 \\ f_1 \\ f_2 \end{bmatrix} = \frac{EA}{L} \begin{bmatrix} 0.072 & 0.096 & -0.072 & -0.096 \\ 0.096 & 0.128 & -0.096 & -0.128 \\ -0.072 & -0.096 & 0.072 & 0.096 \\ -0.096 & -0.128 & 0.096 & 0.128 \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 14.59 \\ -23.74 \end{bmatrix} \frac{1}{EA}$$

$$\begin{bmatrix} D_1 \\ D_2 \end{bmatrix} = \frac{1}{EA} \begin{bmatrix} 14.59 \\ -23.74 \end{bmatrix}$$

$$\begin{bmatrix} f_3 \\ f_4 \\ f_1 \\ f_2 \end{bmatrix} = \begin{bmatrix} 1.23 \\ 1.64 \\ -1.23 \\ -1.64 \end{bmatrix}$$





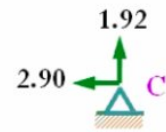
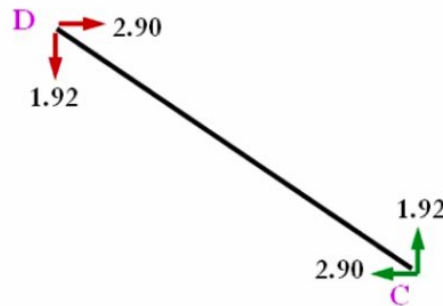
# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020 Talking:

Member	CD	Component Forces
Joint	C	Component Reactions

$$\begin{bmatrix} f_7 \\ f_8 \\ f_1 \\ f_2 \end{bmatrix} = \frac{EA}{L} \begin{bmatrix} 0.096 & -0.063 & -0.096 & 0.063 \\ -0.063 & 0.042 & 0.063 & -0.042 \\ -0.096 & 0.063 & 0.096 & -0.063 \\ 0.063 & -0.042 & -0.063 & 0.042 \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 14.59 \\ -23.74 \end{bmatrix} \frac{1}{EA}$$

$$\begin{bmatrix} f_7 \\ f_8 \\ f_1 \\ f_2 \end{bmatrix} = \begin{bmatrix} -2.90 \\ 1.92 \\ 2.90 \\ -1.92 \end{bmatrix}$$

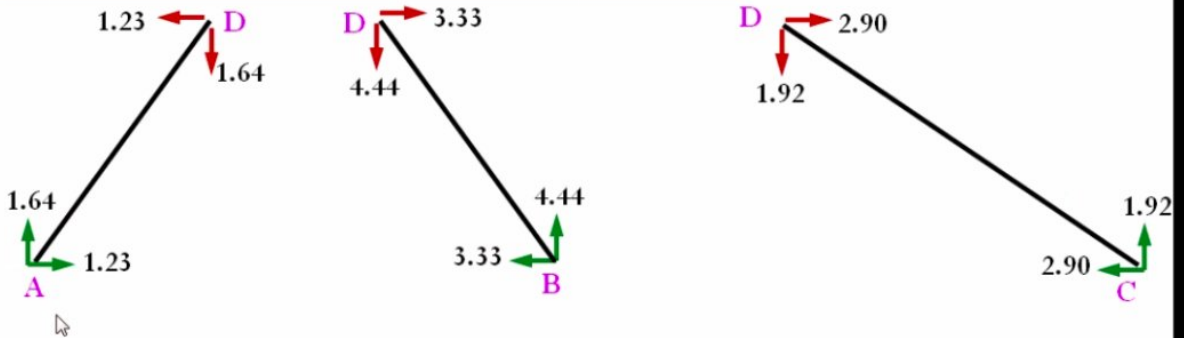




# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020

Talking:

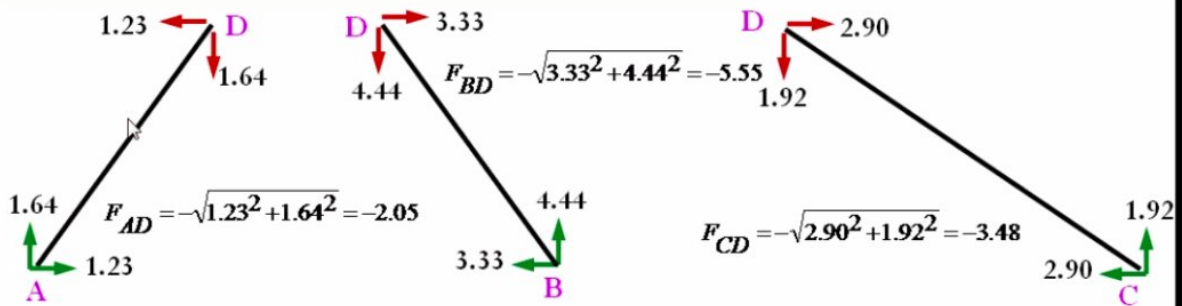




# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020

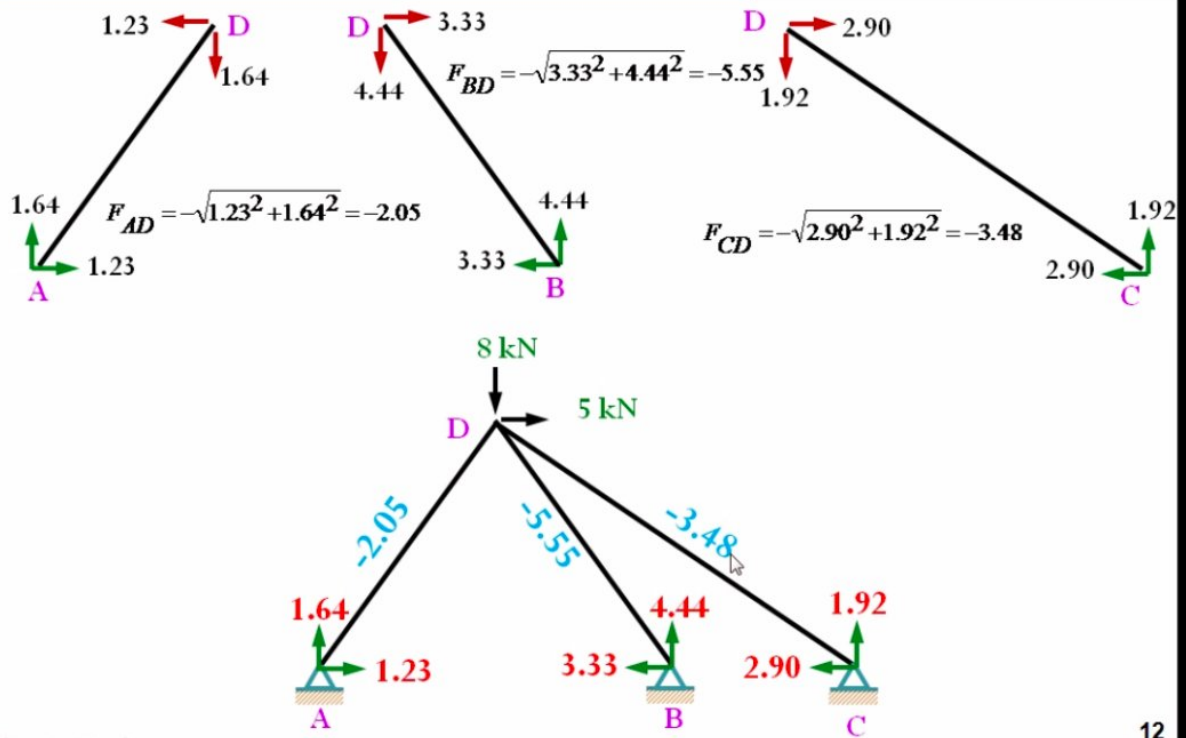
Talking: Dr. Md. Shafiqul ISLAM





# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020 Talking:





# Stiffness Matrix-TRUSS

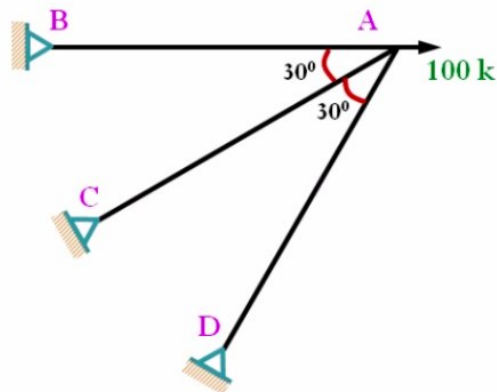
9<sup>th</sup> May, 2020 Talking:

## PROBLEM

09

Using Stiffness Matrix Method solve the followings, EA is constant

- a) Find the **Member Forces**



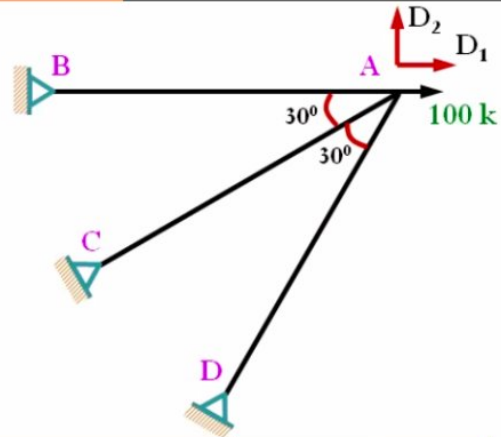


# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020 Talking:

Load Matrix

$$\begin{bmatrix} A_1 \\ A_2 \end{bmatrix} = \begin{bmatrix} 100 \\ 0 \end{bmatrix}$$



Member	Inclination ( $^{\circ}$ ) with respect to 'x'	Length (ft)
AB	180	L
AC	210	L
AD	240	L



# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

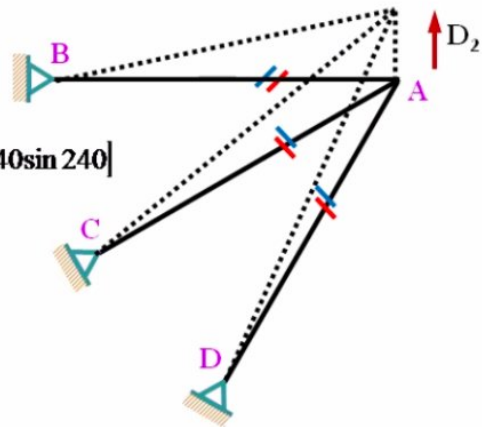
$$D_2 = 1, D_1 = 0$$

$$S_{12} = \frac{EA}{L} [\cos 180 \sin 180 + \cos 210 \sin 210 + \cos 240 \sin 240]$$

$$S_{12} = 0.867 \frac{EA}{L}$$

$$S_{22} = \frac{EA}{L} [\sin^2 180 + \sin^2 210 + \sin^2 240]$$

$$S_{22} = \frac{EA}{L}$$





# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020 Talking:

$$[A]=[S][D]$$

$$\begin{bmatrix} 100 \\ 0 \end{bmatrix} = \frac{EA}{L} \begin{bmatrix} 2 & 0.867 \\ 0.867 & 1 \end{bmatrix} \begin{bmatrix} D_1 \\ D_2 \end{bmatrix}$$



# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020 Talking:

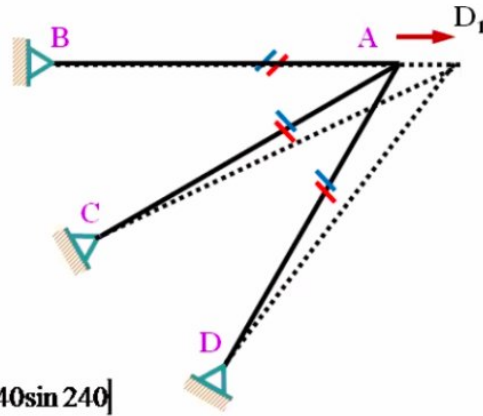
$$D_1 = 1, D_2 = 0$$

$$S_{11} = \frac{EA}{L} [\cos^2 180 + \cos^2 210 + \cos^2 240]$$

$$S_{11} = 2 \frac{EA}{L}$$

$$S_{21} = \frac{EA}{L} [\cos 180 \sin 180 + \cos 210 \sin 210 + \cos 240 \sin 240]$$

$$S_{21} = 0.867 \frac{EA}{L}$$





# Stiffness Matrix-TRUSS

9<sup>th</sup> May, 2020 Talking:

$$[F] = [S][D]$$
$$\begin{bmatrix} F_{AB} \\ F_{AC} \\ F_{AD} \end{bmatrix} = \frac{EA}{L} \begin{bmatrix} -\cos 180 & -\sin 180 \\ -\cos 210 & -\sin 210 \\ -\cos 240 & -\sin 240 \end{bmatrix} \begin{bmatrix} L \\ EA \end{bmatrix} \begin{bmatrix} 80 \\ -6936 \end{bmatrix}$$



## Stiffness Matrix Method

9<sup>th</sup> May, 2020

Talking:1600094\_IRFAN

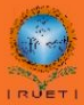
### Force along the member of truss element

Shortening due to displacement of *joint A*  $= \Delta_{Ax} \cos \theta + \Delta_{Ay} \sin \theta$

Elongation due to displacement of *joint B*  $= \Delta_{Bx} \cos \theta + \Delta_{By} \sin \theta$

Net Shortening  $= \left( \Delta_{Ax} - \Delta_{Bx} \right) \cos \theta + \left( \Delta_{Ay} - \Delta_{By} \right) \sin \theta$

$$F_{AB} = -\frac{AE}{L} \left[ \left( \Delta_{Ax} - \Delta_{Bx} \right) \cos \theta + \left( \Delta_{Ay} - \Delta_{By} \right) \sin \theta \right]$$

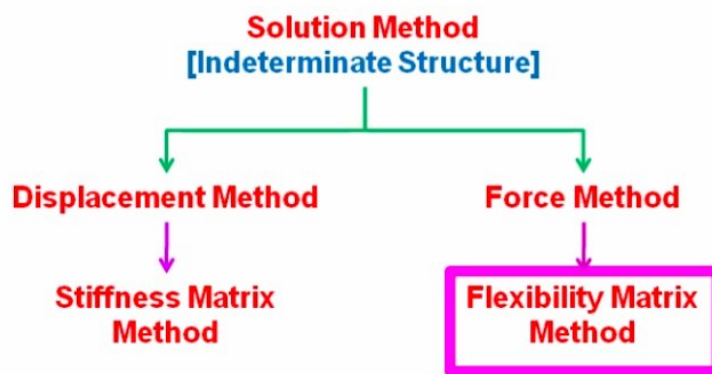


# Flexibility Matrix Method

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Analysis of Statically Indeterminate Beams





# Flexibility Matrix Method

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Definition

### Flexibility

It may be defined as the **displacement** due to application of **unit load**

### Degree of Indeterminacy (DoI)

Number of possible **Redundants** or number of **Releases** to make structure determinate

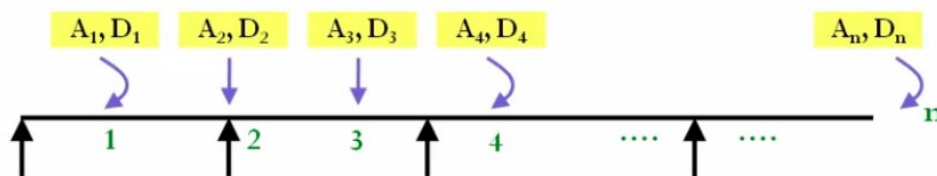


# Flexibility Matrix Method

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Basic Formula for Flexibility Matrix



$A_1, A_2, A_3, \dots, A_n$	<b>Forces</b> applied at nodes
$D_1, D_2, D_3, \dots, D_n$	<b>Displacements</b> produce at the nodes



# Flexibility Matrix Method

9<sup>th</sup> May, 2020

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Displacements at 1, 2, 3, ....., n are

$$D_1 = D_{11} + D_{12} + D_{13} + \dots + D_{1n}$$

$$D_1 = F_{11}A_1 + F_{12}A_2 + F_{13}A_3 + \dots + F_{1n}A_n$$

$$D_2 = F_{21}A_1 + F_{22}A_2 + F_{23}A_3 + \dots + F_{2n}A_n$$

.....  
.....

$$D_n = F_{n1}A_1 + F_{n2}A_2 + F_{n3}A_3 + \dots + F_{nn}A_n$$



# Flexibility Matrix Method

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Matrix formation

$$\begin{bmatrix} D_1 \\ D_2 \\ \cdot \\ \cdot \\ D_n \end{bmatrix} = \begin{bmatrix} F_{11} & F_{12} & \cdot & F_{1n} \\ F_{21} & F_{22} & \cdot & F_{2n} \\ \cdot & \cdot & \cdot & \cdot \\ \cdot & \cdot & \cdot & \cdot \\ F_{n1} & F_{n2} & \cdot & F_{nn} \end{bmatrix} \begin{bmatrix} A_1 \\ A_2 \\ \cdot \\ \cdot \\ A_n \end{bmatrix}$$

$$[D] = [F][A]$$

[Displacement Matrix] = [Flexibility Matrix] [Force Matrix]

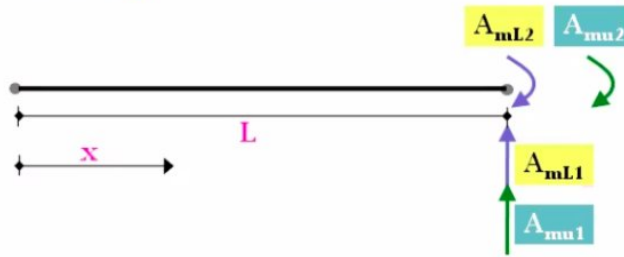


# Flexibility Matrix Method

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Deflection of Single Beam by Flexibility Matrix



Consider a single beam with	$A_{mL1}$	Shear	For Applied Load
	$A_{mL2}$	Moment	
Length = L			
Modulus of Elasticity = E	$A_{mu1}$	Shear	For Unit Load
Moment of Inertia = I	$A_{mu2}$	Moment	

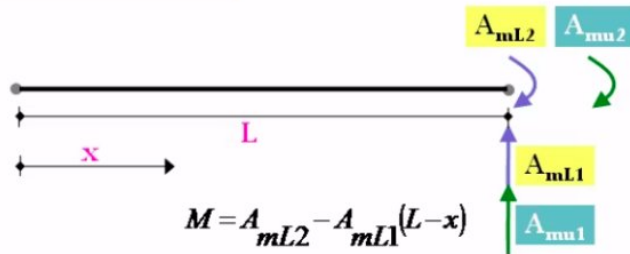


# Flexibility Matrix Method

9<sup>th</sup> May, 2020

Talking: 1600012

$$D = \int_0^L \frac{Mm dx}{EI} \dots (1)$$



putting values in (1)

$$M = A_{mL2} - A_{mL1}(L-x)$$

$$m = A_{mu2} - A_{mu1}(L-x)$$

$$D = \int_0^L \frac{[A_{mL2} - A_{mL1}(L-x)][A_{mu2} - A_{mu1}(L-x)]}{EI} dx$$

$$D = \frac{1}{EI} \int_0^L [A_{mL2}A_{mu2} - A_{mL1}A_{mu2}(L-x) - A_{mL2}A_{mu1}(L-x) + A_{mL1}A_{mu1}(L-x)^2] dx$$

$$D = \frac{1}{EI} \left[ A_{mL2}A_{mu2}x - A_{mL1}A_{mu2} \left( Lx - \frac{x^2}{2} \right) - A_{mL2}A_{mu1} \left( Lx - \frac{x^2}{2} \right) + A_{mL1}A_{mu1} \left( L^2x - 2\frac{x^2}{2}L + \frac{x^3}{3} \right) \right]_0^L$$



# Flexibility Matrix Method

9<sup>th</sup> May, 2020

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$$D = \frac{1}{EI} \left[ A_{mL2} A_{mu2} L - A_{mL1} A_{mu2} \frac{L^2}{2} - A_{mL2} A_{mu1} \frac{L^2}{2} + A_{mL1} A_{mu1} \frac{L^3}{3} \right]$$

*Matrix formation*

$$D = \begin{bmatrix} A_{mu1} & A_{mu2} \end{bmatrix} \begin{bmatrix} L^3/3EI & -L^2/2EI \\ -L^2/2EI & L/EI \end{bmatrix} \begin{bmatrix} A_{mL1} \\ A_{mL2} \end{bmatrix}$$

$$D = [A_{mu}]^T [F_m] [A_{mL}]$$



# Flexibility Matrix Method

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

For Multiple Member

$$D = \sum_{m=1}^n [A_{mu}]^T [F_m] [A_{mL}]$$

$$D = \begin{bmatrix} (A_{mu1}, A_{mu2})_1 & & & & \\ & (A_{mu1}, A_{mu2})_2 & & & \\ & & \dots & & \\ & & & \dots & \\ & & & & (A_{mu1}, A_{mu2})_n \end{bmatrix} \times$$

$$\begin{bmatrix} F_{m1} & & & & \\ & F_{m2} & & & \\ & & \dots & & \\ & & & \dots & \\ & & & & F_{mn} \end{bmatrix} \begin{bmatrix} A_{mL1} \\ A_{mL2} \\ \vdots \\ A_{mLn} \end{bmatrix}$$

## Compatibility Equation

$$[D_{QL}] + [F][Q] = 0$$

$$[Q] = -[F]^{-1}[D_{QL}]$$



## Sequential Procedure

1	Find Degree of Indeterminacy ( <b>DoI</b> )
2	Select Redundant Force ( <b>Q</b> )
3	Replacement of equivalent joint load and find <b>A<sub>mL</sub></b>
4	Calculate <b>A<sub>mu</sub></b>
5	Calculate <b>F<sub>m</sub></b>
6	Calculate <b>D<sub>QL</sub> = A<sub>mu</sub><sup>T</sup> F<sub>m</sub> A<sub>mL</sub></b>
7	Calculate <b>F = A<sub>mu</sub><sup>T</sup> F<sub>m</sub> A<sub>mu</sub></b>
8	Calculate Redundant Force ( <b>Q</b> )
9	Calculate End Shear / End Moment ( <b>A<sub>m</sub></b> )



# Flexibility Matrix-BEAM

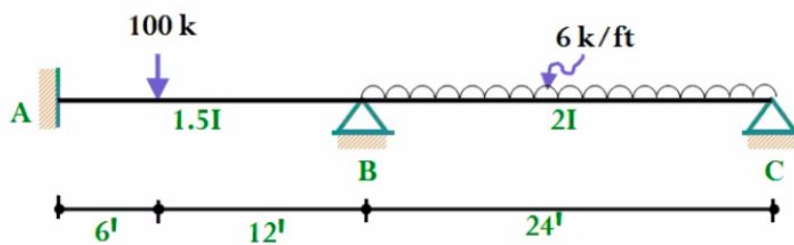
9<sup>th</sup> May, 2020

## PROBLEM

11

Using Flexibility Matrix Method solve the followings, EI is constant

- Determine the Redundant Forces
- Draw Shear Force and Bending Moment Diagrams



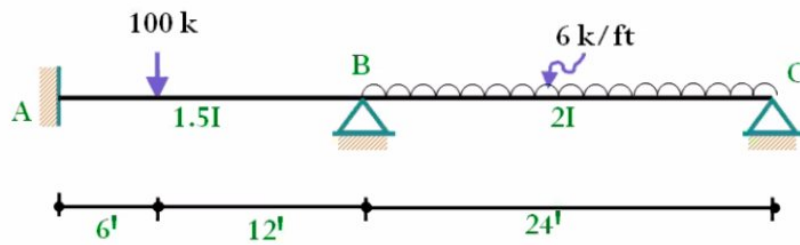


# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

Degree of Support Restrained	$7-3 = 4$
Static Equilibrium Equation	$3-1 = 2$
Degree of Indeterminacy (DoI)	$4-2 = 2$
Let the Redundant Forces	$Q_1$ and $Q_2$



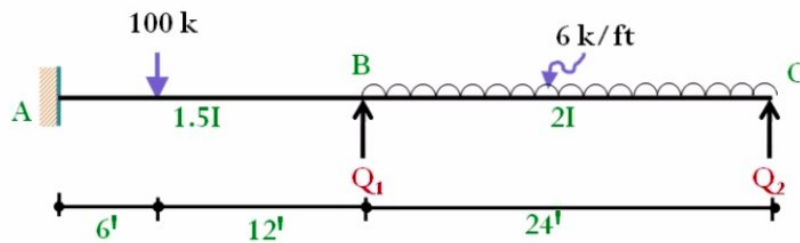


# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

Degree of Support Restrained	$7-3 = 4$
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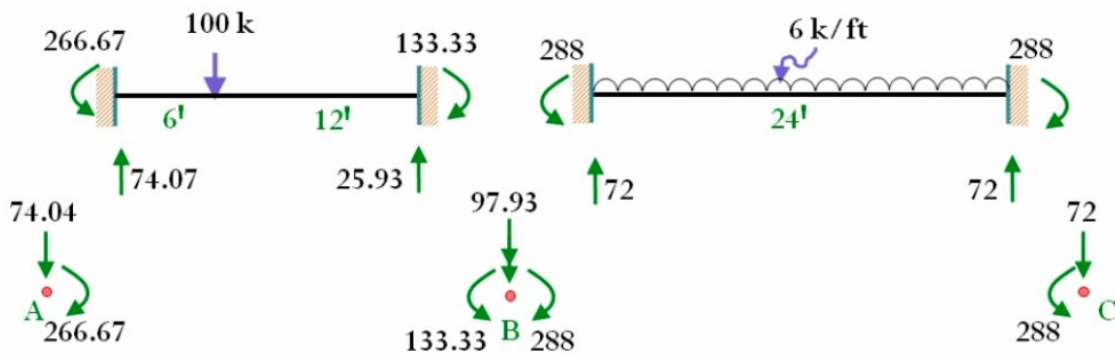


# Flexibility Matrix-BEAM

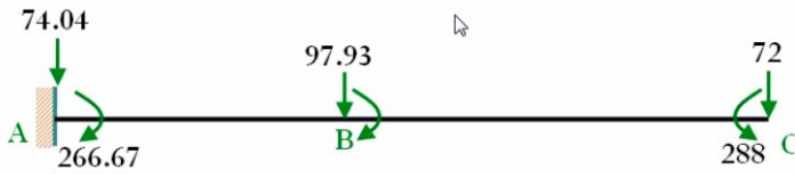
9<sup>th</sup> May, 2020

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## Release Redundant and Taking FEM



## Equivalent Load Transfer



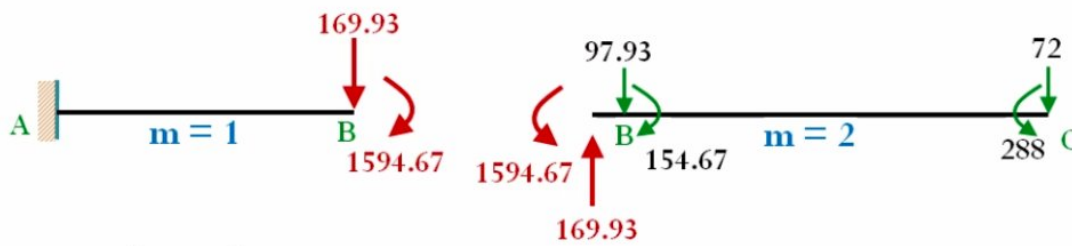
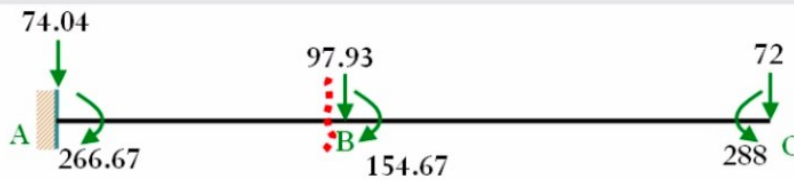


# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Calculation of $A_{mL}$



$$A_{mL} = \begin{bmatrix} \phantom{0} \\ \phantom{0} \end{bmatrix}$$

$$\sum f_y = 0 \quad \sum M_B = 0$$

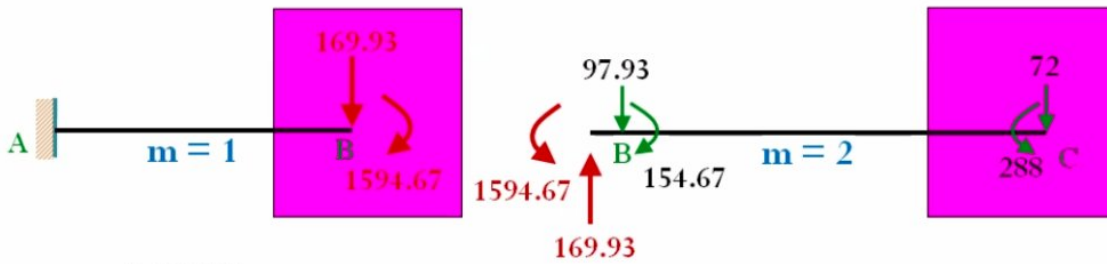
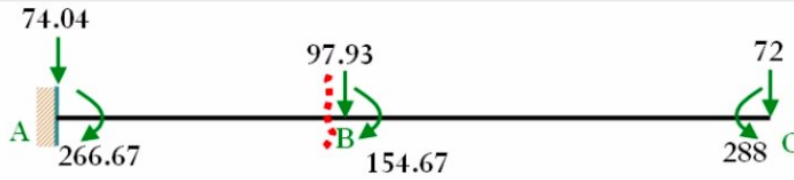


# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

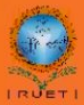
Talking:

## Calculation of $A_{mL}$



$$A_{mL} = \begin{bmatrix} -169.93 \\ 1594.67 \\ -72 \\ -288 \end{bmatrix}$$

$$\sum f_y = 0 \quad \sum M_B = 0$$



# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

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Calculation of  $A_{mu}$

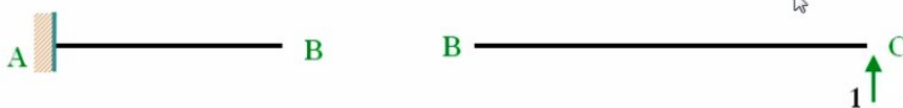
Apply **unit load** in place of  $Q_1$



Apply **unit load** in place of  $Q_2$



$$A_{mu} = \begin{bmatrix} 1 \\ 0 \\ 0 \\ 0 \end{bmatrix}$$





# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

Calculation of  $A_{mu}$

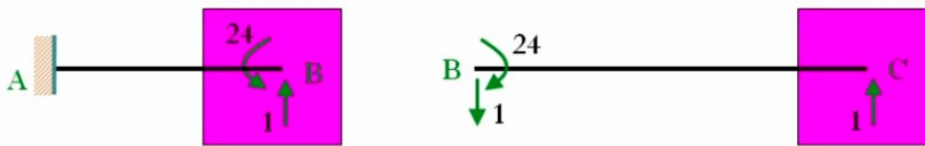
Apply **unit load** in place of  $Q_1$



Apply **unit load** in place of  $Q_2$



$$A_{mu} = \begin{bmatrix} 1 \\ 0 \\ 0 \\ 0 \end{bmatrix}$$





# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Calculation of $F_m$

$$F_m = \frac{1}{EI} \begin{bmatrix} L^3/3 & -L^2/2 \\ -L^2/2 & L \end{bmatrix}$$

$$F_m = \frac{1}{EI} \begin{bmatrix} 18^3/(3 \times 1.5) & -18^2/(2 \times 1.5) & 0 & 0 \\ -18^2/(2 \times 1.5) & 18/(1.5) & 0 & 0 \\ 0 & 0 & 24^3/(3 \times 2) & -24^2/(2 \times 2) \\ 0 & 0 & -24^2/(2 \times 2) & 24/(2) \end{bmatrix}$$

$$F_m = \frac{1}{EI} \begin{bmatrix} 1296 & -108 & 0 & 0 \\ -108 & 12 & 0 & 0 \\ 0 & 0 & 2304 & -144 \\ 0 & 0 & -144 & 12 \end{bmatrix}$$

7



# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

Talking:

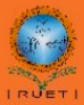
## Calculation of $D_{QL}$

$$D_{QL} = A_{mu}^T F_{m} A_{mL}$$

$$D_{QL} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 1 & -24 & 1 & 0 \end{bmatrix} \frac{1}{EI} \begin{bmatrix} 1296 & -108 & 0 & 0 \\ -108 & 12 & 0 & 0 \\ 0 & 0 & 2304 & -144 \\ 0 & 0 & -144 & 12 \end{bmatrix} \begin{bmatrix} -169.93 \\ 1594.67 \\ -72 \\ -288 \end{bmatrix}$$

$$D_{QL} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 1 & -24 & 1 & 0 \end{bmatrix} \frac{1}{EI} \begin{bmatrix} -392453.64 \\ 37488.48 \\ -124416 \\ 6912 \end{bmatrix}$$

$$\begin{bmatrix} D_{QL1} \\ D_{QL2} \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} -392453.64 \\ -1416593.16 \end{bmatrix}$$



# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Calculation of F

$$F = A_{mu}^T F_m A_{mu}$$

$$F = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 1 & -24 & 1 & 0 \end{bmatrix} \frac{1}{EI} \begin{bmatrix} 1296 & -108 & 0 & 0 \\ -108 & 12 & 0 & 0 \\ 0 & 0 & 2304 & -144 \\ 0 & 0 & -144 & 12 \end{bmatrix} \begin{bmatrix} 1 & 1 \\ 0 & -24 \\ 0 & 1 \\ 0 & 0 \end{bmatrix}$$

$$F = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 1 & -24 & 1 & 0 \end{bmatrix} \frac{1}{EI} \begin{bmatrix} 1296 & 3888 \\ -108 & -396 \\ 0 & 2304 \\ 0 & -144 \end{bmatrix}$$

$$F = \frac{1}{EI} \begin{bmatrix} 1296 & 3888 \\ 3888 & 15696 \end{bmatrix}$$

9



# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Calculation of Redundant Force Q

$$Q = -[F]^{-1} [D_{QL}]$$

$$Q = -\cancel{EI} \begin{bmatrix} 1296 & 3888 \\ 3888 & 15696 \end{bmatrix}^{-1} \frac{1}{\cancel{EI}} \begin{bmatrix} -392453.64 \\ -1416593.16 \end{bmatrix}$$

$$Q = -\frac{1}{5225472} \begin{bmatrix} 15696 & -3888 \\ -3888 & 1296 \end{bmatrix} \begin{bmatrix} -392453.64 \\ -1416593.16 \end{bmatrix}$$

$$\begin{bmatrix} Q_1 \\ Q_2 \end{bmatrix} = \begin{bmatrix} 124.82 \\ 59.33 \end{bmatrix}$$

10



# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Calculation of Shear and Moment $A_m$

$$[A_m] = [A_{mL}] + [A_{mu}]Q + [A_{mR}]$$

$$[A_m] = \begin{bmatrix} -169.93 \\ 1594.67 \\ -72 \\ -288 \end{bmatrix} + \begin{bmatrix} 1 & 1 \\ 0 & -24 \\ 0 & 1 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} 124.82 \\ 59.33 \end{bmatrix} + \begin{bmatrix} 25.93 \\ 133.33 \\ 72 \\ 288 \end{bmatrix}$$

$$[A_m] = \begin{bmatrix} -169.93 \\ 1594.67 \\ -72 \\ -288 \end{bmatrix} + \begin{bmatrix} 184.15 \\ -1423.92 \\ 59.33 \\ 0 \end{bmatrix} + \begin{bmatrix} 25.93 \\ 133.33 \\ 72 \\ 288 \end{bmatrix}$$

$$[A_m] = \begin{bmatrix} 40.75 \\ 304.08 \\ 59.33 \\ 0 \end{bmatrix}$$

Member	1	End	Right	Property	Shear
Member	1	End	Right	Property	Moment
Member	2	End	Right	Property	Shear
Member	2	End	Right	Property	Moment

11



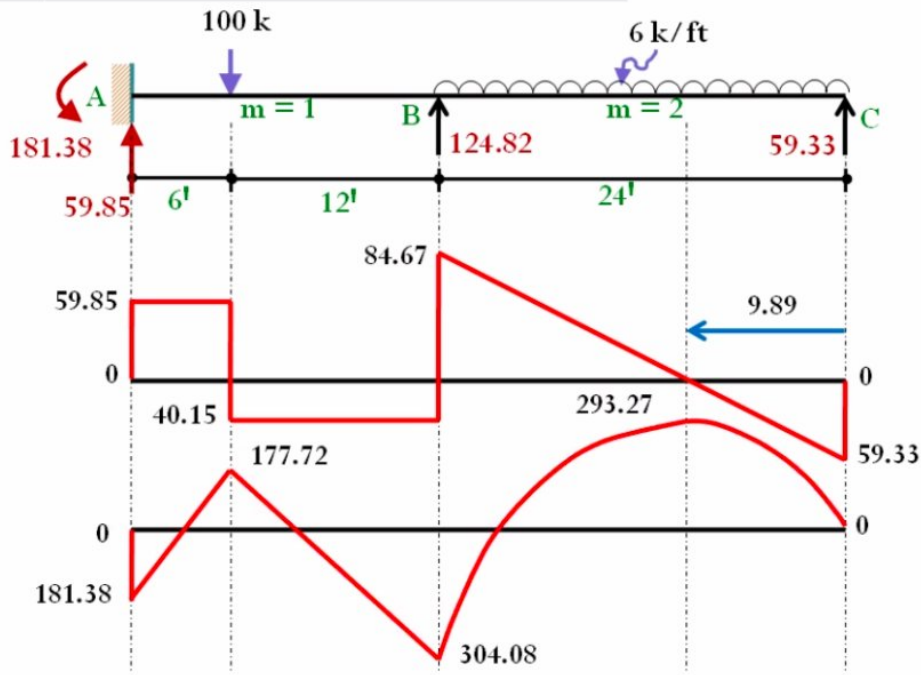
# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

Talking:

## DRAW SFD and BMD

$$\begin{bmatrix} Q_1 \\ Q_2 \end{bmatrix} = \begin{bmatrix} 124.82 \\ 59.33 \end{bmatrix}$$





CE 4111

9<sup>th</sup> May, 2020

# STATICALLY INDETERMINATE STRUCTURE

Dr. Md. Shafiqul ISLAM

Professor, CE, RUET



# Flexibility Matrix-BEAM

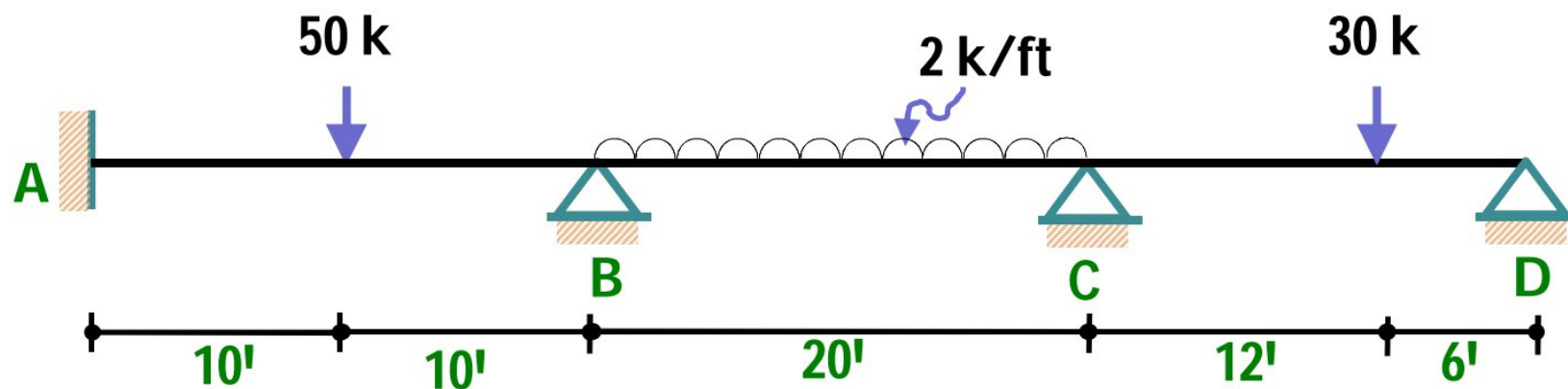
9<sup>th</sup> May, 2020

## PROBLEM

12

Using Flexibility Matrix Method solve the followings, EI is constant

- Determine the Redundant Forces
- Draw Shear Force and Bending Moment Diagrams





# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

Degree of Support Restrained

$$9 - 4 = 5$$

Static Equilibrium Equation

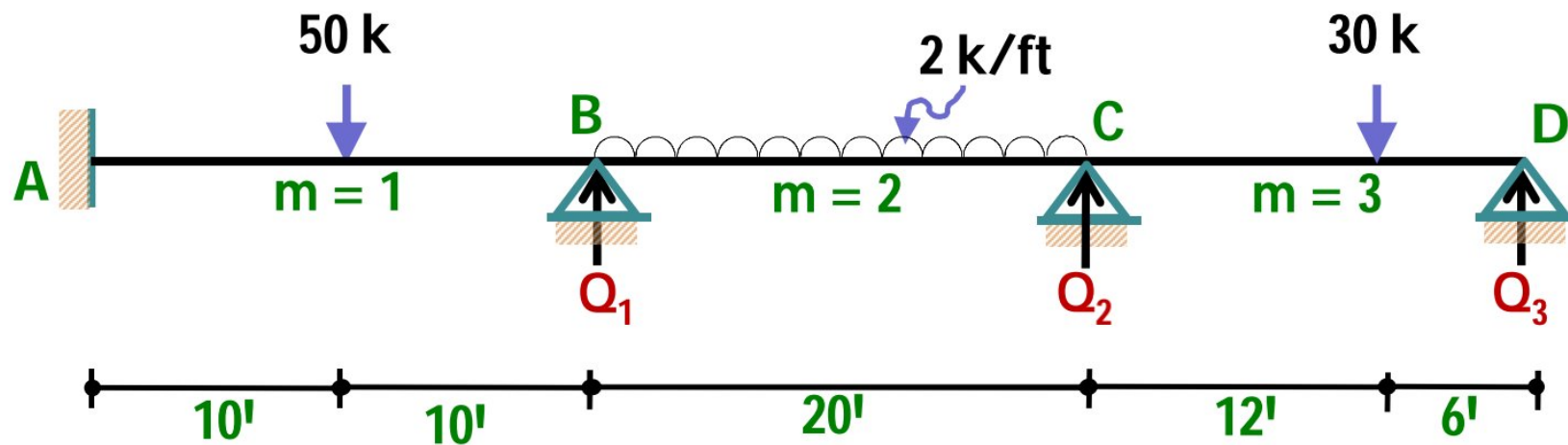
$$3 - 1 = 2$$

Degree of Indeterminacy (DoI)

$$5 - 2 = 3$$

Let the Redundant Forces

$Q_1$ ,  $Q_2$  and  $Q_3$

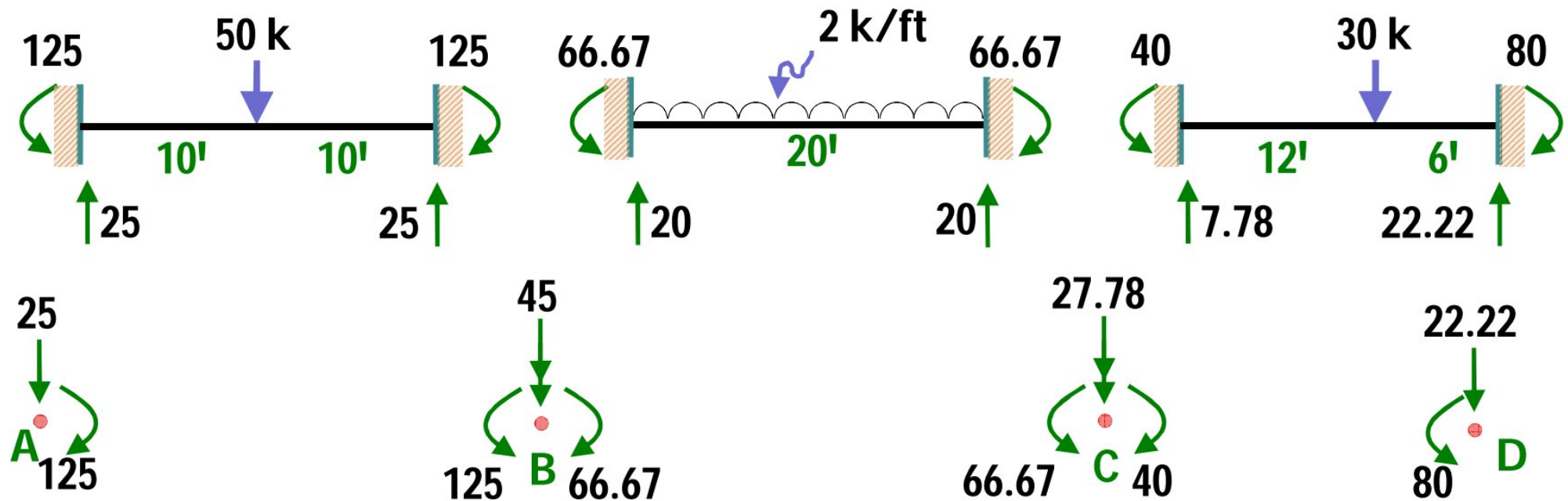




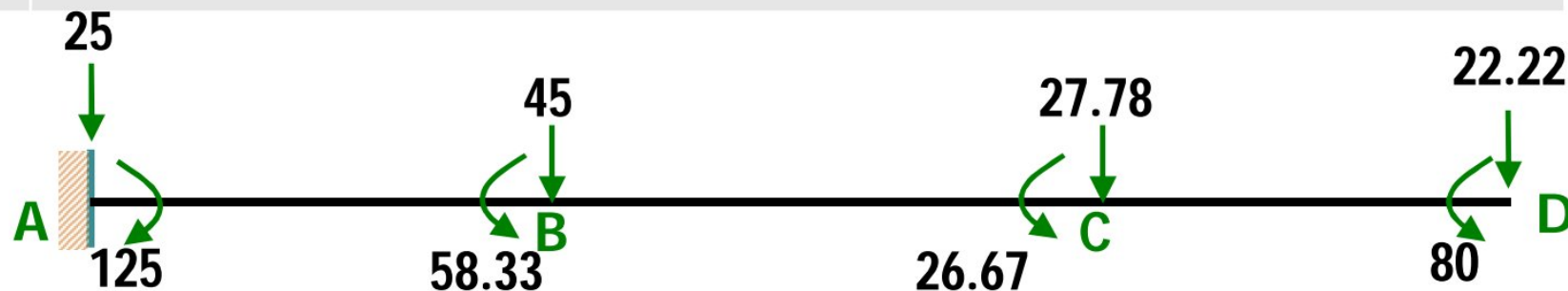
# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

## Release Redundant and Taking FEM



## Equivalent Load Transfer

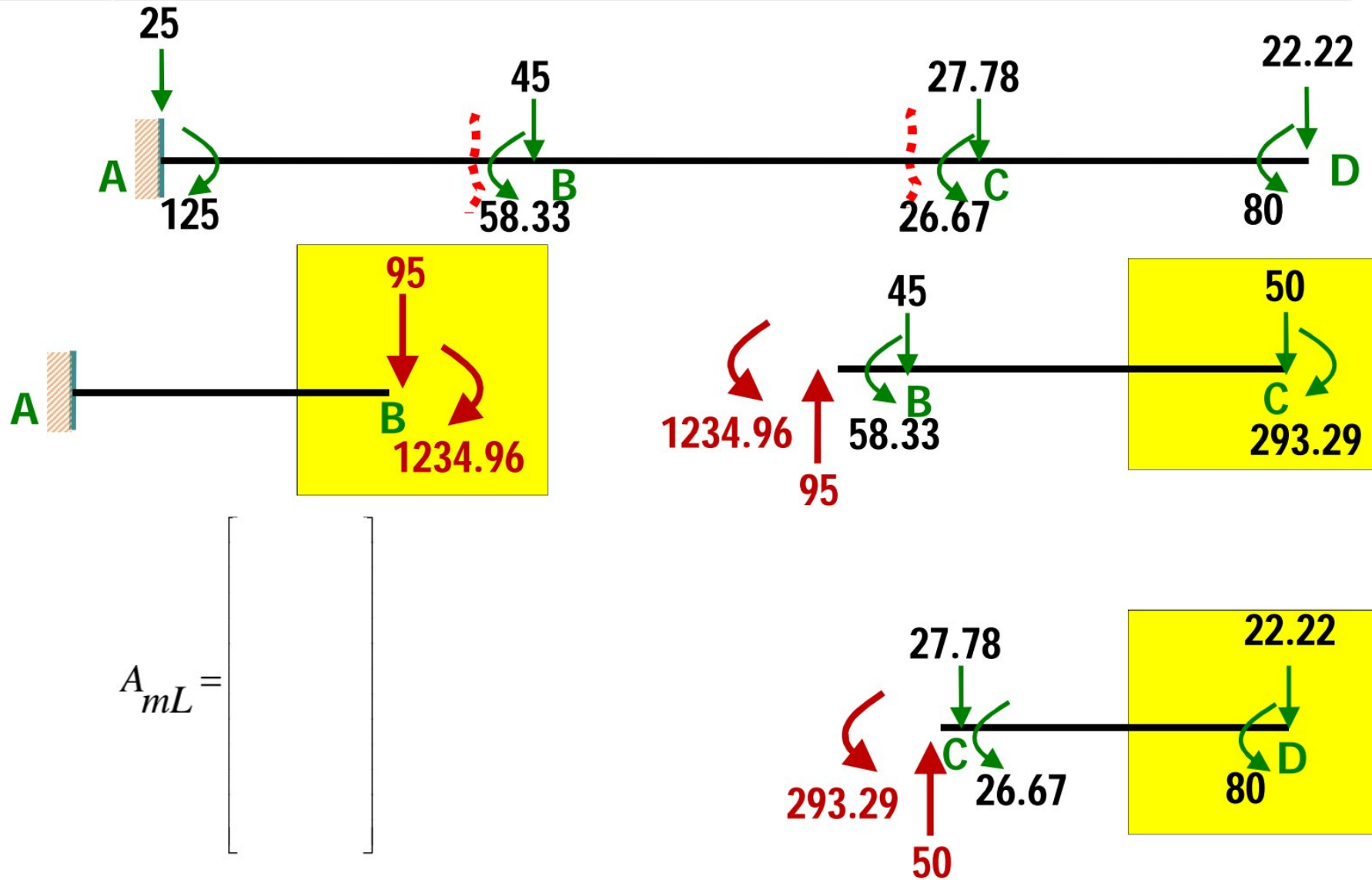




# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

## Calculation of $A_{mL}$



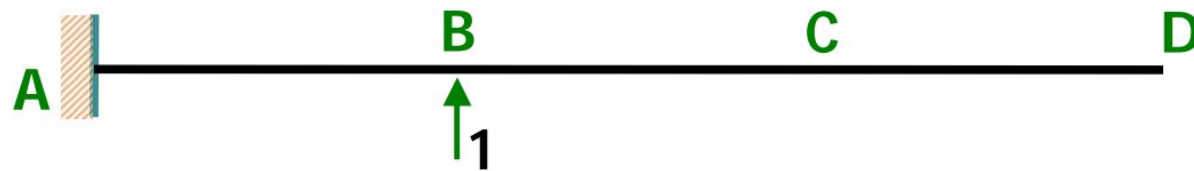


# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

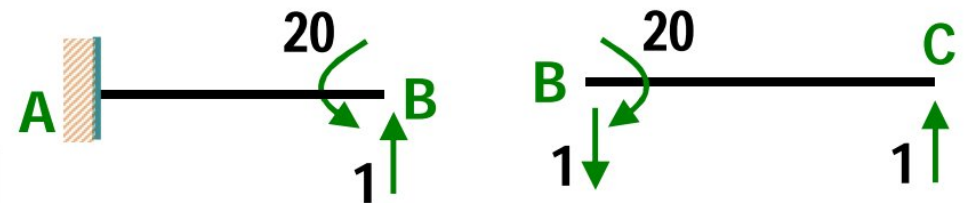
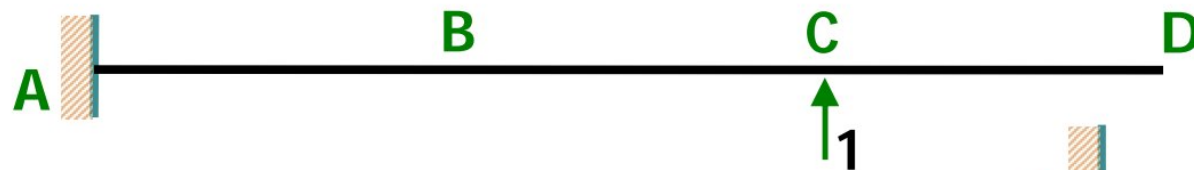
Calculation of  $A_{mu}$

Apply **unit load** in place of  $Q_1$

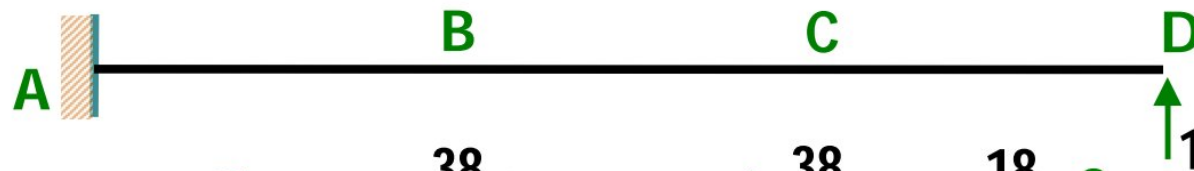


$A_{mu} =$

Apply **unit load** in place of  $Q_2$



Apply **unit load** in place of  $Q_3$





# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

## Calculation of $F_m$

$$F_m = \frac{1}{EI} \begin{bmatrix} L^3/3 & -L^2/2 \\ -L^2/2 & L \end{bmatrix}$$

$$F_m = \frac{1}{EI} \begin{bmatrix} 20^3/3 & -20^2/2 & 0 & 0 & 0 & 0 \\ -20^2/2 & 20 & 0 & 0 & 0 & 0 \\ 0 & 0 & 20^3/3 & -20^2/2 & 0 & 0 \\ 0 & 0 & -20^2/2 & 20 & 0 & 0 \\ 0 & 0 & 0 & 0 & 18^3/3 & -18^2/2 \\ 0 & 0 & 0 & 0 & -18^2/2 & 18 \end{bmatrix}$$

$$F_m = \frac{1}{EI} \begin{bmatrix} 2666.67 & -200 & 0 & 0 & 0 & 0 \\ -200 & 20 & 0 & 0 & 0 & 0 \\ 0 & 0 & 2666.67 & -200 & 0 & 0 \\ 0 & 0 & -200 & 20 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1944 & -162 \\ 0 & 0 & 0 & 0 & -162 & 18 \end{bmatrix}$$



# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

## Calculation of $D_{QL}$

$$D_{QL} = A_{mu}^T F_m A_{mL}$$

$$D_{QL} = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 1 & -20 & 1 & 0 & 0 & 0 \\ 1 & -38 & 1 & -18 & 1 & 0 \end{bmatrix} \frac{1}{EI} \begin{bmatrix} 2666.67 & -200 & 0 & 0 & 0 & 0 \\ -200 & 20 & 0 & 0 & 0 & 0 \\ 0 & 0 & 2666.67 & -200 & 0 & 0 \\ 0 & 0 & -200 & 20 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1944 & -162 \\ 0 & 0 & 0 & 0 & -162 & 18 \end{bmatrix} \begin{bmatrix} -95 \\ 1234.96 \\ -50 \\ 293.29 \\ -22.22 \\ -80 \end{bmatrix}$$

$$D_{QL} = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 1 & -20 & 1 & 0 & 0 & 0 \\ 1 & -38 & 1 & -18 & 1 & 0 \end{bmatrix} \begin{bmatrix} -500325.65 \\ 43699.20 \\ -191991.50 \\ 15865.80 \\ -30235.68 \\ 2159.64 \end{bmatrix}$$

$$\begin{bmatrix} D_{QL1} \\ D_{QL2} \\ D_{QL3} \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} -500325.65 \\ -1566301.15 \\ -2668706.83 \end{bmatrix}$$



# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

## Calculation of F

$$F = A_{mu}^T F_m A_{mu}$$

$$F = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 1 & -20 & 1 & 0 & 0 & 0 \\ 1 & -38 & 1 & -18 & 1 & 0 \end{bmatrix} \frac{1}{EI} \begin{bmatrix} 2666.67 & -200 & 0 & 0 & 0 & 0 \\ -200 & 20 & 0 & 0 & 0 & 0 \\ 0 & 0 & 2666.67 & -200 & 0 & 0 \\ 0 & 0 & -200 & 20 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1944 & -162 \\ 0 & 0 & 0 & 0 & -162 & 18 \end{bmatrix} \begin{bmatrix} 1 & 1 & 1 \\ 0 & -20 & -38 \\ 0 & 1 & 1 \\ 0 & 0 & -18 \\ 0 & 0 & 1 \\ 0 & 0 & 0 \end{bmatrix}$$

$$F = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 1 & -20 & 1 & 0 & 0 & 0 \\ 1 & -38 & 1 & -18 & 1 & 0 \end{bmatrix} \frac{1}{EI} \begin{bmatrix} 2666.67 & 6666.67 & 10266.67 \\ -200 & -600 & -960 \\ 0 & 2666.67 & 6266.67 \\ 0 & -200 & -560 \\ 0 & 0 & 1944 \\ 0 & 0 & -162 \end{bmatrix} F = \frac{1}{EI} \begin{bmatrix} 2666.67 & 6666.67 & 10266.67 \\ 6666.67 & 21333.34 & 35733.34 \\ 10266.67 & 35733.34 & 65037.34 \end{bmatrix}$$



## Calculation of Redundant Force Q

$$Q = -[F]^{-1} [D_{QL}]$$

$$Q = -EI \begin{bmatrix} 2666.67 & 6666.67 & 10266.67 \\ 6666.67 & 21333.34 & 35733.34 \\ 10266.67 & 35733.34 & 65037.34 \end{bmatrix}^{-1} \frac{1}{EI} \begin{bmatrix} -500325.65 \\ -1566301.15 \\ -2668706.83 \end{bmatrix}$$

$$Q = \begin{bmatrix} 0.002341 & -0.001413 & 0.000407 \\ -0.001413 & 0.001440 & -0.000568 \\ 0.000407 & -0.000568 & 0.000263 \end{bmatrix} \begin{bmatrix} 500325.65 \\ 1566301.15 \\ 2668706.83 \end{bmatrix}$$

$$\begin{bmatrix} Q_1 \\ Q_2 \\ Q_3 \end{bmatrix} = \begin{bmatrix} 44.24 \\ 32.69 \\ 15.84 \end{bmatrix}$$



## Calculation of Shear and Moment $A_m$

$$[A_m] = [A_{mL}] + [A_{mu}][Q] + [A_{mR}]$$

$$A_m = \begin{bmatrix} -95 \\ 1234.96 \\ -50 \\ 293.29 \\ -22.22 \\ -80 \end{bmatrix} + \begin{bmatrix} 1 & 1 & 1 \\ 0 & -20 & -38 \\ 0 & 1 & 1 \\ 0 & 0 & -18 \\ 0 & 0 & 1 \\ 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} 44.24 \\ 32.69 \\ 15.84 \end{bmatrix} + \begin{bmatrix} 25 \\ 125 \\ 20 \\ 66.67 \\ 22.22 \\ 80 \end{bmatrix}$$

$$[A_m] = \begin{bmatrix} 22.77 \\ 104.24 \\ 18.53 \\ 74.84 \\ 15.84 \\ 0 \end{bmatrix}$$

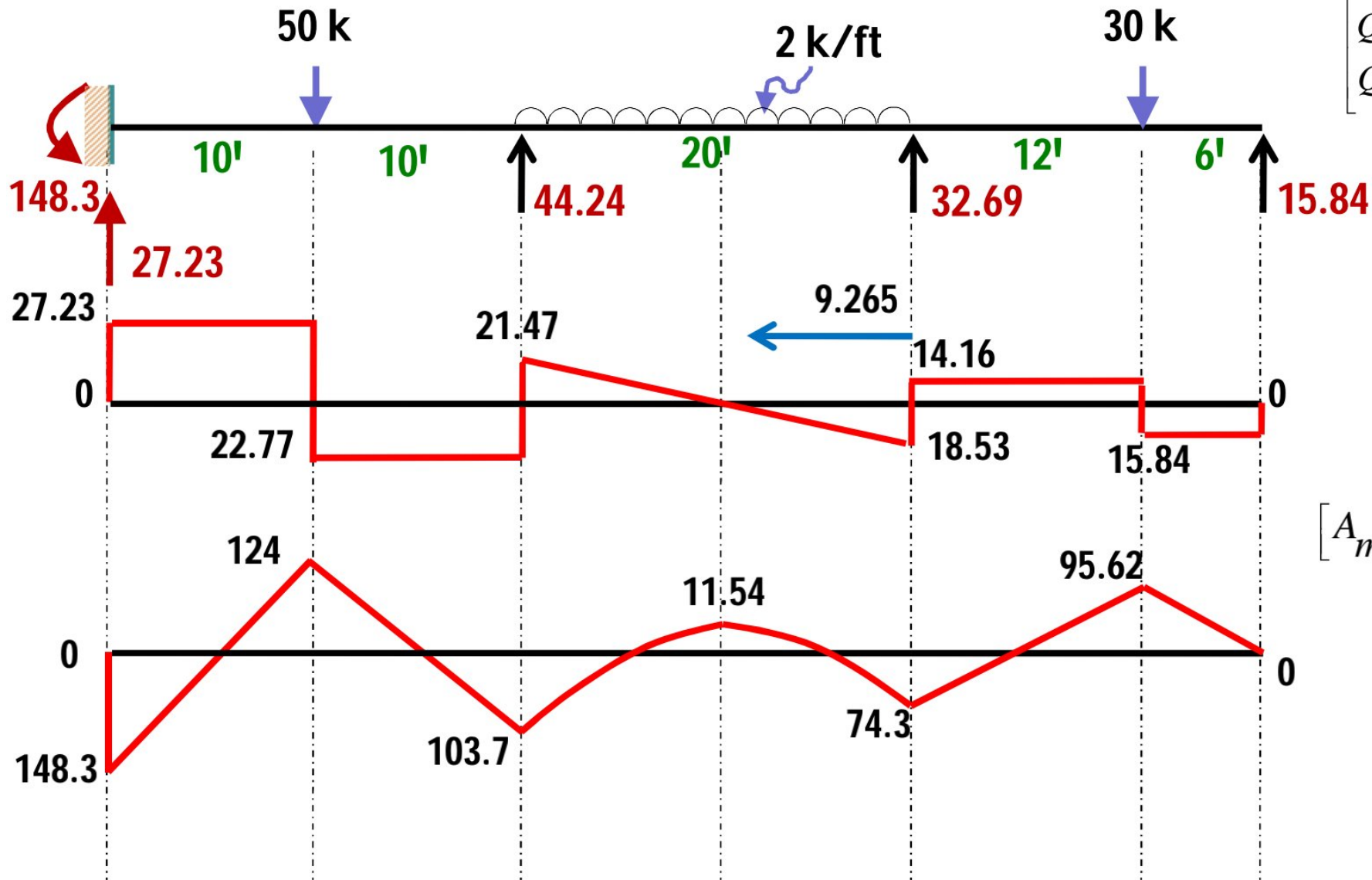
Member	<b>1</b>	End	<b>Right</b>	Property	<b>Shear</b>
Member	<b>1</b>	End	<b>Right</b>	Property	<b>Moment</b>
Member	<b>2</b>	End	<b>Right</b>	Property	<b>Shear</b>
Member	<b>2</b>	End	<b>Right</b>	Property	<b>Moment</b>
Member	<b>3</b>	End	<b>Right</b>	Property	<b>Shear</b>
Member	<b>3</b>	End	<b>Right</b>	Property	<b>Moment</b>



# Flexibility Matrix-BEAM

9<sup>th</sup> May, 2020

DRAW SFD and BMD



$$\begin{bmatrix} Q_1 \\ Q_2 \\ Q_3 \end{bmatrix} = \begin{bmatrix} 44.24 \\ 32.69 \\ 15.84 \end{bmatrix}$$

$$[A_m] = \begin{bmatrix} 22.77 \\ 104.24 \\ 18.53 \\ 74.84 \\ 15.84 \\ 0 \end{bmatrix}$$



**Students/CE/  
RUET**



**Professor/CE/  
RUET**



## Flexibility Matrix-FRAME

9<sup>th</sup> May, 2020

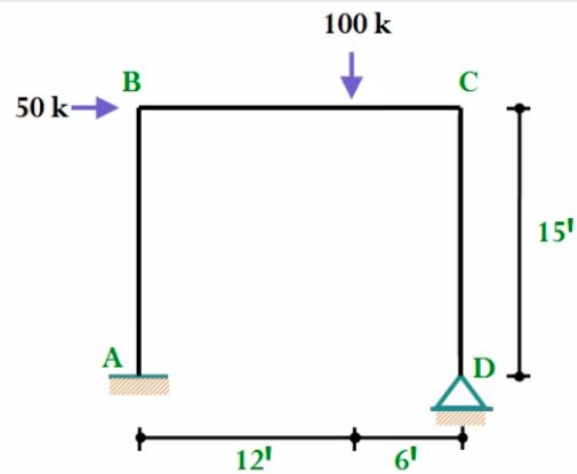
Talking:

### PROBLEM

13

Using Flexibility Matrix Method solve the followings, EI is constant

- Determine the Redundant Forces
- Find Shear Force and Bending Moments



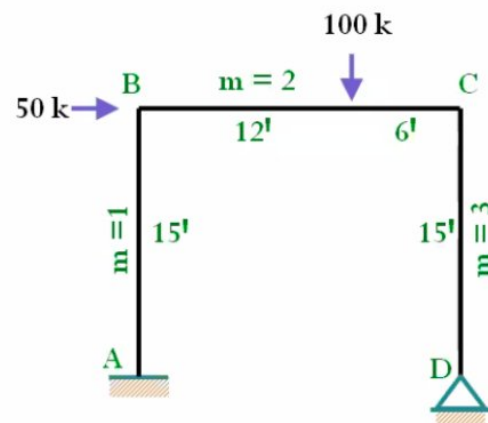
2



# Flexibility Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM





# Flexibility Matrix-FRAME

9<sup>th</sup> May, 2020

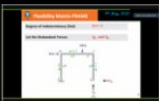
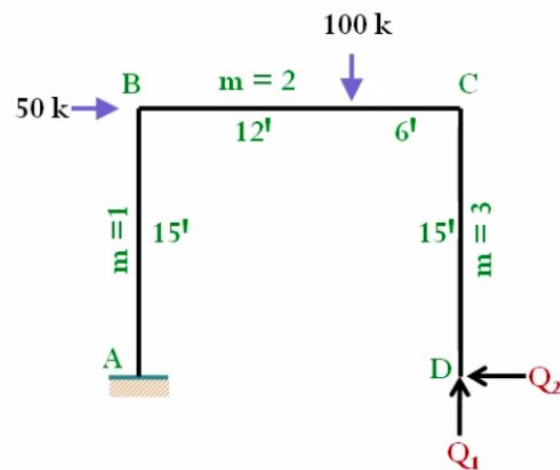
Talking: Dr. Md. Shafiqul ISLAM

Degree of Indeterminacy (DoI)

$$5 - 3 = 2$$

Let the Redundant Forces

$Q_1$  and  $Q_2$



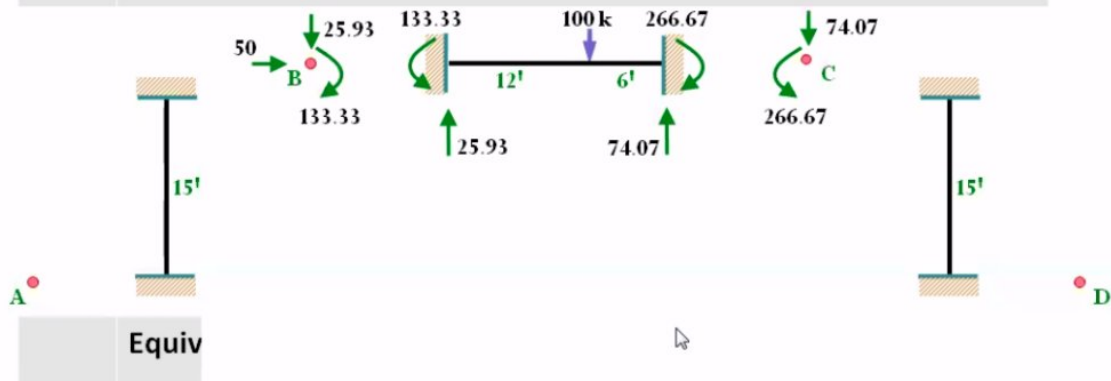


# Flexibility Matrix-FRAME

9<sup>th</sup> May, 2020

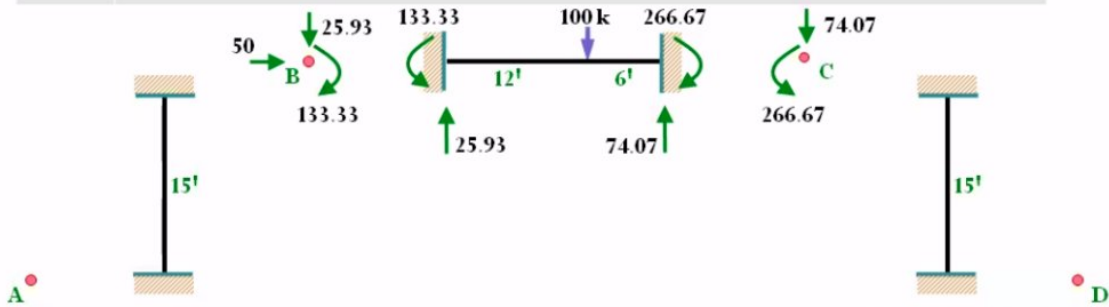
Talking: Dr. Md. Shafiqul ISLAM

## Release Redundant and Taking FEM

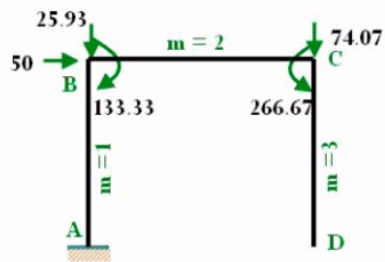




## Release Redundant and Taking FEM

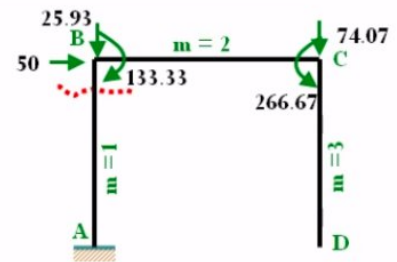
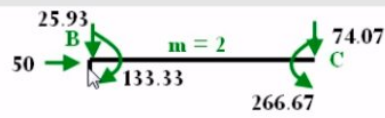


## Equivalent Load Transfer





## Calculation of $A_{mL}$



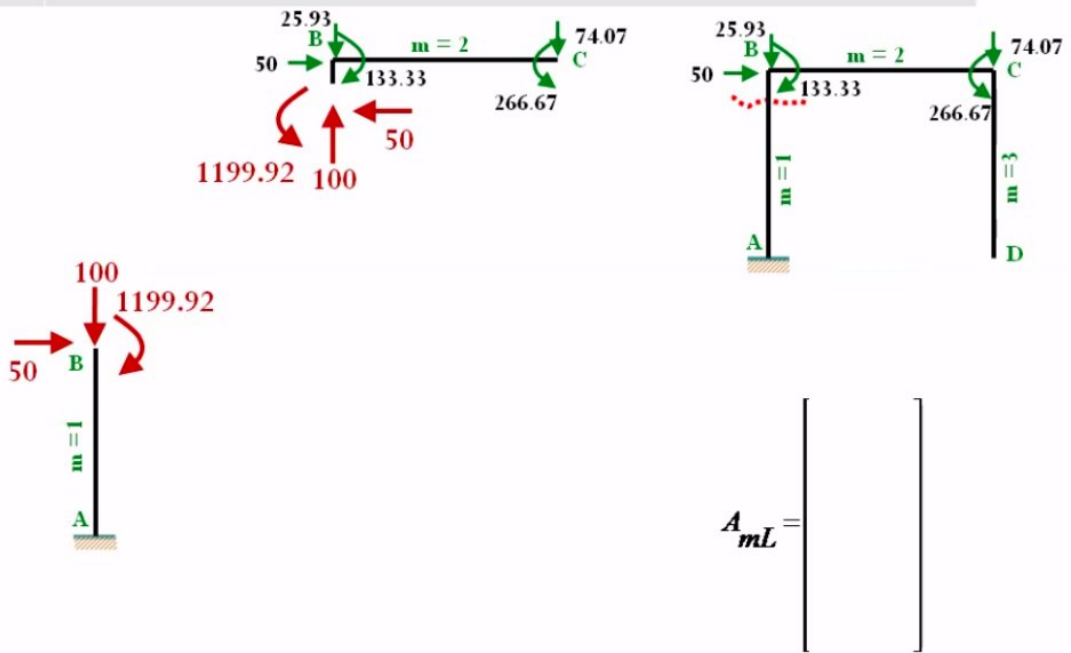


# Flexibility Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Calculation of $A_{mL}$



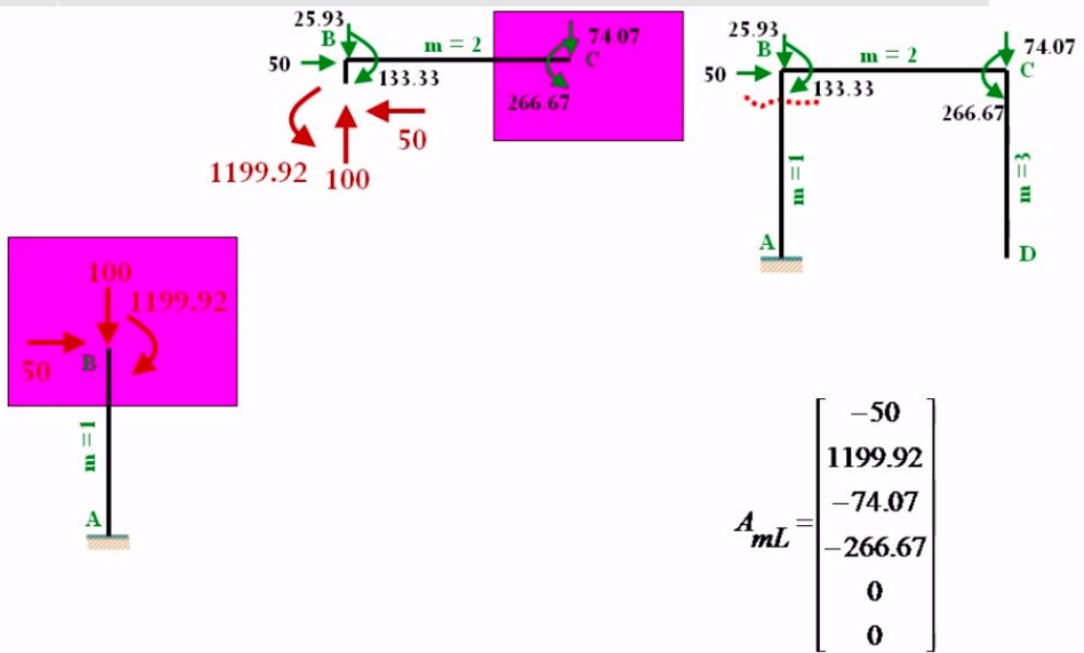


# Flexibility Matrix-FRAME

9<sup>th</sup> May, 2020

Talking:

## Calculation of $A_{mL}$





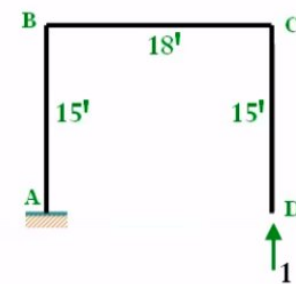
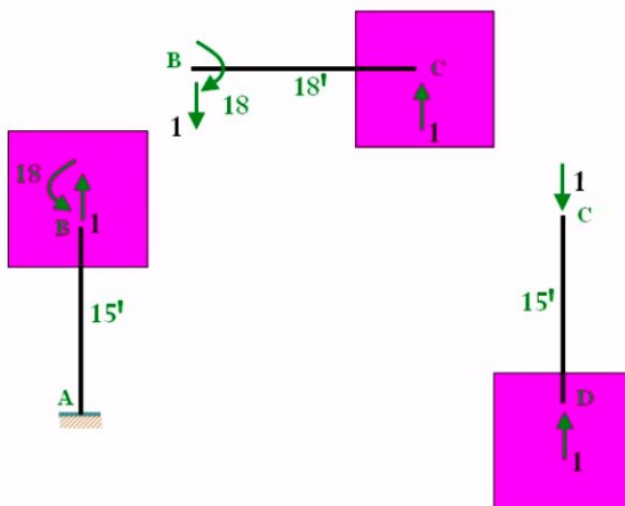
# Flexibility Matrix-FRAME

9<sup>th</sup> May, 2020

Talking:

Calculation of  $A_{mu}$

Apply **unit load** in place of  $Q_i$



$$A_{mu} = \begin{bmatrix} 0 \\ -18 \\ 1 \\ 0 \\ 0 \\ 0 \end{bmatrix}$$



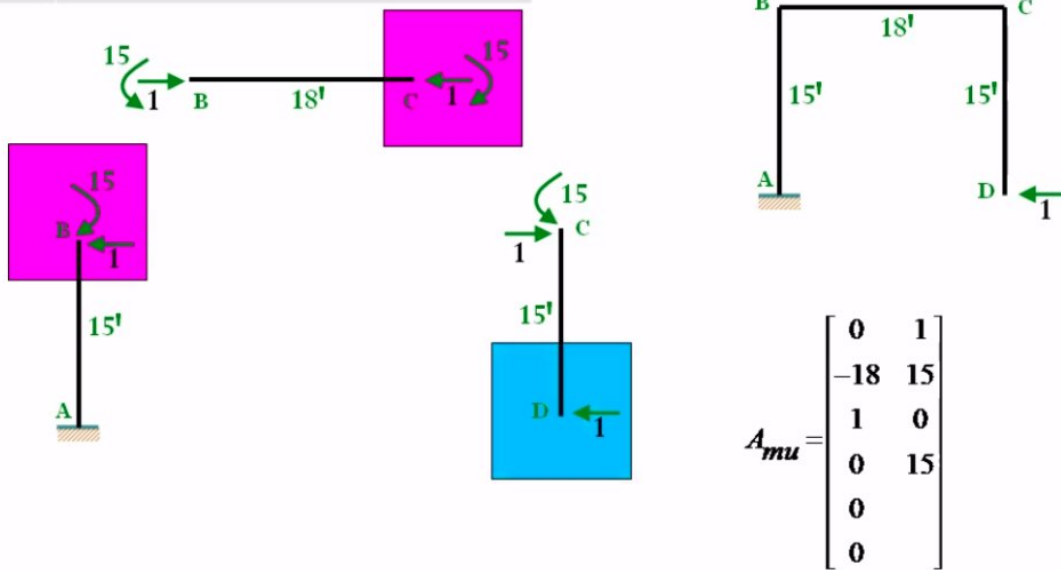
# Flexibility Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

Calculation of  $A_{mu}$

Apply **unit load** in place of  $Q_2$





# Flexibility Matrix-FRAME

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Calculation of $F_m$

$$F_m = \frac{1}{EI} \begin{bmatrix} L^3/3 & -L^2/2 \\ -L^2/2 & L \end{bmatrix}$$

$$F_m = \frac{1}{EI} \begin{bmatrix} 15^3/3 & -15^2/2 & 0 & 0 & 0 & 0 \\ -15^2/2 & 15 & 0 & 0 & 0 & 0 \\ 0 & 0 & 18^3/3 & -18^2/2 & 0 & 0 \\ 0 & 0 & -18^2/2 & 18 & 0 & 0 \\ 0 & 0 & 0 & 0 & 15^3/3 & -15^2/2 \\ 0 & 0 & 0 & 0 & -15^2/2 & 15 \end{bmatrix}$$

$$F_m = \frac{1}{EI} \begin{bmatrix} 1125 & -112.5 & 0 & 0 & 0 & 0 \\ -112.5 & 15 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1944 & -162 & 0 & 0 \\ 0 & 0 & -162 & 18 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1125 & -112.5 \\ 0 & 0 & 0 & 0 & -112.5 & 15 \end{bmatrix}$$





## Calculation of $D_{QL}$

$$D_{QL} = A_{mu}^T F_m A_{mL}$$

$$D_{QL} = \begin{bmatrix} 0 & -18 & 1 & 0 & 0 & 0 \\ 1 & 15 & 0 & 15 & 1 & 0 \end{bmatrix} \frac{1}{EI} \begin{bmatrix} 1125 & -112.5 & 0 & 0 & 0 & 0 \\ -112.5 & 15 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1944 & -162 & 0 & 0 \\ 0 & 0 & -162 & 18 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1125 & -112.5 \\ 0 & 0 & 0 & 0 & -112.5 & 15 \end{bmatrix} \begin{bmatrix} -50 \\ 1199.92 \\ -74.07 \\ -266.67 \\ 0 \\ 0 \end{bmatrix}$$





## Calculation of $D_{QL}$

$$D_{QL} = A_{mu}^T F_m A_{mL}$$

$$D_{QL} = \begin{bmatrix} 0 & -18 & 1 & 0 & 0 & 0 \\ 1 & 15 & 0 & 15 & 1 & 0 \end{bmatrix} \frac{1}{EI} \begin{bmatrix} 1125 & -112.5 & 0 & 0 & 0 & 0 \\ -112.5 & 15 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1944 & -162 & 0 & 0 \\ 0 & 0 & -162 & 18 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1125 & -112.5 \\ 0 & 0 & 0 & 0 & -112.5 & 15 \end{bmatrix} \begin{bmatrix} -50 \\ 1199.92 \\ -74.07 \\ -266.67 \\ 0 \\ 0 \end{bmatrix}$$

$$D_{QL} = \begin{bmatrix} 0 & -18 & 1 & 0 & 0 & 0 \\ 1 & 15 & 0 & 15 & 1 & 0 \end{bmatrix} \frac{1}{EI} \begin{bmatrix} -191241 \\ 23623.8 \\ -100791.54 \\ 7199.28 \\ 0 \\ 0 \end{bmatrix}$$

$$\begin{bmatrix} D_{QL1} \\ D_{QL2} \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} -526019.94 \\ 271105.2 \end{bmatrix}$$



## Calculation of Redundant Force Q

$$Q = -[F]^{-1} [D_{QL}]$$

$$Q = -\cancel{I} \begin{bmatrix} 6804 & -4455 \\ -4455 & 6300 \end{bmatrix}^{-1} \frac{1}{\cancel{I}} \begin{bmatrix} -526019.94 \\ 271105.2 \end{bmatrix}$$

$$Q = -\frac{1}{23018175} \begin{bmatrix} 6300 & 4455 \\ 4455 & 6804 \end{bmatrix} \begin{bmatrix} -526019.94 \\ 271105.2 \end{bmatrix}$$

$$\begin{bmatrix} Q_1 \\ Q_2 \end{bmatrix} = \begin{bmatrix} 91.50 \\ 21.67 \end{bmatrix}$$

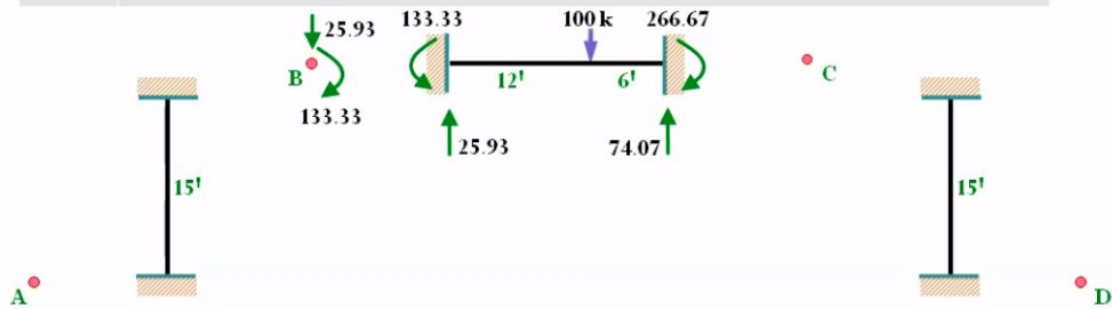


## Calculation of Shear and Moment $A_m$

$$[A_m] = [A_{mL}] + [A_{mu}]Q + [A_{mR}]$$
$$[A_m] = \begin{bmatrix} -50 \\ 1199.92 \\ -74.07 \\ -266.67 \\ 0 \\ 0 \end{bmatrix} + \begin{bmatrix} 0 & 1 \\ -18 & 15 \\ 1 & 0 \\ 0 & 15 \\ 0 & 1 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} 91.50 \\ 21.67 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 74.07 \\ 266.67 \\ 0 \\ 0 \end{bmatrix}$$



## Release Redundant and Taking FEM





## Calculation of Shear and Moment $A_m$

$$[A_m] = [A_{mL}] + [A_{mu}]Q + [A_{mR}]$$

$$[A_m] = \begin{bmatrix} -50 \\ 1199.92 \\ -74.07 \\ -266.67 \\ 0 \\ 0 \end{bmatrix} + \begin{bmatrix} 0 & 1 \\ -18 & 15 \\ 1 & 0 \\ 0 & 15 \\ 0 & 1 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} 91.50 \\ 21.67 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 74.07 \\ 266.67 \\ 0 \\ 0 \end{bmatrix}$$

$$[A_m] = \begin{bmatrix} -50 \\ 1199.92 \\ -74.07 \\ -266.67 \\ 0 \\ 0 \end{bmatrix} + \begin{bmatrix} 21.67 \\ -1321.95 \\ 91.50 \\ 325.05 \\ 21.67 \\ 0 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 74.07 \\ 266.67 \\ 0 \\ 0 \end{bmatrix}$$

$[A_m] =$	$-28.33$	Member	1	End	Right	Property	Shear
	$-122.03$	Member	1	End	Right	Property	Moment
	$91.50$	Member	2	End	Right	Property	Shear
	$325.05$	Member	2	End	Right	Property	Moment
	$21.67$	Member	3	End	Right	Property	Shear
	$0$	Member	3	End	Right	Property	Moment



# Flexibility Matrix-TRUSS

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## FORMULATION

$$\Delta = \sum \frac{SuL}{AE}$$

Member Force due to Applied Load	S	$A_{mL}$
Member Force due to Unit Load	u	$A_{mu}$
Member Properties	L/AE	$F_m$

$$D_{QL} = \sum A_{mu}^T F_m A_{mL}$$



# Flexibility Matrix-TRUSS

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## FORMULATION

$$\Delta = \sum \frac{SuL}{AE}$$

Member Force due to Applied Load  
Member Force due to Unit Load  
Member Properties

S  
u  
L/AE

$A_{mL}$   
 $A_{mu}$   
 $F_m$

$$D_{QL} = \sum A_{mu}^T F_m A_{mL}$$

.....

.....

$$[A_m] = [A_{mL}] + [A_{mu}] [Q]$$

2



## Flexibility Matrix-TRUSS

9<sup>th</sup> May, 2020

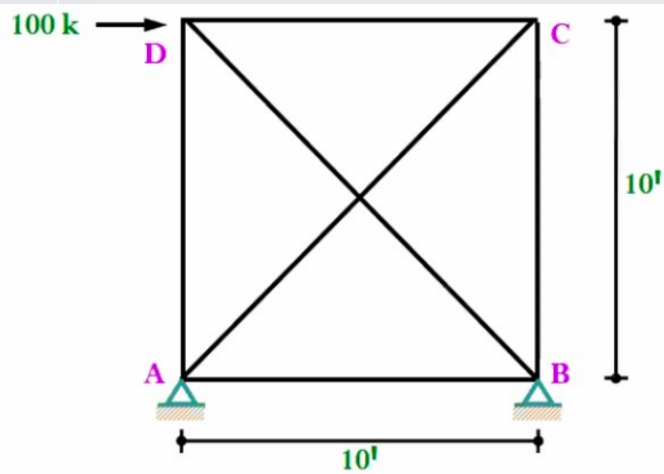
Talking: Dr. Md. Shafiqul ISLAM

### PROBLEM

15

Using Flexibility Matrix Method solve the followings, EA is constant

- Determine the Redundant Forces
- Find out Member Forces



3



# Flexibility Matrix-TRUSS

9<sup>th</sup> May, 2020

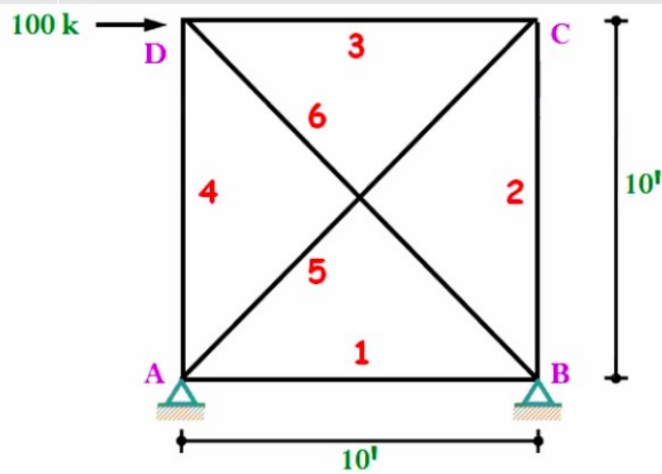
Talking: Dr. Md. Shafiqul ISLAM

## PROBLEM

15

Using Flexibility Matrix Method solve the followings, EA is constant

- Determine the Redundant Forces
- Find out Member Forces



3

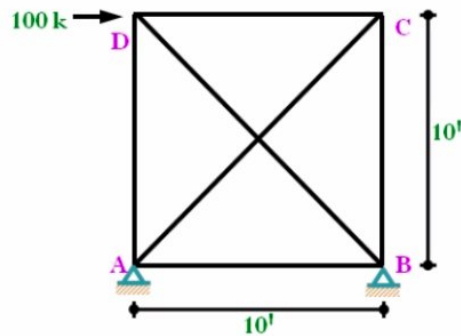


# Flexibility Matrix-TRUSS

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

Degree of Internal Indeterminacy (DoI)	$= [m-(2j-3)]$
	$= [6-(2 \times 4-3)] = 1$
Degree of External Indeterminacy (DoI)	$= r-3$
	$= 4-3 = 1$



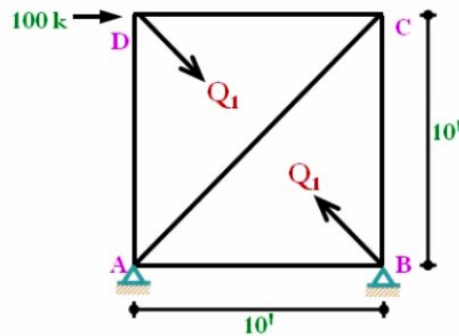


# Flexibility Matrix-TRUSS

9<sup>th</sup> May, 2020

Talking:

Degree of Internal Indeterminacy (DoI)	$= [m-(2j-3)]$
	$= [6-(2 \times 4-3)] = 1$
Degree of External Indeterminacy (DoI)	$= r-3$
	$= 4-3 = 1$
Let the Redundant Forces	$Q_1$ and $Q_2$



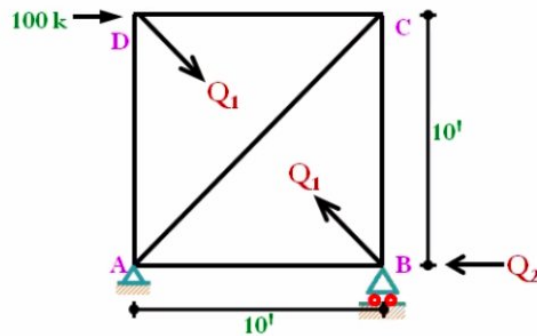


# Flexibility Matrix-TRUSS

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

Degree of Internal Indeterminacy (DoI)	$= [m-(2j-3)]$
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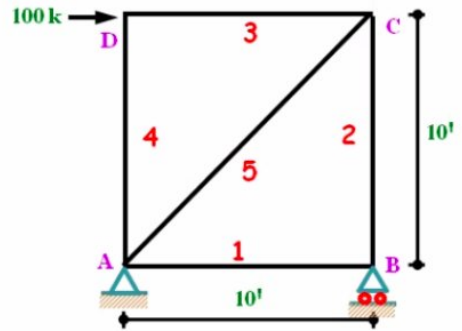


# Flexibility Matrix-TRUSS

9<sup>th</sup> May, 2020

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## Release Redundant and calculation of $A_{mL}$





# Flexibility Matrix-TRUSS

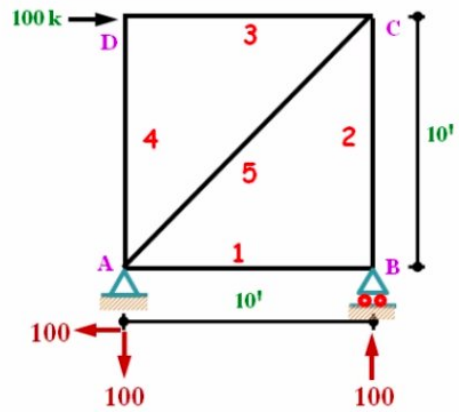
9<sup>th</sup> May, 2020

Talking:

## Release Redundant and calculation of $A_{mL}$

$$\sum M_A = 0 \quad R_B = 100 \quad \sum f_y = 0 \quad R_A = 100$$

$$\sum f_x = 0 \quad H_A = 100$$





# Flexibility Matrix-TRUSS

9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Release Redundant and calculation of $A_{mL}$

$$\Sigma M_A = 0 \quad R_B = 100 \quad \Sigma f_y = 0 \quad R_A = 100$$

$$\Sigma f_x = 0 \quad H_A = 100$$

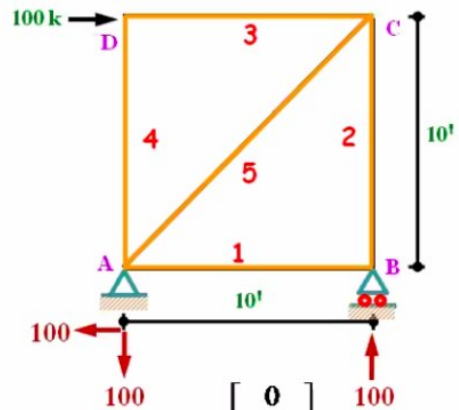
$$\Sigma f_{xB} = 0 \quad AB = 0$$

$$\Sigma f_{yB} = 0 \quad BC = -100$$

$$\Sigma f_{xD} = 0 \quad CD = -100$$

$$\Sigma f_{yD} = 0 \quad AD = 0$$

$$\Sigma f_{yC} = 0 \quad AC = \frac{100 \times 14.14}{10} = 141.4$$



$$A_{mL} = \begin{bmatrix} 0 \\ -100 \\ -100 \\ 0 \\ 141.4 \\ 0 \end{bmatrix}$$



# Flexibility Matrix-TRUSS

9<sup>th</sup> May, 2020

Talking:

## Calculation of $A_{mu}$

Apply **unit load** in place of  $Q_1$

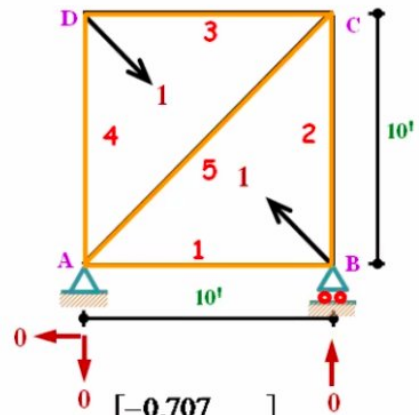
$$\Sigma f_{xB} = 0 \quad AB = -\frac{1 \times 10}{14.14} = -0.707$$

$$\Sigma f_{yB} = 0 \quad BC = -\frac{1 \times 10}{14.14} = -0.707$$

$$\Sigma f_{xD} = 0 \quad CD = -\frac{1 \times 10}{14.14} = -0.707$$

$$\Sigma f_{yD} = 0 \quad AD = -\frac{1 \times 10}{14.14} = -0.707$$

$$\Sigma f_{yC} = 0 \quad AC = \frac{0.707 \times 14.14}{10} = 1$$



$$A_{mu} = \begin{bmatrix} -0.707 \\ -0.707 \\ -0.707 \\ -0.707 \\ 1 \\ 1 \end{bmatrix}$$



# Flexibility Matrix-TRUSS

9<sup>th</sup> May, 2020

Talking:

## Calculation of $A_{mu}$

Apply **unit load** in place of  $Q_u$

$$\sum M_A = 0 \quad R_B = 0 \quad \sum f_y = 0 \quad R_A = 0$$

$$\sum f_x = 0 \quad H_A = 1$$

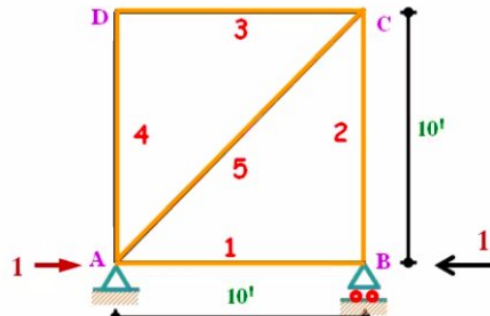
$$\sum f_{xB} = 0 \quad AB = -1$$

$$\sum f_{yB} = 0 \quad BC = 0$$

$$\sum f_{xD} = 0 \quad CD = 0$$

$$\sum f_{yD} = 0 \quad AD = 0$$

$$\sum f_{yC} = 0 \quad AC = \frac{0 \times 14.14}{10} = 0$$



$$A_{mu} = \begin{bmatrix} -0.707 \\ -0.707 \\ -0.707 \\ -0.707 \\ 1 \\ 1 \end{bmatrix}$$



# Flexibility Matrix-TRUSS

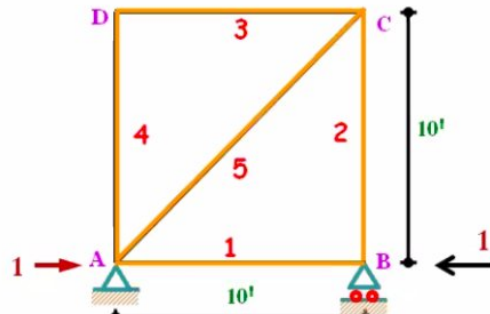
9<sup>th</sup> May, 2020

Talking: Dr. Md. Shafiqul ISLAM

## Calculation of $A_{mu}$

Apply **unit load** in place of  $Q_u$

$$\begin{aligned} \Sigma M_A = 0 & \quad R_B = 0 & \quad \Sigma f_y = 0 & \quad R_A = 0 \\ \Sigma f_x = 0 & \quad H_A = 1 & & \\ \Sigma f_{xB} = 0 & & AB = -1 & \\ \Sigma f_{yB} = 0 & & BC = 0 & \\ \Sigma f_{xD} = 0 & & CD = 0 & \\ \Sigma f_{yD} = 0 & & AD = 0 & \\ \Sigma f_{yC} = 0 & & AC = \frac{0 \times 14.14}{10} = 0 & \end{aligned}$$



$$A_{mu} = \begin{bmatrix} -0.707 & -1 \\ -0.707 & 0 \\ -0.707 & 0 \\ -0.707 & 0 \\ 1 & 0 \\ 1 & 0 \end{bmatrix}$$





# Flexibility Matrix-TRUSS

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## Calculation of $F_m$

$$F_m = \frac{L}{AE}$$

$$F_m = \frac{1}{AE} \begin{bmatrix} 10 & 0 & 0 & 0 & 0 & 0 \\ 0 & 10 & 0 & 0 & 0 & 0 \\ 0 & 0 & 10 & 0 & 0 & 0 \\ 0 & 0 & 0 & 10 & 0 & 0 \\ 0 & 0 & 0 & 0 & 14.14 & 0 \\ 0 & 0 & 0 & 0 & 0 & 14.14 \end{bmatrix}$$



# Flexibility Matrix-TRUSS

9<sup>th</sup> May, 2020

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## Calculation of $D_{QL}$

$$D_{QL} = A_{mu}^T F_m A_{mL}$$

$$D_{QL} = \begin{bmatrix} -0.707 & -0.707 & -0.707 & -0.707 & 1 & 1 \\ -1 & 0 & 0 & 0 & 0 & 0 \end{bmatrix} \frac{1}{AE} \begin{bmatrix} 10 & 0 & 0 & 0 & 0 & 0 \\ 0 & 10 & 0 & 0 & 0 & 0 \\ 0 & 0 & 10 & 0 & 0 & 0 \\ 0 & 0 & 0 & 10 & 0 & 0 \\ 0 & 0 & 0 & 0 & 14.14 & 0 \\ 0 & 0 & 0 & 0 & 0 & 14.14 \end{bmatrix} \begin{bmatrix} 0 \\ -100 \\ -100 \\ 0 \\ 141.4 \\ 0 \end{bmatrix}$$



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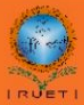
## Calculation of $D_{QL}$

$$D_{QL} = A_{mu}^T F_m A_{mL}$$

$$D_{QL} = \begin{bmatrix} -0.707 & -0.707 & -0.707 & -0.707 & 1 & 1 \\ -1 & 0 & 0 & 0 & 0 & 0 \end{bmatrix} \frac{1}{AE} \begin{bmatrix} 10 & 0 & 0 & 0 & 0 & 0 \\ 0 & 10 & 0 & 0 & 0 & 0 \\ 0 & 0 & 10 & 0 & 0 & 0 \\ 0 & 0 & 0 & 10 & 0 & 0 \\ 0 & 0 & 0 & 0 & 14.14 & 0 \\ 0 & 0 & 0 & 0 & 0 & 14.14 \end{bmatrix} \begin{bmatrix} 0 \\ -100 \\ -100 \\ 0 \\ 141.4 \\ 0 \end{bmatrix}$$

$$D_{QL} = \begin{bmatrix} -0.707 & -0.707 & -0.707 & -0.707 & 1 & 1 \\ -1 & 0 & 0 & 0 & 0 & 0 \end{bmatrix} \frac{1}{AE} \begin{bmatrix} 0 \\ -1000 \\ -1000 \\ 0 \\ 1999.40 \\ 0 \end{bmatrix}$$

$$\begin{bmatrix} D_{QL1} \\ D_{QL2} \end{bmatrix} = \frac{1}{AE} \begin{bmatrix} 3413.4 \\ 0 \end{bmatrix}$$



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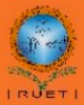
## Calculation of F

$$F = A_{mu}^T F_m A_{mu}$$

$$F = \begin{bmatrix} -0.707 & -0.707 & -0.707 & -0.707 & 1 & 1 \\ -1 & 0 & 0 & 0 & 0 & 0 \end{bmatrix} \frac{1}{AE} \begin{bmatrix} 10 & 0 & 0 & 0 & 0 & 0 \\ 0 & 10 & 0 & 0 & 0 & 0 \\ 0 & 0 & 10 & 0 & 0 & 0 \\ 0 & 0 & 0 & 10 & 0 & 0 \\ 0 & 0 & 0 & 0 & 14.14 & 0 \\ 0 & 0 & 0 & 0 & 0 & 14.14 \end{bmatrix} \begin{bmatrix} -0.707 & -1 \\ -0.707 & 0 \\ -0.707 & 0 \\ -0.707 & 0 \\ 1 & 0 \\ 1 & 0 \end{bmatrix}$$

$$F = \begin{bmatrix} -0.707 & -0.707 & -0.707 & -0.707 & 1 & 1 \\ -1 & 0 & 0 & 0 & 0 & 0 \end{bmatrix} \frac{1}{AE} \begin{bmatrix} -7.07 & -10 \\ -7.07 & 0 \\ -7.07 & 0 \\ -7.07 & 0 \\ 14.14 & 0 \\ 14.14 & 0 \end{bmatrix} \Rightarrow \frac{1}{AE} \begin{bmatrix} 48.27 & 7.07 \\ 7.07 & 10 \end{bmatrix}$$

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# Flexibility Matrix-TRUSS

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## Calculation of Redundant Force Q

$$Q = -[F]^{-1} [D_{QL}]$$

$$Q = -\cancel{AE} \begin{bmatrix} 48.27 & 7.07 \\ 7.07 & 10 \end{bmatrix}^{-1} \frac{1}{\cancel{AE}} \begin{bmatrix} 3413.4 \\ 0 \end{bmatrix}$$

$$Q = -\frac{1}{432.72} \begin{bmatrix} 10 & -7.07 \\ -7.07 & 48.27 \end{bmatrix} \begin{bmatrix} 3413.4 \\ 0 \end{bmatrix}$$

$$\begin{bmatrix} Q_1 \\ Q_2 \end{bmatrix} = \begin{bmatrix} -78.88 \\ 55.77 \end{bmatrix}$$

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# Flexibility Matrix-TRUSS

9<sup>th</sup> May, 2020

Talking:

## Calculation of Shear and Moment $A_m$

$$[A_m] = [A_{mL}] + [A_{mu}] [Q]$$

$$[A_m] = \begin{bmatrix} 0 \\ -100 \\ -100 \\ 0 \\ 141.4 \\ 0 \end{bmatrix} + \begin{bmatrix} -0.707 & -1 \\ -0.707 & 0 \\ -0.707 & 0 \\ -0.707 & 0 \\ 1 & 0 \\ 1 & 0 \end{bmatrix} \begin{bmatrix} -78.88 \\ 55.77 \end{bmatrix}$$

$$[A_m] = \begin{bmatrix} 0 \\ -44.23 \\ -44.23 \\ 55.77 \\ 62.52 \\ -78.88 \end{bmatrix}$$

$$[A_m] = \begin{bmatrix} 0 \\ -100 \\ -100 \\ 0 \\ 141.4 \\ 0 \end{bmatrix} + \begin{bmatrix} 0 \\ 55.77 \\ 55.77 \\ 55.77 \\ -78.88 \\ -78.88 \end{bmatrix}$$

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